Abstract: The invention refers to a method for producing a near-surface cooling passage (17) in a thermally highly stressed component (14), which comprises the following steps: a) providing a component (14) which has a surface (18) on a hot side in a region which is to be cooled; b) letting a channel (19) into the surface (18); c) inserting a cooling tube (20) into the channel (19); d) filling the channel (19), with the cooling tube (20) inserted, with a temperature-resistant filling material (21) in such a way that the inserted cooling tube (20) is embedded into the filling material (21), leaving free an inlet (17i) and an outlet (17o); and e) covering the channel (19), with the cooling tube (20) embedded, with an anti-oxidation, temperature-stable cover layer (22). The method is inexpensive and can be used in a flexible manner in the most diverse situations in order to save cooling medium or to reduce the thermal load.
METHOD FOR PRODUCING A NEAR-SURFACE COOLING PASSAGE IN A THERMALLY HIGHLY STRESSED COMPONENT, AND COMPONENT HAVING SUCH A PASSAGE

TECHNICAL FIELD

The present invention relates to the field of thermal machines. It refers to a method for producing a near-surface cooling passage in a thermally highly stressed component according to the preamble of claim 1. It also refers to a component which is produced according to the method.

BACKGROUND OF THE INVENTION

In thermal machines, efficiency which is as high as possible has always been the target in order to use the applied fuels more effectively for power generation. In the case of gas turbines, the aim is an efficiency of 63%, for example, for which higher combustion temperatures in the region of 1850K would be required. In order to achieve this, thermally highly loaded components of the machine have to be cooled by means of complex cooling devices and configurations. On account of the increasing complexity, problems in the production of such components increase and lead to high scrap rates.
In the case of gas turbines, on account of an irregular profile of the combustion chamber exit temperature, critical hot zones in the subsequently arranged components, such as stator blades or rotor blades or wall elements of the hot gas passage, occur, resulting in local overheating so that in such components working temperatures which are approximately 80 - 130K higher than the hot gas temperature are to be taken into consideration in the future.

For this reason, very efficient local cooling of the thermally highly loaded components is required in the case of gas turbines and comparable thermal machines.

One possible way, in which such efficient local cooling can be developed, is near-surface or near-wall cooling which is shown in two variants in Figs. 1 and 2. The component 10' (tubular in the example) from Fig. 1 has a wall 11 with a thickness t which is 4 mm, for example. Hot gas impinges upon the component 10' from the outside (block arrow). Cooling medium, mostly air or steam, flows through the interior space 12 of the component 10' and at least partially dissipates the externally introduced heat from the wall 11.

An improved alternative cooling configuration is reproduced in Fig. 2 for the component 10. In this case, parallel cooling passages 13, through which flows cooling medium, with an inside diameter d1 of 1 mm, for example, extend directly in the wall 11 and are only at a distance d2 of 0.5 mm, for example, from the outer surface of the wall 11.

A transition from the configuration in Fig. 1 to the configuration of Fig. 2 enables a reduction of the cooling medium mass flow by 40 - 55%, or an increase of the hot gas temperatures by 50 - 125K, on account of the reduced distance between cooling medium and hot gas.
Such a configuration can be achieved in components with effusion cooling in the following way: the basis is a component which according to Fig. 3 has an effusion-cooled component wall 14' (with a thickness of 2.0 mm - 5.3 mm, for example) through which oblique cooling holes 15 (with an inside diameter of 0.8 mm, for example) extend from a cool side CS of the component wall 14' to a hot side HS, through which cooling holes cooling medium 16 flows and discharges on the thermally loaded surface 18.

In the case of a component according to Fig. 4 with a comparable wall 14, instead of cooling holes 15 cooling passages 17 are formed in the component wall 14 and with an inside diameter of 1.0 mm, for example, comprise a plurality of sections 17a, 17b and 17c. The first passage section 17a extends from the inlet on the cool side CS into the interior of the component wall 14. A second passage section 17b adjoins the first passage section 17a and (in the manner of the cooling passages 13 in Fig. 2) extends essentially parallel (at a distance of 0.6 mm, for example) to the surface 18 which is to be cooled. A third passage section 17c then adjoins the second cooling passage 17b and terminates in an outlet on the hot side HS. The first passage section 17a and the third passage section 17c are oriented obliquely to the surface 18 in this case (similar to the cooling holes 15 in Fig. 3).

A cooling configuration of the type shown in Fig. 4, as near-surface or near-wall cooling, would bring significant advantages compared with conventional cooling configurations.

Such a cooling configuration, however, poses problems with regard to the difficulties related to production engineering, which lead to high costs and high scrap rates.

It is certainly conceivable to realize such cooling configurations by casting methods in the hollow core technique. In this case, after the casting of the component the core forming the network of internal cooling passages is removed.
The remaining cavities form the passages. Although this method is practical as regards production engineering, it is expensive owing to the complexity and is afflicted with high scrap rates. Furthermore, a component cannot be reworked with this technology or be subsequently altered.

SUMMARY OF THE INVENTION

It is therefore an object of the invention to disclose a method for producing near-surface cooling passages for thermally loaded components of a thermal machine, especially of a gas turbine, which method can be applied to different components and is to be carried out at comparatively low cost and with a low scrap rate, even in retrospect on already existing components, and provides components with significantly improved cooling effect and correspondingly increased service life.

It is also an object of the invention to disclose a corresponding component.

These and other objects are achieved by the total features of claims 1 and 13.

The method according to the invention for producing a near-surface cooling passage in a thermally highly stressed component comprises the following steps:

a) providing a component which has a surface on a hot side in a region which is to be cooled;
b) letting at least one channel into this surface;
c) inserting a cooling tube into the channel;
d) filling the channel, with the cooling tube inserted, with a temperature-resistant filling material in such a way that the inserted cooling tube is embedded into the filling material, leaving free an inlet and an outlet; and
e) covering the channel, with the cooling tube embedded, with an anti-oxidation, temperature-stable cover layer.
One embodiment of the method according to the invention is characterized in that in step (b) the channel in the component is hollowed out by means of a material-removing process.

In this case, the channel can especially be hollowed out in the component by spark erosion by means of an EDM electrode.

The EDM electrode in its shape preferably corresponds to the channel which is to be hollowed out.

Another embodiment of the method according to the invention is characterized in that the component has a wall with a hot side and an oppositely disposed cool side, and in that the channel is introduced into the component wall in such a way that it extends through the wall from the cool side towards the hot side and has an inlet on the cool side and an outlet on the hot side.

It is especially favorable in this case if the channel, and consequently also the finished cooling passage, comprise a first passage section which extends from the inlet on the cool side into the interior of the component wall, a second passage section which adjoins the first passage section and extends essentially parallel to the surface which is to be cooled, and a third passage section which adjoins the second passage section and terminates in the outlet on the hot side.

The first cooling passage and the third cooling passage are preferably oriented obliquely to the surface, that is to say at an acute angle.

In this case, the cooling passage can especially have an inside diameter of approximately 1 mm and the second passage section can be at a distance which is less than or equal to 1 mm from the surface which is to be cooled.

A further embodiment of the method according to the invention is characterized in that the channel is let into the component to such a depth, or hollowed out from
the component to such a depth, that the inserted cooling tube, apart from inlet and outlet, is located well below the surface.

Another embodiment of the method according to the invention is characterized in that the channel, with the cooling tube inserted, is filled with a high-temperature solder as filling material.

Yet another embodiment of the method according to the invention is characterized in that the anti-oxidation, temperature-stable cover layer is applied by deposition welding by means of a laser metal forming process (LMF). In this case, the cover layer is preferably formed by consecutive application of a plurality of overlapping coatings.

Thermal spraying constitutes an alternative preferred coating process.

The thermally highly stressed component according to the invention, having a hot side delimited by a surface and at least one near-surface cooling passage, is characterized in that the cooling passage is produced by a method according to the invention.

One embodiment of the component according to the invention is characterized in that the component has a wall with a hot side and an oppositely disposed cool side, and in that the cooling passage extends through the component wall from the cool side to the hot side and has an inlet on the cool side and an outlet on the hot side.

Another embodiment of the component according to the invention is characterized in that the cooling passage comprises a first passage section which extends from the inlet on the cool side into the interior of the component wall, a second passage section which adjoins the first passage section and extends essentially parallel to the surface which is to be cooled, and a third passage section which adjoins the second passage section and terminates in the outlet on the hot side.
The first passage section and the third passage section are especially oriented obliquely to the surface and preferably include an angle of between $15^\circ$ and $30^\circ$, especially preferably an angle of approximately $18^\circ$, with the surface normal.

A further embodiment of the component according to the invention is characterized in that the cooling passage has a cooling tube which lies in a channel let into the surface and is embedded into a temperature-resistant filling material, especially a high-temperature solder.

The cooling tube preferably has an inside diameter of approximately 1 mm and an outside diameter of approximately 1.5 mm, and the second passage section is at a distance which is less than or equal to 1 mm from the surface which is to be cooled.

Another embodiment of the component according to the invention is characterized in that the cooling passage has a length of approximately 20 mm.

Yet another embodiment of the component according to the invention is characterized in that a plurality of cooling passages are arranged in the component in parallel and/or in series and at a distance from each other. In this case, cooling medium can flow through the plurality of cooling passages in the same or opposite directions.

Other cooling arrangements, with differently oriented or dimensioned cooling passages, which are optimally adapted to the cooling requirements of the component, are also conceivable.

**BRIEF EXPLANATION OF THE FIGURES**

The invention shall subsequently be explained in more detail based on exemplary embodiments in conjunction with the drawing. In the drawing
Fig. 1 shows in cross section a tubular component in which the thermally loaded wall is cooled by means of cooling medium flowing inside the tube;

Fig. 2 shows in cross section and in an enlarged detail a tubular component in which the thermally loaded wall is cooled close to the surface by means of cooling passages extending inside the wall;

Fig. 3 shows the section through a component wall with cooling passages for conventional effusion cooling;

Fig. 4 shows in a view comparable to Fig. 3 a component wall with near-surface cooling passages in addition to effusion cooling;

Fig. 5 shows in a view comparable to Fig. 4 a component wall with near-surface cooling passages, according to an exemplary embodiment of the invention;

Fig. 6 shows the section through a cooling passage from Fig. 5 in the plane VI - VI;

Fig. 7 shows in a photographic representation various steps for producing near-surface cooling passages in a plate-like component, according an exemplary embodiment of the invention;

Fig. 9 shows in a perspective side view an example of an EDM electrode which can be used in the invention;
Fig. 10 shows the inserting of correspondingly bent tubes into the channels which have been hollowed out in the component, according to another exemplary embodiment of the invention;

Fig. 11 shows in a view comparable to Fig. 6 a plurality of steps during the production of the cover layer by means of deposition welding (LMF), according to another exemplary embodiment of the invention; and

Fig. 12 shows an exemplary embodiment for a component according to the invention in the form of a stator blade with cooling passages introduced into the leading edge of the blade airfoil, according to the invention.

WAYS OF IMPLEMENTING THE INVENTION

The invention discloses a new alternative to already known production methods for near-surface cooling configurations. Instead of attempting to form corresponding cooling passages in the base material or to form cooling passages by the combination of two or more parts, the subsequently explained solution for producing near-surface or near-wall cooling passages is based on the embedding of complete passages into the surface of the component.

A sequence of production steps for this method comprises the following: in a first step, the base material is prepared in a suitable manner, especially by hollowing out a channel, in order to accommodate a tube which is later let into the surface. The configuration of such a channel can be straight, but other configurations, such as meander configurations, are also conceivable in order to optimize the cooling effect in a specific manner depending upon the application case.
The channels are usually introduced into the component or into the wall from the hot gas side or hot side (see Fig. 7(a)). It is also conceivable, however, to introduce the channels from the other side if this location is accessible for the machine being used. In parallel with the introduction of the channel(s), passage inserts in the form of closed bodies, preferably in the form of tubes with an inside diameter of approximately 1 mm and outside diameters of between 1.5 mm and 2.5 mm, are prefabricated. A round cross-sectional shape assists in minimizing crack development.

The tubes are then introduced into the channels in the component or in the component wall which is to be cooled (see Figs. 7(b) and 10). The introduction of closed forms, such as tubes, ensures stabilization of the molten pool during the later deposition welding of the cover layer.

For fixing the tubes in the channel and for achieving an optimum heat transfer, the tubes are embedded into a filling material, especially in the form of a high-temperature solder, in the channel and the surface is smoothed off by means of grinding (see Fig. 7(c)).

Finally, an anti-oxidation cover layer is applied by means of laser metal forming (LMF) or by means another coating process (see Figs. 7(d) and 11). For final thermal insulation, a thermal barrier coating (TBC) can also be applied on top of it.

The ends of the inserted tubes form an inlet and an outlet for the through-flowing cooling air. It is of great importance, therefore, that these openings are not closed off or constricted during the embedding with high-temperature solder.

In a view comparable to Fig. 4, Fig. 5 shows a component wall with near-surface cooling air passages according to an exemplary embodiment of the invention. Fig. 6 shows the section through a cooling passage from Fig. 5 in the plane VI - VI. A cooling passage 17, which comprises a plurality of sections 17a, 17b and 17c, extends through the component wall 14 of Fig. 5, and cooling medium 16, for
example cooling air 16, flows through the cooling passage during operation from an inlet 17i on the cool side to an outlet 17o on the hot side and discharges there on the thermally loaded surface 18.

The cooling passage 17 is formed essentially by a cooling tube 20 which is inserted into a channel 19 introduced into the component wall 14 and embedded there into a filling material 21 consisting of high-temperature solder. A cover layer 22 consisting of oxidation-resistant material is applied on top of the (smoothed) layer of filling material 21 by means of LMF. The cross section of the arrangement is reproduced in Fig. 6. The round cross-sectional geometry of the tube 20 is less susceptible to crack development.

The cooling passage 17 does not have any undercuts. The inside diameter of the cooling tube 20 is, for example, 1.0 mm and the outside diameter is 1.5 mm. The center passage section 17b extends parallel to the surface 18, whereas the passage sections 17a and 17c are oriented obliquely to the surface normal by an angle of approximately 18°. The length of the cooling passage 17 is approximately 20 mm. The depth of the channel 19 in the center passage section 17b is approximately 1.6 mm. The tube 20 extends at least over the center passage section 17b and the passage section 17c on the hot side, as shown in Fig. 5. It can also extend, however, over a part of, or the entirety of, the passage section 17a on the cold side.

Fig. 7 shows in a photographic representation various steps (a) to (e) for producing near-surface cooling passages in a plate-form component according to exemplary embodiments of the invention. Fig. 7(a) shows the channels 24 or 29 which are introduced into the components 23 or 28 by means of EDM. Correspondingly formed cooling tubes 25 or 30 are then introduced (inserted) into these channels 24, 29 according to Fig. 7(b). The inserted tubes are then embedded into high-temperature solder according to Fig. 7(c) and the surface in the region of the filled channels is ground smooth. The remaining outlets 26 or 31 of the cooling passages are clearly visible. Finally, an oxidation-resistant cover
layer 27 or 32 consisting of suitable material is applied in overlapping widths by means of LMF according to Fig. 7(d).

For introducing the channels (19 in Figs. 5, 6) into the surface of the component, use is made of an EDM electrode 33 according to Fig. 9, having a plurality of electrode sections 33a - c which correspond to the subsequent passage sections 17a - c. With such an electrode, the channels are hollowed out by means of countersink erosion. In conformance with the configuration of the channels 35, comprising three sections, in a component 34, the cooling tubes 36 which are to be inserted are also divided into three sections 36a - c according to Fig. 10.

The application of the cover layer 22 by means of LMF is carried out according to Fig. 11 preferably by overlapping, consecutive application of cover layer coatings 1-R to 3-C. In a first step (Fig. 11(a)), a first right-hand cover layer coating 1-R is applied. In a second step (Fig. 11(b)), a first left-hand cover layer coating 1-L is applied in an overlapping manner. In further steps (Fig. 11(c)), further right-hand and left-hand cover layer coatings 2-RR and 2-LL and a third central cover layer coating 3-C are then applied.

As an exemplary embodiment of a component according to the invention, Fig. 12 finally shows a stator blade 43 of a gas turbine, which stator blade has a cooled blade airfoil 38 between a lower platform 39 and an upper platform 40, the blade airfoil having a trailing edge 41 and leading edge 42. In the leading edge 42, instead of simple effusion cooling holes, parallel cooling passages 44, which are offset in relation to each other in a plurality of rows, are arranged according to the invention. With regard to the flow direction of the cooling medium, in this case the cooling passages 44 of adjacent rows, also such a row itself, can be differently oriented corresponding to the requirements of the specific individual case. As a result of this, some of the cooling medium flowing through the blade can be saved with cooling remaining constant.
Overall, using the method according to the invention a near-surface or near-wall cooling passage of any shape can be arranged on any customarily convection-cooled hot gas surface in order to improve the cooling effect and to save cooling medium. If necessary, larger surfaces can also be equipped with such cooling passages. The described technology can also be applied if a component has to be reconditioned or if an existing component has to be improved or replaced.

The invention has a number of advantages:

• The near-wall cooling system can be used locally in hot zones;
• It can be introduced from the hot outer side;
• Already installed components can be reworked (retrofit);
• The production method enables reconditioning of used components;
• The high cooling effect reduces the consumption of cooling medium;
• Under certain conditions, the hot gas temperature in the machine can be increased;
• The method is a favorable alternative to double-wall casting; and
• The shape of the introduced cooling passages minimizes the risk of crack development.

LIST OF DESIGNATIONS

10, 10' Component (e.g. tube)
11 Wall
12 Interior space
13 Cooling passage (near-surface)
14, 14' Component
15 Cooling hole
16 Cooling medium, for example air
17 Cooling passage (near-surface)
17a - c Passage section
17i Inlet
Outlet
Surface
Channel
Cooling tube
Filling material (e.g. high-temperature solder)
Cover layer (e.g. deposition welded)
Component
Channel
Cooling tube
Outlet
Cover layer
EDM electrode
Electrode section
Tube section
LMF device
Blade airfoil
Platform
Trailing edge
Leading edge
Stator blade (e.g. gas turbine)
Cooling passages
Inside diameter
Distance
Hot side
Cool side
Wall thickness
Cover layer coating
Cover layer coating
Cover layer coating
PATENT CLAIMS

1. A method for producing a near-surface cooling passage (17, 44) in a thermally highly stressed component (14, 23, 28, 34), characterized by the following steps:
   a) providing a component (14, 23, 28, 34) which has a surface (18) on a hot side (HS) in a region which is to be cooled;
   b) letting a channel (19, 24, 29, 35) into the surface (18);
   c) inserting a cooling tube (20, 25, 30, 36) into the channel (19, 24, 29, 35);
   d) filling the channel (19, 24, 29, 35), with the cooling tube (20, 25, 30, 36) inserted, with a temperature-resistant filling material (21) in such a way that the inserted cooling tube (20, 25, 30, 36) is embedded into the filling material (21), leaving free an inlet (17i) and an outlet (17o, 26, 31); and
   e) covering the channel (19, 24, 29, 35), with the cooling tube (20, 25, 30, 36) embedded, with an anti-oxidation, temperature-stable cover layer (22, 27, 32).

2. The method as claimed in claim 1, characterized in that in step (b) the channel (19, 24, 29, 35) in the component (14, 23, 28, 34) is hollowed out by means of a material-removing process.

3. The method as claimed in claim 2, characterized in that the channel (19, 24, 29, 35) is hollowed out in the component (14, 23, 28, 34) by spark erosion by means of an EDM electrode (33).

4. The method as claimed in claim 3, characterized in that the EDM electrode (33) in its shape corresponds to the channel (19, 24, 29, 35) which is to be hollowed out.

5. The method as claimed in one of claims 2 - 4, characterized in that the component (14, 23, 28, 34) has a wall (14) with a hot side (HS) and an oppositely
disposed cool side (CS), and in that the channel (19, 24, 29, 35) is introduced into the component wall (14) in such a way that it extends through the wall (14) from the cool side (CS) to the hot side (HS) and has an inlet (17i) on the cool side (CS) and an outlet (17o) on the hot side (HS).

6. The method as claimed in claim 5, characterized in that the channel (19, 24, 29, 35), and consequently also the finished cooling passage (17, 44), comprise a first passage section (17a) which extends from the inlet (17i) on the cool side (CS) into the interior of the component wall (14), a second passage section (17b) which adjoins the first passage section (17a) and extends essentially parallel to the surface (18) which is to be cooled, and a third passage section (17c) which adjoins the second passage section (17b) and terminates in the outlet (17o) on the hot side (HS).

7. The method as claimed in claim 6, characterized in that the first passage section (17a) and the third passage section (17c) are oriented obliquely to the surface (18), that is to say at an acute angle.

8. The method as claimed in claim 6 or 7, characterized in that the cooling air passage (17, 44) has an inside diameter of approximately 1 mm and the second passage section (17b) is at a distance (d2) which is less than or equal to 1 mm from the surface (18) which is to be cooled.

9. The method as claimed in one of claims 1 - 8, characterized in that the channel (19, 24, 29, 35) is let into the component (14, 23, 28, 34) to such a depth, or hollowed out of the component (14, 23, 28, 34) to such a depth, that the inserted cooling tube (20, 25, 30, 36), apart from inlet (17i) and outlet (17o), is located well below the surface (18).

10. The method as claimed in one of claims 1 - 9, characterized in that the channel (19, 24, 29, 35), with the cooling tube (20, 25, 30, 36) inserted, is filled with a high-temperature solder as filling material (21).
11. The method as claimed in one of claims 1 - 10, characterized in that the anti-oxidation, temperature-stable cover layer (22) is applied by deposition welding by means of a laser metal forming process (LMF).

12. The method as claimed in claim 11, characterized in that the cover layer (22) is formed by consecutive application of a plurality of overlapping cover layer coatings (1-R, 1-L, 2-RR, 2-LL, 3-C).

13. A thermally highly stressed component (14, 23, 28, 34), with a hot side delimited by a surface (18) and at least one near-surface cooling passage (17, 44), characterized in that the cooling passage (17, 44) is produced according to a method as claimed in one of claims 1 - 12.

14. The component as claimed in claim 13, characterized in that the component (14, 23, 28, 34) has a wall (14) with a hot side (HS) and an oppositely disposed cool side (CS), and in that the cooling passage (14, 44) extends through the component wall (14) from the cool side (CS) to the hot side (HS) and has an inlet (17i) on the cool side (CS) and an outlet (17o) on the hot side (HS).

15. The component as claimed in claim 14, characterized in that the cooling passage (17, 44) comprises a first passage section (17a) which extends from the inlet (17i) on the cool side (CS) into the interior of the component wall (14), a second passage section (17b) which adjoins the first passage section (17a) and extends essentially parallel to the surface (18) which is to be cooled, and a third passage section (17c) which adjoins the second passage section (17b) and terminates in the outlet (17o) on the hot side (HS).

16. The component as claimed in claim 15, characterized in that the first passage section (17a) and the third passage section (17c) are oriented obliquely to the surface (18), that is to say at an acute angle, and especially include an
angle of between 15° and 30°, preferably an angle of approximately 18°, with the surface normal.

17. The component as claimed in claim 6 or 7, characterized in that the cooling passage (17, 44) has a cooling tube (20, 25, 30, 36) which lies in a channel (19, 24, 29, 35) let into the surface (18) and is embedded into a temperature-resistant filling material (21), especially a high-temperature solder.

18. The component as claimed in claim 17, characterized in that the cooling tube has an inside diameter of approximately 1 mm and an outside diameter of approximately 1.5 mm, and in that the second passage section (17b) is at a distance (d2) which is less than or equal to 1 mm from the surface (18) which is to be cooled.

19. The component as claimed in one of claims 13 - 18, characterized in that the cooling passage (17, 44) has a length of approximately 20 mm.

20. The component as claimed in one of claims 13 - 19, characterized in that a plurality of passages (44) are arranged in the component (43) in parallel and/or in series and at a distance from each other.
Fig. 7
INTERNATIONAL SEARCH REPORT

A. CLASSIFICATION OF SUBJECT MATTER

INV. F01D5/18
ADD.

According to International Patent Classification (IPC) or to both national classification and IPC

B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols)

F01D F02K

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched

Electronic database consulted during the international search (name of database and, where practicable, search terms used)

EPO-Internal

C. DOCUMENTS CONSIDERED TO BE RELEVANT

<table>
<thead>
<tr>
<th>Category</th>
<th>Citation of document, with indication, where appropriate, of the relevant passages</th>
<th>Relevant to claim No.</th>
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<td>X</td>
<td>DE 27 54 896 Al (GEN ELECTRIC) 15 June 1978 (1978-06-15) the whole document page 6, line 31 - page 7, line 34 page 11, lines 1-6 figures 1-3 -----</td>
<td>1-4, 9-11, 13, 17-20</td>
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Further documents are listed in the continuation of Box C. See patent family annex.

* Special categories of cited documents:

*A* document defining the general state of the art which is not considered to be of particular relevance

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*X* document of particular relevance; the claimed invention cannot be considered novel or cannot be considered to involve an inventive step when the document is taken alone

*Y* document of particular relevance; the claimed invention cannot be considered to involve an inventive step when the document is combined with one or more other such documents, such combination being obvious to a person skilled in the art

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Date of the actual completion of the international search

8 May 2013

Date of mailing of the international search report

16/05/2013

Name and mailing address of the ISA

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Authorized officer

Gombert, Raf
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