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(54) **Turbocharger with turbine nozzle vanes and an annular rotary bypass valve**

Turbolader mit Turbinendüsenschaufeln und einem ringförmigen drehbaren Umleitungsventil

Turbocompresseur à aubes de turbine et soupape de dérivation rotative annulaire

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EP 2 818 666 B1

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Description

BACKGROUND OF THE INVENTION

[0001] The present disclosure relates to exhaust gas-driven turbochargers, and particularly to turbine arrangements that allow exhaust gas to bypass the turbine under certain engine operating conditions.

[0002] Patent document number EP2213865A2 describes a turbocharger with an exhaust gas-driven turbine which includes a variable nozzle formed by a stator supporting fixed vanes and defining bypass openings, and a rotor supporting movable vanes and defining bypass channels. Each fixed vane overlaps with a movable vane by a varying extent dependent on rotational position of the rotor. In a fully open position of the rotor, the movable vanes have a maximum extent of overlap with the fixed vanes, and the bypass channels align with the bypass openings so that some of the exhaust gas flows through the aligned bypass openings and channels, bypassing the turbine wheel.

[0003] Patent document number US2013/247566A1 describes a turbocharger and engine cylinder head assembly which includes a center housing rotating assembly (CHRA) and an engine cylinder head defining a receptacle for receiving the CHRA, the engine cylinder head defining a compressor volute. A diffuser is defined for receiving and diffusing the compressed air from the compressor wheel and supplying the compressed air to the compressor volute. The assembly includes a cap formed separately from the engine cylinder head and removably secured to the engine cylinder head, the cap defining one wall of the diffuser. An opposite wall of the diffuser is defined either by the engine cylinder head or a portion of the center housing, depending on whether the CHRA slides into the receptacle compressor-wheel-first or turbine-wheel-first. The compressor volute can be located on a turbine side of the diffuser. DE102005045387 A1 discloses a rotary drive member for a bypass valve, which is aligned with the axis about which the rotary valve member rotates.

[0004] In a conventional turbocharger, the turbine housing defines a bypass conduit located generally to one side of the main bore through the housing, and the bypass conduit is connected to the exhaust gas inlet or the volute of the housing via a bypass valve. The bypass valve typically is a swing or poppet style valve comprising a circular valve member that is urged against a flat valve seat surrounding the bypass volute opening. The valve usually is arranged such that the exhaust gas pressure acts on the valve member in a direction tending to open the valve. One drawback associated with such an arrangement is that it is difficult to completely seal the valve in the closed position, since gas pressure tends to open the valve. Leakage past the closed bypass valve is a cause of performance degradation of the turbine and, hence, the turbocharger and its associated engine. The typical solution to the leakage issue is to preload the by-

pass valve member against the valve seat, but often this does not fully eliminate leakage, and in any event it causes additional problems such as an increase in the required actuation force for opening the valve.

[0005] Furthermore, swing or poppet valves tend to be poor in terms of controllability, especially at the crack-open point, and it is common for the bypass flow rate to be highly nonlinear with valve position, which makes it very difficult to properly regulate the bypass flow rate. This leads to problems such as poor transient response of the turbocharger and engine system.

BRIEF SUMMARY OF THE DISCLOSURE

[0006] The present invention in its various aspects is as set out in the appended claims. The present disclosure describes various embodiments of turbochargers having a novel bypass arrangement that seeks to address issues such as the ones noted above. In one embodiment, a turbocharger comprises a compressor including a compressor housing, and a turbine including a turbine wheel mounted in a turbine housing. The turbine housing defines at least part of a generally annular chamber extending circumferentially about a longitudinal axis of the turbine housing, and defines a bore extending along the longitudinal axis. A nozzle passage leads from the chamber radially inwardly into the turbine wheel. The turbine housing defines an annular bypass volute surrounding the bore and arranged to allow exhaust gas to bypass the turbine wheel. The turbocharger also includes an annular bypass valve disposed in the bypass volute, the bypass valve comprising a fixed annular valve seat and a rotary annular valve member. The valve member is disposed against the valve seat and is rotatable about the longitudinal axis for selectively varying a degree of alignment between respective orifices defined through each of the valve seat and valve member, ranging from no alignment defining a closed condition of the bypass valve, to at least partial alignment defining an open condition of the bypass valve.

[0007] A rotary drive member penetrates through the turbine housing along a drive axis that is generally parallel to and transversely offset from the longitudinal axis about which the valve member rotates, and a drive arm is attached to a distal end of the rotary drive member. A distal end of the drive arm engages the valve member such that rotation of the rotary drive member about the drive axis causes the drive arm to rotate the valve member about the longitudinal axis.

[0008] The turbocharger also includes a rotary actuator coupled to the rotary drive member and operable to rotatably drive the rotary drive member about the drive axis.

[0009] In one embodiment, the rotary drive member includes a lengthwise section having a substantially greater bending flexibility than that of a remainder of the rotary drive member. Preferably the lengthwise section has the substantially greater bending flexibility about a

plurality of axes that are not parallel to the drive axis. This bending flexibility allows a certain degree of angular misalignment between the rotary output shaft of the rotary actuator and the drive arm that drives the valve member.

[0010] In one embodiment the flexible lengthwise section of the rotary drive member comprises a bellows. An exemplary embodiment described herein is characterized by the bellows having an integral first hollow cylindrical portion at a first end of the bellows and an integral second hollow cylindrical portion at an opposite second end of the bellows. A first drive shaft is rigidly affixed to the first hollow cylindrical portion (e.g., by inserting the first drive shaft into the first hollow cylindrical portion and welding these parts together) and a second drive shaft is rigidly affixed to the second hollow cylindrical portion (e.g., by a similar welding operation). The first drive shaft is connected to the drive arm and the second drive shaft is connected to the rotary actuator.

[0011] In one embodiment, the first drive shaft and the drive arm together constitute a single integral monolithic structure.

[0012] Additionally, the second drive shaft can be an output shaft of the rotary actuator.

[0013] In addition to being flexible in bending, the bellows is also able to accommodate thermal expansion and contraction along the direction of the rotary drive axis. Furthermore, because the bellows acts like a compression spring along the drive axis, it is possible to axially compress the bellows during assembly (e.g., before welding the bellows to the output shaft of the rotary actuator) so as to create an axial compressive pre-load in the bellows. The pre-load will tend to bias the drive arm and the actuator output shaft axially away from each other. This can have the effect of eliminating undesirable axial gaps or "play" in the linkage.

[0014] The turbocharger advantageously includes a bushing mounted in the turbine housing. The bushing defines a passage through which the first drive shaft passes, and an inner surface of the passage constitutes a bearing surface allowing the first drive shaft to rotate about the drive axis.

[0015] If desired or required in a particular application, the bushing can define a first mechanical stop for the drive arm to limit rotation of the drive arm in a first direction. Additionally, the bushing can further define a second mechanical stop for the drive arm to limit rotation of the drive arm in a second direction.

[0016] In accordance with a further embodiment described herein, a turbocharger comprises a compressor including a compressor housing, and a turbine housing defining an exhaust gas inlet and defining at least part of a generally annular chamber extending circumferentially about a longitudinal axis of the turbine housing, the exhaust gas inlet leading into the chamber, the turbine housing further defining a bore extending along the longitudinal axis; a turbine wheel disposed in the turbine housing; a nozzle leading from the chamber radially inwardly into the turbine wheel; the turbine housing defining

an annular bypass volute surrounding the bore and a passage leading from the exhaust gas inlet into the bypass volute, the bypass volute being arranged to allow exhaust gas to bypass the turbine wheel; and an annular bypass valve disposed in the bypass volute. The bypass volute has a non-uniform cross-sectional flow area around a circumference of the bypass volute.

[0017] In one embodiment the cross-sectional flow area of the bypass volute has a maximum value where the passage leads from the exhaust gas inlet into the bypass volute, and decreases from that maximum value with increasing circumferential distance away from the passage.

BRIEF DESCRIPTION OF THE SEVERAL VIEWS OF THE DRAWING(S)

[0018] Having thus described the disclosure in general terms, reference will now be made to the accompanying drawings, which are not necessarily drawn to scale, and wherein:

FIG. 1 is perspective view of a turbocharger in accordance with one embodiment of the present invention;

FIG. 2 is an exploded perspective view of the turbocharger of FIG. 1;

FIG. 3 is a side view of the turbocharger of FIG. 1, partly in section to show internal details in the region of the turbine;

FIG. 4 is an axial cross-sectional view of a turbine housing assembly for the turbocharger of FIG. 1, also showing a bellows drive member affixed to a monolithic drive shaft and drive arm for a bypass valve member;

FIG. 5 is an exploded perspective view of a bushing and a monolithic drive shaft and drive arm in accordance with one embodiment; and

FIG. 6 is a perspective view of the turbocharger of FIG. 1, with a portion of the turbine housing broken away to show the connection between the drive arm and the rotary valve member.

DETAILED DESCRIPTION OF THE DRAWINGS

[0019] The present disclosure now will be described more fully hereinafter with reference to the accompanying drawings in which some but not all embodiments of the inventions are shown. Indeed, these inventions may be embodied in many different forms and should not be construed as limited to the embodiments set forth herein; rather, these embodiments are provided so that this disclosure will satisfy applicable legal requirements. Like

numbers refer to like elements throughout.

[0020] A turbocharger **20** in accordance with one embodiment of the present invention is shown in FIGS. 1 through 3. As shown, major sub-assemblies of the turbocharger **20** include a compressor assembly **30** and a turbine assembly **50**. The compressor assembly **30** includes a compressor housing **32** and a compressor wheel (not visible in the drawings) mounted therein and attached to one end of a rotary shaft (not visible). A center housing assembly **40** includes a center housing that is affixed to the compressor housing **32** and that contains bearings for the rotary shaft. The turbine assembly **50** includes a turbine housing **52** and a turbine wheel **54** mounted therein and attached to the opposite end of the rotary shaft.

[0021] With particular reference to FIGS. 3 and 4, the turbine housing **52** defines an exhaust gas inlet **56** through which exhaust gas from an internal combustion engine is received, and a volute or annular chamber **58** that receives the exhaust gas from the inlet **56** and distributes the gas around the 360° chamber for feeding into the turbine wheel **54**. The exhaust gas inlet **56** is also connected via a connecting passage **59** to a generally annular bypass volute **60** defined in the turbine housing **52**. The bypass volute **60** surrounds an axial bore **62** defined in the turbine housing. Exhaust gas that has passed through the turbine wheel **54** is exhausted from the turbine housing through the bore **62**. The bypass volute **60** provides an alternative pathway for exhaust gas to flow without first having to pass through the turbine wheel **54**. The gas can flow through the bypass volute **60**, however, only if the bypass valve described below is open.

[0022] With reference to FIGS. 2 and 4, an annular bypass valve **70** is installed in the bypass volute **60** for regulating flow through the bypass volute. The major components of the annular bypass valve **70** include a stationary valve seat **72** and a rotary valve member **74** in abutting engagement with the valve seat. The valve seat **72** and valve member **74** are arranged between an annular outer portion **52a** of the turbine housing **52** and an annular inner member **53**. The inner member **53** is tubular in form. In the illustrated embodiment the inner member **53** is integral with the rest of the turbine housing, but alternatively the inner member can be a separately formed tubular member that is fitted into the turbine housing. Making the inner member **53** as an integral part of the turbine housing can improve rigidity and robustness of the construction. The valve member **74** is prevented from moving axially upstream (forward) by a tapering part of the outer portion **52a** of the turbine housing, although during operation pressure of the exhaust gas urges the valve member **74** in the downstream direction against the valve seat **72**. The valve member **74** is not constrained by the turbine housing but is free to rotate about its axis and to move axially against the valve seat **72**. The valve seat **72** is prevented from moving axially, radially, or rotationally.

[0023] The valve seat **72** (best illustrated in FIG. 2) is a generally flat ring-shaped or annular member having a plurality of orifices **73** circumferentially spaced apart about a circumference of the valve seat, the orifices **73** extending generally axially between the upstream and downstream faces of the valve seat. The orifices **73** in the illustrated embodiment are uniformly spaced about the circumference of the valve seat, but as further described below, non-uniform spacing of the orifices is also possible and can be advantageous in some circumstances. The valve seat **72** can be formed by any of various processes and materials. For example, processes that can be used include casting, casting and machining, and stamping. The illustrated valve seat **72**, which includes solid reinforcing ribs between the orifices **73**, cannot feasibly be made by a stamping process, but can be made by casting and machining.

[0024] The rotary valve member **74** (best illustrated in FIG. 2) is a generally flat ring-shaped or annular member having a plurality of orifices **75** circumferentially spaced apart about a circumference of the valve member, the orifices **75** extending generally axially between the upstream and downstream faces of the valve member. The orifices **75** in the illustrated embodiment are uniformly spaced about the circumference of the valve member, and the number and spacing of the orifices **75** in the valve member are the same as the number and spacing of the orifices **73** in the valve seat. However, as further described below, non-uniform spacing of the orifices **75** is also possible and can be advantageous in some circumstances; furthermore, the spacings of the orifices **73** and **75** do not have to be the same, and in some cases it can be advantageous for the spacings to be different. The valve member **74** has a substantially circular outer edge **76** and a substantially circular inner edge **78**, the outer and inner edges **76**, **78** being coaxial with respect to a central longitudinal axis of the valve member, which axis is also substantially coincident with a central longitudinal axis of the valve seat **72**. The outer portion **52a** of the turbine housing and the inner member **53** both define substantially circular bearing surfaces for the outer and inner edges **76**, **78** of the rotary valve member **74** and there are clearances therebetween, so that the valve member can be rotated in one direction or the opposite direction about its central longitudinal axis in order to vary a degree of alignment between the valve member orifices **75** and the valve seat orifices **73**, as further described below.

[0025] With reference to FIGS. 2 through 6, rotation of the valve member **74** is accomplished by a rotary actuator **80** having an output shaft **82** that is connected to one end of a rotary drive member **90**. Connected to the opposite end of the drive member **90** is a drive shaft **100**. The output shaft **82**, rotary drive member **90**, and drive shaft **100** are generally collinear, having a common axis, referred to herein as the drive axis, that is parallel to and transversely offset from the longitudinal axis about which the valve member **74** rotates. A drive arm **110** is connected to the drive shaft **100**. The drive arm **110** is gen-

erally "L"-shaped, having a portion that extends generally perpendicular to the drive axis, and a distal end (i.e., the end remote from the end that is connected to the drive shaft **100**) that defines a pin or rod portion **112** that extends generally parallel to the drive axis. The valve member **74** includes a fork **79** that receives the pin portion **112** of the drive arm **110**. Rotation of the rotary actuator's output shaft **82** causes the drive member **90** to rotate about the drive axis, which causes the drive shaft **100** to rotate and therefore the distal end of the drive arm **110** sweeps through an arc, thereby causing the valve member **74** to rotate about its longitudinal axis. Thus, rotation of the actuator in one direction will rotate the valve member in a first direction (opposite to that of the actuator), and rotation of the actuator in the other direction will cause the valve member to rotate in a second direction.

[0026] In accordance with an embodiment of the invention, as illustrated in the figures, the rotary drive member **90** can include a lengthwise section whose bending flexibility is substantially greater than that of the remaining portions of the drive member. The bending flexibility preferably is substantially greater about multiple axes that are not parallel to the drive axis about which the drive member rotates to impart movement to the drive arm **110**. In one embodiment, as shown, the section of greater flexibility is a bellows **92**. The drive member **90** is preferably formed of a resilient metal such that the bellows **92** can act as a spring in axial compression and will also return to a straight (i.e., unbent) condition after any bending force is removed. The rotary drive member has an integral first hollow cylindrical portion **94** at a first end of the bellows **92**, and an integral second hollow cylindrical portion **96** at an opposite second end of the bellows. The drive shaft **100** is rigidly affixed to the first hollow cylindrical portion **94** and the actuator output shaft **82** is rigidly affixed to the second hollow cylindrical portion **96**. For example, each of the shafts **82** and **100** can be inserted into the respective hollow cylindrical portions and can be welded in place.

[0027] As noted, the bellows **92** can act like a compression spring along the drive axis. This can be used to advantage for taking up any axial play in the linkage between the actuator **80** and the drive arm **110**. Accordingly, the bellows can be axially compressed so as to create an axial compressive pre-load in the bellows. For example, with reference to FIG. 3, with the turbocharger substantially assembled and the hollow cylindrical portion **94** already welded to the drive shaft **100** but the other hollow cylindrical portion **96** not yet welded to the actuator output shaft **82**, the bellows **92** can be axially compressed by a determined amount, and then the hollow cylindrical portion **96** can be welded to the actuator output shaft **82**.

[0028] With reference to FIGS. 3 through 5, the turbocharger includes a bushing **120** for the drive shaft **100**. The bushing is installed in a cavity **55** defined in the turbine housing **52**. The bushing defines a through passage **122** for the drive shaft **100**. The through passage **122** has a cylindrical inner surface of a diameter sized to fit

closely about the drive shaft **100** while still allowing the drive shaft to freely rotate about the axis defined by the inner surface. An end of the drive shaft **100** extends out the end of the through passage **122** and connects to the drive arm **110**.

[0029] With reference to FIG. 4, it will be appreciated that because the drive arm **110** is oriented substantially radially and rotates about the drive axis at the radially outer end of the drive arm, for a given angular rotation of the drive shaft **100**, the radially inner or distal end of the drive arm swings through an arc whose circumferential extent is related to the radial distance between the drive axis and the distal end of the drive arm, but is not related to the axial length of the pin **112**. In other words, the drive arm **110** can occupy a short axial length without compromising the ability to achieve the desired circumferential movement range of the pin **112**. This is not true of an arrangement such as shown in Applicant's U.S. Patent No. 8,353,664 issued on January 15, 2013. With reference to FIG. 1 of that patent, because the drive member **100** rotates about an axis that is radial, for a given angular rotation of the drive member **100** the circumferential extent of movement of the end of the attached drive arm **90** is related directly to the axial length of the arm **90**. Thus, the drive arm cannot be reduced in axial length below a certain value or else an inadequate amount of movement of the drive arm would result. In contrast, the arrangement described and claimed herein can enable an axial length reduction of the drive arm **110** and consequently an axial length reduction of the turbine housing.

[0030] In one embodiment, the drive shaft **100** and drive arm **110** together constitute a single integral, monolithic part. Thus, the drive shaft is configured so that it can be inserted (right-to-left in FIG. 3) through the passage **122** of the bushing **120**, after which the end of the drive shaft is affixed to one end of the drive member **90**. The opposite end of the drive member **90** is affixed to the output shaft **82** of the actuator **80**.

[0031] The bushing **120** can define one or two mechanical stops for the drive arm **110**. Thus, as shown in FIGS. 5 and 6, a mechanical stop **124** is defined on the end face of the bushing **120** for limiting the rotation of the drive arm in a counterclockwise direction of those figures. An additional mechanical stop **126** is defined for limiting the rotation of the drive arm in the clockwise direction. FIG. 6 shows the arm **110** against the stop **126**, which may define, for example, the fully open position of the bypass valve **70**. The stop **124** may define the fully closed position.

[0032] With reference to FIG. 3, a further feature of the invention is now described. It will be seen that the bypass volute **60** does not have a uniform cross-sectional flow area around the circumference of the volute. Rather, the bypass volute **60** has a non-uniform cross-sectional flow area characterized by a maximum flow area where the passage **59** leads from the inlet **56** into the bypass volute **60**, and the flow area decreases from that maximum with

increasing circumferential distance away from the passage **59**. It will also be noted that the bypass volute **60** has a relatively small flow area at circumferential locations where the main volute **58** has a relatively large flow area, and has a relatively large flow area at circumferential locations where the main volute **58** has a relatively small flow area. The non-uniform flow area of the bypass volute **60** may allow substantially the same flow velocity at each of the bypass valve orifices.

[0033] In operation, bypass flow regulation in the turbocharger of FIGS. 1 through 6 is accomplished by rotation of the rotary actuator **80**, which changes the position of the bypass valve **70**. The valve **70** can be configured such that it is normally closed, i.e., there is no overlap between the valve seat orifices **73** and the valve member orifices **75**. The valve then begins to open only when the orifices **73**, **75** begin to overlap as the actuator **80** rotates the valve member **74**. Alternatively, the valve can be configured such that it is normally open, and rotation by the actuator moves the valve toward its closed position. In accordance with present disclosure, changing from a normally open to a normally closed valve configuration can be accomplished simply by rotating the valve seat **72** to the desired position and then fixing it in that position, for example by a set screw **77** (FIG. 6).

[0034] With the described annular bypass valve **70**, exhaust gas pressure acts on the valve member **74** in a direction toward the fixed valve seat **72**, thereby tending to improve sealing between the valve member and valve seat. Furthermore, the gas pressure does not tend to open the valve, in contrast to the aforementioned swing and poppet style bypass valve arrangements in which gas pressure acts in a direction tending to open the valve and cause leakage. The improved sealing made possible by the valve is thought to be significant because it can improve the transient response time of the turbocharger, by making better use of instantaneous engine pulses in the exhaust gas stream, especially at low engine speeds and gas flow rates where the pulse impact is most significant in regard to turbine efficiency.

[0035] A further advantage is that the valve **70** can achieve better controllability than is typically possible with swing or poppet valves, particularly at the crack-open point. In particular, the evolution of the shape and size of the flow passages through the valve as the valve member **74** is rotated can be tailored to the needs of a particular application simply by suitably configuring the sizes, angular locations (e.g., whether uniformly or non-uniformly spaced apart), and shapes of the orifices in the valve member and valve seat. Thus, while the orifices **73**, **75** are shown as being circular in the drawings, alternatively they can be made non-circular as a way of altering the evolution of the flow passages as the valve opens. For example, the orifices could be made generally rectangular with their edges extending generally radially (possibly with a larger dimension in the radial direction than in the circumferential direction), which would result in a greater change in flow passage size per degree of

valve member rotation, in comparison with the circular orifice shape.

[0036] As another example of the fine-tuning of the evolution of the valve flow passages made possible by the invention, the valve seat orifices **73** could have a first circumferential spacing (e.g., uniform) about the circumference, and the valve member orifices **75** could have a second circumferential spacing (e.g., non-uniform) different from the first circumferential spacing. It is further possible (though not essential) in such an embodiment for the orifices in the valve member to be of different sizes and/or shapes from the orifices in the valve seat. This could result in, for example, one flow passage (or some other subset of the total number of flow passages) beginning to open before any of the other flow passages begin to open, thereby achieving a very gradual cracking open of the bypass valve. Further rotation of the valve member would then cause the other flow passages to open (perhaps in a sequential or staged fashion, e.g., one flow passage opening at a time until finally all flow passages are open). These are merely some examples of the many different ways the orifices can be configured so as to achieve a desired flow passage evolution as a function of valve member rotation.

[0037] In accordance with the present disclosure, the rotary actuator **80** can be an electric actuator. The inline axial orientation of the rotary actuator and drive linkage enables a particularly compact arrangement of the turbocharger **20** relative to prior turbochargers employing pneumatic linear actuators whose linkage by necessity extended transversely to the turbocharger axis in order to convert the linear actuator movement into a rotation of the bypass valve member. A further advantage of the arrangement disclosed herein is the substantial elimination of free play in the kinematic chain between the actuator and the bypass valve member. This in turn means that the position of the actuator output shaft is well-correlated with the position of the valve member, which facilitates the accurate sensing and control of the bypass valve position.

[0038] The elimination of play may be further helped, in accordance with another embodiment, by ensuring that the pin **112** of the drive arm **110** is always abutting one of the legs of the fork **79** (see FIG. 6). It will be appreciated that it is not possible to make the width of the fork exactly the same as the diameter of the pin **112** and therefore there will generally be some amount of play in this connection. By ensuring that the pin is always held against just one of the legs, however, the play is taken out. This can be done in various ways. As one example, a suitable clip (not shown) can connect the pin to the leg and keep them in abutting contact. As another option, a vane (not shown) of suitable configuration can be provided on the valve member **74** for creating an aerodynamic loading from the exhaust gases so as to always urge the valve member in a rotational direction to keep one of the legs of the fork against the pin.

[0039] Many modifications and other embodiments of

the inventions set forth herein will come to mind to one skilled in the art to which these inventions pertain having the benefit of the teachings presented in the foregoing descriptions and the associated drawings. Therefore, it is to be understood that the inventions are not to be limited to the specific embodiments disclosed and that modifications and other embodiments are intended to be included within the scope of the appended claims. Although specific terms are employed herein, they are used in a generic and descriptive sense only and not for purposes of limitation.

Claims

1. A turbocharger comprising:

a compressor (30) including a compressor housing (32);
 a turbine housing (52) defining at least part of a generally annular chamber (58) extending circumferentially about a longitudinal axis of the turbine housing, the turbine housing defining a bore (62) extending along the longitudinal axis;
 a turbine wheel (54) disposed in the turbine housing;
 a nozzle passage leading from the chamber radially inwardly into the turbine wheel;
 the turbine housing defining an annular bypass volute (60) surrounding the bore and arranged to allow exhaust gas to bypass the turbine wheel;
 an annular bypass valve (70) disposed in the bypass volute, the bypass valve comprising a fixed annular valve seat (72) and a rotary annular valve member (74), the valve member being disposed against the valve seat and being rotatable about the longitudinal axis for selectively varying a degree of alignment between respective orifices (73, 75) defined through each of the valve seat and valve member, ranging from no alignment defining a closed condition of the bypass valve, to at least partial alignment defining an open condition of the bypass valve;
 a rotary drive member (90) penetrating through the turbine housing along a drive axis that is generally parallel to and transversely offset from the longitudinal axis about which the valve member rotates, and a drive arm (110) attached to a distal end of the rotary drive member, a distal end of the drive arm engaging the valve member such that rotation of the rotary drive member about the drive axis causes the drive arm to rotate the valve member about the longitudinal axis, wherein the rotary drive member (90) includes a lengthwise section having a substantially greater bending flexibility than that of a remainder of the rotary drive member; and

a rotary actuator (80) coupled to the rotary drive member and operable to rotatably drive the rotary drive member about the drive axis for rotating the valve member.

2. The turbocharger of claim 1, wherein the lengthwise section has the substantially greater bending flexibility about a plurality of axes that are not parallel to the drive axis.
3. The turbocharger of claim 2, wherein the lengthwise section comprises a bellows (92).
4. The turbocharger of claim 3, wherein the rotary drive member has an integral first hollow cylindrical portion (94) at a first end of the bellows (92) and an integral second hollow cylindrical portion (96) at an opposite second end of the bellows, and further comprising a first drive shaft (100) rigidly affixed to the first hollow cylindrical portion and a second drive shaft (82) rigidly affixed to the second hollow cylindrical portion, wherein the first drive shaft is connected to the drive arm (110) and the second drive shaft is connected to the rotary actuator (80).
5. The turbocharger of claim 4, wherein the bellows (92) acts like a compression spring along the drive axis, and the bellows is axially compressed so as to create an axial compressive pre-load in the bellows.
6. The turbocharger of claim 4, wherein the first drive shaft (100) and the drive arm (110) together constitute a single integral monolithic structure.
7. The turbocharger of claim 4, wherein the second drive shaft (82) is an output shaft of the rotary actuator (80).
8. The turbocharger of claim 4, further comprising a bushing (120) mounted in the turbine housing (52), the bushing defining a passage (122) through which the first drive shaft (100) passes, an inner surface of the passage constituting a bearing surface allowing the first drive shaft to rotate about the drive axis.
9. The turbocharger of claim 8, wherein the bushing defines a first mechanical stop (124) for the drive arm (110) to limit rotation of the drive arm in a first direction.
10. The turbocharger of claim 9, wherein the bushing further defines a second mechanical stop (126) for the drive arm (110) to limit rotation of the drive arm in a second direction.
11. The turbocharger of claim 1, the valve member (74) defining a plurality of first orifices (75) therethrough, the valve seat (72) defining a plurality of second or-

ifices (73) therethrough, and each first orifice having a corresponding second orifice.

12. The turbocharger of claim 11, the first orifices (75) having a first circumferential spacing therebetween and the second orifices (73) having a second circumferential spacing therebetween, wherein the first circumferential spacing is different from the second circumferential spacing, and wherein at least partial alignment between one of the first orifices and a corresponding one of the second orifices creates a flow passage for flow of exhaust gas therethrough, rotation of the valve member (74) about the axis causing variation in a degree of alignment between the first orifices (75) defined through the valve member and the second orifices (73) defined through the valve seat (72), ranging from no alignment defining a closed condition of the bypass valve (70), to at least partial alignment defining an open condition of the bypass valve.
13. The turbocharger of claim 12, wherein the first and second orifices (75, 73) are configured and arranged such that one subset of all flow passages begins to open before any of the other flow passages begin to open.
14. The turbocharger of claim 13, wherein the first and second orifices (75, 73) are configured and arranged such that one flow passage opens at a time, until finally all flow passages are open.

Patentansprüche

1. Turbolader, umfassend:

einen Kompressor (30), der ein Kompressorgehäuse (32) beinhaltet;
 ein Turbinengehäuse (52), das mindestens einen Teil einer allgemein ringförmigen Kammer (58) definiert, die sich umlaufend um eine Längsachse des Turbinengehäuses erstreckt, wobei das Turbinengehäuse eine Bohrung (62) definiert, die sich entlang der Längsachse erstreckt;
 ein Turbinenrad (54), das in dem Turbinengehäuse angeordnet ist;
 eine Düsenpassage, die von der Kammer radial nach innen in das Turbinenrad führt;
 wobei das Turbinengehäuse eine ringförmige Umleitungsschnecke (60) definiert, welche die Bohrung umgibt und angeordnet ist, um Abgas zu gestatten, das Turbinenrad zu umgehen;
 ein ringförmiges Umleitungsventil (70), das in der Umleitungsschnecke angeordnet ist, wobei das Umleitungsventil einen befestigten ringförmigen Ventilsitz (72) und ein drehendes ringförmiges Ventilelement (74) umfasst, wobei das Ventilelement gegen den Ventilsitz angeordnet ist und um die Längsachse drehbar ist, um einen Grad von Ausrichtung zwischen entsprechenden Öffnungen (73, 75), die durch jeden von dem Ventilsitz und dem Ventilelement definiert sind, wahlweise zu variieren, reichend von keiner Ausrichtung, die einen geschlossenen Zustand des Umleitungsventils definiert, bis zu einer mindestens teilweisen Ausrichtung, die einen offenen Zustand des Umleitungsventils definiert;
 ein Drehantriebsselement (90), das das Turbinengehäuse entlang einer Antriebsachse penetriert, die zu der Längsachse, um die sich das Ventilelement dreht, im Allgemeinen parallel und quer von ihr versetzt ist, und einen Antriebsarm (110), der an einem distalen Ende des Drehantriebsselements angebracht ist, wobei ein distales Ende des Antriebsarms in das Ventilelement eingreift, sodass eine Drehung des Drehantriebsselements um die Drehachse den Antriebsarm veranlasst, das Ventilelement um die Längsachse zu drehen, wobei das Drehantriebsselement (90) einen Längsabschnitt beinhaltet, der eine wesentlich größere Biegeflexibilität als jene von einem Rest des Drehantriebsselements aufweist; und
 einen Drehaktor (80), der an das Drehantriebsselement gekoppelt und betreibbar ist, um das Drehantriebsselement um die Drehachse drehbar anzutreiben, um das Ventilelement zu drehen.

miges Ventilelement (74) umfasst, wobei das Ventilelement gegen den Ventilsitz angeordnet ist und um die Längsachse drehbar ist, um einen Grad von Ausrichtung zwischen entsprechenden Öffnungen (73, 75), die durch jeden von dem Ventilsitz und dem Ventilelement definiert sind, wahlweise zu variieren, reichend von keiner Ausrichtung, die einen geschlossenen Zustand des Umleitungsventils definiert, bis zu einer mindestens teilweisen Ausrichtung, die einen offenen Zustand des Umleitungsventils definiert;
 ein Drehantriebsselement (90), das das Turbinengehäuse entlang einer Antriebsachse penetriert, die zu der Längsachse, um die sich das Ventilelement dreht, im Allgemeinen parallel und quer von ihr versetzt ist, und einen Antriebsarm (110), der an einem distalen Ende des Drehantriebsselements angebracht ist, wobei ein distales Ende des Antriebsarms in das Ventilelement eingreift, sodass eine Drehung des Drehantriebsselements um die Drehachse den Antriebsarm veranlasst, das Ventilelement um die Längsachse zu drehen, wobei das Drehantriebsselement (90) einen Längsabschnitt beinhaltet, der eine wesentlich größere Biegeflexibilität als jene von einem Rest des Drehantriebsselements aufweist; und
 einen Drehaktor (80), der an das Drehantriebsselement gekoppelt und betreibbar ist, um das Drehantriebsselement um die Drehachse drehbar anzutreiben, um das Ventilelement zu drehen.

2. Turbolader nach Anspruch 1, wobei der Längsabschnitt die wesentlich größere Biegeflexibilität um eine Vielzahl von Achsen aufweist, die nicht parallel zu der Antriebsachse sind.
3. Turbolader nach Anspruch 2, wobei der Längsabschnitt einen Balg (92) umfasst.
4. Turbolader nach Anspruch 3, wobei das Drehantriebsselement an einem ersten Ende des Balgs (92) einen einstückigen ersten hohlen zylindrischen Abschnitt (94) und an einem entgegengesetzten zweiten Ende des Balgs einen einstückigen zweiten hohlen zylindrischen Abschnitt (96) aufweist, und ferner umfassend eine erste Antriebswelle (100), die starr an dem ersten hohlen zylindrischen Abschnitt befestigt ist, und eine zweite Antriebswelle (82), die starr an dem zweiten hohlen zylindrischen Abschnitt befestigt ist, wobei die erste Antriebswelle mit dem Antriebsarm (110) verbunden ist und die zweite Antriebswelle mit dem Drehaktor (80) verbunden ist.
5. Turbolader nach Anspruch 4, wobei der Balg (92) entlang der Antriebsachse wie eine Kompressions-

feder wirkt und der Balg axial zusammengedrückt wird, um in dem Balg eine axiale, kompressive Vorlast herzustellen.

6. Turbolader nach Anspruch 4, wobei die erste Antriebswelle (100) und der Antriebsarm (110) gemeinsam eine einzelne einstückige monolithische Struktur ausbilden.
7. Turbolader nach Anspruch 4, wobei die zweite Antriebswelle (82) eine Ausgangswelle des Drehaktors (80) ist.
8. Turbolader nach Anspruch 4, ferner umfassend eine Buchse (120), die in dem Turbinengehäuse (52) montiert ist, wobei die Buchse einen Durchgang (122) definiert, durch den die erste Antriebswelle (100) hindurch verläuft, wobei eine Innenfläche des Durchgangs eine Lagerfläche bildet, die der ersten Antriebswelle gestattet, sich um die Antriebsachse zu drehen.
9. Turbolader nach Anspruch 8, wobei die Buchse einen ersten mechanischen Anschlag (124) für den Antriebsarm (110) definiert, um eine Drehung des Antriebsarms in einer ersten Richtung zu begrenzen.
10. Turbolader nach Anspruch 9, wobei die Buchse ferner einen zweiten mechanischen Anschlag (126) für den Antriebsarm (110) definiert, um eine Drehung des Antriebsarms in einer zweiten Richtung zu begrenzen.
11. Turbolader nach Anspruch 1, wobei das Ventilelement (74) eine Vielzahl von ersten Öffnungen (75) dort hindurch definiert, der Ventil Sitz (72) eine Vielzahl von zweiten Öffnungen (73) dort hindurch definiert und jede erste Öffnung eine entsprechende zweite Öffnung aufweist.
12. Turbolader nach Anspruch 11, wobei die ersten Öffnungen (75) einen ersten umlaufenden Abstand dazwischen aufweisen und die zweiten Öffnungen (73) einen zweiten umlaufenden Abstand dazwischen aufweisen, wobei sich der erste umlaufende Abstand von dem zweiten umlaufenden Abstand unterscheidet, und wobei eine mindestens teilweise Ausrichtung zwischen einer von den ersten Öffnungen und einer entsprechenden einen von den zweiten Öffnungen einen Strömungsdurchgang zum Strömen von Abgas dort hindurch schafft, wobei Drehung des Ventilelements (74) um die Achse Variation in einem Grad von Ausrichtung zwischen den ersten Öffnungen (75), die durch das Ventilelement definiert sind, und den zweiten Öffnungen (73), die durch den Ventil Sitz (72) definiert sind, verursacht, reichend von keiner Ausrichtung, die einen geschlossenen Zustand des Umleitungsventils (70)

definiert, bis zu einer mindestens teilweisen Ausrichtung, die einen offenen Zustand des Umleitungsventils definiert.

13. Turbolader nach Anspruch 12, wobei die erste und zweite Öffnung (75, 73) dazu ausgelegt und so angeordnet sind, dass ein Teilsatz aller Strömungsdurchgänge sich zu öffnen beginnt, ehe einer der anderen Strömungsdurchgänge sich zu öffnen beginnt.
14. Turbolader nach Anspruch 13, wobei die erste und zweite Öffnung (75, 73) dazu ausgelegt und so angeordnet sind, dass sich jeweils ein Strömungsdurchgang öffnet, bis schließlich alle Strömungsdurchgänge offen sind.

Revendications

1. Turbocompresseur comprenant :

un compresseur (30) comprenant un carter de compresseur (32) ;
 un carter de turbine (52) définissant au moins une partie d'une chambre généralement annulaire (58) s'étendant circonférentiellement autour d'un axe longitudinal du carter de turbine, le carter de turbine définissant un alésage (62) s'étendant le long de l'axe longitudinal ;
 une roue de turbine (54) disposée dans le carter de turbine ;
 un passage de buse menant de la chambre radialement vers l'intérieur dans la roue de turbine ;
 le carter de turbine définissant une volute de dérivation annulaire (60) entourant l'alésage et conçue pour permettre aux gaz d'échappement de contourner la roue de turbine ;
 une soupape de dérivation annulaire (70) disposée dans la volute de dérivation, la soupape de dérivation comprenant un siège de soupape annulaire fixe (72) et un élément de soupape annulaire rotatif (74), l'élément de soupape étant disposé contre le siège de soupape et pouvant tourner autour de l'axe longitudinal pour faire varier de manière sélective un degré d'alignement entre des orifices respectifs (73, 75) définis à travers le siège de soupape et l'élément de soupape, allant d'aucun alignement définissant une condition fermée de la soupape de dérivation à un alignement au moins partiel définissant une condition ouverte de la soupape de dérivation ;
 un élément d'entraînement rotatif (90) pénétrant à travers le carter de turbine le long d'un axe d'entraînement qui est généralement parallèle à l'axe longitudinal, et décalé transversalement

- de l'axe longitudinal autour duquel l'élément de soupape tourne, et un bras d'entraînement (110) fixé à une extrémité distale de l'élément d'entraînement rotatif, une extrémité distale du bras d'entraînement venant en prise avec l'élément de soupape de telle sorte que la rotation de l'élément d'entraînement rotatif autour de l'axe d'entraînement amène le bras d'entraînement à faire tourner l'élément de soupape autour de l'axe longitudinal, dans lequel l'élément d'entraînement rotatif (90) comprend une section longitudinale ayant une flexibilité de pliage sensiblement plus importante que celle d'un reste de l'élément d'entraînement rotatif ; un actionneur rotatif (80) couplé à l'élément d'entraînement rotatif et permettant d'entraîner en rotation l'élément d'entraînement rotatif autour de l'axe d'entraînement pour faire tourner l'élément de soupape.
2. Turbocompresseur selon la revendication 1, dans lequel la section longitudinale présente une flexibilité de pliage sensiblement plus importante autour d'une pluralité d'axes qui ne sont pas parallèles à l'axe d'entraînement.
 3. Turbocompresseur selon la revendication 2, dans lequel la section longitudinale comprend un soufflet (92).
 4. Turbocompresseur selon la revendication 3, dans lequel l'élément d'entraînement rotatif comporte une première partie cylindrique creuse intégrée (94) au niveau d'une première extrémité du soufflet (92) et une seconde partie cylindrique creuse intégrée (96) au niveau d'une seconde extrémité opposée du soufflet et comprenant en outre un premier arbre d'entraînement (100) fixé de manière rigide à la première partie cylindrique creuse et un second arbre d'entraînement (82) fixé de manière rigide à la seconde partie cylindrique creuse, dans lequel le premier arbre d'entraînement est raccordé au bras d'entraînement (110) et le second arbre d'entraînement est raccordé à l'actionneur rotatif (80).
 5. Turbocompresseur selon la revendication 4, dans lequel le soufflet (92) fait office de ressort de compression le long de l'axe d'entraînement et le soufflet est comprimé axialement de sorte à créer une précharge de compression axiale dans le soufflet.
 6. Turbocompresseur selon la revendication 4, dans lequel le premier arbre d'entraînement (100) et le bras d'entraînement (110) constituent ensemble une seule structure monolithique intégrée.
 7. Turbocompresseur selon la revendication 4, dans lequel le second arbre d'entraînement (82) est un arbre de sortie de l'actionneur rotatif (80).
 8. Turbocompresseur selon la revendication 4, comprenant en outre une bague (120) montée dans le carter de turbine (52), la bague définissant un passage (122) à travers lequel le premier arbre d'entraînement (100) passe, une surface interne du passage constituant une surface de bague permettant au premier arbre d'entraînement de tourner autour de l'axe d'entraînement.
 9. Turbocompresseur selon la revendication 8, dans lequel la bague définit un premier arrêt mécanique (124) pour le bras d'entraînement (110) pour limiter la rotation du bras d'entraînement dans une première direction.
 10. Turbocompresseur selon la revendication 9, dans lequel la bague définit en outre un second arrêt mécanique (126) pour le bras d'entraînement (110) pour limiter la rotation du bras d'entraînement dans une seconde direction.
 11. Turbocompresseur selon la revendication 1, l'élément de soupape (74) définissant une pluralité de premiers orifices (75) à travers ce dernier, le siège de soupape (72) définissant une pluralité de seconds orifices (73) à travers ce dernier, et chaque premier orifice ayant un second orifice correspondant.
 12. Turbocompresseur selon la revendication 11, les premiers orifices (75) ayant un premier espacement circonférentiel entre ces derniers et les seconds orifices (73) ayant un second espacement circonférentiel entre ces derniers, dans lequel le premier espacement circonférentiel est différent du second espacement circonférentiel et dans lequel au moins un alignement partiel entre l'un des premiers orifices et un orifice correspondant des seconds orifices crée un passage d'écoulement pour l'écoulement d'un gaz d'échappement à travers ce dernier, la rotation de l'élément de soupape (74) autour de l'axe provoquant une variation d'un degré d'alignement entre les premiers orifices (75) définis à travers l'élément de soupape et les seconds orifices (73) définis à travers le siège de soupape (72), allant d'aucun alignement définissant une condition fermée de la soupape de dérivation (70) à un alignement au moins partiel définissant une condition ouverte de la soupape de dérivation.
 13. Turbocompresseur selon la revendication 12, dans lequel les premiers et seconds orifices (75, 73) sont configurés et agencés de telle sorte qu'un sous-ensemble de tous les passages d'écoulement commence à s'ouvrir avant que l'un quelconque des autres passages d'écoulement ne commence à s'ouvrir.

14. Turbocompresseur selon la revendication 13, dans lequel les premiers et seconds orifices (75, 73) sont configurés et agencés de telle sorte qu'un seul passage d'écoulement s'ouvre à un moment jusqu'à ce que finalement tous les passages d'écoulement soient ouverts. 5

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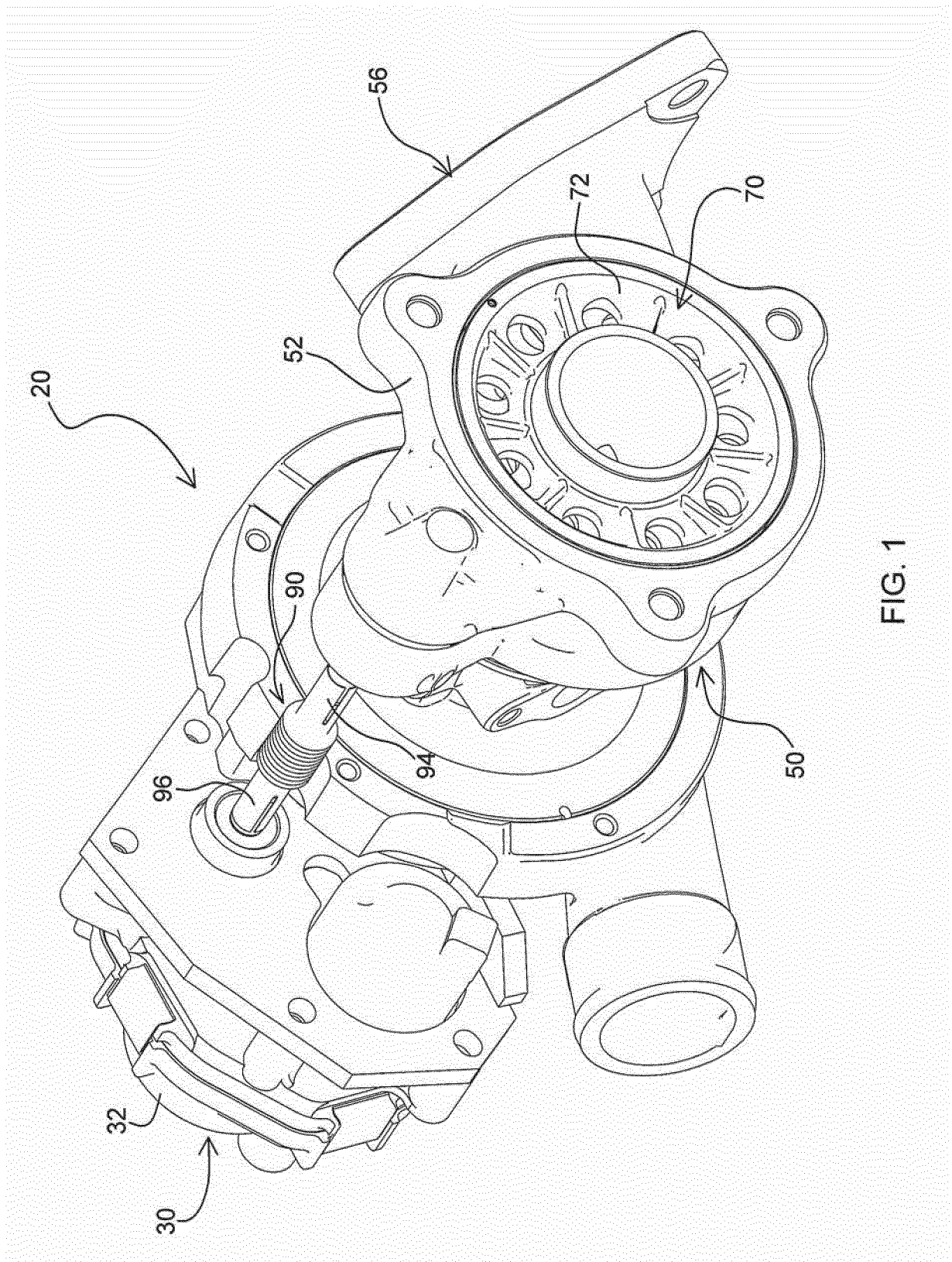
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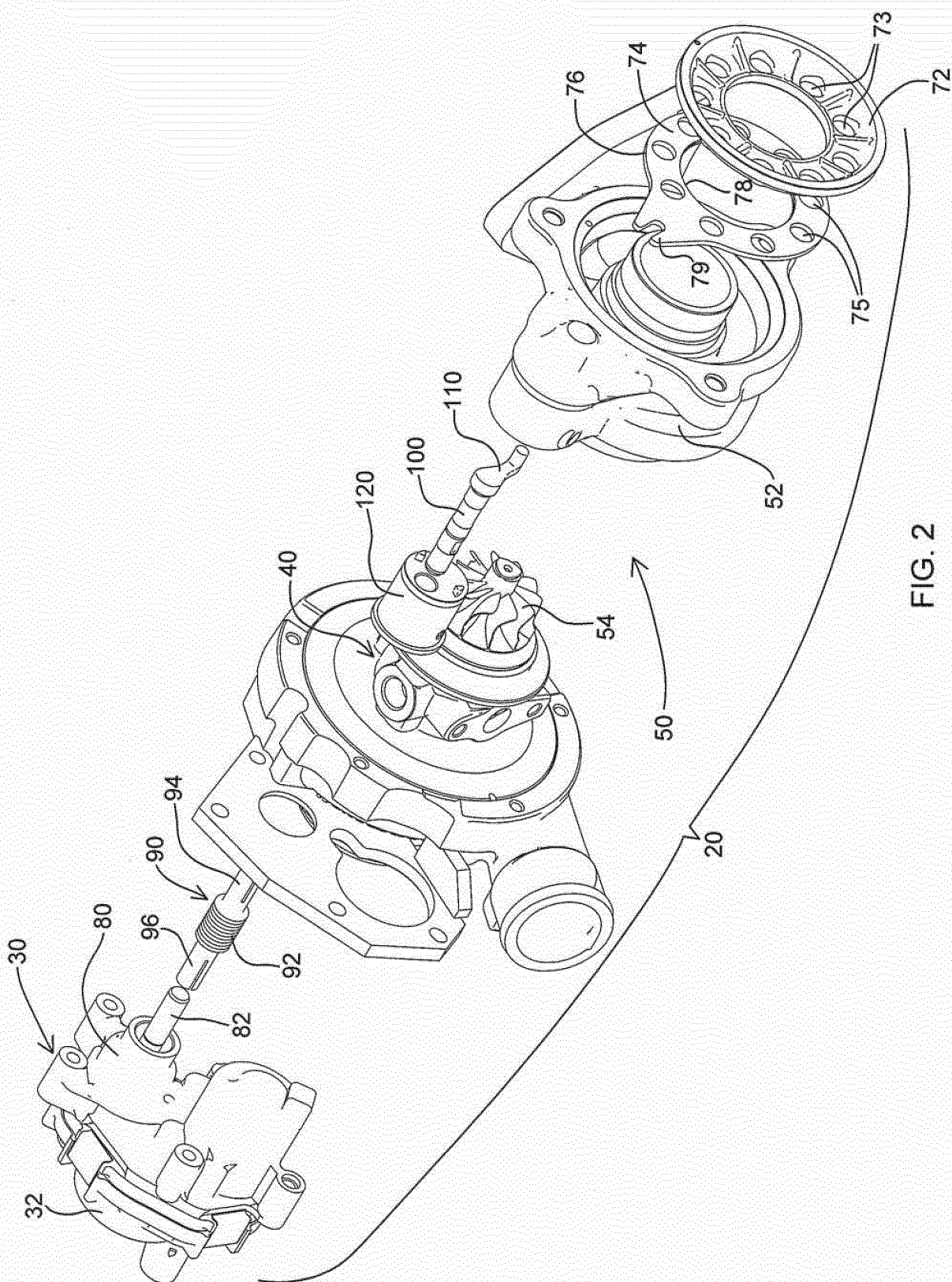
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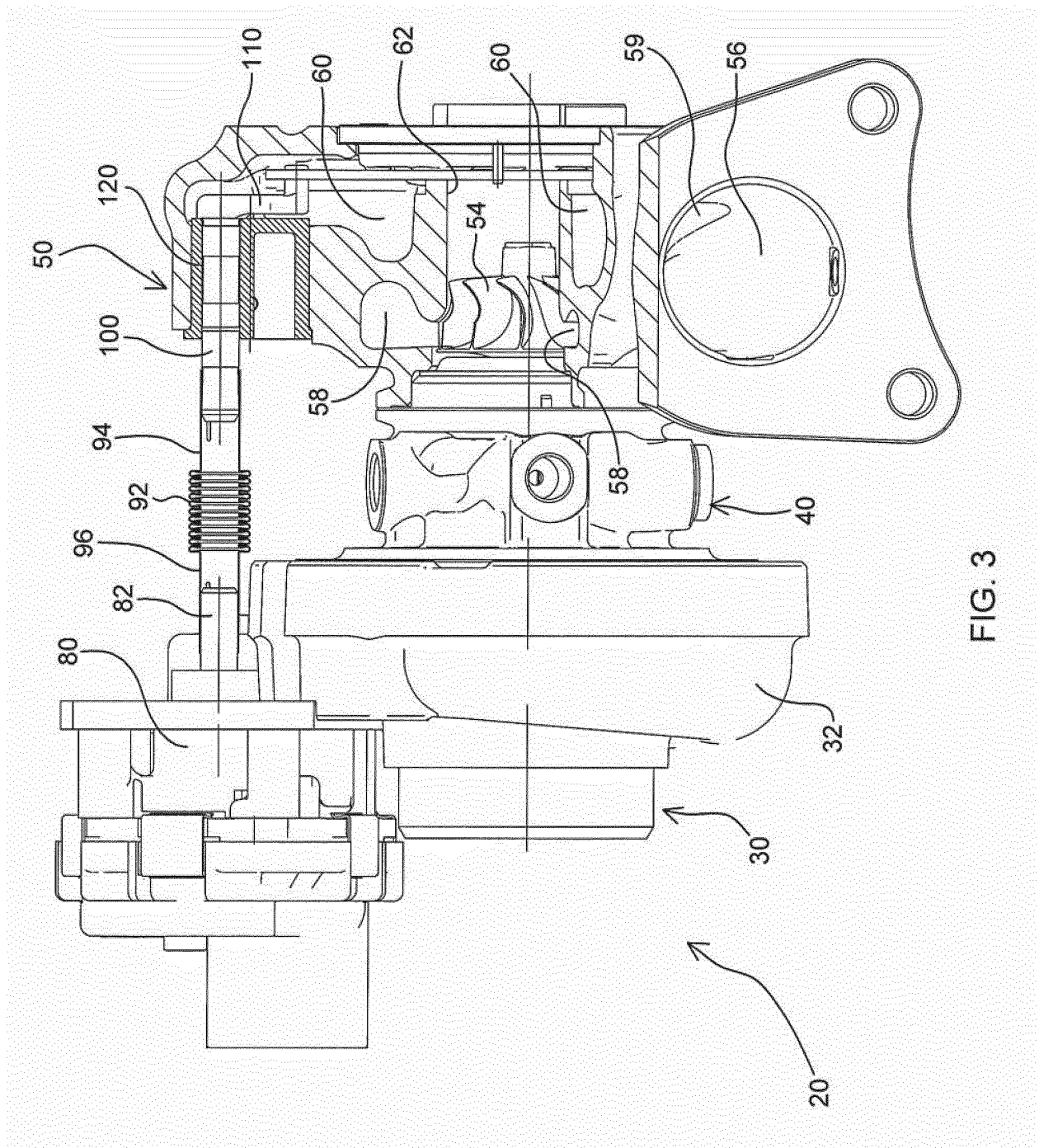
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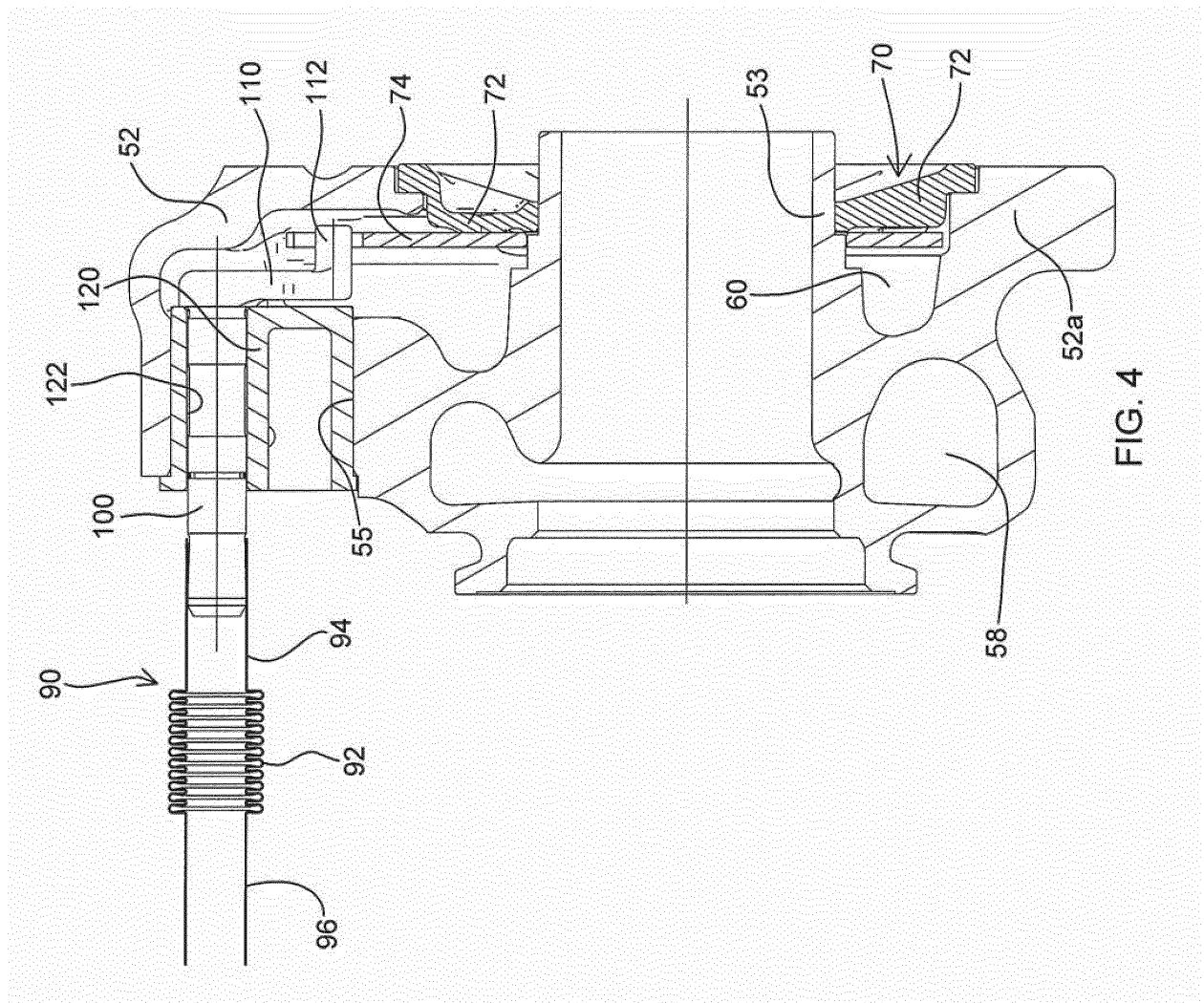
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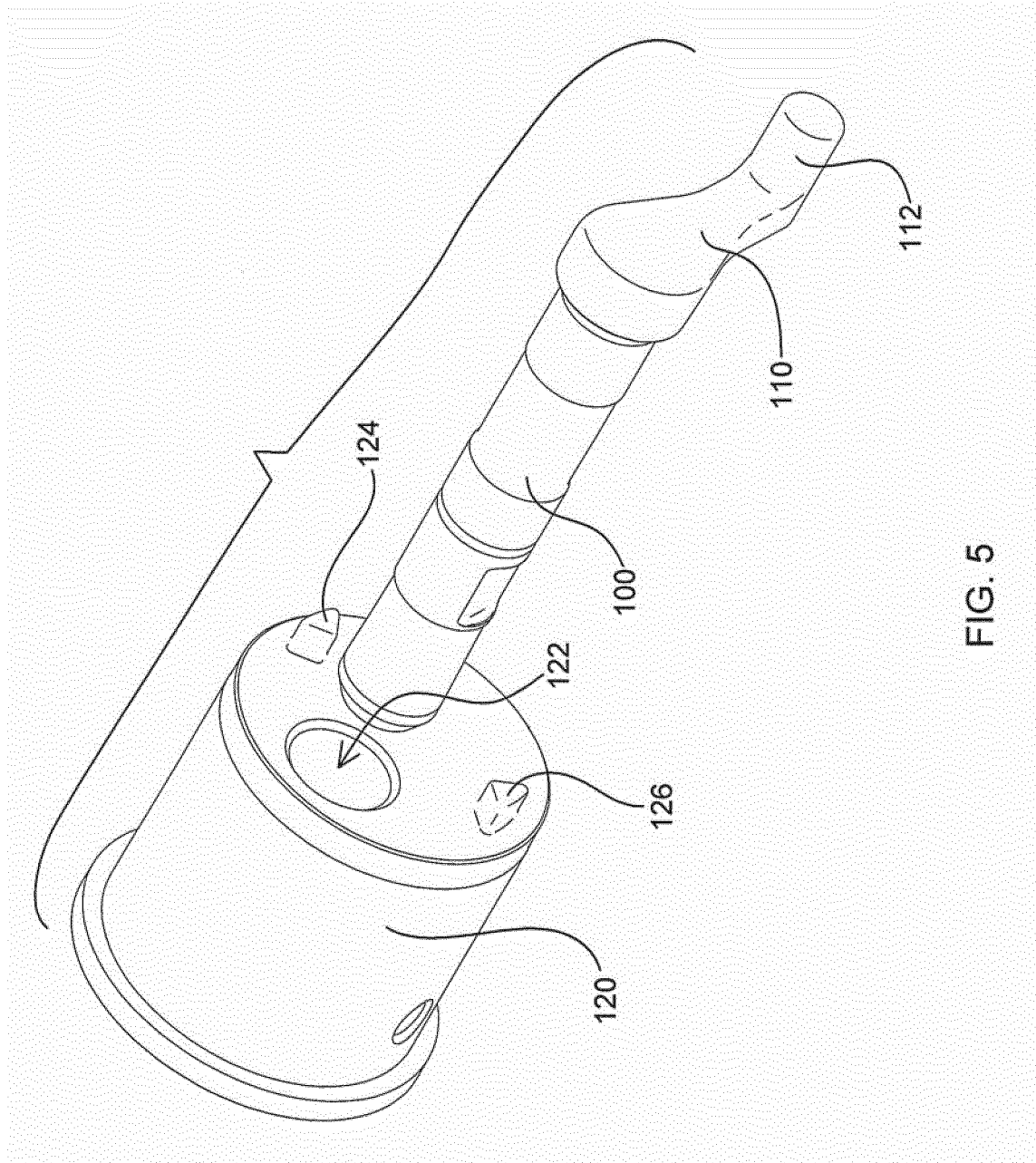
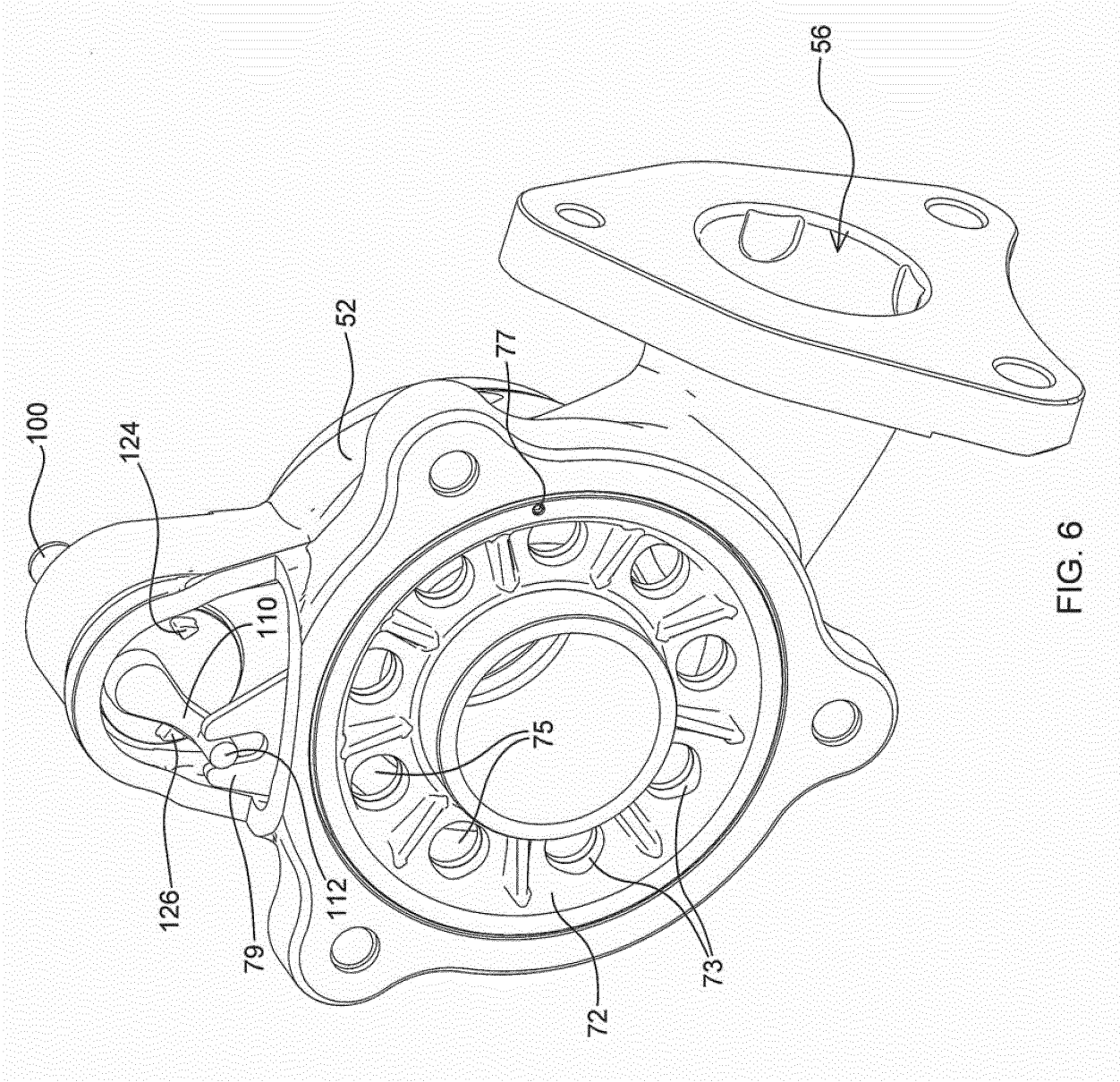


FIG. 5



REFERENCES CITED IN THE DESCRIPTION

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