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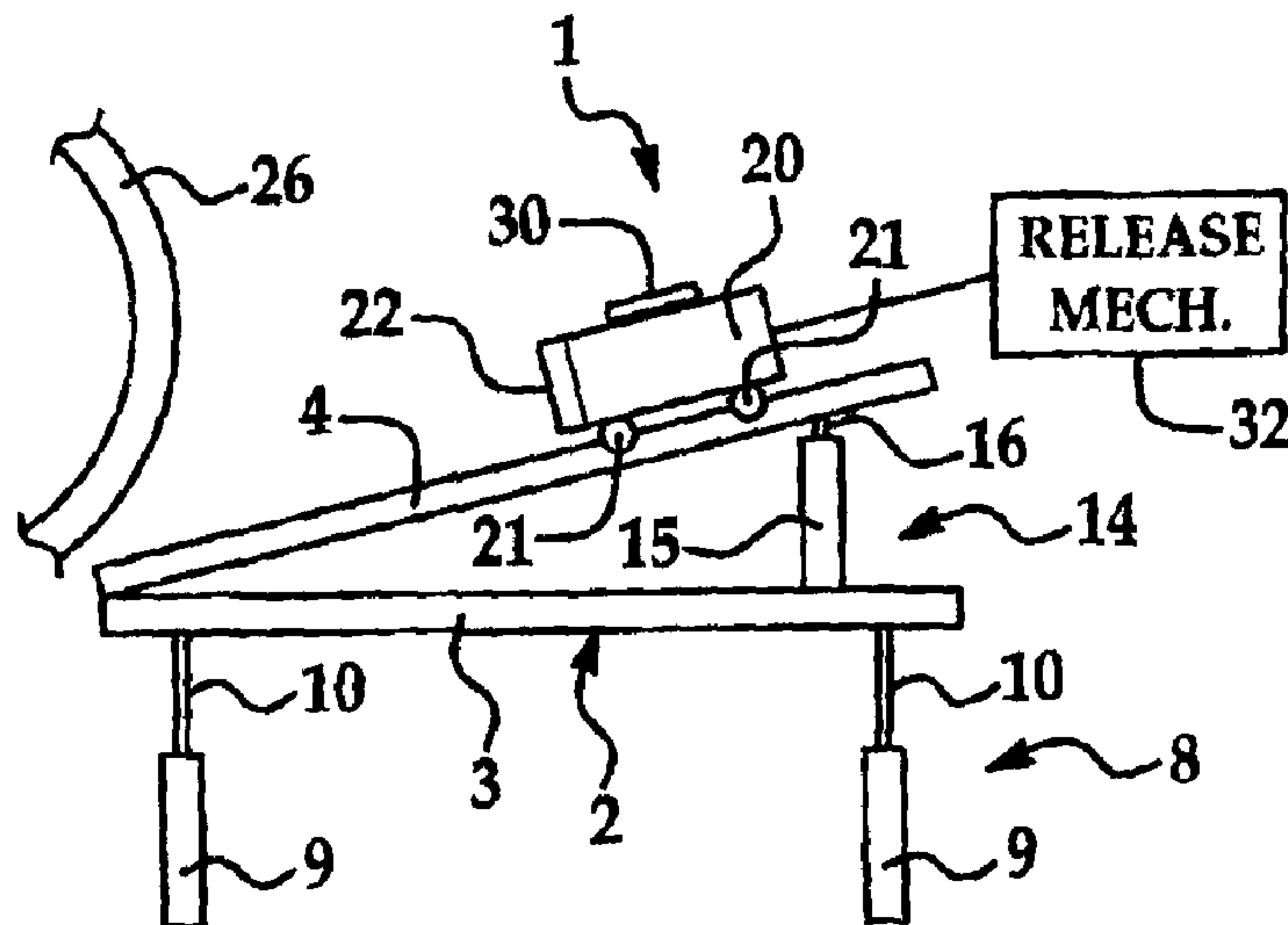
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(54) **Titre : APPAREIL ET METHODE D'ESSAI D'IMPACT PAR OBJET**

(54) **Title: BLUNT IMPACT TEST APPARATUS AND METHOD**



(57) **Abrégé/Abstract:**

A blunt impact test apparatus includes an apparatus frame having track rails for positioning proximate the test structure; a track angle positioning mechanism engaging the track rails to control a slope of the track rails; and an impact cart adapted to roll on the track rails.



ABSTRACT OF THE DISCLOSURE

[0036] A blunt impact test apparatus includes an apparatus frame having track rails for positioning proximate the test structure; a track angle positioning mechanism engaging the track rails to control a slope of the track rails; and an impact cart adapted to roll on the track rails.

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BLUNT IMPACT TEST APPARATUS AND METHOD

TECHNICAL FIELD

[001] The disclosure generally relates to testing of blunt impact forces on composite materials. More particularly, the disclosure relates to a blunt impact test apparatus and method which can be used to test the effect of blunt impact having a measured magnitude on composite materials.

BACKGROUND

[002] Aircraft and other structures built of composite materials may be inadvertently impacted by a service vehicle or other object without any external indications of the impact. Assessment of blunt impact to a composite material structure may be necessary or desirable to determine the necessity or feasibility of making repairs to the structure. Therefore, various methods used to assess the effect of blunt impact on a composite material structure have been devised.

[003] Some techniques which have been used to assess the effect of blunt impact on composite material structures include the use of drop tubes, spring-actuated "guns" and pendulums. Drop tubes include relatively small masses (typically <100 pounds) which are dropped vertically through a tube onto a horizontal surface from a height which produces the desired energy level at impact. Spring-actuated guns use a spring to accelerate a small mass against a surface and can be used in any orientation. However, both of these techniques are limited to a small impact area (typically <~12

square inches). Pendulums use larger masses and are used to perform impacts on vertical surfaces.

[004] A drawback of conventional drop tubes and gun-type impact devices is that their relatively small masses and relatively high velocities do not replicate the damage which is sustained from large masses at relatively low velocities. Additionally, use of such devices may render acquisition of data during the impact difficult. Pendulums are generally much lower in mass than the vehicles that cause the impact which they attempt to replicate and typically cannot be used to reliably simulate impacts low on the body of an aircraft fuselage (particularly when an attempt is made to simulate a vehicle scraping under the belly or the cargo door of an aircraft). Pendulums may also require large or tall support structures; therefore, secondary impacts caused by such structures may be difficult if not impossible to prevent.

[005] Using any of these conventional impact simulation methods on a complete aircraft in such a manner that the aircraft mass and moments of inertia are valid would be difficult at best. As an alternative, an actual service vehicle (or other powered vehicle with the appropriate bumper shape attached to it) may be used to impact the structure but such a solution would require that a human operator control the vehicle's velocity, direction and angle of impact, which would be unsafe and have poor repeatability. Elimination of the human operator factor in such a scenario would require a complex and expensive electronic control system or external thrust system.

[006] Accordingly, a blunt impact test apparatus and method which can be used to test the effect of blunt impact having a measured magnitude on composite materials is needed.

SUMMARY

5 **[007]** The disclosure is generally directed to a blunt impact test apparatus for testing blunt impact against a test structure. An illustrative embodiment of the apparatus includes an apparatus frame having track rails for positioning proximate the test structure; a track angle positioning mechanism engaging the distal end of the track rails to control a slope of the track rails; and an impact cart
10 adapted to roll on the track rails.

[008] The disclosure is further generally directed to a method of testing a structure for survivability after impact. An illustrative embodiment of the method includes providing a test structure for testing; providing an inclined ramp placed distal to and coupled to a level ramp placed proximate to the test structure;
15 placing a cart on the ramp, the cart configured to roll on the ramp; and rolling the cart down the ramp against the test structure.

[009] The disclosure is further generally directed to a method of predicting the effect of impact of a service vehicle on a composite aircraft structure. An illustrative embodiment of the method includes providing a test structure for
20 testing; providing a level ramp placed proximate to the test structure coupled to an inclined ramp placed distal and coupled to the level ramp; placing a cart on the ramp, the cart configured to roll on the ramp; and rolling the cart down the ramp against the test structure.

[009A] In accordance with one aspect of the invention, there is provided a blunt
25 impact test apparatus for testing blunt impact against a test structure. The apparatus includes a frame having track rails operable to be positioned proximate the test structure and operable to guide an impact cart toward the test structure to cause the impact cart to impact the test structure. The apparatus

also includes a track angle positioning mechanism engaged with the track rails to control a slope of the track rails, and a restraining mechanism operable to restrain the impact cart after impacting the test structure to prevent multiple impacts of the impact cart against the test structure.

5 **[009B]** The test structure may include an aircraft composite structure.

[009C] The apparatus may further include the impact cart.

[009D] The impact cart may be operable to roll on the track rails.

[009E] The impact cart may include an impact bumper.

10 **[009F]** The apparatus may further include frame height adjusting means for adjusting a height of the frame.

[009G] The frame height adjusting means may include a hydraulic system.

[009H] The track rails may be cylindrical and may be configured to guide grooved cart wheels of the impact cart when the grooved cart wheels are engaged with the track rails.

15 **[009I]** The impact cart may include at least one weight plate.

[009J] The track angle positioning mechanism may include a hydraulic system.

[009K] The apparatus may further include a release mechanism operable to hold the impact cart in an elevated position and to release the impact cart to enable the impact cart to travel along the track rails.

20 **[009L]** The track rails may be configured to enable the impact cart to roll along the track rails when the impact cart is positioned on the track rails.

25 **[009M]** In accordance with one aspect of the invention, there is provided a method of testing a structure for survivability after impact. The method involves placing an inclined ramp proximate to the structure to be tested, and placing a cart on the ramp. The method also involves rolling the cart down the ramp to

cause the cart to impact the structure, and restraining the cart after the cart impacts the structure to prevent multiple impacts of the cart against the structure.

[009N] The structure may include an aircraft composite structure.

[009O] The method may further involve raising the cart to a predetermined height on the ramp prior to rolling the cart down the ramp.

[009P] The method may further involve carrying at least one weight plate on the cart.

[009Q] Rolling the cart down the ramp to cause the cart to impact the test structure may involve rolling the cart down the ramp to cause an impact bumper of the cart to impact the test structure.

[009R] The test structure may include an aircraft.

[009S] The cart may have a mass about the same as an aircraft service vehicle.

[009T] In accordance with another aspect of the invention, there is provided a method of testing a structure for survivability of an impact. The method involves placing a ramp proximate to the structure, and adjusting an incline of the ramp based on a desired impact of a cart on the structure. The method also involves rolling the cart down the ramp to cause the cart to impact the structure, and restraining the cart after the cart impacts the structure to prevent multiple impacts of the cart against the structure.

[009U] The structure may include an aircraft composite structure.

[009V] The method may further involve raising the cart to a predetermined height on the ramp prior to rolling the cart down the ramp.

[009W] The method may further involve placing the cart on the ramp and subsequently adding mass to the cart.

[009X] Rolling the cart down the ramp to cause the cart to impact the test structure may involve rolling the cart down the ramp to cause an impact bumper of the cart to impact the test structure.

BRIEF DESCRIPTION OF THE ILLUSTRATIONS

[0010] FIG. 1 is a side view of an illustrative embodiment of the blunt impact test apparatus.

5 [0011] FIG. 1A is a cross-sectional view of a track rail of an illustrative embodiment of the blunt test apparatus, with an impact cart wheel engaging the track rail.

[0012] FIG. 2 is a side view of an illustrative embodiment of the blunt impact test apparatus, with a track angle positioning mechanism deployed in a raised configuration to increase the angle of the track rails.

10 [0013] FIG. 3 is a side view of an illustrative embodiment of the blunt impact test apparatus, with the impact cart released from the cart release mechanism and an impact bumper on the impact cart impacting a test structure for testing.

[0014] FIG. 4 is a flow diagram which illustrates an illustrative embodiment of a method of testing a structure for survivability after impact.

15 [0015] FIG. 5 is a flow diagram which illustrates an illustrative embodiment of a method of predicting the effect of impact of a service vehicle on a composite aircraft structure.

[0016] FIG. 6 is a flow diagram of an aircraft production and service methodology.

[0017] FIG. 7 is a block diagram of an aircraft.

DETAILED DESCRIPTION

[0018] The following detailed description is merely exemplary in nature and is not intended to limit the described embodiments or the application and uses of the described embodiments. As used herein, the word “exemplary” or “illustrative” means “serving as an example, instance, or illustration.” Any implementation described herein as “exemplary” or “illustrative” is not necessarily to be construed as preferred or advantageous over other implementations. All of the implementations described below are exemplary implementations provided to enable persons skilled in the art to practice the disclosure and are not intended to limit the scope of the appended claims. Furthermore, there is no intention to be bound by any expressed or implied theory presented in the preceding technical field, background, brief summary or the following detailed description.

[0019] Referring initially to FIGS. 1-3, an illustrative embodiment of the blunt impact test apparatus, hereinafter apparatus, is generally indicated by reference numeral 1. The apparatus 1 may include an apparatus frame 2. In some embodiments, the apparatus frame 2 may include a frame base 3 and a pair of track rails 4 provided on the frame base 3. The track rails 4 may be adjustable at an angle with respect to the frame base 3. Accordingly, a track angle positioning mechanism 14 may engage the track rails 4 for the purpose. In some embodiments, the track angle positioning mechanism 14 may include a track angle positioning cylinder 15 on the frame base 3 and a track angle positioning piston 16 which is extendable from the track angle positioning cylinder 15 and engages the track rails 4. Therefore, the angle of the track rails 4 relative to the frame base 3 may be controlled by extension or retraction of the piston 16 relative to the

cylinder **15**. In other embodiments, the track angle positioning mechanism **14** may be any other type of mechanism which can be used to selectively vary the angle of the track rails **4** relative to the frame base **3**.

[0020] In some embodiments, a second pair of track rails (not shown) may be attached to the track rails **4**, respectively, at the lower end of the track rails **4**. The second pair of track rails may be generally level or horizontal and disposed adjacent to a test structure **26**.

[0021] In some embodiments, a frame lifting mechanism **8** may engage the frame base **3** of the apparatus frame **2**. In some embodiments, the frame lifting mechanism **8** may include at least one frame lifting cylinder **9** and a frame lifting piston **10** which is extendable from the frame lifting cylinder **9** and engages the frame base **3**. Accordingly, the height or vertical position of the apparatus frame **2** can be selectively controlled by extension or retraction of the pistons **10** relative to the respective cylinders **9**. In other embodiments, the frame lifting mechanism **8** may be any other mechanism which can be used to selectively raise and lower the apparatus frame **2**. In still other embodiments, the frame itself can be placed atop sub-frames (not shown) which may have a fixed height to achieve the required height of the track.

[0022] An impact cart **20** may be provided on the track rails **4** of the apparatus frame **2**. The impact cart **20** may include multiple cart wheels **21** which engage the track rails **4**. As shown in FIG. 1A, in some embodiments, each of the track rails **4** may be generally circular in cross-section. Each cart wheel **21** of the impact cart **20** may have a grooved

or concave contour to complement the cylindrical shape of each track rail 4. Accordingly, the impact cart 20 may be adapted to roll on the track rails 4 of the apparatus frame 2 from the pre-release position shown in FIG. 2 to the impact position shown in FIG. 3 for purposes which will be hereinafter described.

5 [0023] A release mechanism 32 may be adapted to engage the impact cart 20 and hold or secure the impact cart 20 in the pre-release position shown in FIG. 2. The release mechanism 32 may have any design which is suitable for the purpose. In some embodiments, the release mechanism 32 may be a hydraulically-actuated release mechanism.

10 [0024] As further shown in FIGS. 1-3, the impact cart 20 may be adapted to receive one or multiple weight plates 30. The weight plates 30 may be added to the impact cart 20 to select the weight of the impact cart 20 for impact testing as will be hereinafter further described. In some embodiments, a sufficient number of the weight plates 30 may be added to the impact cart 20 to achieve a weight of up to 11,000 pounds or more for the
15 impact cart 20.

[0025] An impact bumper 22 may be attached to the front of the impact cart 20. The impact bumper 22 may have any design which allows variation of (1) bumper width (can be greater than the width of the mass and track); (2) angle of impact with respect to the longitudinal axis of a test structure 26 such as an aircraft (can impact normal to the
20 surface or obliquely); and (3) orientation with respect to the circumference of the test

structure 26 such as an aircraft (can be horizontal, vertical or any given angle with respect to the structure being impacted).

[0026] In application, the apparatus 1 is placed adjacent to a test structure 26 which is to be tested. In some applications, the test structure 26 may be an aircraft. The impact cart 20 is raised to a selected height or position on the track rails 4 of the apparatus frame 2. The pre-release position (FIG. 2) of the impact cart 20 on the track rails 4 may be selected depending on the desired velocity of the impact cart 20 against the test structure 26. The frame lifting mechanism 8 can be operated to adjust the vertical position of the apparatus frame 2 and thus, the vertical position of the point of impact of the impact cart 20 against the test structure 26. The track angle positioning mechanism 14 can be operated to adjust the angle or slope of the track rails 4 in order to achieve a desired impact speed of the impact cart 20 on the track rails 4. A mechanism (not shown) to prevent multiple impacts may be installed which restrains the cart after rebounding from the initial impact with the structure under test.

[0027] The impact cart 20 is raised to the pre-release position on the track rails 4 and attached to the release mechanism 32. A selected number of weight plates 30 may be placed on the impact cart 20 to achieve a desired weight of the impact cart 20. An impact bumper 22 of selected configuration and design can be attached to the front of the impact cart 20. In some embodiments, the apparatus 1 may be used to determine the impact which would be created in the event that an aircraft service vehicle inadvertently strikes an aircraft. Therefore, the height of the apparatus frame 2, the slope of the track rails 4, the position of the impact cart 20 on the track rails 4 and the

weight of the impact cart **20** may be adjusted to strike the test structure **26** at a kinetic energy level which replicates the impact kinetic energy level of the aircraft service vehicle against the aircraft.

[0028] The release mechanism **32** is operated to release the impact cart **20** such that the impact cart **20** rolls down the track rails **4** from the pre-release position shown in FIG. **2** to the impact position shown in FIG. **3** under the influence of gravity. Accordingly, the impact bumper **22** on the front of the impact cart **20** strikes the test structure **26** and may damage the test structure **26**. The test structure **26** may be examined using conventional methods to determine the extent of damage to the test structure **26**. The extent of damage to the test structure **26** may be correlated to the extent of damage which would be exerted on an aircraft in the event that an aircraft service vehicle inadvertently strikes the aircraft. Accordingly, the apparatus **1** may enable electronic data acquisition of impact loads in multiple axes, acceleration, displacement and velocity of the impact cart **20** throughout the duration of the impact testing.

[0029] Referring next to FIG. **4**, a flow diagram **400** which illustrates an illustrative embodiment of a method of testing a structure for survivability after impact is shown. In block **402**, a composite test structure for testing is provided. In block **404**, an inclined ramp is placed proximate to the composite structure for testing. In block **406**, a cart which is configured to roll on the ramp is placed on the ramp. In block **408**, an impact device is attached to the cart. In block **410**, the cart is raised to a predetermined height

on the ramp. In block **412**, the cart is released and allowed to impact the test structure for impact testing.

[0030] Referring next to FIG. 5, a flow diagram **500** which illustrates an illustrative embodiment of a method of predicting the effect of impact of a service vehicle on a composite aircraft structure is shown. In block **502**, a composite test structure for testing is provided. In block **504**, an inclined ramp is placed proximate to the composite structure for testing. In block **506**, a cart which is configured to roll on the ramp is placed on the ramp. In some embodiments, the cart may have a mass of the order of magnitude of an aircraft service vehicle. In block **508**, an impact device is attached to the cart. In block **510**, the cart is raised to a predetermined height on the ramp. In block **512**, the cart is released and allowed to impact the test structure for impact testing.

[0031] Referring next to FIGS. 6 and 7, embodiments of the disclosure may be used in the context of an aircraft manufacturing and service method **78** as shown in FIG. 6 and an aircraft **94** as shown in FIG. 7. During pre-production, exemplary method **78** may include specification and design **80** of the aircraft **94** and material procurement **82**. During production, component and subassembly manufacturing **84** and system integration **86** of the aircraft **94** takes place. Thereafter, the aircraft **94** may go through certification and delivery **88** in order to be placed in service **90**. While in service by a customer, the aircraft **94** may be scheduled for routine maintenance and service **92** (which may also include modification, reconfiguration, refurbishment, and so on).

[0032] Each of the processes of method **78** may be performed or carried out by a system integrator, a third party, and/or an operator (e.g., a customer). For the purposes of this description, a system integrator may include without limitation any number of aircraft manufacturers and major-system subcontractors; a third party may include without limitation any number of vendors, subcontractors, and suppliers; and an operator may be an airline, leasing company, military entity, service organization, and so on.

[0033] As shown in FIG. 7, the aircraft **94** produced by exemplary method **78** may include an airframe **98** with a plurality of systems **96** and an interior **100**. Examples of high-level systems **96** include one or more of a propulsion system **102**, an electrical system **104**, a hydraulic system **106**, and an environmental system **108**. Any number of other systems may be included. Although an aerospace example is shown, the principles of the invention may be applied to other industries, such as the automotive industry.

[0034] The apparatus embodied herein may be employed during any one or more of the stages of the production and service method **78**. For example, components or subassemblies corresponding to production process **84** may be fabricated or manufactured in a manner similar to components or subassemblies produced while the aircraft **94** is in service. Also one or more apparatus embodiments may be utilized during the production stages **84** and **86**, for example, by substantially expediting assembly of or reducing the cost of an aircraft **94**. Similarly, one or more apparatus

embodiments may be utilized while the aircraft **94** is in service, for example and without limitation, to maintenance and service **92**.

[0035] Although the embodiments of this disclosure have been described with respect to certain exemplary embodiments, it is to be understood that the specific embodiments
5 are for purposes of illustration and not limitation, as other variations will occur to those of skill in the art.

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**THE EMBODIMENTS OF THE INVENTION IN WHICH AN EXCLUSIVE
PROPERTY OR PRIVILEGE IS CLAIMED ARE DEFINED AS FOLLOWS:**

1. A blunt impact test apparatus for testing blunt impact against a test
5 structure, the apparatus comprising:

a frame having track rails operable to be positioned proximate the
test structure and operable to guide an impact cart toward the test
structure to cause the impact cart to impact the test structure;

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a track angle positioning mechanism engaged with said track rails
to control a slope of said track rails; and

15

a restraining mechanism operable to restrain the impact cart after
impacting the test structure to prevent multiple impacts of the
impact cart against the test structure.

2. The apparatus of claim 1 wherein the test structure comprises an aircraft
composite structure.

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3. The apparatus of claim 1 or 2 further comprising the impact cart.

4. The apparatus of any one of claims 1 to 3, wherein the impact cart is operable to roll on the track rails.
- 5 5. The apparatus of claim 3 wherein the impact cart comprises an impact bumper.
6. The apparatus of any one of claims 1 to 4 further comprising frame height adjusting means for adjusting a height of the frame.
- 10 7. The apparatus of claim 6 wherein said frame height adjusting means comprises a hydraulic system.
8. The apparatus of any one of claims 1 to 4 wherein said track rails are cylindrical and are configured to guide grooved cart wheels of the impact
15 cart when the grooved cart wheels are engaged with said track rails.
9. The apparatus of any one of claims 1 to 4 wherein said impact cart comprises at least one weight plate.
- 20 10. The apparatus of any one of claims 1 to 4 wherein said track angle positioning mechanism comprises a hydraulic system.

11. The apparatus of any one of claims 1 to 4 further comprising a release mechanism operable to hold the impact cart in an elevated position and to release the impact cart to enable the impact cart to travel along the track rails.

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12. The apparatus of any one of claims 1 to 4 wherein the track rails are configured to enable the impact cart to roll along the track rails when the impact cart is positioned on the track rails.

- 10 13. A method of testing a structure for survivability after impact, comprising:

placing an inclined ramp proximate to said structure to be tested;

placing a cart on said ramp; and

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rolling said cart down said ramp to cause the cart to impact said structure; and

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restraining the cart after the cart impacts the structure to prevent multiple impacts of the cart against the structure.

14. The method of claim **13** wherein the structure comprises an aircraft composite structure.

15. The method of claim **13** or **14** further comprising raising said cart to a predetermined height on said ramp prior to rolling said cart down said ramp.

16. The method of claim **13** or **14** further comprising carrying at least one weight plate on said cart.

17. The method of claim **13** or **14** wherein rolling said cart down said ramp to cause the cart to impact said test structure comprises rolling said cart down said ramp to cause an impact bumper of the cart to impact said test structure.

18. The method of claim **13** or **14** wherein the test structure comprises an aircraft.

19. The method of claim **18** wherein the cart has a mass about the same as an aircraft service vehicle.

20. A method of testing a structure for survivability of an impact, comprising:

placing a ramp proximate to said structure;

adjusting an incline of said ramp based on a desired impact of a
5 cart on said structure; and

rolling said cart down said ramp to cause said cart to impact said
structure; and

10 restraining the cart after the cart impacts the structure to prevent
multiple impacts of the cart against the structure.

21. The method of claim **20** wherein the structure comprises an aircraft
composite structure.

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22. The method of claim **20** or **21** further comprising raising said cart to a
predetermined height on said ramp prior to rolling said cart down the
ramp.

20 **23.** The method of claim **20** or **21** further comprising placing the cart on the
ramp and subsequently adding mass to the cart.

- 24.** The method of claim **20** or **21** wherein rolling said cart down said ramp to cause the cart to impact said test structure comprises rolling said cart down said ramp to cause an impact bumper of the cart to impact said test structure.

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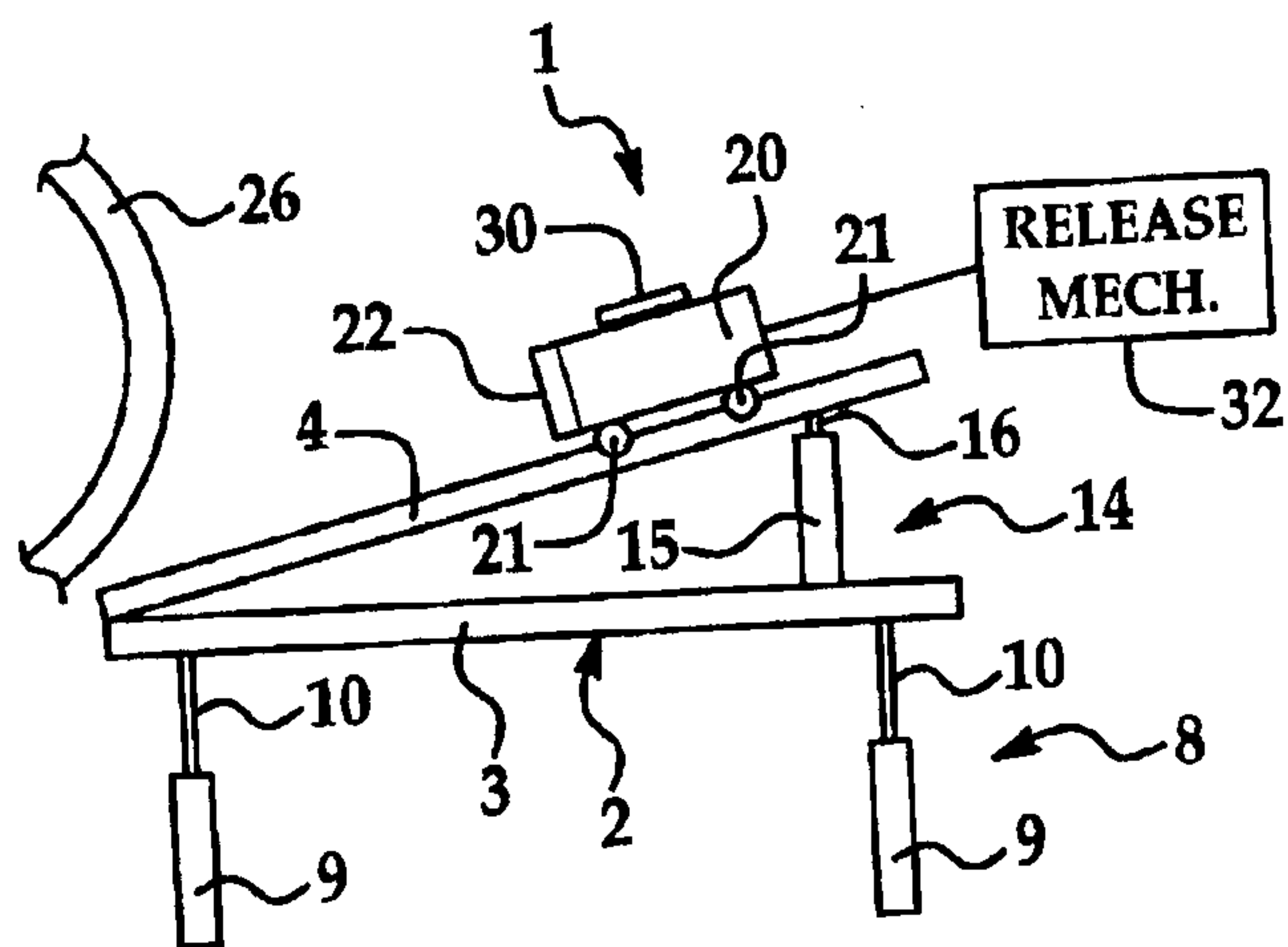


FIG. 1

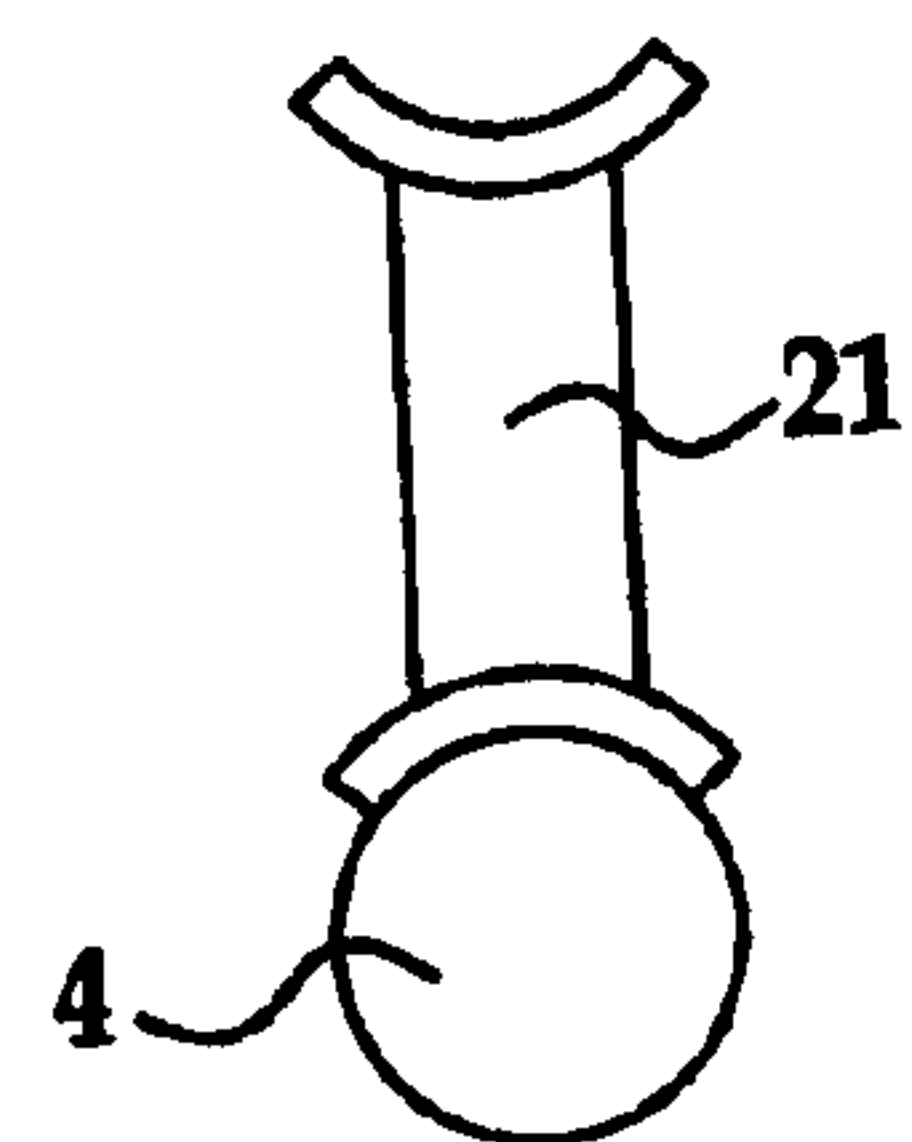


FIG. 1A

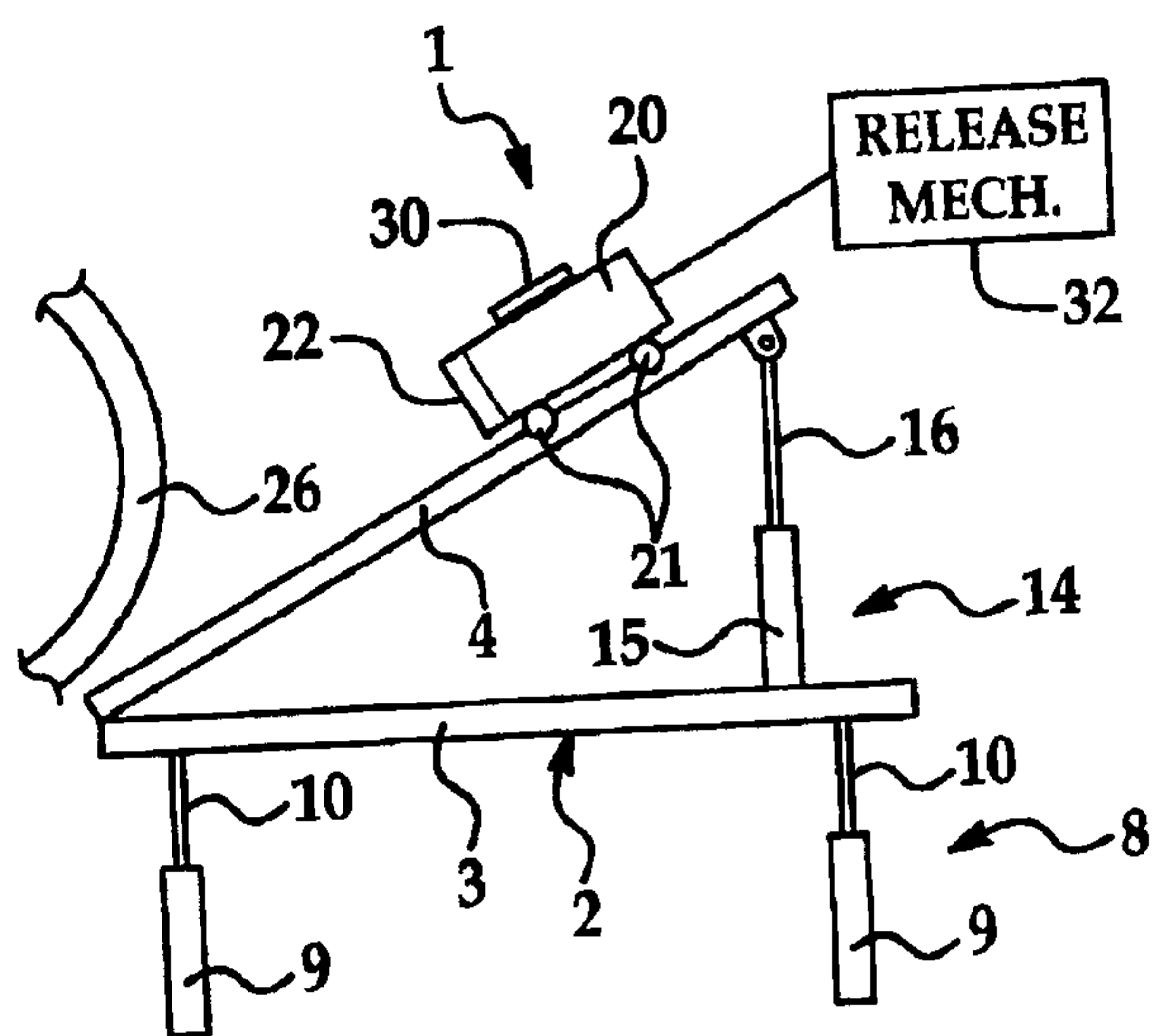
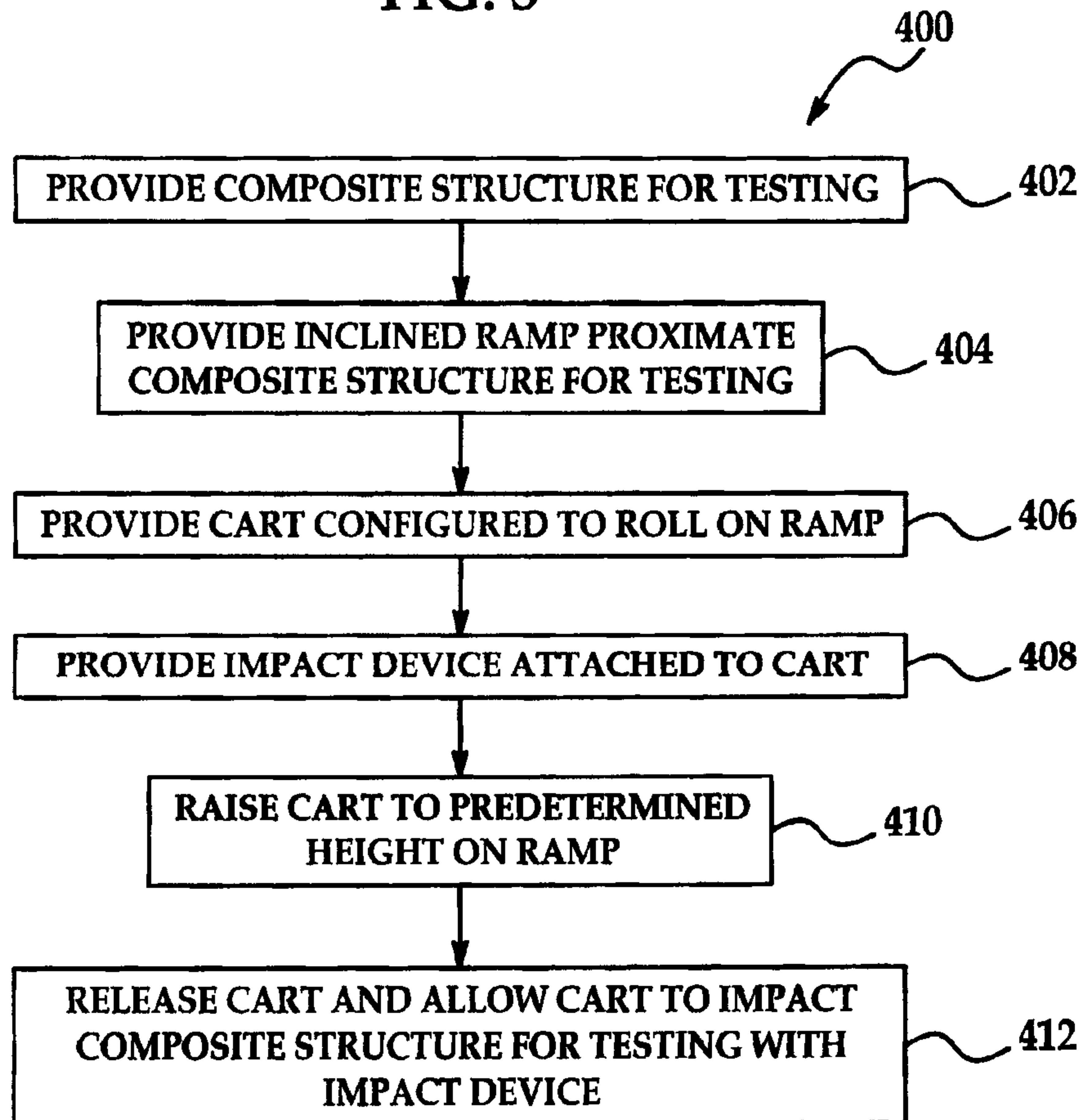
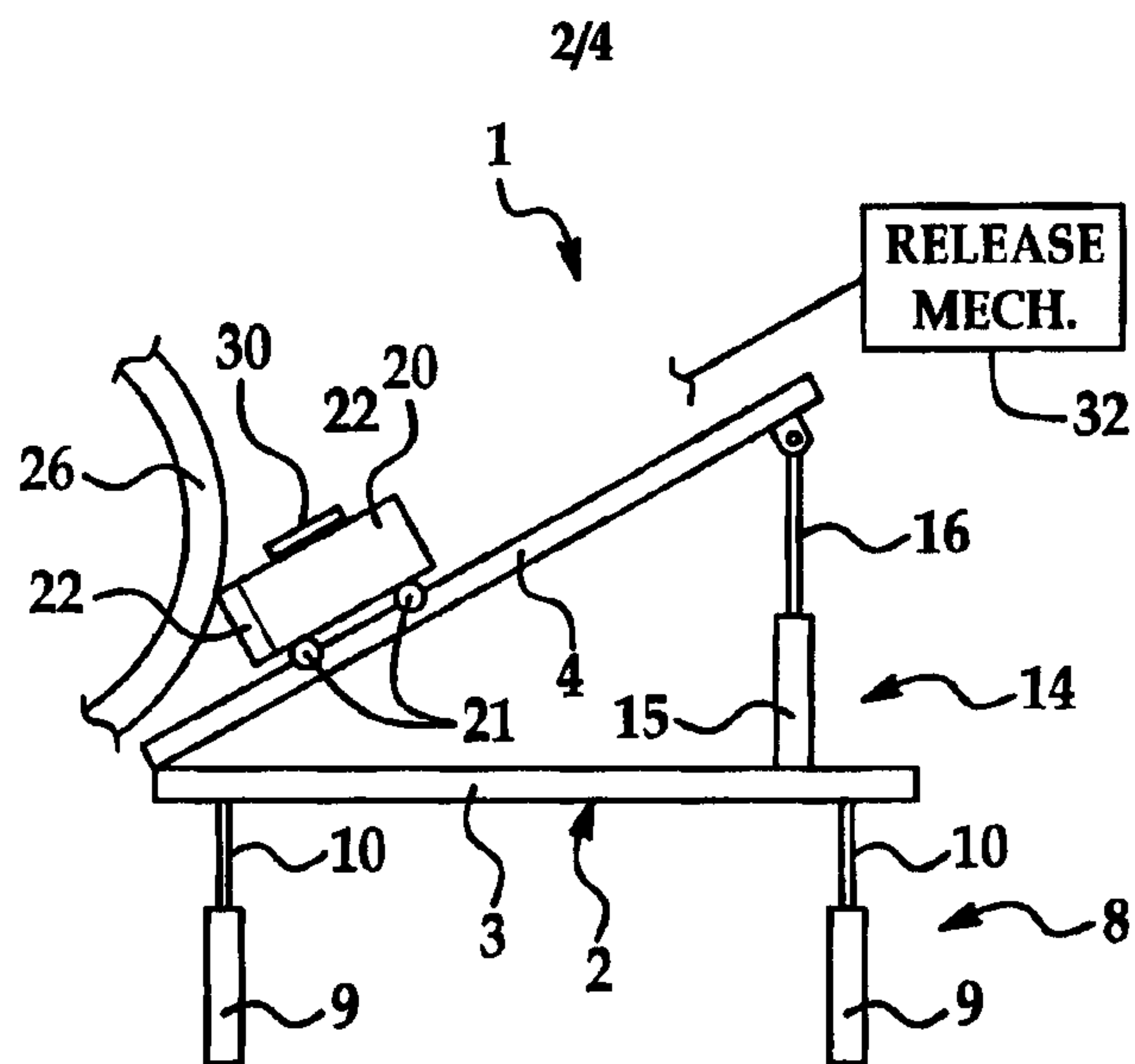


FIG. 2



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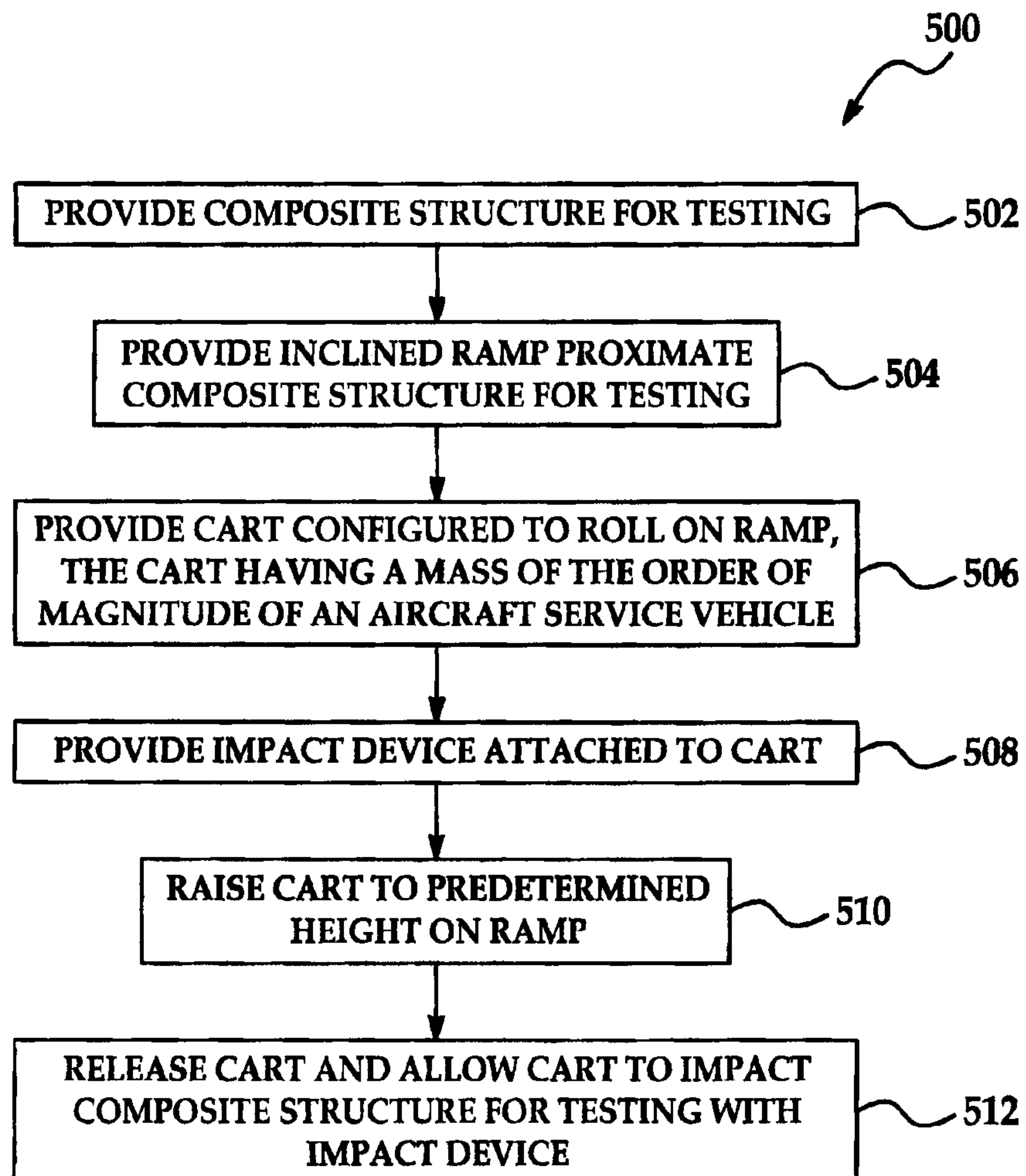


FIG. 5

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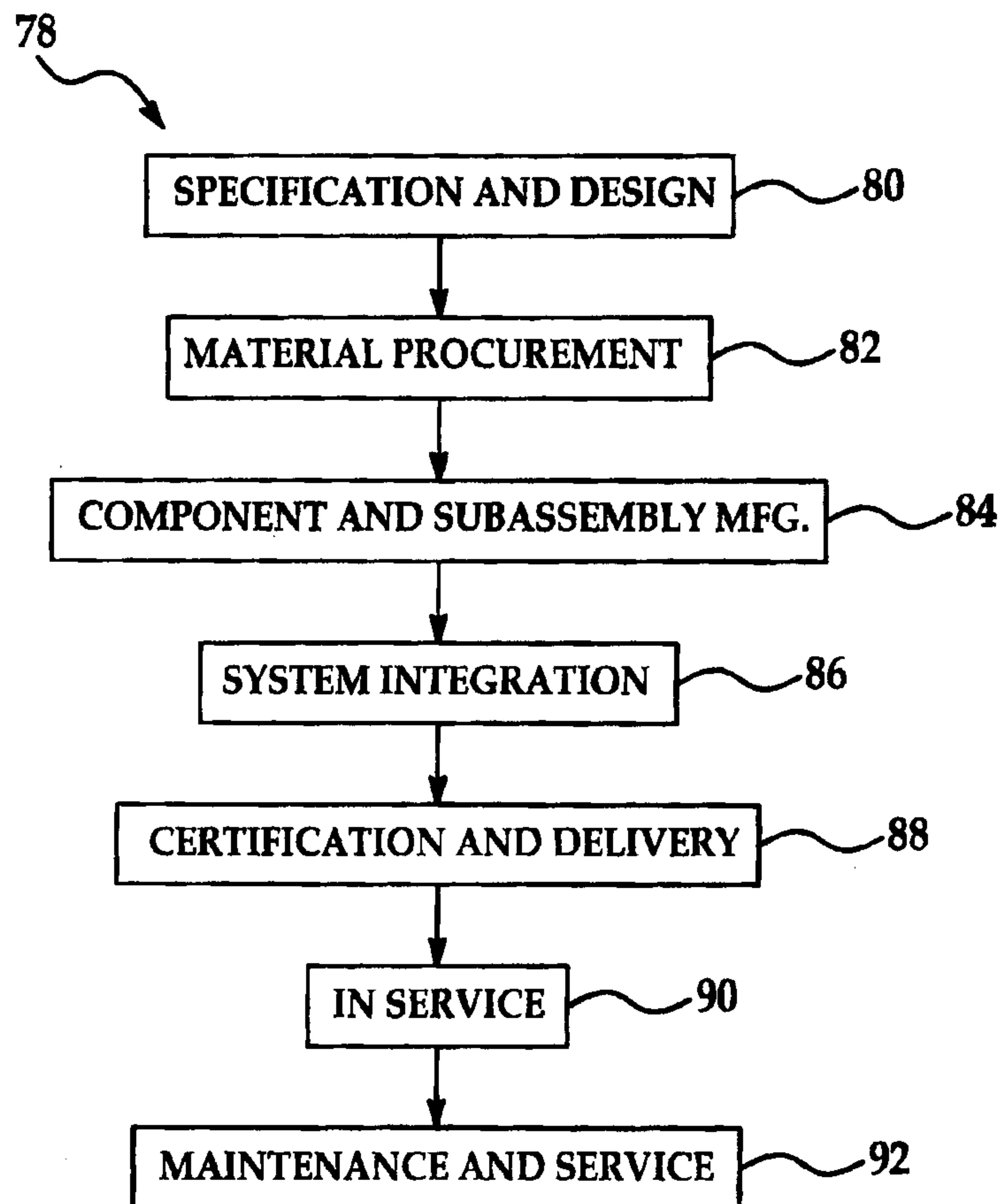


FIG. 6

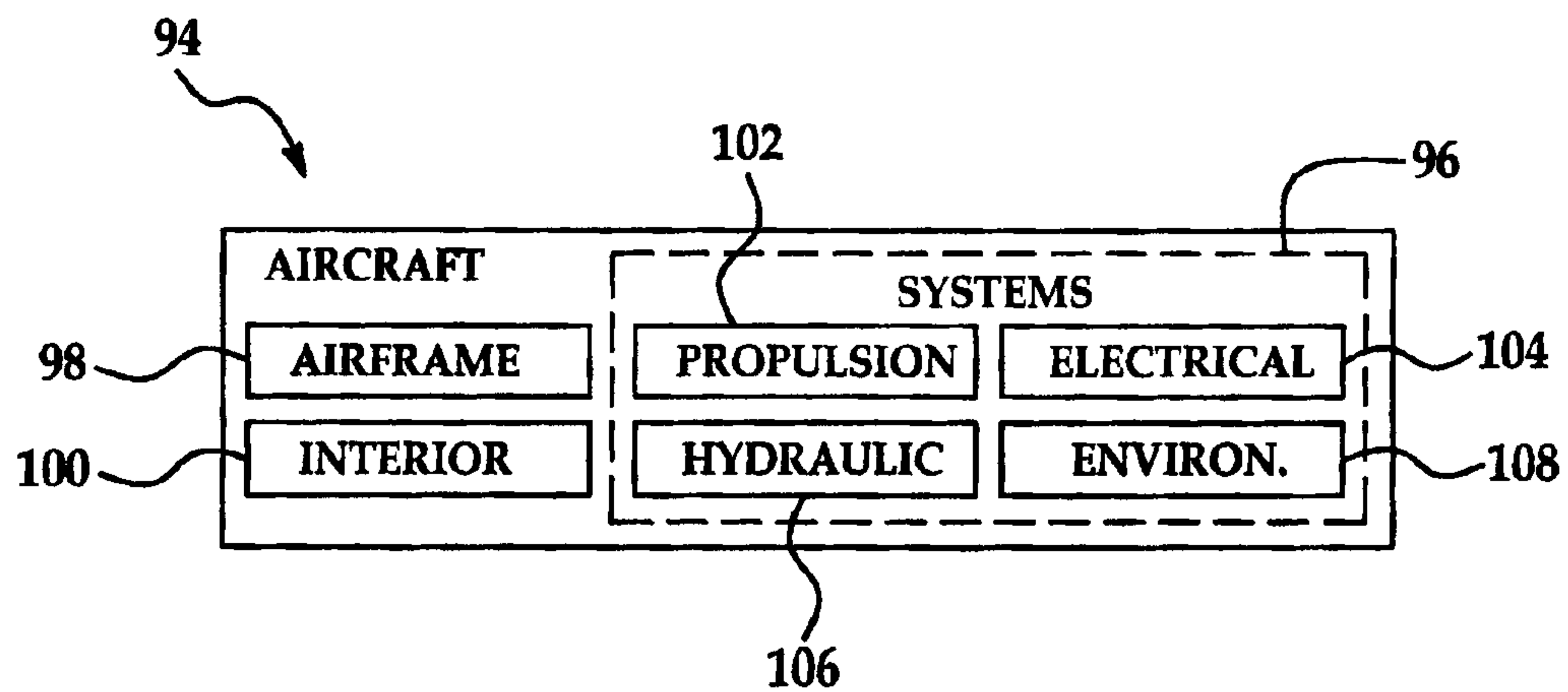


FIG. 7

