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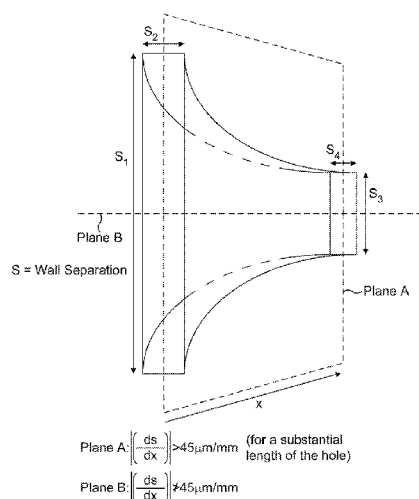


Fig. 7c

$$\left(\frac{dS}{dx}\right) \quad (\text{Formula})$$

(57) Abstract: An injection nozzle for injecting a fluid, the injection nozzle comprising: a nozzle body and a nozzle hole defining a flow passage for fluid, the flow passage comprising passage walls and the nozzle hole having an inlet in fluid communication via the flow passage with an outlet, wherein, the inlet is larger than the outlet and for at least one section through the inlet and outlet along the flow passage the nozzle hole is defined, for all distances x within a substantial length of the flow passage, by the condition: (Formula) $>45 \mu\text{m/mm}$, where S = passage wall separation and x is the distance from the inlet.

INJECTION NOZZLE

Field of the Invention

5 The present invention relates to an injection nozzle. In particular, the present invention relates to the formation and profile of an improved nozzle for the injection of a fluid from an internal nozzle volume into an external volume. The invention has particular application to fuel injection systems but may be applied to any device that utilises a nozzle arrangement to inject a fluid from a first volume to a second volume.

10

Background to the Invention

For internal combustion engines that use direct injection, fuel is typically injected from an injection nozzle which utilises multi-hole nozzle design in which each individual hole
15 (nozzle outlet) has an internal geometry that has been precision manufactured from dedicated tooling. This internal hole geometry is defined and optimized in order to reach an efficient liquid fuel atomization allowing a rapid fuel and air mixture within the combustion chamber. Such optimisation leads to lower exhaust emissions, optimized combustion noise and lower fuel consumption.

20

Prior efforts to improve fuel/air mixing have included rounding of the hole entry orifice, the understanding being that rounding of the hole entry increased the nozzle discharge coefficient, thereby increasing the spray momentum and leading to better fuel mixing within the combustion chamber. Rounding of this type was achieved using a paste with
25 abrasive particles but this had the disadvantage of being a lengthy manufacturing process which impacted upon the overall manufacturing cost for the injection nozzle.

More recently (see, for example, Applicant's EP0352926, EP1669157 and EP1669158) it has been suggested that the use of tapered holes gives equivalent nozzle efficiency
30 performances (compared to injection nozzles with rounded hole orifices) while reducing the manufacturing process time and cost. The tapered hole angle (convergent) has, in the past, been characterised by a factor k_{factor} defined as follows:

35

$$k_{factor} = \frac{(D_{in} - D_{out})}{10}$$

where D_{in} and D_{out} are respectively the inlet and outlet nozzle orifice diameters given in microns (μm).

5

Production injection nozzles currently available have typical k_{factor} values of between 1 and 2.5, which equates to a reduction of hole diameter between the hole inlet and the hole outlet of 10 to 25 μm (typically, the length of the nozzle hole itself is 1mm=1000 μm). It is noted that these k_{factor} values have been determined through existing knowledge of the physical processes involved in injection and also by current manufacturing equipment arrangements.

10

Nozzle hole efficiency may be characterised by a nozzle discharge coefficient C_d which is calculated using the Bernoulli formula as:

$$C_d = \frac{Q}{S_{out} \sqrt{\frac{2(P_{in} - P_{out})}{\rho}}}$$

15

where Q is the measured hole flow rate, P_{in} and P_{out} are respectively inlet and outlet hole pressure (fuel injection pressure and back pressure which could be combustion chamber gas pressure), S_{out} is the hole outlet section and ρ is the liquid fuel density at the inlet hole pressure and temperature conditions.

20

C_d values for automotive applications typically are measured during manufacture as being between 0.80 and 0.88 (for nozzle upstream and downstream pressures of 101 bar and 1 bar respectively) and it is noted that current, known hole designs do not provide for nozzle hole discharge coefficients of more than 0.88.

25

A further factor in the design of nozzle holes is the accuracy to which the hole needs to be manufactured in order for the nozzle hole to operate effectively. In this regard it is noted that holes designed with k_{factor} values of between 1 and 2.5 are sensitive to the length of the hole such that variations in hole length can potentially adversely affect the performance of the injection nozzle. As a consequence the machining of nozzle holes in current injection nozzles requires a high degree of accuracy which results in lengthy and costly manufacturing processes.

30

It is therefore an object of the present invention to provide an injection nozzle that overcomes or substantially mitigates the above-mentioned problems.

5 Statements of Invention

According to a first aspect of the present invention there is provided an injection nozzle for injecting a fluid, the injection nozzle comprising: a nozzle body and a nozzle hole defining a flow passage for fluid, the flow passage comprising passage walls and the
10 nozzle hole having an inlet in fluid communication via the flow passage with an outlet, wherein,

the inlet is larger than the outlet and for at least one section through the inlet and outlet along the flow passage the nozzle hole is defined, for all distances x within a substantial length of the flow passage, by the condition:

15 $\left| \left(\frac{dS}{dx} \right) \right| > 45 \text{ microns/millimetre,}$ where S = passage wall separation and x is the distance from the inlet.

The present invention provides for an injection nozzle with a tapered injection hole (the inlet being larger than the outlet) that has a far greater level of tapering than in
20 conventional nozzle designs. In particular it is noted that if a slice (section) is taken along the length of the hole then, for a substantial portion of that section, the condition $|dS/dx|$ (i.e. magnitude of the rate of change of wall separation (opposing internal hole walls) with distance) will be greater than 45 microns per millimetre for all distances x within that substantial portion.

25

In other words the magnitude of the condition $\left(\frac{dS}{dx} \right)$ (or $\left| \left(\frac{dS}{dx} \right) \right|$) at any given distance x

along a substantial portion of the nozzle hole is greater than 45 microns per millimetre. It is noted that the profile of the passage walls within the section may be linear.

Alternatively the profile of the walls may be parabolic or otherwise curved or a mixture of
30 sections of curved and linear profile. Within the section through the hole however the minimum value of the condition, along a substantial portion of the length of the hole,

always exceeds 45 microns per millimetre, i.e. $\left| \left(\frac{dS}{dx} \right) \right| > 45 \text{ } \mu\text{m/mm.}$

It is noted that compared to traditional nozzle hole designs, injection nozzles in accordance with embodiments of the present invention demonstrate improved discharge coefficients, better fuel atomisation performance and improved pressure and velocity flows within the hole itself. It is also noted that in traditional hole designs which

5 incorporate hole rounding the local wall separation values may exceed the wall condition stated above. However, this occurs over an extremely localised part of the traditional nozzle hole and is in contrast to the present invention in which the wall condition holds along a substantial length of the hole's length.

10 An injection nozzle in accordance with an embodiment of the present invention may be used in a fuel injection system such as those described in the Applicant's patent applications EP0352926, EP1669157, EP1669158, EP1081374, EP1180596, EP1344931, EP1496246, EP1498602, EP1522721, EP1553287, EP1645749, EP1703117, EP1744051 and EP1643117. However, it is noted that the present invention

15 is applicable to any fluid delivery system where a fluid is injected from a first volume to a second volume.

Preferably, the nozzle hole is defined, at any given x along a substantial length of the hole, by the condition $\left| \left(\frac{dS}{dx} \right) \right| > 60 \mu\text{m/mm}$. It is noted that a nozzle hole satisfying this

20 condition exhibits around a 5% performance increase based on an analysis of the discharge coefficient C_d compared to known tapered injection holes.

Preferably, the nozzle hole is defined, at any given x along a substantial length of the hole, by the condition $\left| \left(\frac{dS}{dx} \right) \right| > 80 \mu\text{m/mm}$. It is noted that such a condition reduces the

25 effects of variations in the length of the injection hole on its performance. A nozzle hole satisfying such a condition will not therefore need to be manufactured to such high manufacturing tolerance levels as for current injection holes.

Conveniently, it is noted that the improved performance of nozzle holes in accordance

30 with embodiments of the present invention is observed when the wall condition holds for at least 40% of the length of the hole. Preferably, the condition should hold for the final 60% to 90% of the length of the hole.

Conveniently, if the hole inlet and outlet define a nozzle hole axis then the at least one section may be taken through the axis. Conveniently, the wall separation condition may be satisfied for all sections through the axis regardless of their orientation about the axis.

- 5 Conveniently, the cross section of the nozzle hole may be circular or elliptical. Where the cross section is elliptical then sections taken through the hole axis and either the major or minor axes of the ellipse may satisfy the wall separation condition. As a further alternative, the cross section of the nozzle hole may be triangular, rectangular, square or any other polygon.

10

It is noted that the nozzle body may be provided with a bore which is in communication with a source of fluid (e.g. pressurised fuel) and the injection nozzle may be arranged to inject fluid from the bore through the nozzle hole to a volume outside the nozzle, e.g. a combustion volume of an engine system. In this arrangement it is noted that the hole
15 inlet opens into the bore and the hole outlet opens into the volume outside the injection nozzle.

- Preferably, the injection nozzle comprises a plurality of nozzle holes in accordance with the nozzle hole described above and this plurality of holes may be arranged in one or
20 more rows of holes such as those described in the Applicant's patent applications EP1645749, EP1703117, EP1744051 and EP1643117.

- The passage walls of the flow passage within the at least one section may comprise linear and non-linear arrangements, e.g. the walls may form a straight line taper, a
25 parabola, a mixture of linear and non-linear profiles etc.

The invention extends to a fuel injector for an internal combustion engine comprising an injection nozzle according to the first aspect of the present invention.

30 Brief Description of the Drawings

Figures 1 and 2 show sections through known fuel injector arrangements;

Figure 3 shows a section through a typical injection nozzle outlet hole;

Figures 4 and 5 show known injection hole arrangements in an injection nozzle;

Figure 6 shows sections through an injection nozzle outlet hole in accordance with an embodiment of the present invention;

Figure 7 shows cross sections through injection nozzle outlet holes that may be used in conjunction with an embodiment of the present invention;

5 Figure 8 shows a plot of discharge coefficient C_d versus hole inlet radius;

Figures 9a to 9j show the effects of nozzle hole taper on internal hole fluid pressure and velocity;

Figure 10a is a plot of internal nozzle hole pressure with distance from the hole inlet;

Figure 10b is a plot of internal fluid velocity; with distance from the hole inlet;

10 Figure 10c is a plot of internal fluid velocity with distance from the hole axis;

Figure 11a shows a plot of discharge coefficient improvement versus internal hole geometry for two nozzle holes of different lengths;

Figure 11b shows a plot of discharge coefficient versus internal hole geometry for a first nozzle hole having no inlet rounding and for a second nozzle hole having inlet rounding;

15 Figures 12a to 12f show a comparison in internal pressure and velocity fields for known hole geometries and hole geometries in accordance with embodiments of the present invention;

Figures 13a to 13d show the effects of increasing hole taper on fluid exit velocity for two holes of different lengths;

20

Figures 14a to 14f show the effect of hole taper on spray penetration into the combustion volume;

Figure 15 shows a plot of emission and particulate levels for a known hole geometry and
25 a hole geometry in accordance with embodiments of the present invention;

Figure 16 shows a comparison of CO_2 emission levels for a known hole geometry and a hole geometry in accordance with embodiments of the present invention;

Figures 17a to 17d show plots of fuel consumption, filter smoke number (FSN), boost pressure and exhaust temperature for a known hole geometry and a hole geometry in accordance with embodiments of the present invention.

5 In the following description the present invention is discussed in relation to its application to fuel injection nozzles. It is to be noted however that the present invention may be applied to any type of injection nozzle used to inject a fluid from a first volume into a second volume. For example, the injection nozzle may be used to inject liquid fuel from a supply volume into a heating/combustion chamber in a domestic heating system. Other
10 applications for the present invention include gasoline direct injection systems and furnaces.

It is further noted that the use of the injection nozzle in accordance with embodiments of the present invention described below are not limited to any particular type of engine.

15

In the following description it is noted that like numerals are used to denote like features.

It is also noted that the terminology $Average\left(\left|\frac{dS}{dx}\right|\right)$ is used as a shorthand notation in the

description below to describe the manner in which the separation of the walls of an injection hole change along the length of the injection hole. In the above expression, S
20 relates to the separation of the walls of the injection nozzle within a section taken along the passage way formed by the injection hole and the expression is taken to mean that at any given point along the section (or at any given point along a substantial length of the hole length) the "gradient" of the wall separation will always exceed the stated value. It is noted that non-linear wall profiles are therefore included within this expression but that
25 the minimum value of the value $|dS/dx|$ will always exceed the stated value (even though the value may vary along the length of the injection hole or may vary along the substantial portion of the injection hole for which the condition is defined).

Turning to Figures 1 and 2, a fuel injection nozzle 1 is shown comprising an injection
30 needle 3 located in a bore 5 of the nozzle body 7. The nozzle further comprises a feedhole 9 for the supply of fuel to a fuel gallery 11. The needle 3 is constrained to move by an upper guide 13 and lower guide 15. A series of injection holes 17 in the tip of the body 7 allow fuel to be injected from a nozzle sac 19 at the base of the injection nozzle 1 into a combustion space (not shown) when the needle lifts from its seat 21.

35

Figure 3 shows a section through a nozzle hole. It is noted that the hole inlet 25 has a diameter D_{in} and the hole outlet 27 a diameter D_{out} and that $D_{in} > D_{out}$. It is noted that as the distance x along the hole axis 29 increases, the walls 31 of the hole converge to form a tapered internal geometry. The dimensions of Figure 3 have been exaggerated for illustrative purposes but it is noted that typically the hole will have a length in the order of 1 millimetre (1000 μ m) and the difference between D_{in} and D_{out} will be in the range 10 μ m to 25 μ m.

Figure 4 shows a section through an injection nozzle 1 with a single row of injection holes 17. Figure 5 shows an alternative arrangement in which there are two rows of injection holes.

Figure 6 shows a section through a nozzle hole 17 in accordance with an embodiment of the present invention. Three separate hole internal geometries are shown in Figure 6 (denoted by the three wall positions 31a, 31b and 31c). It is noted that in comparison to the injection nozzle of Figure 3, the hole inlet 25 in Figure 6 is significantly larger than the hole outlet 27.

In Figure 6 the diameter, D , of the hole at a position x along the hole axis is designated as $D(x)$ and it is noted that $Average\left(\left|\frac{dS}{dx}\right|\right) > 45 \mu\text{m/mm}$. In other words, the minimum value of $|dD/dx|$ along the central hole axis is > 45 microns per millimetre. It is noted however that the gradient of $|dD/dx|$ may vary along the axis such that the profile of the hole walls is non-linear.

As is described below all the various hole geometries shown in Figure 6 provide improved injector performance in comparison to known injection nozzles if the rate of change of the hole diameter (or hole wall separation for non-circular cross sections) exceeds 45 microns per millimetre.

As noted above in Figure 6, the cross sectional profile of the hole need not be circular. As shown in Figures 7a to 7d, circular, elliptical, rectangular and even semi-circular hole cross sections may also be used in conjunction with embodiments of the present invention as long as, for at least one section along the hole axis, the wall separation of the hole, along a substantial length of the hole, satisfies the condition that $Average(|dS/dx|) > 45 \mu\text{m/mm}$, where S = wall separation.

Non-circular hole cross sections may offer performance advantages, e.g. a rectangular hole design may inject a sheet of fuel into a combustion chamber which may be preferable in certain circumstances to a jet as would be injected with a circular hole.

5

Figure 8 shows a plot of discharge coefficient C_d versus the hole internal geometry for a circular cross-sectional nozzle hole. It can be seen that the Figure covers internal hole geometries that vary from cylindrical ($dD/dx=0$) upto an extreme hole design in which the hole diameter changes by the equivalent of $180\mu\text{m}$ per $1000\mu\text{m}$. Results for five different hole inlet radii are shown.

10

For the purposes of Figure 8 the reference hole design equates to a discharge coefficient of between 0.85-0.88 and the y axis indicates percentage improvements relative to this design.

15

Current nozzle designs fall within the region indicated 50 and, for nozzle holes of length 1 millimetre, it can be seen that these hole geometries equate to a k_{factor} of between 0 and 3.

20

It can be seen from the figure that internal hole geometries whose wall separation increases at a rate of approximately $45\mu\text{m}/\text{mm}$ or more show a noticeable increase in discharge coefficient compared to current designs. It is also noted that the hole taper has a greater effect on the discharge coefficient of the hole than the inlet radius (i.e. the taper has a greater effect than local rounding of the hole inlet). It is further noted that once the wall separation increases at a rate greater than $60\mu\text{m}/\text{mm}$, the injection nozzle demonstrates a 5% performance increase.

25

Figures 9a to 9j show the effects of nozzle hole taper on internal hole fluid pressure and velocity. In Figures 9, three different hole geometries are tested and it can be seen from Figure 9a that the hole taper increases from left to right across the figure. In each hole tested the exit diameter of the hole is a constant.

30

Figures 9b, 9c and 9d relate to a cylindrical hole, i.e. hole taper = 0. Figure 9b shows the internal pressure field within the hole. The area to the far left of Figure 9b is the pressure within the bore 5 of the injection nozzle and it can be seen that for the taper=0 design there is a sudden and significant pressure drop at the inlet to the nozzle hole.

35

Figures 9c and 9d show the internal hole velocity field. Figure 9c shows the velocity field along the axis of the hole. Figure 9d shows the velocity field through a cross section through the hole outlet. It can be seen from Figures 9c and 9d that the maximum fluid velocity occurs at the hole inlet and that the maximum velocities concentrate around the hole axis. Towards the hole walls the velocity drops off towards lower values.

Figures 9e, 9f and 9g relate to a tapered nozzle hole in accordance with current known nozzle arrangements, i.e. hole taper = 10-25 μ m/mm. Figure 9e shows the internal hole pressure field for this hole arrangement and it can be seen that the pressure drop in the hole is more progressive than for the cylindrical hole geometry. The velocity field for this arrangement is shown in Figure 9f and this shows a more gradual flow acceleration than for the cylindrical hole arrangement. However, as can be seen from Figure 9g, the velocity field at the outlet is still concentrated about the hole axis.

Figures 9h, 9i and 9j relate to a tapered nozzle hole in accordance with an embodiment of the present invention, i.e. hole taper = 90 μ m/mm (hole length = 0.6mm in this example). In Figure 9h it can be seen that the nozzle arrangement in accordance with an embodiment of the present invention now shows a gradual pressure drop along the entire length of the nozzle hole. Furthermore, as can be seen from Figure 9i the velocity of the fluid accelerates towards the hole outlet and from Figure 9j it can be seen that the boundary layer in the outlet cross section is significantly thinner than in the first two hole geometries. This has the effect that the average speed of fluid exiting the hole is increased in comparison to the first two hole geometries.

Figures 10a to 10c show the data from Figure 9 in the form of graphical plots. Figure 10a confirms that the pressure drop along the hole axis is more gradual for the hole designed in accordance with an embodiment of the present invention (labelled "extreme design" in Figure 10a).

Figure 10b shows that for the cylindrical and current reference hole geometries there is an initial acceleration at the hole inlet followed by an extended period of substantially constant fluid velocity. In the geometry in accordance with an embodiment of the present invention by contrast there is a gradual acceleration along the entire hole length.

Figure 10c confirms that the fluid velocity at across the hole outlet is more uniform with a hole geometry in accordance with an embodiment of the present invention.

Figure 11a shows a plot of improvement in discharge coefficient (compared to a
5 reference geometry) versus internal hole geometry. Two separate plots are shown, the first for a nozzle hole of length 0.6mm and the second for a nozzle hole of length 1.2mm.

It can be seen that for hole taper values in accordance with current known production designs the length of the hole has a noticeable effect on the performance of the nozzle.
10 However, for higher values of $|dD/dx|$ (i.e. for values in accordance with an embodiment of the present invention) the hole length becomes less important and from a value of approximately $80\mu\text{m}/\text{mm}$ the nozzle performance appears to be independent of nozzle hole length.

15 Figure 11b a plot of discharge coefficient versus hole geometry for a hole without inlet rounding and a hole with inlet rounding. It can be seen that for lower hole taper values hole rounding is more significant than at higher hole taper values.

Figures 12a to 12f show a comparison in internal pressure and velocity fields for known
20 hole geometries and hole geometries in accordance with embodiments of the present invention.

Figures 12a and 12b relate to a hole with a $|dD/dx|$ value of approximately $30\mu\text{m}/\text{mm}$. It can be seen that there is a large and sudden pressure drop within the hole and the
25 velocity field shows a large high velocity area which leads to high energy losses.

Figures 12c to 12f show two hole geometries with a $|dD/dx|$ value of $180\mu\text{m}/\text{mm}$. Figures 12c and 12d relate to a hole that has a linear wall profile along the hole axis. Figures 12e and 12f relate to a hole that is initially parabolic in profile and then subsequently linear in
30 profile. In both cases the $|dD/dx|$ value is equal to or exceeds $180\mu\text{m}/\text{mm}$ along the entire section of the hole.

It can be seen that the two hole profiles shown in Figures 12c to 12f exhibit similar behaviour indicating that the actual profile of the hole along the axis does not affect the
35 performance of the nozzle. In both cases it can be seen that there is a smooth discharge area and the higher fluid velocities are located in the vicinity of the hole outlet.

Figures 13a and 13b show the effect of increasing the taper of a hole of length 0.6mm from 0 to 50 μ m/mm . It can be seen from Figure 13a that the velocity field within the hole is substantially “U” shaped. In Figure 13b by contrast the velocity field is more uniform at the hole outlet.

Figures 13c and 13d show a similar velocity field plot for a hole of length 0.9mm. Again, the increased taper geometry shows an improvement in homogenous velocity at the exit of the hole.

Figures 14a to 14f show the effect of hole taper on spray penetration into a combustion volume. Figures 14a to 14c show spray penetration at three different crank angles (6 degrees before top dead centre; 24 degrees after top dead centre; and, 44 degrees after top dead centre) for a cylindrical nozzle hole. It can be seen that the spray does not mix well, especially in Figure 14c where there is an area of unused air (circled in Figure 14c).

Figures 14d to 14f show spray penetration at the same three crank angles for a nozzle hole with relatively high taper (in this example the taper is 50 μ m/mm). It can be seen that compared to the hole design of Figures 14a to 14c there is an improvement in spray penetration and mixing.

Figures 15, 16 and 17a to 17d show results that compare a reference hole and a high performance hole geometry. It is noted that in each case the reference nozzle comprises a design at the limit of current production values (e.g. 25 μ m/mm) and the high performance nozzle comprises a hole taper of approximately 100 μ m/mm. In all cases the nozzles are 6 hole nozzles.

Figure 15 shows a comparison of particulate emissions and NO_x emissions for a reference (i.e. known) nozzle design and a nozzle in accordance with embodiments of the present invention. It can be seen that the nozzle in accordance with embodiments of the present invention demonstrates a reduction of particulate emissions of upto 40% compared to the known design.

Figure 16 shows that a reduction in CO₂ emissions may also be achieved with nozzles in accordance with embodiments of the present invention in comparison to known nozzle hole geometries.

Figure 17a to 17d illustrate an assessment of a nozzle in accordance with embodiments of the present invention on a multi-cylinder engine operating at full load. At full load an improved global combustion efficiency was observed in comparison to known nozzle hole designs. At the same power point the engine comprising nozzle designs in accordance with the present invention demonstrated lower fuel consumption (approximately a 1.5 % improvement compared to the reference system); lower smoke emissions (-1 FSN) and a lower exhaust temperature (approximately 10°C compared to the reference system).

The present invention may be implemented in a fuel injector, such as a common rail injector, in which a common supply (rail) delivers fuel to at least one injector of the engine, or may be implemented in an electronic unit injector (EUI) in which each injector of the engine is provided with its own dedicated pump and, hence, high pressure fuel supply. The invention may also be implemented in a hybrid scheme, having dual common rail/EUI functionality.

The invention may also be implemented in any system where a fluid is injected from a first volume to a second volume.

It will be understood that the embodiments described above are given by way of example only and are not intended to limit the invention, the scope of which is defined in the appended claims. It will also be understood that the embodiments described may be used individually or in combination.

CLAIMS

1. An injection nozzle for injecting a fluid, the injection nozzle comprising: a nozzle body and a nozzle hole defining a flow passage for fluid, the flow passage comprising passage walls and the nozzle hole having an inlet in fluid communication via the flow passage with an outlet, wherein,

the inlet is larger than the output and for at least one section through the inlet and outlet along the flow passage the nozzle hole is defined, for all distances x within a substantial length of the flow passage, by the condition:

$$\left| \left(\frac{dS}{dx} \right) \right| > 45 \text{ microns/millimetre,} \quad \text{where } S = \text{passage wall separation}$$

and x is the distance from the inlet.

2. An injection nozzle as claimed in Claim 1, wherein the nozzle hole is defined by the condition $\left| \left(\frac{dS}{dx} \right) \right| > 60 \text{ microns/millimetre.}$

3. An injection nozzle as claimed in Claim 1 or Claim 2, wherein the nozzle hole is defined by the condition $\left| \left(\frac{dS}{dx} \right) \right| > 80 \text{ microns/millimetre.}$

4. An injection nozzle as claimed in any preceding claim, wherein the inlet and outlet define a nozzle hole axis and the at least one section is taken through the axis.

5. An injection nozzle as claimed in Claim 4, wherein the condition is satisfied for all sections through the axis.

6. An injection nozzle as claimed in any preceding claim, wherein the wall condition holds for at least 40% of the length of the flow passage.

7. An injection nozzle as claimed in any preceding claim, wherein the nozzle hole has a circular cross section along the length of the flow passage.

8. An injection nozzle as claimed in any one of Claims 1 to 6, wherein the nozzle hole has an elliptical cross section along the length of the flow passage.

9. An injection nozzle as claimed in Claim 8, wherein sections taken through either the major and minor axes or both axes of the ellipse satisfy the condition.

5 10. An injection nozzle as claimed in any one of Claims 1 to 6, wherein the nozzle hole has a substantially rectangular cross section along the length of the flow passage.

11. An injection nozzle as claimed in any preceding claim, wherein the nozzle body is provided with a bore in communication with a source of fluid and the injection nozzle is
10 arranged to inject fluid from the bore through the nozzle hole to a volume outside the injection nozzle.

12. An injection nozzle as claimed in any preceding claim, wherein the nozzle comprises a plurality of nozzle holes in accordance with the nozzle hole of Claims 1 to 11.

15 13. An injection nozzle as claimed in Claim 12, wherein the plurality of nozzle holes are arranged in one or more rows of holes.

14. An injection nozzle as claimed in any preceding claim, wherein the passage walls in
20 the at least one section define: a parabolic profile; or, a linear profile; or, a mixture of curved and linear profiles.

15. A fuel injector for an internal combustion engine comprising an injection nozzle as claimed in any preceding claim.

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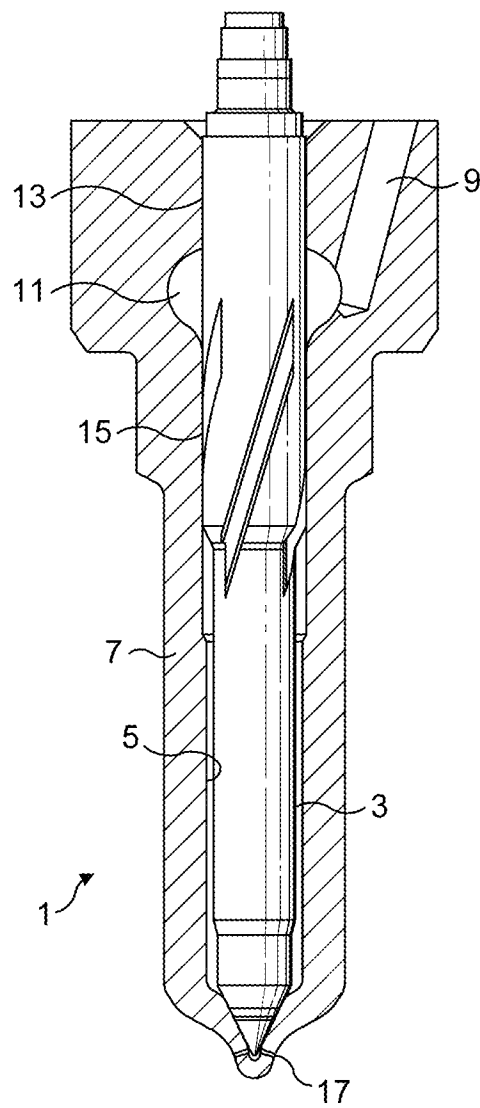


Fig. 1

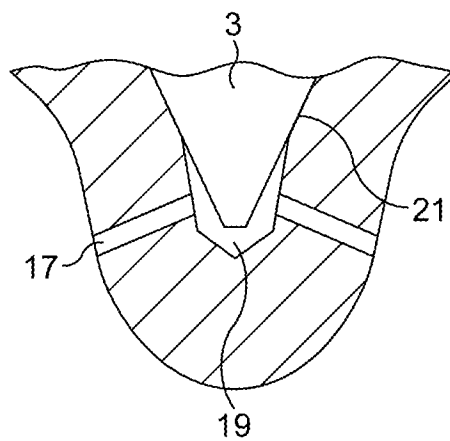


Fig. 2

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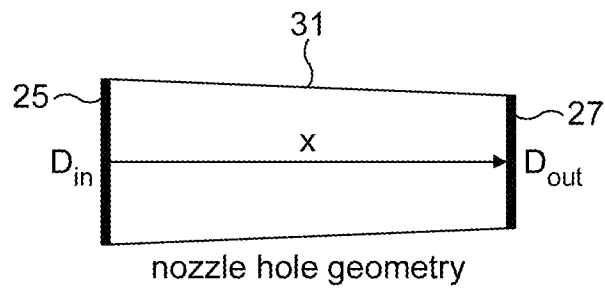


Fig. 3

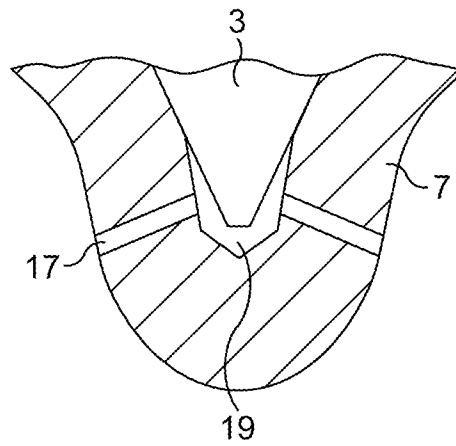


Fig. 4

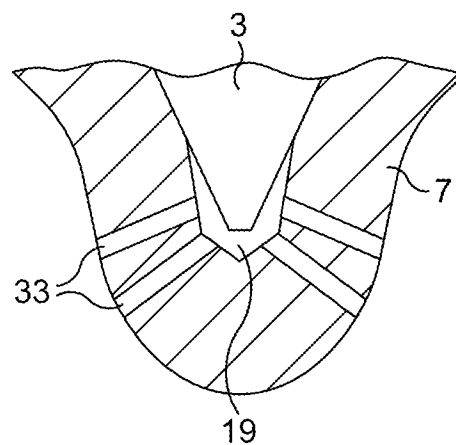
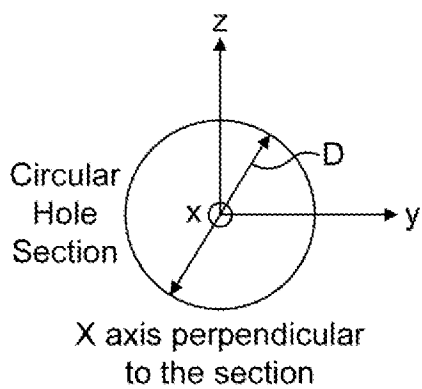
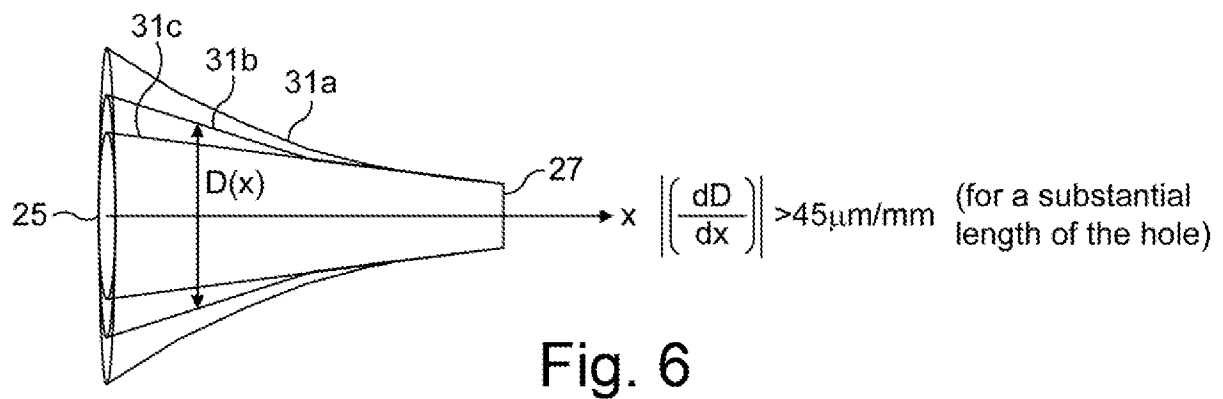


Fig. 5

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or

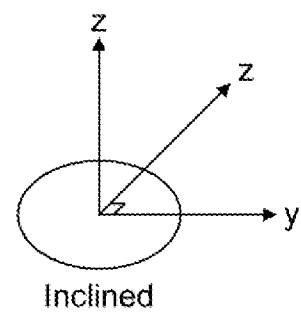
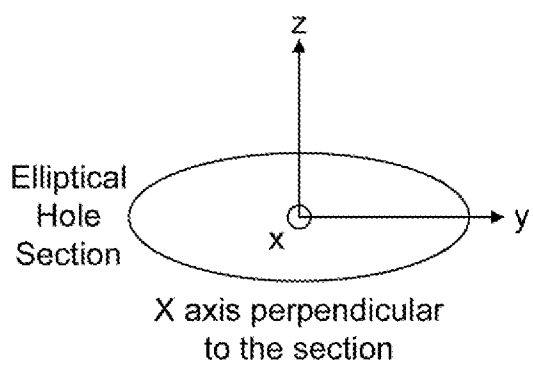


Fig. 7a



or

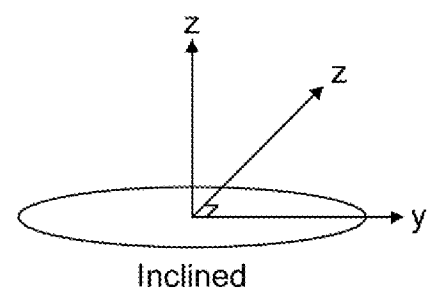


Fig. 7b

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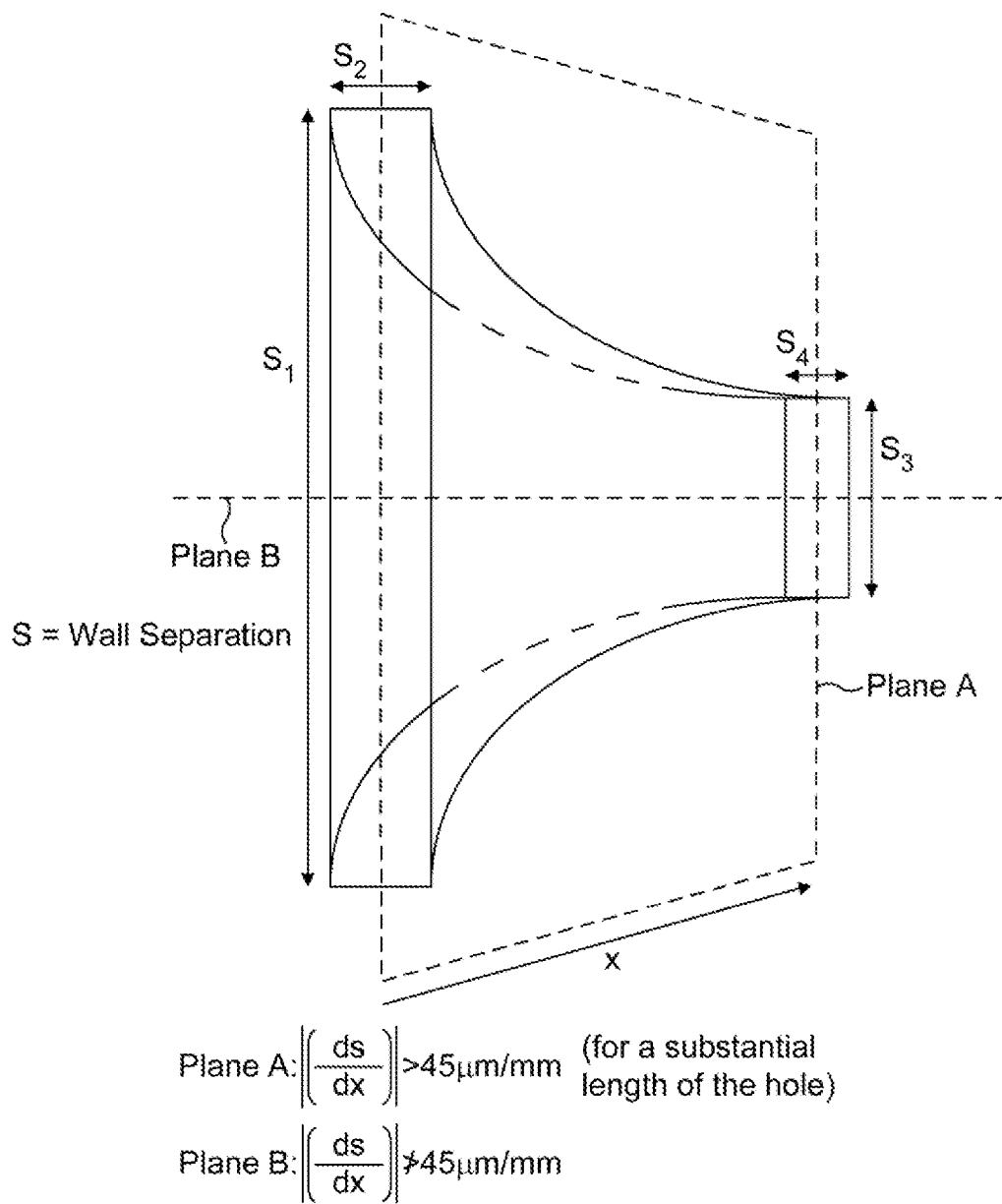


Fig. 7c

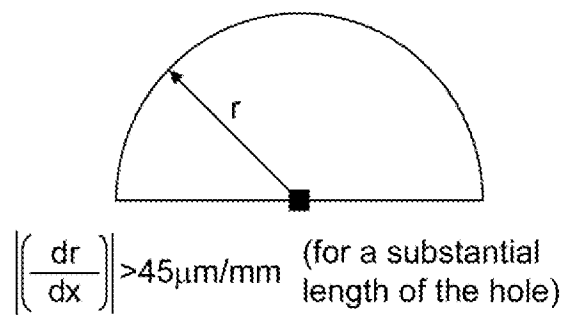


Fig. 7d

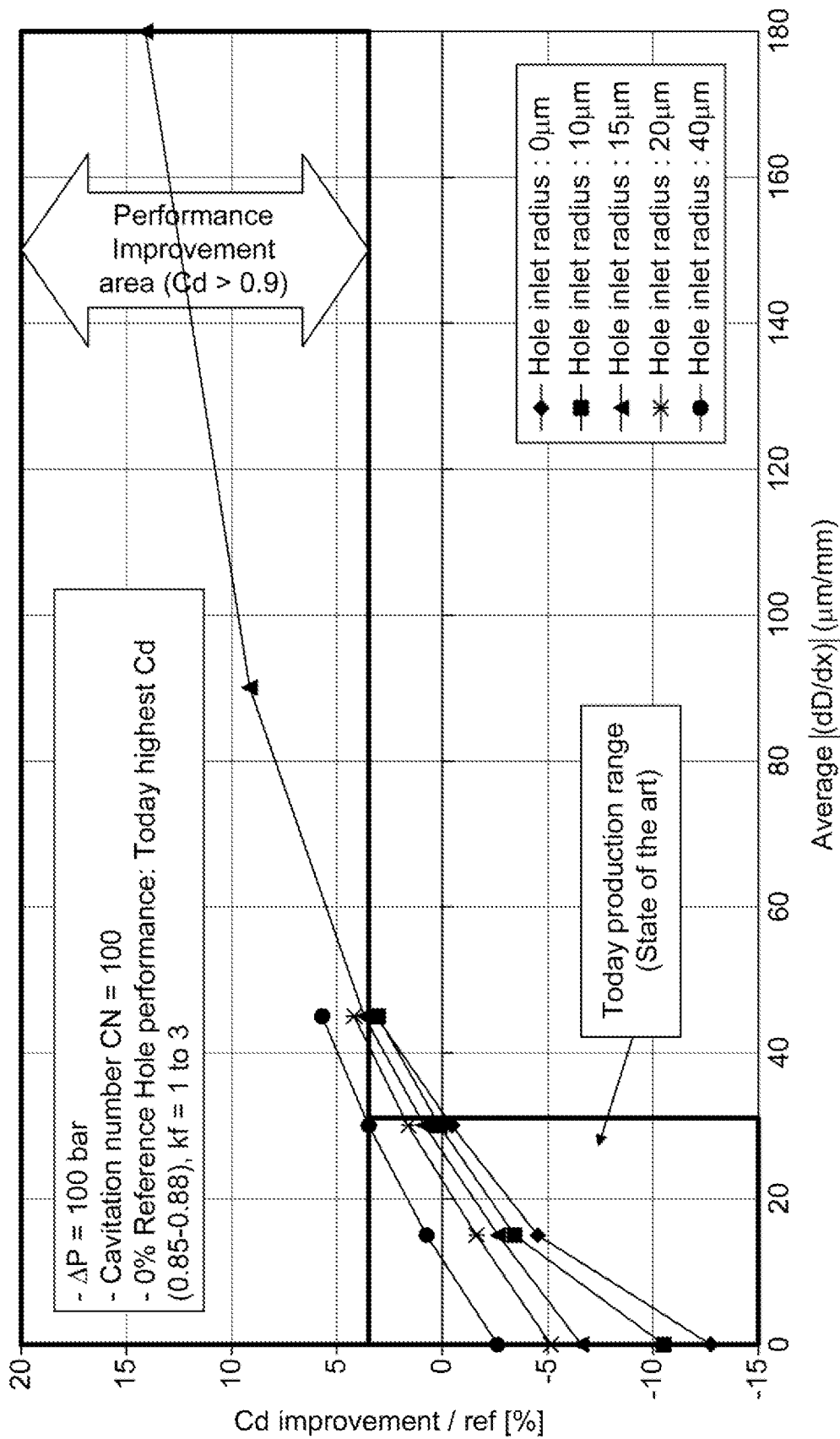


Fig. 8

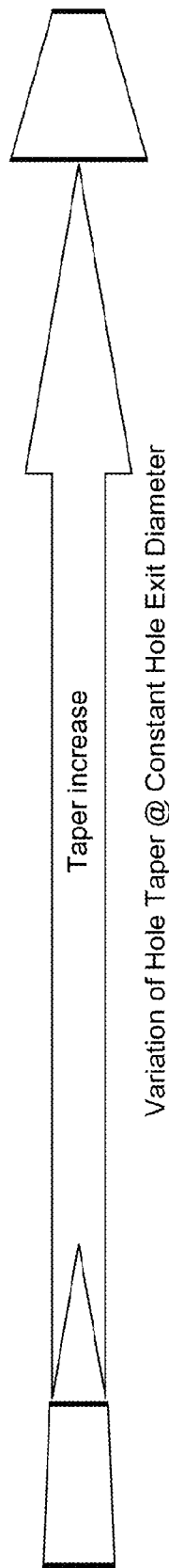


Fig. 9a

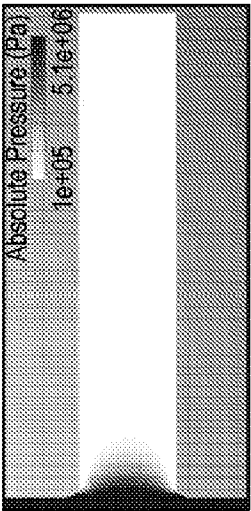


Fig. 9b

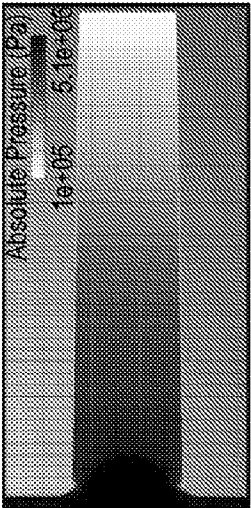


Fig. 9e

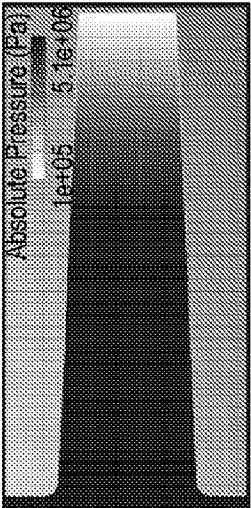


Fig. 9h

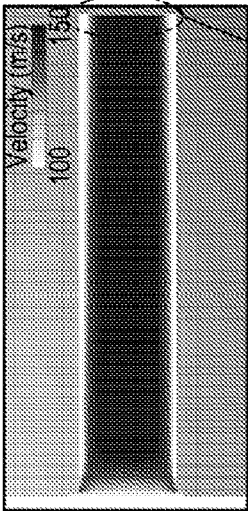


Fig. 9c

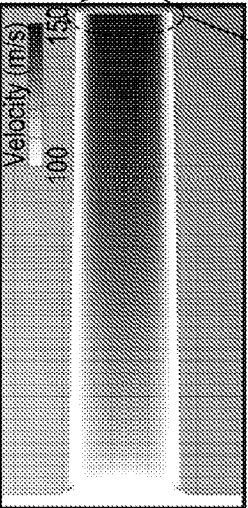


Fig. 9f

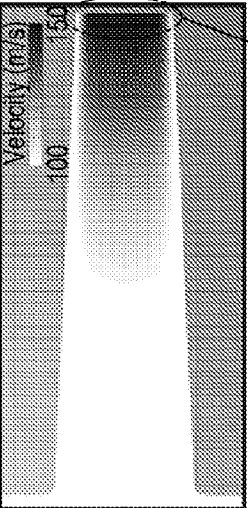


Fig. 9i

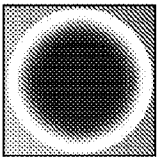


Fig. 9d

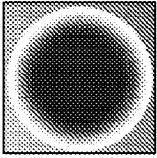


Fig. 9g

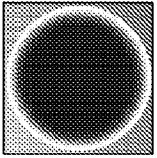


Fig. 9j

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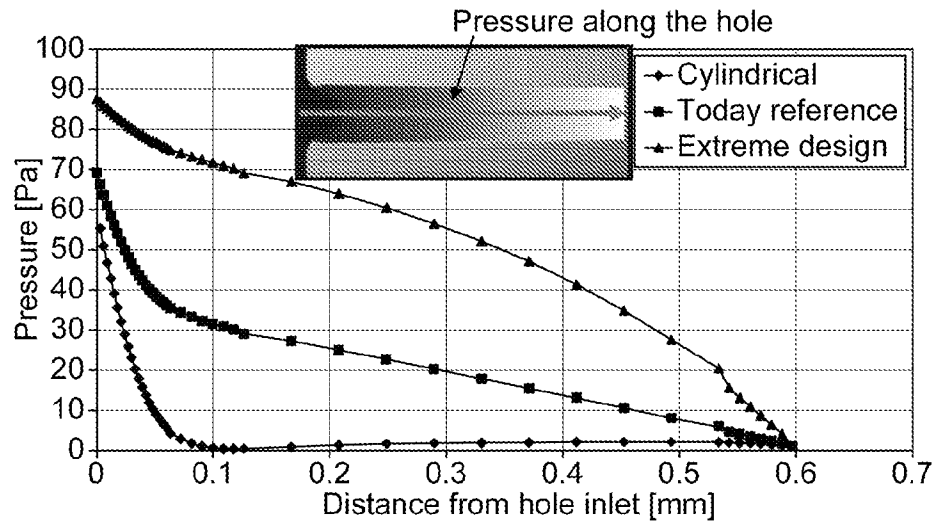


Fig. 10a

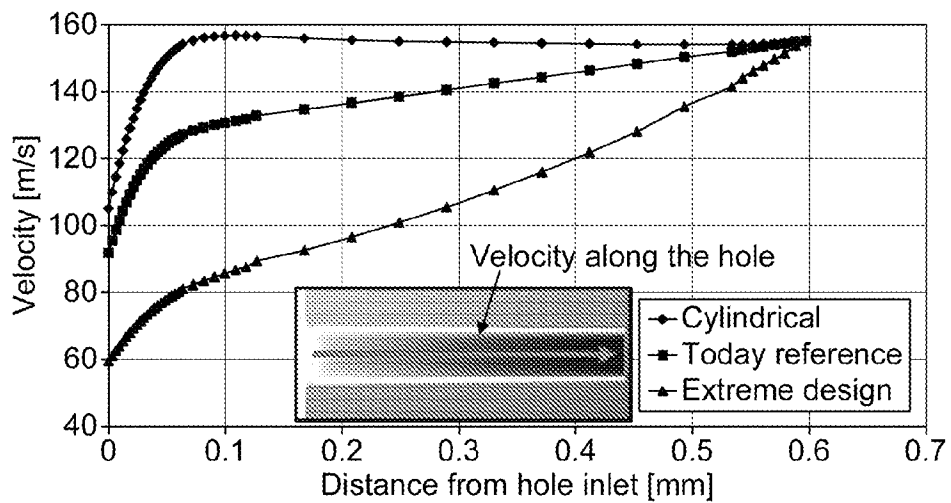


Fig. 10b

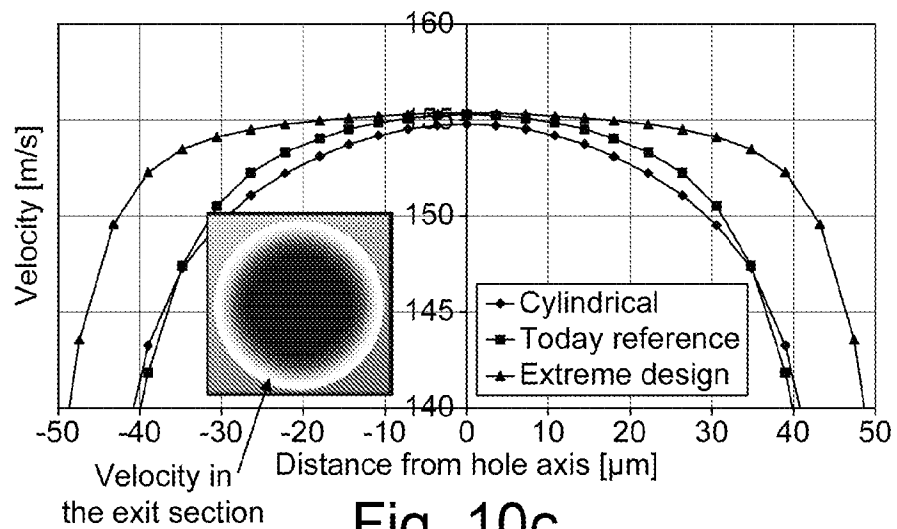


Fig. 10c

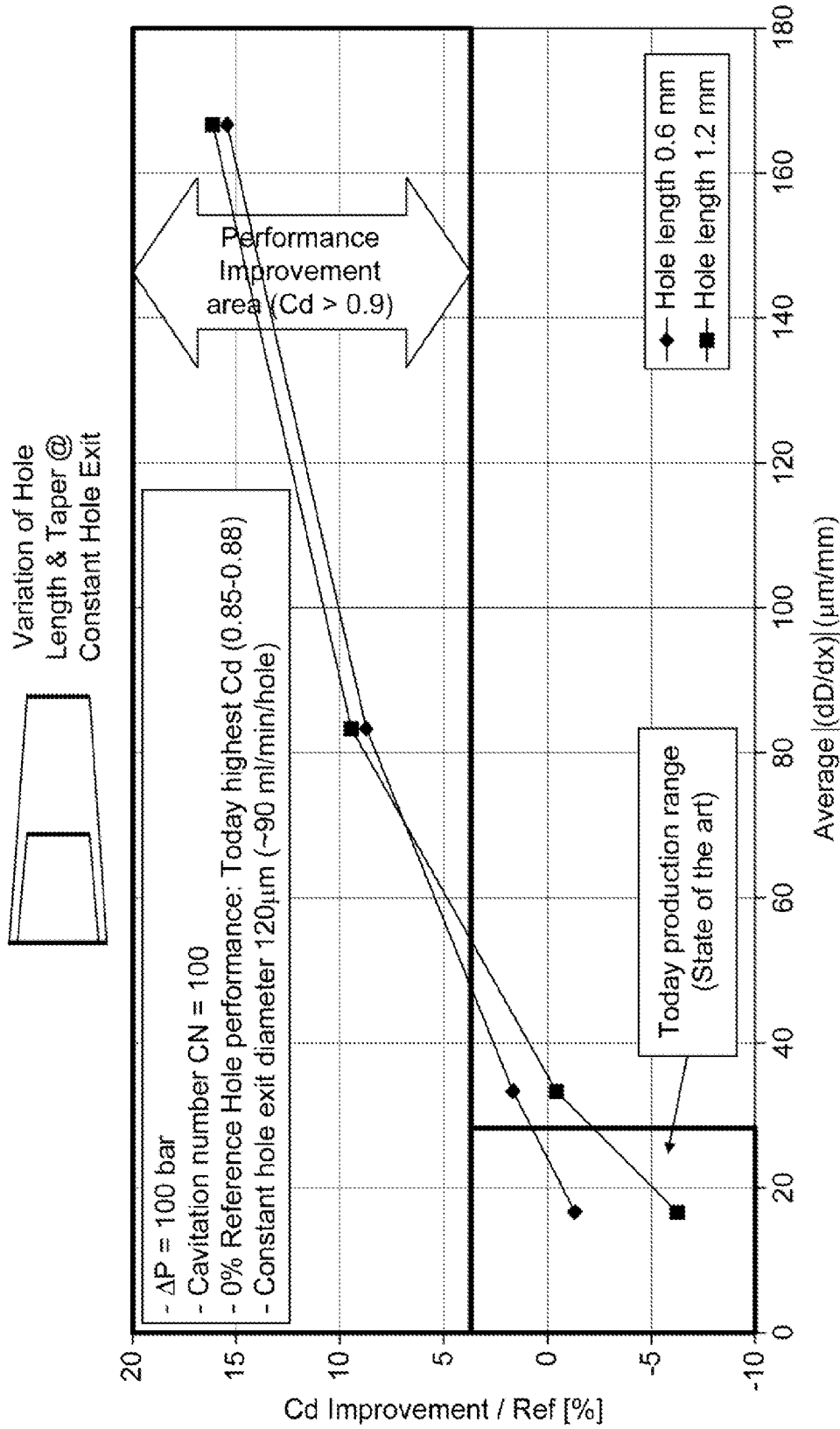


Fig. 11a

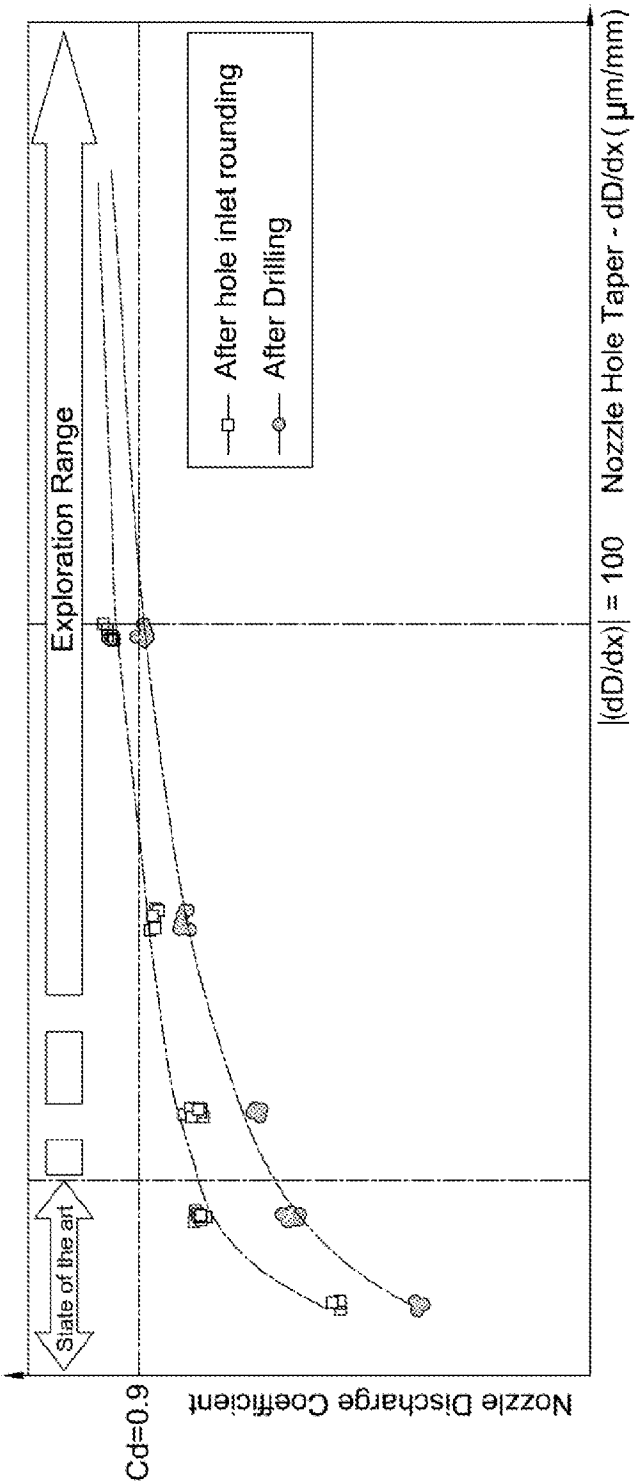


Fig. 11b

Average $|(dD/dx)| = 30\mu\text{m/mm}$

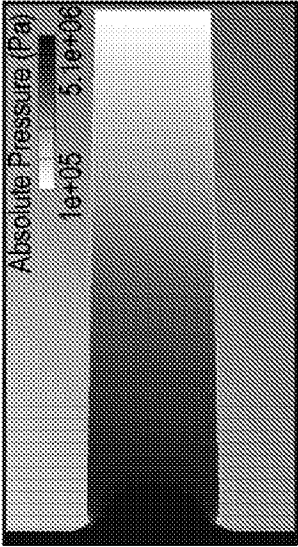


Fig. 12a

Average $|(dD/dx)| = 180\mu\text{m/mm}$

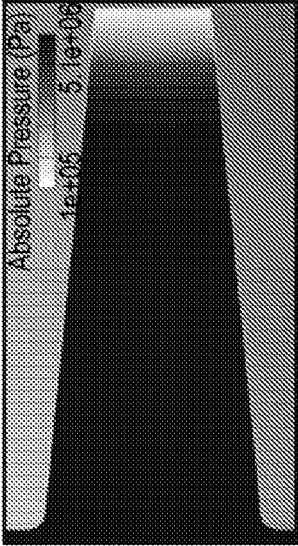


Fig. 12c

Average $|(dD/dx)| = 180\mu\text{m/mm}$

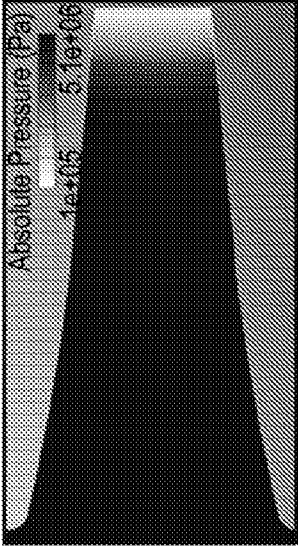


Fig. 12e

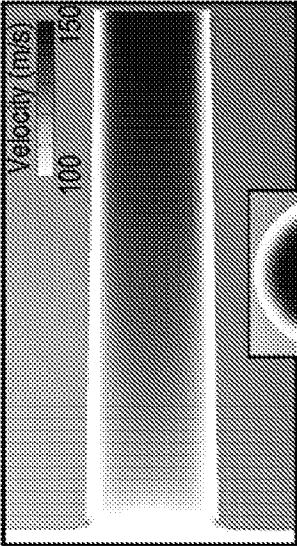


Fig. 12b

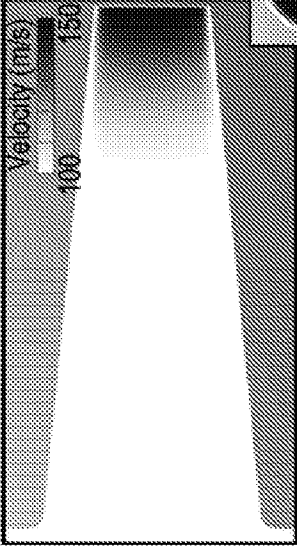


Fig. 12d

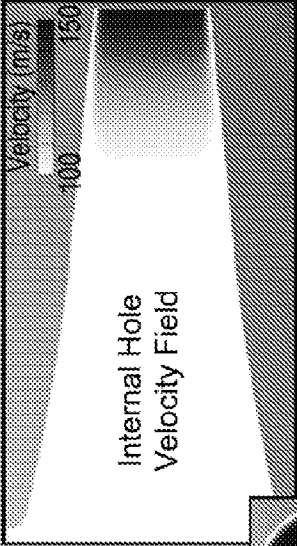
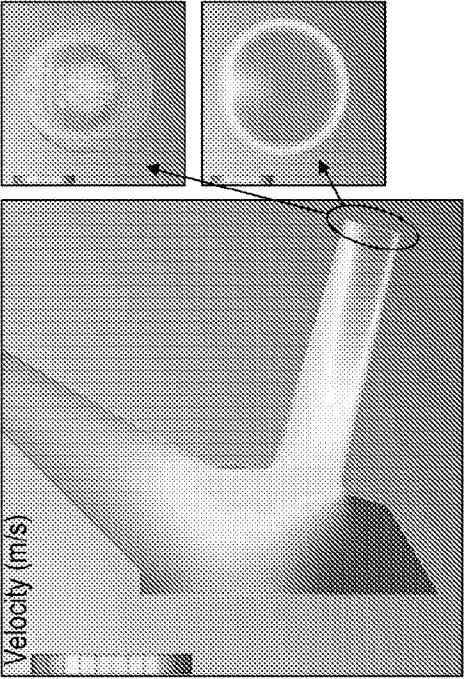


Fig. 12f



0.6 mm Hole Length
Taper / Honing
increase

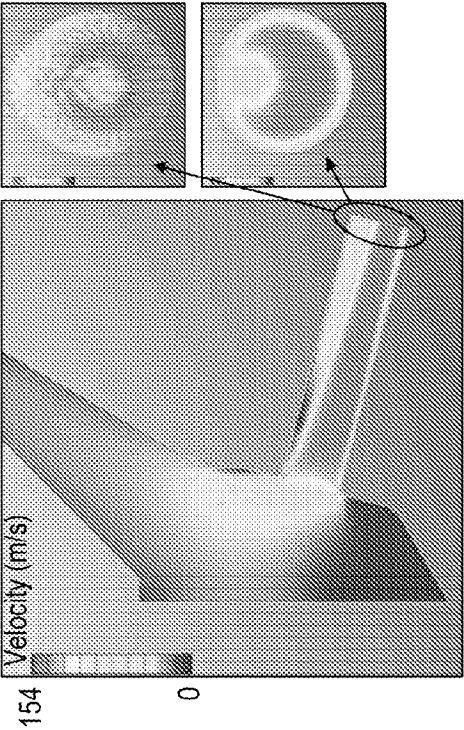
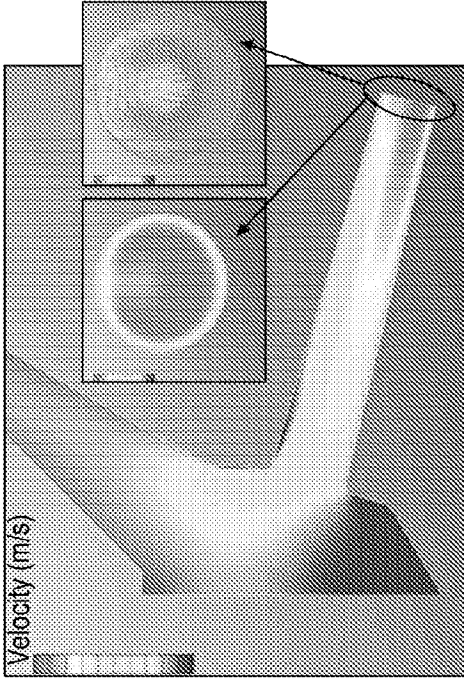


Fig. 13a



0.9 mm Hole Length
Taper / Honing
increase

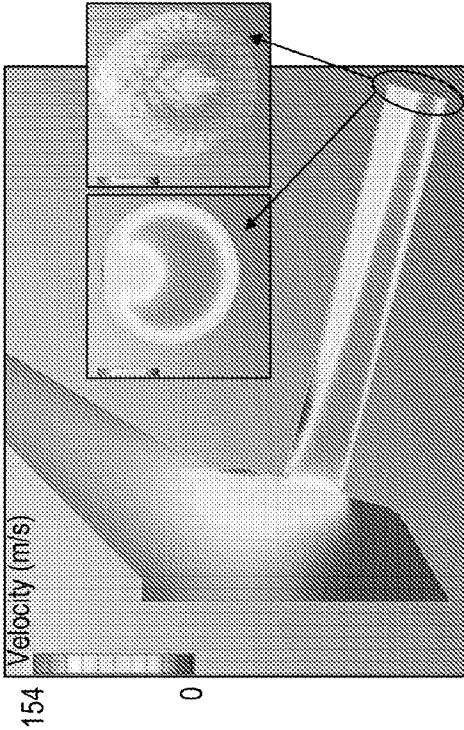


Fig. 13b

Fig. 13c

Fig. 13d

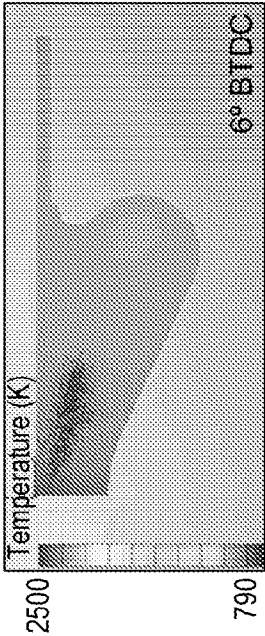


Fig. 14a

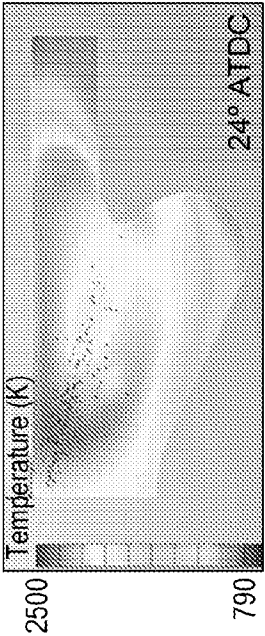


Fig. 14b

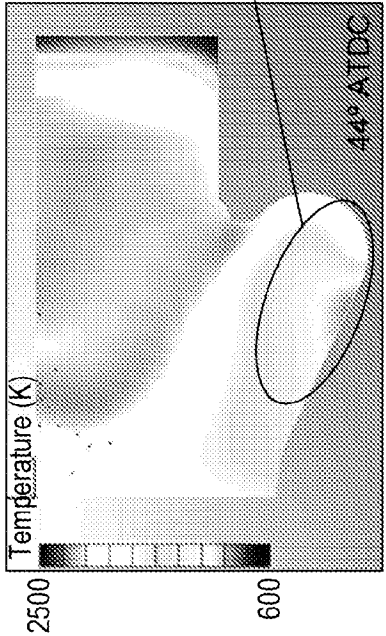


Fig. 14c

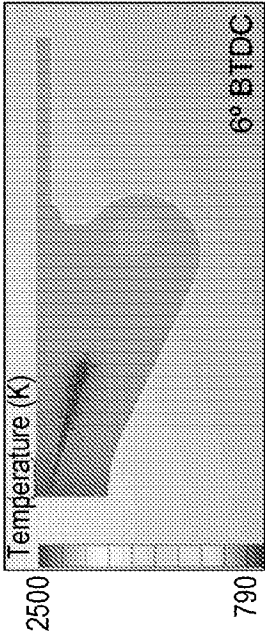


Fig. 14d

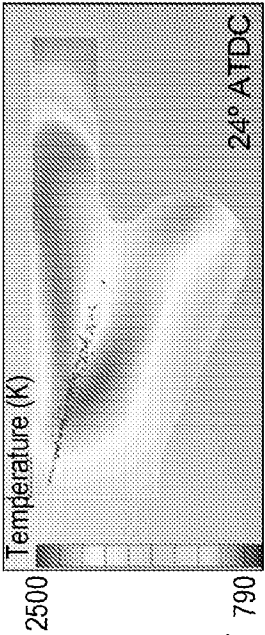


Fig. 14e

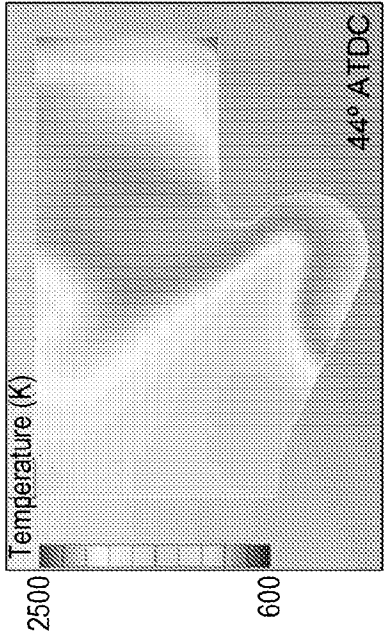
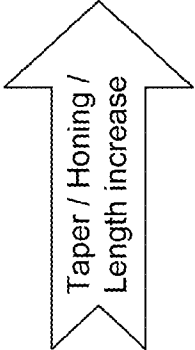


Fig. 14f



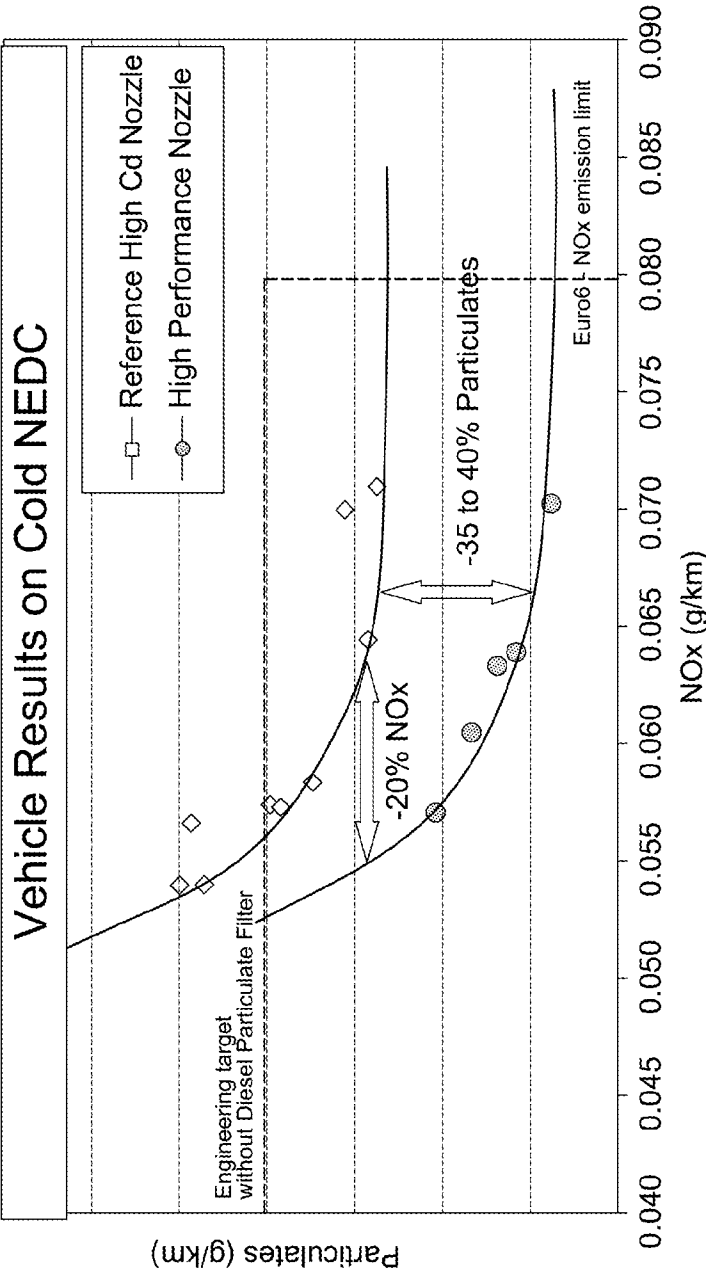


Fig. 15

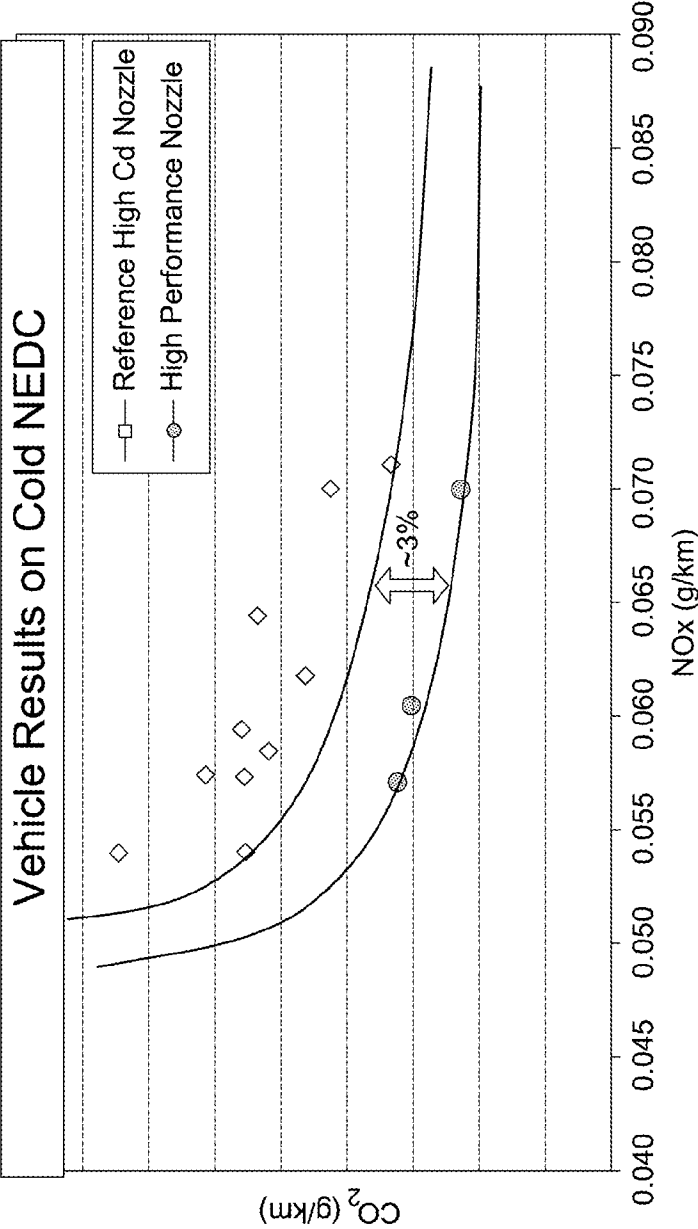


Fig. 16

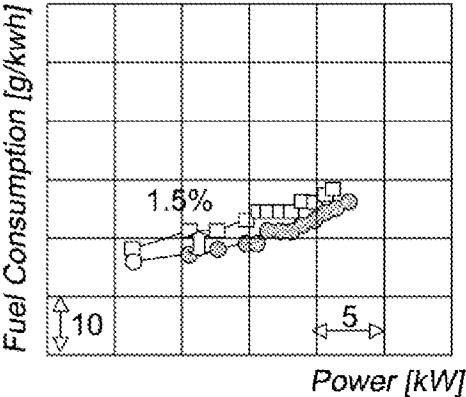
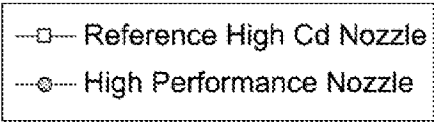


Fig. 17a

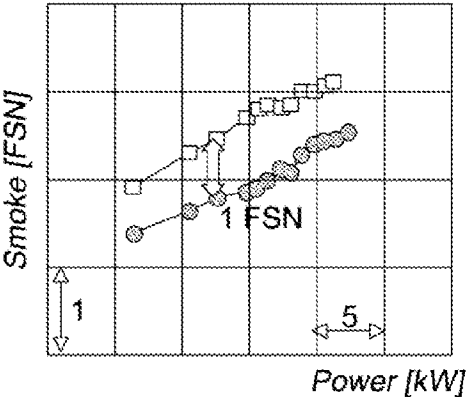


Fig. 17b

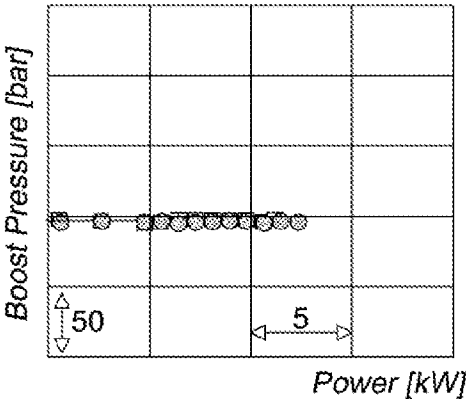


Fig. 17c

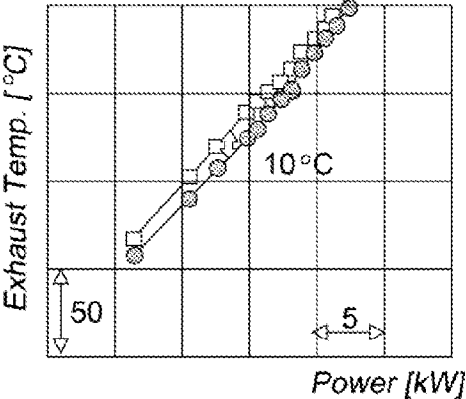


Fig. 17d

INTERNATIONAL SEARCH REPORT

International application No

PCT/EP2009/065070

A. CLASSIFICATION OF SUBJECT MATTER
INV. F02M61/18

According to International Patent Classification (IPC) or to both national classification and IPC

B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols)

F02M

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched

Electronic data base consulted during the international search (name of data base and, where practical, search terms used)

EP0-Internal

C. DOCUMENTS CONSIDERED TO BE RELEVANT

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X Y	DE 103 15 967 A1 (BOSCH GMBH ROBERT [DE]) 21 October 2004 (2004-10-21) paragraph [0020]; figures	1-7, 11-15 8-10, 13-14
X	----- YALCIN H ET AL: "KONISCHE EINSPRITZLOCHER EINES KRAFTSTOFFEINSPRITZVENTILS FUER BRENNKRAFTMASCHINEN" 1 January 1999 (1999-01-01), SIEMENS TECHNIK REPORT, SIEMENS AG., ERLANGEN, AT, PAGE(S) 73/74, XP000828544 ISSN: 1436-7777 the whole document	1-7, 11-15
X	----- EP 0 352 926 A (LUCAS IND PLC [GB]) 31 January 1990 (1990-01-31) cited in the application column 3, line 2 - line 9; figure 2 ----- -/--	1-7, 11-15

☒ Further documents are listed in the continuation of Box C.

☒ See patent family annex.

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"&" document member of the same patent family

Date of the actual completion of the international search

10 December 2009

Date of mailing of the international search report

21/12/2009

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Landriscina, V

INTERNATIONAL SEARCH REPORT

International application No
PCT/EP2009/065070

C(Continuation). DOCUMENTS CONSIDERED TO BE RELEVANT

Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
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Y	US 2004/237929 A1 (CAVANAGH MARK S [US] ET AL CAVANAGH MARK S [US] ET AL) 2 December 2004 (2004-12-02) figures 2-5	13
Y	WO 2004/040125 A (BOSCH GMBH ROBERT [DE]; HAFNER UDO [DE]; HANS WALDEMAR [DE]; BRENNER F) 13 May 2004 (2004-05-13) page 15, last paragraph - page 16, paragraph 1; figures 4,5	14
A	US 2007/040053 A1 (DATE KENJI [JP]) 22 February 2007 (2007-02-22) paragraph [0065]; figure 4	1-7, 11-15

INTERNATIONAL SEARCH REPORT

Information on patent family members

International application No

PCT/EP2009/065070

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