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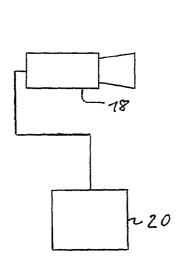
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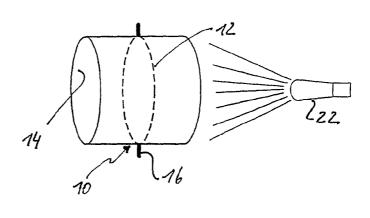
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(54) Title: SUPPORT SYSTEM FOR THE MONITORING OF AN AIR OUTLET VALVE IN AN AIRCRAFT





(57) **Abstract:** The invention proposes that a camera (18) takes a technical picture of an air outlet valve (10) of an aircraft, and the picture taken of the outlet valve (10) is displayed on a picture display unit (20), for example in the cockpit of the aircraft. By shining light onto the outlet valve from a light source (22), a clear picture with a high level of contrast is produced. The valve setting of the outlet valve (10) can thus be monitored visually, in a reliable manner, by the pilot.

Support system for the monitoring of an air outlet valve in an aircraft

The invention relates to a support system for the monitoring of an air outlet valve in an aircraft.

The outlet valve being considered here is a valve which serves to control the cabin pressure of a transport aircraft for passengers or goods. Air from the cabin flows through the outlet valve into the external surrounds. The cabin pressure depends upon the valve setting of the outlet valve.

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After landing the aircraft doors can only be opened if the outlet valve (just one outlet valve is referred to here; it is, however, clear that aircraft can also be equipped with two or more outlet valves) has been fully opened previously so that total equalisation of the cabin pressure and the atmospheric level on the ground can take place. For this, the outlet valve must not be covered or blocked. Incomplete opening of the outlet valve or an impaired flow of air through the valve can result in residual low pressure in the cabin, which can be dangerous. There have been numerous injuries in the past as a result of residual high pressure when the cabin doors were opened which can cause the doors to swing open towards the outside in an uncontrolled manner.

Generally, the valve setting of the outlet valve is determined by sensors and established in the cockpit of the aircraft. Despite monitoring by sensors, accidents have still happened because the sensors used are unreliable and indicated a fully open valve setting, even though this was not the case, or the valve was fully open, but covered by objects.

The aim of the invention, therefore, is to improve safety when opening aircraft doors.

As a solution to this problem, the invention proposes a support system for the monitoring of an air outlet valve in an aircraft, whereby this system includes a camera directed at the outlet valve and a picture display unit to show pictures of the outlet valve taken by the camera.

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By means of the technical picture representation of the outlet valve, an observer can identify clearly, and without doubt, the valve setting of the outlet valve shown by the picture display unit, and so be reliably certain that the outlet valve is actually in its fully open setting. The invention therefore follows on from the previous task of identifying the valve setting of the outlet valve with appropriate sensors and communicating the results identified to the pilot, the accuracy of which he has to depend upon. Instead of this, the invention offers the pilot (or another person who is responsible for checking the valve setting of the outlet valve) the possibility to check the valve with his own eyes — an unusual way in a world which tries more and more to relieve people of tasks by the use of technical devices. It has, however, been shown that this way provides greater safety, and accidents associated with opening the doors of aircraft can be better avoided.

The picture display unit is specifically positioned within the cockpit of the aircraft. It is not, however, fundamentally out of the question to position the picture display unit elsewhere in the aircraft, for example so that the personnel responsible for opening the aircraft doors have the possibility of checking the valve setting of the outlet valve.

In a preferred form, the outlet valve is provided with a valve lighting system, whereby the light is directed onto the outlet valve. By illuminating the outlet valve, an even higher level of safety is offered because the valve setting is more easily identified on the picture displayed. It can be advantageous to arrange the valve lighting system in such a way that, seen from the camera, rear lighting is created for the outlet valve. Such rear lighting can be helpful for more clearly identifying the edge contours of the valve elements of the outlet valve on the picture displayed, and so to be able to make a better assessment of the valve setting. In order to provide rear lighting, the camera and the valve lighting system can be directed, approximately at least, from the opposite side onto the outlet valve.

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The valve lighting system can have any number of lighting elements, for example just one, or several lighting elements distributed in an appropriate arrangement. With regard to the spectral composition of the light emanating from the valve lighting system, various solutions are possible here as well. The valve lighting system can be designed in such a way that the outlet valve is illuminated with infra-red light. Alternatively, black light can be used to illuminate the outlet valve. It is, of course, also possible to use visible light for the illumination of the outlet valve.

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Identification of the outlet valve on the picture displayed can be improved further by providing at least parts of the outlet valve with reflective coating for at least one part of the light spectrum of the valve lighting system. It is also possible to provide at least parts of the outlet valve with a coating which absorbs part of the light spectrum of the valve lighting system. The spectral sensitivity of the camera is specifically suited to the spectrum of the light emitted and appropriate to the absorptive and/or reflective properties of the outlet valve. For example, the camera can be a monochrome camera.

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Independent of the provision of the camera and the picture display unit, heightened safety when opening aircraft doors can also be offered by the valve lighting system alone. The outlet valve can, indeed, also be checked by direct observation without transferring a picture of the outlet valve to a place some distance away. Here, the valve lighting system improves identification of the outlet valve which is often installed in a relatively dark position at the stern of the aircraft.

An example of the invention is illustrated on the attached drawing in fig. 1. This

gives a schematic representation of an outlet valve 10 of an aircraft which is not illustrated. The outlet valve 10 is in the form of a flap valve with a so-called butterfly valve flap 12, which is built into an air outlet channel 14 so as to rotate around an axis 16. A camera 18 is directed onto the outlet valve 10 from one side, and this takes a picture of the outlet valve 10. The camera 18 sends its picture to a monitor 20, on which the picture of the outlet valve 10 is displayed. The monitor is positioned, for example, in the cockpit of the aircraft. From the opposite side, light, which can be infra-red light, black light, or another light, is shone onto the outlet valve 10 by a lighting source 22. By means of this background lighting, a particularly

contrast-rich picture can be obtained which allows the contours of the valve flap 12

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to be clearly identified.

In fig. 2 the valve flap 12 is shown in a partially closed setting. If the camera 18 takes a picture of the outlet valve 10 more or less from directly in front, the valve flap 12 in a partially closed setting appears as an ellipse on the picture displayed. The wider the ellipse, the more firmly shut the outlet valve 10. From this, the pilot can safely and reliably assess the valve setting of the outlet valve 10. Only when the valve flap 12 appears on the picture displayed more or less just as a line, as shown in fig. 3, the pilot knows that the outlet valve 10 is fully open.

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Patent Claims

- Support system for the monitoring of an air outlet valve (10) in an aircraft, characterised by a camera (18) directed onto the outlet valve (10), and a picture display unit (20) for the display of a picture of the outlet valve taken by the camera.
- 2. System in accordance with claim 1, characterised in that the picture display unit (20) is positioned within the cockpit of the aircraft.

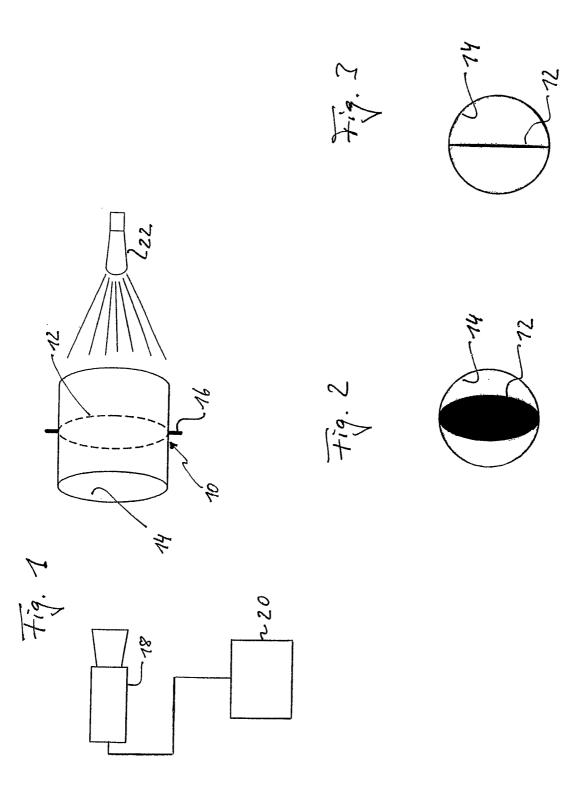
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- 3. System in accordance with claim 1 or 2, characterised by a valve lighting system (22) assigned to the outlet valve, the light of which is directed onto the outlet valve (10).
- 4. System in accordance with claim 3, characterised in that the camera (18) and the valve lighting system (22) are directed onto the outlet valve (10) from more or less the opposite side.
- 5. System in accordance with claim 3 or 4, characterised in that the valve lighting system (22) for the lighting of the outlet valve (10) uses infra-red light.
- 25 6. System in accordance with claim 3 or 4, characterised in that the valve lighting system (22) for the lighting of the outlet valve (10) uses black light.
- 7. System in accordance with any of the claims 3 to 6,
 characterised in that at least parts of the outlet valve (10) are provided with a
 covering which reflects at least part of the light spectrum of the valve lighting system
 (22).
- 8. System in accordance with any of the claims 3 to 6,
 characterised in that at least parts of the outlet valve (10) are provided with a
 coating that absorbs part of the light spectrum of the valve lighting system (22).

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- 9. System in accordance with any of the previous claims, characterised in that the camera (18) is a monochrome camera.
- 10. Support system for the monitoring of an air outlet valve (10) in an aircraft, characterised by a valve lighting system (22) assigned to an outlet valve, the light of which is directed onto the outlet valve (10), in particular in connection with other features of at least one of the claims 1, 2 and 4 to 9.



INTERNATIONAL SEARCH REPORT

Internal Application No

A. CLASSIFICATION OF SUBJECT MATTER IPC 7 B64D13/04 B64E B64D47/08 B64F5/00 According to International Patent Classification (IPC) or to both national classification and IPC **B. FIELDS SEARCHED** Minimum documentation searched (classification system followed by classification symbols) B64D B64F Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched Electronic data base consulted during the international search (name of data base and, where practical, search terms used) EPO-Internal, WPI Data C. DOCUMENTS CONSIDERED TO BE RELEVANT Citation of document, with indication, where appropriate, of the relevant passages Relevant to claim No. A US 4 553 474 A (WONG ET AL) 1,10 19 November 1985 (1985-11-19) column 10, lines 21,22; figure 4 Α US 2003/071899 A1 (JOAO RAYMOND ANTHONY) 1,10 17 April 2003 (2003-04-17) paragraphs '0428!, '0429!, '0446! Α FR 2 686 568 A (AEROSPATIALE STE NATIONALE 1,10 INDUSTRIELLE) 30 July 1993 (1993-07-30) page 2, lines 12-19 Further documents are listed in the continuation of box C. Patent family members are listed in annex. ° Special categories of cited documents : *T* later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle or theory underlying the *A* document defining the general state of the art which is not considered to be of particular relevance invention "E" earlier document but published on or after the international *X* document of particular relevance; the claimed invention cannot be considered novel or cannot be considered to involve an inventive step when the document is taken alone "L" document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another citation or other special reason (as specified) *Y* document of particular relevance; the claimed invention cannot be considered to involve an inventive step when the document is combined with one or more other such docu-ments, such combination being obvious to a person skilled in the art. "O" document referring to an oral disclosure, use, exhibition or *P* document published prior to the international filling date but later than the priority date claimed *&* document member of the same patent family Date of the actual completion of the International search Date of mailing of the international search report 5 April 2005 13/04/2005 Name and mailing address of the ISA Authorized officer European Patent Office, P.B. 5818 Patentlaan 2 NL – 2280 HV Rijswijk Tel. (+31–70) 340–2040, Tx. 31 651 epo nl, Fax: (+31–70) 340–3016 Salentiny, G

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