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(54) **THERMALLY COMPLIANT HEATSHIELD**

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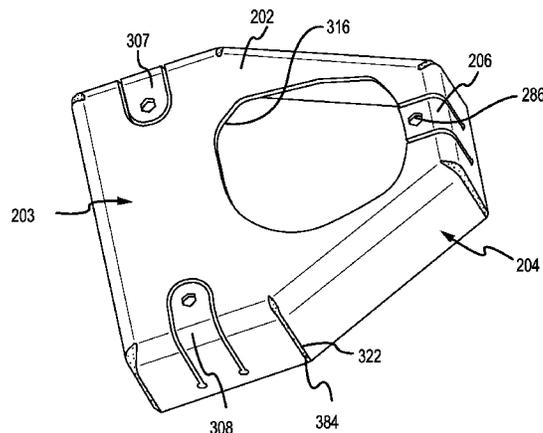
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(57) **ABSTRACT**

A heat shield comprising a sidewall portion, a top portion,
and a plurality of flexible tabs attached to the sidewall
portion is described herein, in accordance with various
embodiments. The top portion may comprise an aperture.
The sidewall portion may extend at an angle between 80
degrees and 100 degrees from the top portion. The sidewall
portion may bound a hexagonal void. The flexible tab may
comprise an angle between 80 degrees and 100 degrees. The
flexible tab may be fixed to the sidewall portion, wherein the
flexible tab is configured to be attached to a fitting.

7 Claims, 7 Drawing Sheets



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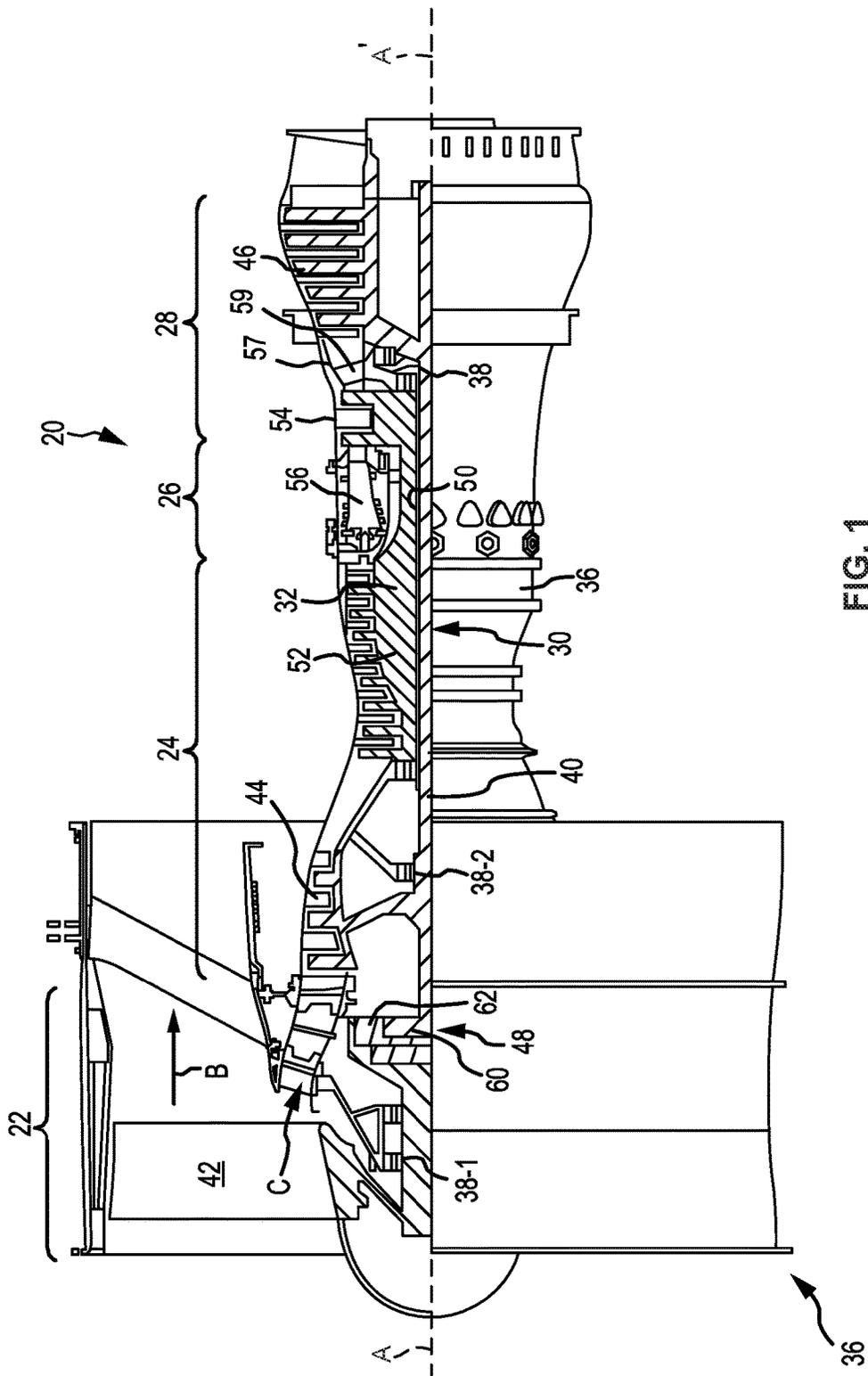


FIG. 1

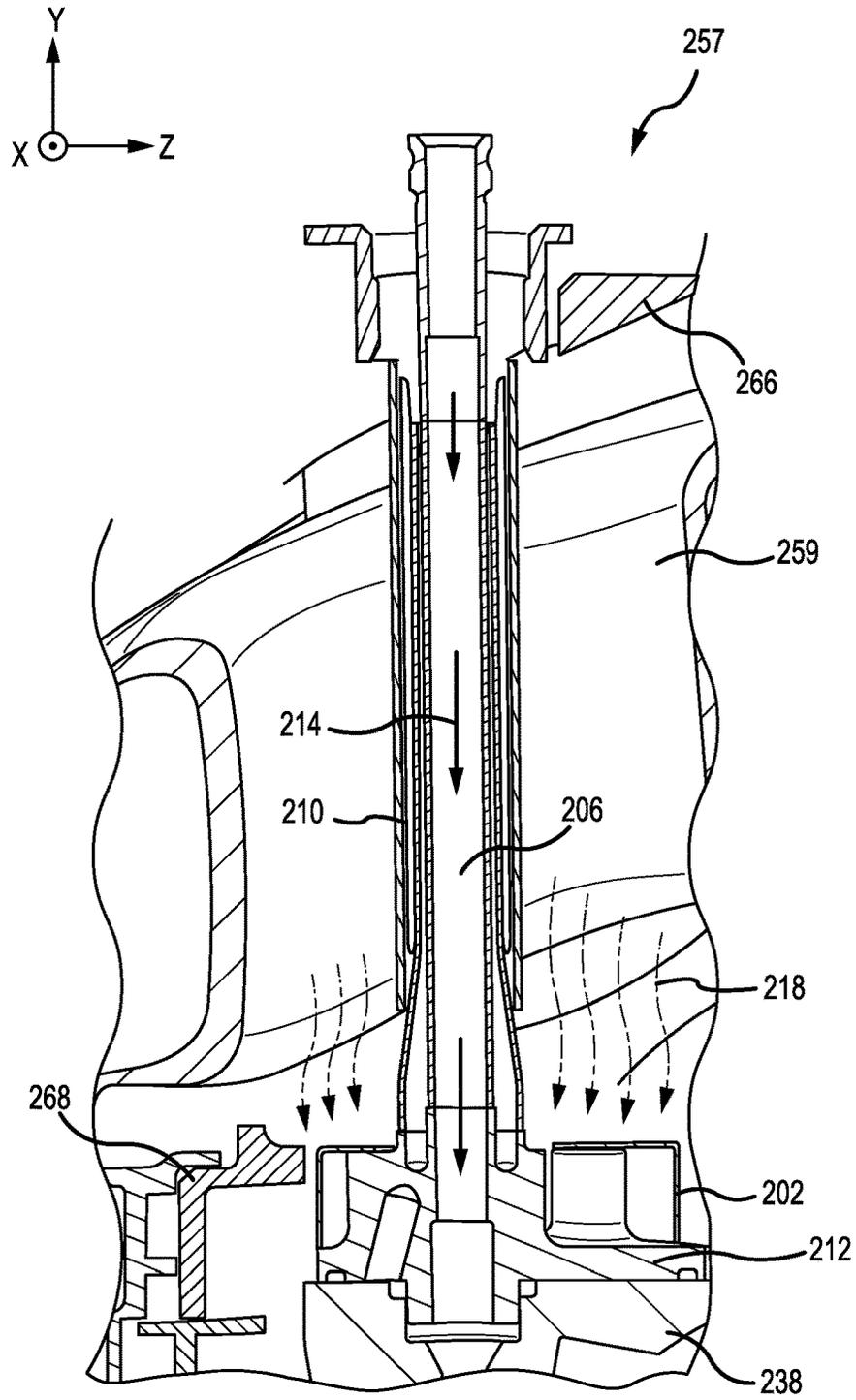


FIG.2A

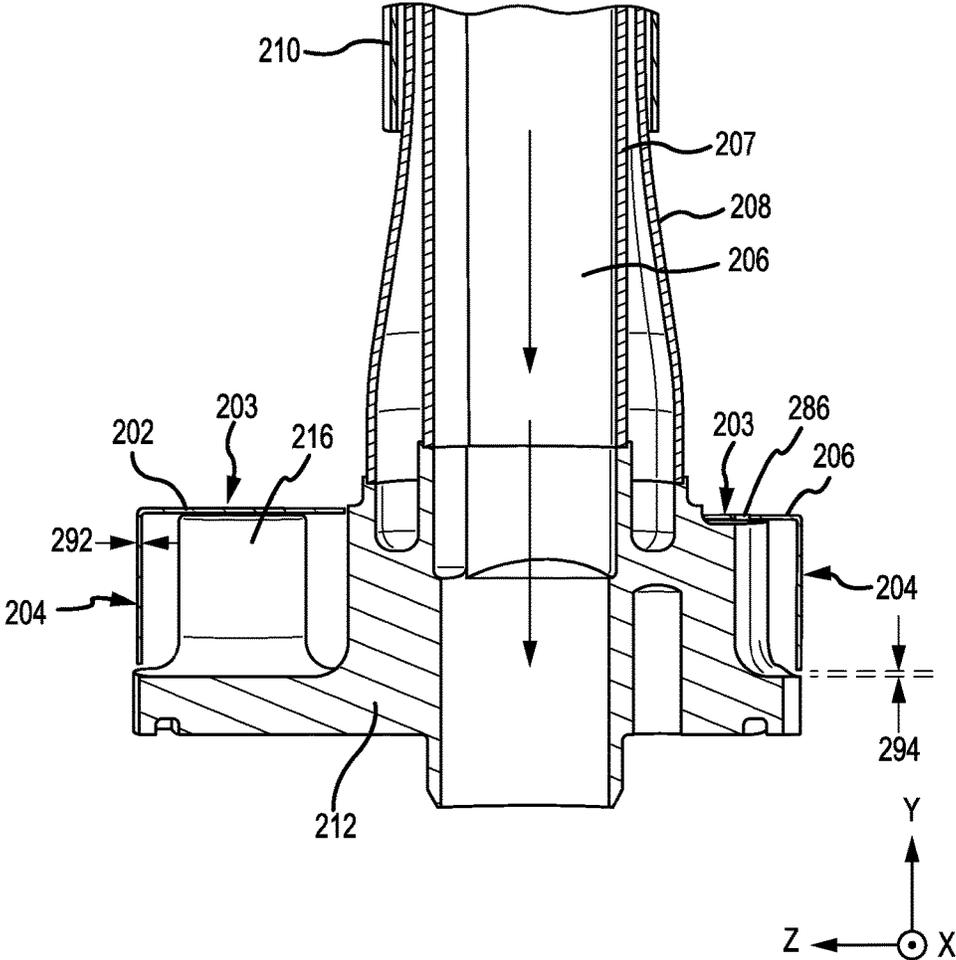


FIG. 2B

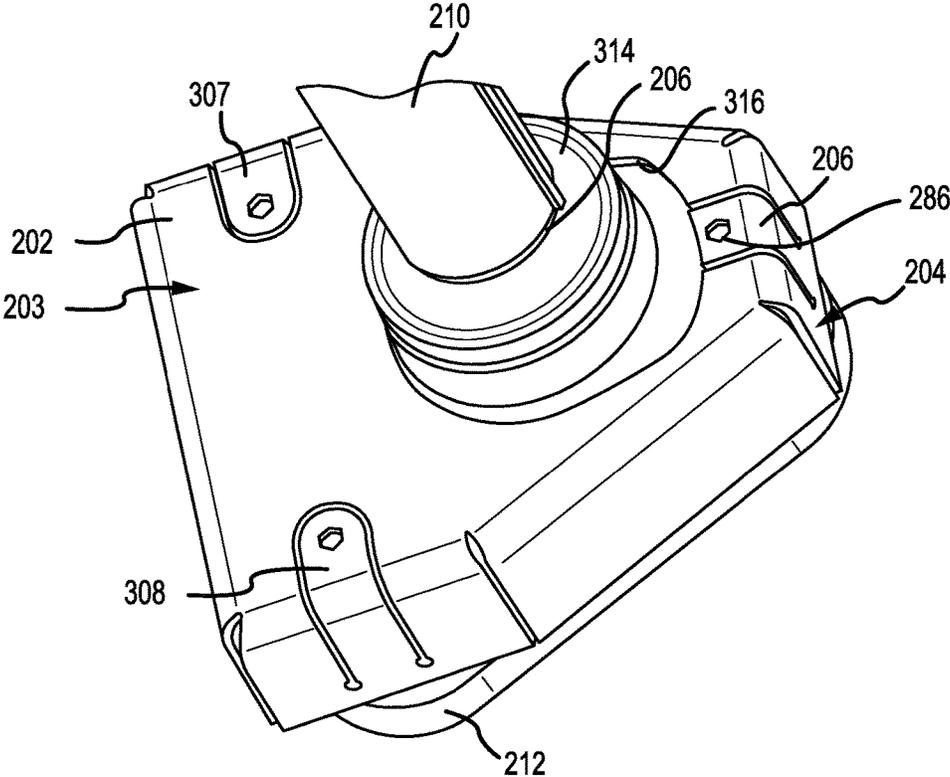


FIG.3A

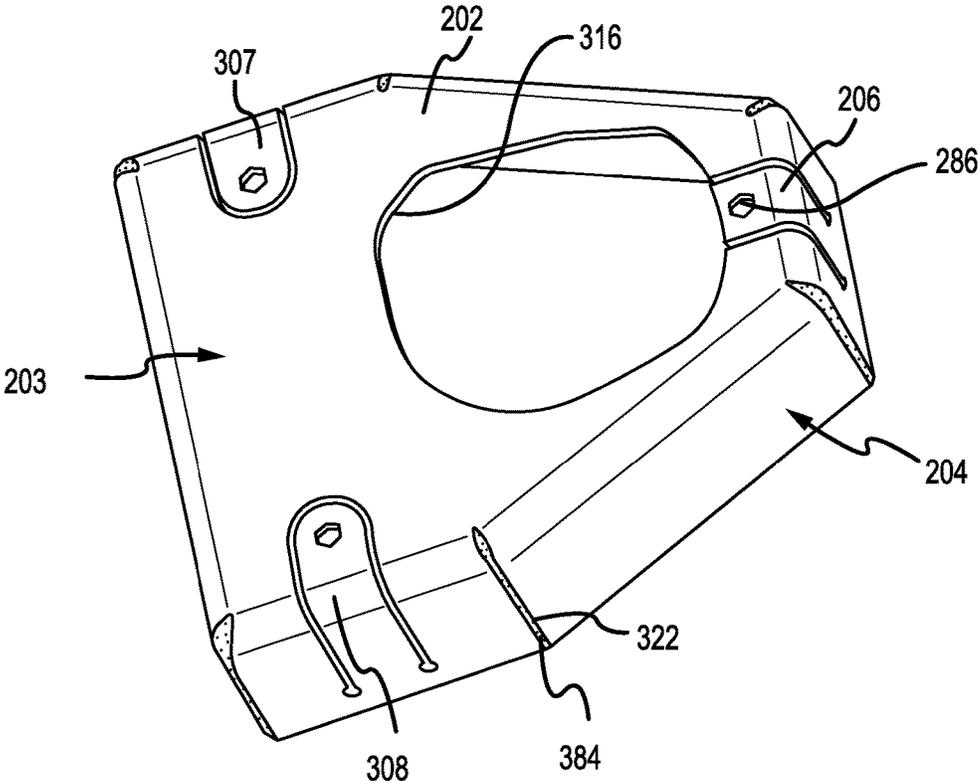


FIG.3B

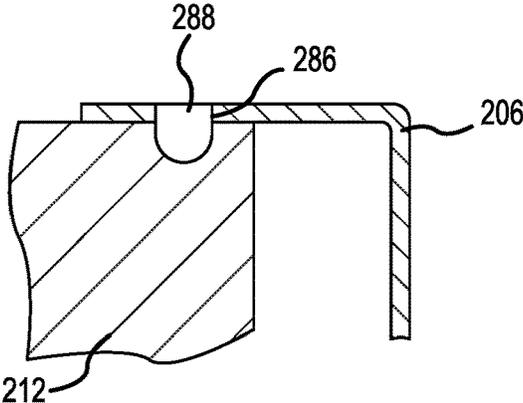


FIG.4A

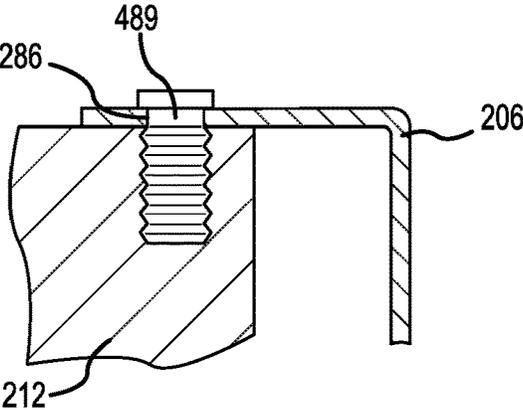


FIG.4B

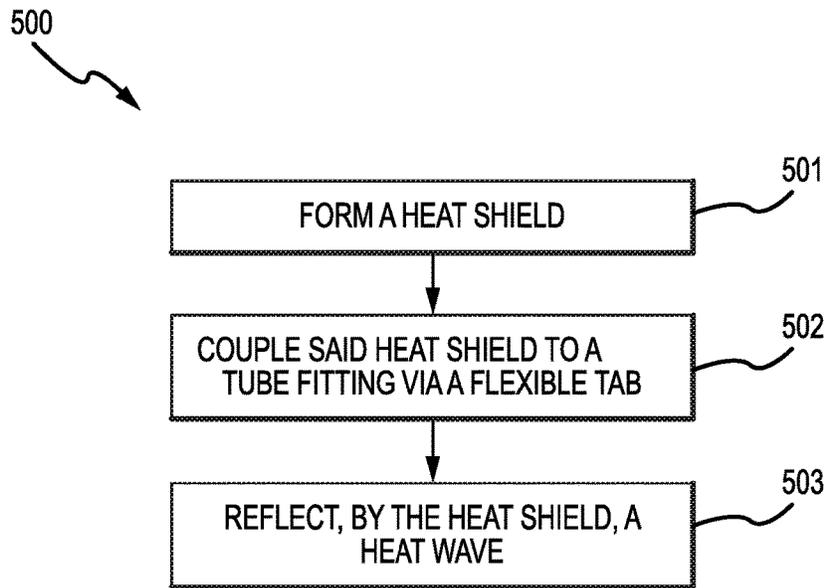


FIG.5

THERMALLY COMPLIANT HEATSHIELD**CROSS REFERENCE TO RELATED APPLICATIONS**

This application is a divisional of, and claims priority to, and the benefit of U.S. patent application Ser. No. 14/793,457, entitled "THERMALLY COMPLIANT HEATSHIELD," filed on Jul. 7, 2015, The '457 Application is hereby incorporated by reference in its entirety for all purposes.

FIELD

This disclosure relates to a gas turbine engine, and more particularly to heat shields for oil tube fittings.

BACKGROUND

Engine oil tubes and fittings may be subjected to relatively high temperatures. Once subjected to excessive heating, oil may undergo coking. Oil coking may cause solid oil deposits to form within oil tubes, causing undesirable effects such as blocked passageways and filters.

SUMMARY

A heat shield is described herein, in accordance with various embodiments. A heat shield may comprise a top portion, the top portion comprising an aperture, a sidewall portion, the sidewall portion extending at an angle between 80 degrees and 100 degrees from the top portion, and a flexible tab. The flexible tab may comprise an angle between 80 degrees and 100 degrees. The flexible tab may be fixed to the sidewall portion. The flexible tab may be configured to be attached to a fitting, wherein said flexible tab is configured to flex in response to at least one of an increase and decrease in temperature. The sidewall portion may bound a void. In various embodiments, the heat shield may be configured to at least partially encase the fitting. In various embodiments, the heat shield may be configured to impede heat transfer between the fitting and surrounding air. In various embodiments, the sidewall portion of the heat shield and the fitting may be separated by a gap. In various embodiments, the flexible tab may comprise an aperture, wherein the flexible tab is configured to be attached to the fitting via the aperture via at least one of a weld, solder, or braze. In various embodiments, the flexible tab may comprise an aperture, wherein the flexible tab is configured to be attached to the fitting via a plug weld via the aperture. In various embodiments, the heat shield may comprise at least one of a nickel-chromium based alloy and a stainless steel.

An assembly is described herein, in accordance with various embodiments. An assembly may include an oil tube, a fitting, wherein the fitting is configured to be attached to the oil tube, and a heat shield, wherein the heat shield is configured to be attached to the fitting. The heat shield may comprise a top portion, the top portion comprising an aperture, a sidewall portion, the sidewall portion extending at an angle between 80 degrees and 100 degrees from the top portion, the sidewall portion bounding a void, and a flexible tab. The flexible tab may comprise an angle between 80 degrees and 100 degrees. The flexible tab may be fixed to the sidewall portion, wherein the flexible tab is configured to be attached to a fitting. In various embodiments, the flexible tab may be configured to flex in response to at least one of an increase and decrease in temperature. In various embodi-

ments, the heat shield may be configured to at least partially encase the fitting. In various embodiments, the heat shield may be configured to prevent heat transfer between the fitting and surrounding air. In various embodiments, the sidewall portion and the fitting may be separated by a gap. In various embodiments, the tube may be a dual wall tube comprising an inner tube and an outer tube. In various embodiments, the sidewall portion may bound a hexagonal void. In various embodiments, the heat shield may be attached to the tube fitting via at least one of a plug weld and a fastener. In various embodiments, the heat shield may be manufactured via a brake bending process. In various embodiments, the heat shield may comprise at least one of a nickel-chromium based alloy and a stainless steel.

A method of cooling a tube fitting is disclosed herein, in accordance with various embodiments. The method of cooling a tube fitting may include forming a heat shield via a brake bending process, and coupling the heat shield to an outer surface of a tube fitting via a flexible tab, the heat shield at least partially encasing the tube fitting. The heat shield may comprise a top portion, a sidewall portion, and a flexible tab. The flexible tab may comprise an angle between 80 and 100 degrees. The flexible tab may be fixed to the sidewall portion. In various embodiments, the method may further comprise reflecting, by the heat shield, a heat wave away from the tube fitting. In various embodiments, the sidewall portion of the heat shield may be configured to be separated from the tube fitting by a gap.

Introducing a heat shield may prevent oil tube fittings from excessively heating, preventing oil coking.

The foregoing features and elements may be combined in various combinations without exclusivity, unless expressly indicated otherwise. These features and elements as well as the operation thereof will become more apparent in light of the following description and the accompanying drawings. It should be understood, however, the following description and drawings are intended to be exemplary in nature and non-limiting.

BRIEF DESCRIPTION OF THE DRAWINGS

FIG. 1 illustrates an example gas turbine engine, in accordance with various embodiments;

FIG. 2A illustrates a schematic view of an example mid-turbine frame assembly, in accordance with various embodiments;

FIG. 2B illustrates a schematic view of an oil tube fitting heat shield assembly, in accordance with various embodiments;

FIG. 3A illustrates a perspective view of an oil tube fitting heat shield assembly, in accordance with various embodiments;

FIG. 3B illustrates a perspective view of an oil tube fitting heat shield, in accordance with various embodiments;

FIG. 4A illustrates a cross-section view of a heat shield attached to an oil tube fitting via a plug weld, in accordance with various embodiments;

FIG. 4B illustrates a cross-section view of a heat shield attached to an oil tube fitting via a fastener, in accordance with various embodiments; and

FIG. 5 illustrates a method of cooling an oil tube fitting, in accordance with various embodiments.

DETAILED DESCRIPTION

The detailed description of exemplary embodiments herein makes reference to the accompanying drawings,

which show exemplary embodiments by way of illustration. While these exemplary embodiments are described in sufficient detail to enable those skilled in the art to practice the inventions, it should be understood that other embodiments may be realized and that logical changes and adaptations in design and construction may be made in accordance with this invention and the teachings herein. Thus, the detailed description herein is presented for purposes of illustration only and not of limitation. The scope of the invention is defined by the appended claims. For example, the steps recited in any of the method or process descriptions may be executed in any order and are not necessarily limited to the order presented. Furthermore, any reference to singular includes plural embodiments, and any reference to more than one component or step may include a singular embodiment or step. Also, any reference to attached, fixed, connected or the like may include permanent, removable, temporary, partial, full and/or any other possible attachment option. Additionally, any reference to without contact (or similar phrases) may also include reduced contact or minimal contact. Surface shading lines may be used throughout the figures to denote different parts but not necessarily to denote the same or different materials. In some cases, reference coordinates may be specific to each figure.

As used herein, "aft" refers to the direction associated with the tail (e.g., the back end) of an aircraft, or generally, to the direction of exhaust of the gas turbine. As used herein, "forward" refers to the direction associated with the nose (e.g., the front end) of an aircraft, or generally, to the direction of flight or motion.

As used herein, "distal" refers to the direction radially outward, or generally, away from the axis of rotation of a turbine engine. As used herein, "proximal" refers to a direction radially inward, or generally, towards the axis of rotation of a turbine engine.

In various embodiments and with reference to FIG. 1, a gas turbine engine 20 is provided. Gas turbine engine 20 may be a two-spool turbofan that generally incorporates a fan section 22, a compressor section 24, a combustor section 26 and a turbine section 28. Alternative engines may include, for example, an augmentor section among other systems or features. In operation, fan section 22 can drive air along a bypass flow-path B while compressor section 24 can drive air for compression and communication into combustor section 26 then expansion through turbine section 28. Although depicted as a turbofan gas turbine engine 20 herein, it should be understood that the concepts described herein are not limited to use with turbofans as the teachings may be applied to other types of gas turbine engines including three-spool architectures.

Gas turbine engine 20 may generally comprise a low speed spool 30 and a high speed spool 32 mounted for rotation about an engine central longitudinal axis A-A' relative to an engine static structure 36 via one or more bearing systems 38 (shown as bearing system 38-1 and bearing system 38-2). It should be understood that various bearing systems 38 at various locations may alternatively or additionally be provided, including for example, bearing system 38, bearing system 38-1, and bearing system 38-2.

Low speed spool 30 may generally comprise an inner shaft 40 that interconnects a fan 42, a low pressure (or first) compressor section 44 (also referred to a low pressure compressor) and a low pressure (or first) turbine section 46. Inner shaft 40 may be connected to fan 42 through a geared architecture 48 that can drive fan 42 at a lower speed than low speed spool 30. Geared architecture 48 may comprise a gear assembly 60 enclosed within a gear housing 62. Gear

assembly 60 couples inner shaft 40 to a rotating fan structure. High speed spool 32 may comprise an outer shaft 50 that interconnects a high pressure compressor 52 (e.g., a second compressor section) and high pressure (or second) turbine section ("HPT") 54. A combustor 56 may be located between high pressure compressor 52 and HPT 54. A mid-turbine frame 57 of engine static structure 36 may be located generally between HPT 54 and low pressure turbine 46. Mid-turbine frame 57 may support one or more bearing systems 38 in turbine section 28. Inner shaft 40 and outer shaft 50 may be concentric and rotate via bearing systems 38 about the engine central longitudinal axis A-A', which is collinear with their longitudinal axes. As used herein, a "high pressure" compressor or turbine experiences a higher pressure than a corresponding "low pressure" compressor or turbine.

The core airflow may be compressed by low pressure compressor 44 then high pressure compressor 52, mixed and burned with fuel in combustor 56, then expanded over HPT 54 and low pressure turbine 46. Mid-turbine frame 57 includes airfoils 59 which are in the core airflow path. Low pressure turbine 46 and HPT 54 rotationally drive the respective low speed spool 30 and high speed spool 32 in response to the expansion.

Gas turbine engine 20 may be, for example, a high-bypass geared aircraft engine. In various embodiments, the bypass ratio of gas turbine engine 20 may be greater than about six (6). In various embodiments, the bypass ratio of gas turbine engine 20 may be greater than ten (10). In various embodiments, geared architecture 48 may be an epicyclic gear train, such as a star gear system (sun gear in meshing engagement with a plurality of star gears supported by a carrier and in meshing engagement with a ring gear) or other gear system. Geared architecture 48 may have a gear reduction ratio of greater than about 2.3 and low pressure turbine 46 may have a pressure ratio that is greater than about 5. In various embodiments, the bypass ratio of gas turbine engine 20 is greater than about ten (10:1). In various embodiments, the diameter of fan 42 may be significantly larger than that of the low pressure compressor 44, and the low pressure turbine 46 may have a pressure ratio that is greater than about (5:1). Low pressure turbine 46 pressure ratio may be measured prior to inlet of low pressure turbine 46 as related to the pressure at the outlet of low pressure turbine 46 prior to an exhaust nozzle. It should be understood, however, that the above parameters are exemplary of various embodiments of a suitable geared architecture engine and that the present disclosure contemplates other gas turbine engines including direct drive turbofans.

In various embodiments, with reference to FIG. 2A, a mid-turbine frame (MTF) assembly is illustrated. An xyz coordinate axis is provided for ease of illustration. MTF assembly 257 may include bearing compartment 238, outer case 266, and inner case 268. MTF vane 259 may be located between inner case 268 and outer case 266. Oil tube fitting 212 may be attached to a portion of bearing compartment 238. Oil tube 206 may extend between outer case 266 and oil tube fitting 212. Oil 214 may be located within oil tube 206. Oil 214 may be used to lubricate at least a portion of bearing compartment 238. Oil tube 206 may be located at least partially within MTF vane 259. Sleeve 210 may encase at least a portion of oil tube 206. Oil tube fitting heat shield (also referred to herein as heat shield) 202 may be located between MTF vane 259 and oil tube fitting 212, whereby at least a portion of oil tube fitting 212 is located in the negative

y-direction from heat shield **202** and at least a portion of MTF vane **259** is located in the positive y-direction from heat shield **202**.

Hot exhaust may impinge on MTF vane **259** which may cause MTF vane **259** to increase in temperature due to convective heat transfer from the hot exhaust. The hot exhaust may be in the range from about 800° F. (427° C.) to about 1000° F. (538° C.) in one embodiment. Heat waves **218** may radiate from MTF vane **259**. In various embodiments, heat waves may radiate to other nearby components which may cause the nearby components to increase in temperature. In return, the nearby components may transfer heat to other adjacent components and/or fluids. For example, heat waves may radiate from MTF vane **259** to oil tube **206** and may convectively transfer heat from MTF vane **259** to oil tube **206**. Heat may be conductively transferred to oil located inside oil tube **206**. Furthermore, when oil exceeds various threshold temperatures, it may undergo severe oxidative and thermal breakdown which may cause solid deposits to form. These deposits may be undesirable as they may impede the flow of fluid through various components including, for example, tubes and filters. Heat shield **202** may be configured to block heat waves **218** radiating from MTF vane **259** from directly impinging on oil tube fitting **212**. Furthermore, heat shield **202** may help minimize convective heat transfer from hot air surrounding oil tube fitting **212**. Accordingly, heat shield **202** may prevent heat from being transferred to oil tube fitting **212**. In various embodiments, heat shield **202** may prevent oil from coking within oil tube fitting **212**. Sleeve **210** may be configured to block radiating heat waves from MTF vane **259** from impinging on oil tube **206**.

In various embodiments, with reference to FIG. 2B, oil tube **206** may comprise an inner tube **207** and an outer tube **208**. Accordingly, oil tube **206** may be referred to as a dual wall tube. Inner tube **207** may be enclosed by outer tube **208** which may be occupied by air. The outer tube **208** may be configured to contain oil within outer tube **208** in the event that there is an oil leak from inner tube **207**. Outer tube **208** may be configured to prevent heat transfer from surrounding hot air to inner tube **207**. Oil tube **206** may be configured to attach to oil tube fitting **212**. Oil tube **206** may be attached to oil tube fitting **212** via weld, solder, braze, or any other suitable method. Heat shield **202** may comprise a sidewall portion **204** and a top portion **203**. Sidewall portion **204** of heat shield **202** may be configured to at least partially encase oil tube fitting **212**. Top portion **203** of heat shield **202** may be configured to at least partially encase oil tube fitting **212**. Accordingly, heat shield **202** may be configured to at least partially encase oil tube fitting **212**. In various embodiments, sidewall portion **204** may extend in a direction that is normal to top portion **203**. In various embodiments, sidewall portion **204** may extend at an angle between eighty degrees and one hundred degrees (80°-100°) from top portion **203**. Accordingly, sidewall portion **204** and top portion **203** may form an angle between eighty degrees and one hundred degrees (80°-100°).

In various embodiments, oil tube fitting **212** may be separated from sidewall portion **204** of heat shield **202** by a gap **294**. Heat shield **202** may be configured to be attached to oil tube fitting **212** such that there is a gap **294** between sidewall portion **204** of heat shield **202** and oil tube fitting **212**. Sidewall portion **204** may be configured to be separated from oil tube fitting **212** by gap **294** such that a conductive thermal path does not exist between sidewall portion **204** and oil tube fitting **212**. Gap **294** may be configured to be

minimal while allowing thermal expansion of heat shield **202** and oil tube fitting **212** without creating a thermal conduction path between heat shield **202** and oil tube fitting **212**. In various embodiments, gap **294** may comprise a distance in a range from about 0.1 millimeters to about 7 millimeters and in various embodiments, in a range from about 1 millimeter to about 4 millimeters and in various embodiments, gap **294** may comprise a distance of about 2 millimeters. Minimizing gap **294** may allow heat shield **202** to more effectively minimize convective heat transfer between oil tube fitting **212** and surrounding hot air. Minimizing gap **294** may allow heat shield **202** to more effectively minimize convective heat transfer between oil tube fitting **212** and radiated heat from an adjacent MTF vane. Gap **294** may be configured to prevent a thermal conduction path between sidewall portion **204** and oil tube fitting **212** during thermal expansion of oil tube fitting **212** and/or heat shield **202**.

In various embodiments, various components of MTF assemblies may comprise various materials. Various components, including heat shield **202**, may comprise a high temperature metal (e.g., an austenitic nickel-chromium-based alloy such as that available under the trade name INCONEL), a high temperature composite, and/or the like. In various embodiments, heat shield **202** may comprise a high temperature stainless steel (e.g., type 330 stainless steel).

With reference to FIG. 3A and FIG. 3B, elements with like element numbering as depicted in FIG. 2A and FIG. 2B, are intended to be the same and will not be repeated for the sake of clarity.

In various embodiments, with reference to FIG. 2B and FIG. 3A, heat shield **202** may comprise flexible tab **206**. Flexible tab **206** may be fixed to sidewall portion **204**. Flexible tab **206** may comprise a right angle. Flexible tab **206** may comprise an angle between eighty degrees and one hundred degrees (80°-100°). The sidewall portion **204** of flexible tab **206** may extend in a direction that is normal to the top portion **203** of flexible tab **206**. Flexible tab **206** may comprise aperture **286**. In various embodiments, aperture **286** may be configured to accommodate a plug weld. In various embodiments, aperture **286** may be configured to accommodate a fastener. In various embodiments, top portion **203** of heat shield **202** may comprise aperture **316**. Aperture **316** may be configured to allow fitting post **314** to be located within aperture **316** when in the installed position. Aperture **316** may be configured to be large enough to allow various components such as fittings attached to oil tube **206** to slide through aperture **316** in response to sliding heat shield **202** over oil tube **206** to an installed position.

In various embodiments, with reference to FIG. 3B, heat shield **202** may be manufactured via a sheet metal brake bending process. In various embodiments, a brake bending process may include a piece of sheet metal that is cut into a desired shape via a two-axis laser, water jet, or the like. Flexible tab **206**, flexible tab **307**, flexible tab **308**, aperture **286**, aperture **316**, and any other apertures or features may also be cut into the sheet metal at this point. Next, the sheet metal may be bent along a straight line several times until the desired shape is formed, resulting in heat shield **202**. Finally, the newly formed edges, such as edge **322** for example, may be welded together, via weld **384** for example, to fill the gap that may exist between the newly formed edges of sidewall portion **204**. Accordingly, no gap

In various embodiments, with reference now to FIG. 2B, heat shield **202** may comprise a wall thickness **292**. In various embodiments, heat shield **202** may be manufactured

via a hydro-forming process. Wall thickness **292** may be chosen according to various design considerations. In various embodiments, wall thickness **292** may be between 0.010 in (0.25 mm) and 0.030 in (0.76 mm) in thick. During manufacturing, sheet metal of a preferred wall thickness may be chosen to be hydro-formed to the desired heat shield geometry. For example, if a heat shield comprising a wall thickness of 0.5 mm is desired, a piece of sheet metal comprising a wall thickness of 0.5 mm may be used and formed into the desired geometry using high pressure hydraulic fluid to press the sheet metal into a die in a process known as hydro-forming. In various embodiments, a single piece of sheet metal may be hydro-formed into heat shield **202**. In various embodiments, two or more pieces of sheet metal may be hydro-formed into different geometries and welded together to form heat shield **202**.

With reference now to FIG. 2B and FIG. 3A, heat shield **202** may further comprise flexible tab **307** and flexible tab **308**, according to various embodiments. Accordingly, heat shield **202** may comprise a plurality of flexible tabs. In various embodiments, flexible tab **307** and flexible tab **308** may be similar to flexible tab **206**. In various embodiments, oil tube fitting **212** may comprise a plurality of attachment posts, such as attachment post **216**, for example. In various embodiments, flexible tab **307** may be configured to be attached to attachment post **216**. In various embodiments, flexible tab **308** may be similar to flexible tab **307**. Flexible tab **308** may be configured to be attached to an attachment post located adjacent to attachment post **216**.

With reference to FIG. 4A and FIG. 4B, elements with like element numbering as depicted in FIG. 2A through FIG. 3B, are intended to be the same and will not be repeated for the sake of clarity.

With reference now to FIG. 4A, flexible tab **206** may be configured to attach to oil tube fitting **212**. In various embodiments, flexible tab **206** may be configured to attach to oil tube fitting **212** via weld, solder, braze, or any other suitable method. In various embodiments, flexible tab **206** may be configured to attach to oil tube **206** via a plug weld **288**. In various embodiments, flexible tab **206** may be attached to oil tube fitting **212**, whereby a plug weld **288** is applied into aperture **286**, thereby attaching flexible tab **206** to oil tube fitting **212**.

With reference now to FIG. 4B, flexible tab **206** may be configured to attach to oil tube fitting **212** via a fastener **489**, such as a screw or a bolt for example. In various embodiments, fastener **489** may be inserted into aperture **286** and threadingly attach to oil tube fitting **212**, thereby attaching flexible tab **206** to oil tube fitting **212**.

In various embodiments, with further reference to FIG. 2B and FIG. 3A, flexible tab **307** may be attached to oil tube fitting **212** via attachment post **216** in a similar manner as flexible tab **206**. Flexible tab **308** may be attached to oil tube fitting **212** in a similar manner as flexible tab **307**. In various embodiments, flexible tab **206**, flexible tab **307**, and flexible tab **308** may be configured to flex or bend relative to heat shield **202** during thermal expansion and/or contraction of oil tube fitting **212** and/or heat shield **202**, thereby preventing heat shield **202** from bending or distorting during such event. Accordingly, flexible tabs **206**, **307**, and **308** may prevent heat shield **202** from cracking or breaking during operation, thus preventing cooling efficiency of heat shield **202** from decreasing during a thermal expansion and/or contraction event. Accordingly, flexible tabs **206**, **307**, and **308** may be configured to flex in response to an increase and/or decrease in temperature.

In various embodiments, the sidewall portion **204** of heat shield **202** may comprise a hexagonal geometry. For example, FIG. 3B illustrates the sidewall portion **204** of heat shield **202**, wherein the sidewall portion **204** bounds a hexagonal void. In various embodiments, the sidewall portion **204** of heat shield **202** may comprise one of a square, rectangular, pentagon, heptagon, octagon, oblong, round, elliptical, or any other geometry. The geometry of the sidewall portion **204** of heat shield **202** may be driven by the geometry of oil tube fitting **212**. Accordingly, the geometry of oil tube fitting **212** and sidewall portion **204** may be complementary.

With reference to FIG. 5, a method **500** of cooling a tube fitting is disclosed herein, in accordance with various embodiments. The method **500** of cooling a tube fitting may include forming a heat shield in step **501**. Step **502** may include coupling said heat shield to a tube fitting via a flexible tab. Step **503** may include reflecting, by the heat shield, a heat wave. With further reference to FIG. 2A and FIG. 3A, step **501** may include forming heat shield **202** via the brake bending process as described herein. Step **502** may include coupling heat shield **202** to an outer surface of oil tube fitting **212** via flexible tab **206**, **307**, and/or **308**. Heat shield **202** may at least partially encase oil tube fitting **212**. Step **503** may include reflecting, by heat shield **202**, a heat wave **218** away from oil tube fitting **212**.

Benefits, other advantages, and solutions to problems have been described herein with regard to specific embodiments. Furthermore, the connecting lines shown in the various figures contained herein are intended to represent exemplary functional relationships and/or physical couplings between the various elements. It should be noted that many alternative or additional functional relationships or physical connections may be present in a practical system. However, the benefits, advantages, solutions to problems, and any elements that may cause any benefit, advantage, or solution to occur or become more pronounced are not to be construed as critical, required, or essential features or elements of the inventions. The scope of the inventions is accordingly to be limited by nothing other than the appended claims, in which reference to an element in the singular is not intended to mean "one and only one" unless explicitly so stated, but rather "one or more." Moreover, where a phrase similar to "at least one of A, B, or C" is used in the claims, it is intended that the phrase be interpreted to mean that A alone may be present in an embodiment, B alone may be present in an embodiment, C alone may be present in an embodiment, or that any combination of the elements A, B and C may be present in a single embodiment; for example, A and B, A and C, B and C, or A and B and C. Systems, methods and apparatus are provided herein. In the detailed description herein, references to "one embodiment", "an embodiment", "various embodiments", etc., indicate that the embodiment described may include a particular feature, structure, or characteristic, but every embodiment may not necessarily include the particular feature, structure, or characteristic. Moreover, such phrases are not necessarily referring to the same embodiment. Further, when a particular feature, structure, or characteristic is described in connection with an embodiment, it is submitted that it is within the knowledge of one skilled in the art to affect such feature, structure, or characteristic in connection with other embodiments whether or not explicitly described. After reading the description, it will be apparent to one skilled in the relevant art(s) how to implement the disclosure in alternative embodiments.

Furthermore, no element, component, or method step in the present disclosure is intended to be dedicated to the public regardless of whether the element, component, or method step is explicitly recited in the claims. No claim element herein is to be construed under the provisions of 35 U.S.C. 112(f), unless the element is expressly recited using the phrase “means for.” As used herein, the terms “comprises”, “comprising”, or any other variation thereof, are intended to cover a non-exclusive inclusion, such that a process, method, article, or apparatus that comprises a list of elements does not include only those elements but may include other elements not expressly listed or inherent to such process, method, article, or apparatus.

What is claimed is:

1. A heat shield comprising:

- a top portion, the top portion comprising an aperture;
- a sidewall portion, the sidewall portion extending at an angle between 80 and 100 degrees from the top portion, the sidewall portion bounding a void; and
- a flexible tab cut into the top portion and the sidewall portion, the flexible tab comprising an angle between 80 and 100 degrees, the flexible tab being fixed to the sidewall portion, the flexible tab being configured to be

attached to a fitting, wherein said flexible tab is configured to flex in response to at least one of an increase and a decrease in temperature.

2. The heat shield of claim 1, wherein the heat shield is configured to at least partially encase the fitting.

3. The heat shield of claim 2, wherein the heat shield is configured to impede heat transfer between the fitting and surrounding air.

4. The heat shield of claim 3, wherein the sidewall portion of the heat shield and the fitting are separated by a gap.

5. The heat shield of claim 1, wherein the flexible tab comprises a tab aperture, wherein the flexible tab is configured to be attached to the fitting via the tab aperture via at least one of a weld, solder, or braze.

6. The heat shield of claim 1, wherein the flexible tab comprises a tab aperture, wherein the flexible tab is configured to be attached to the fitting via a plug weld via the tab aperture.

7. The heat shield of claim 1, wherein the heat shield comprises at least one of a nickel-chromium based alloy or a stainless steel.

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