



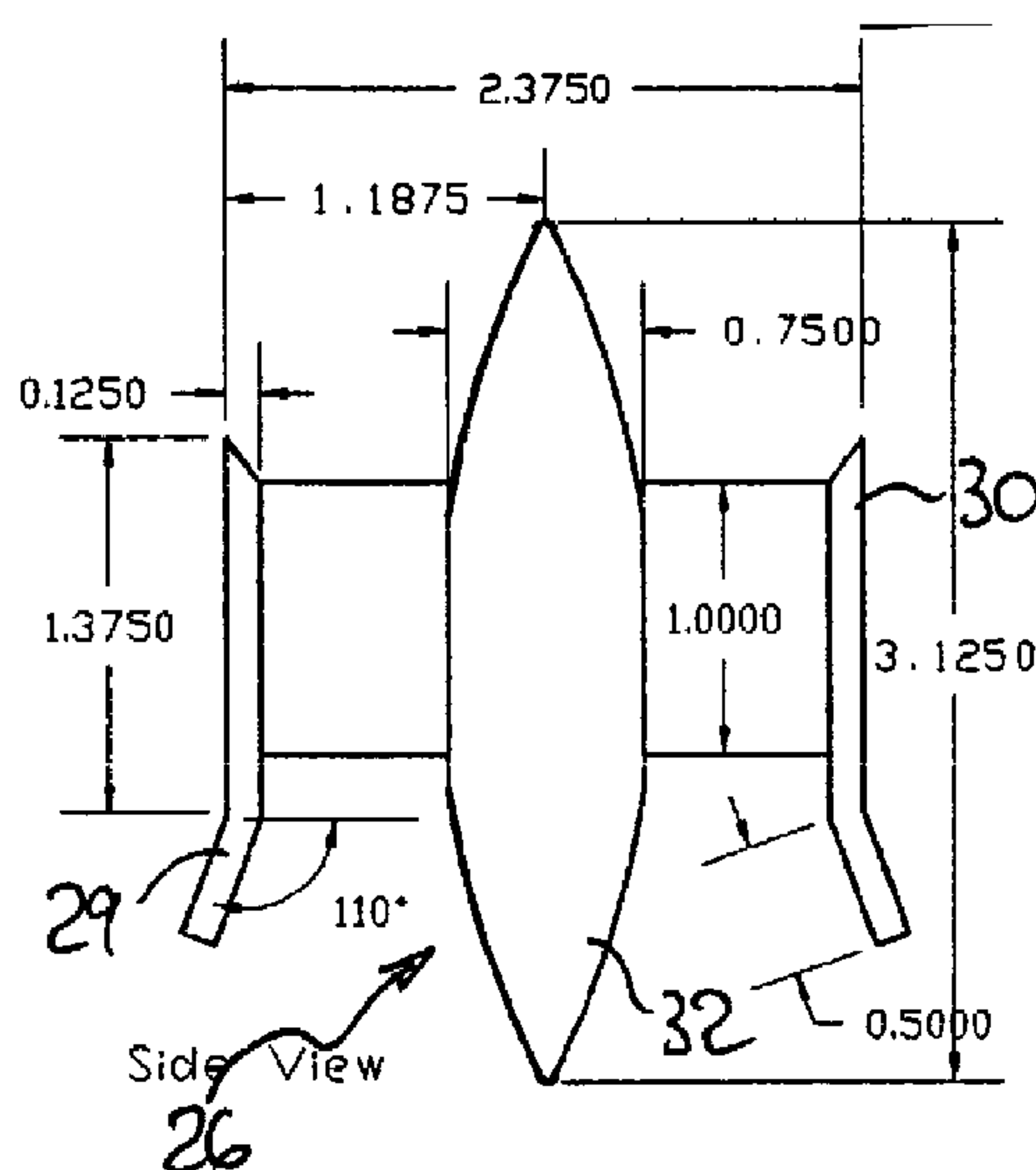
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(54) **BUSE D'AERATION POUR PRODUIRE UN JET D'AIR HAUTE
INTENSITE ET A GRANDE PORTEE EN UTILISANT DE
L'AIR PROPULSE PAR UN VENTILATEUR**

(54) **NOZZLE FOR PRODUCING A HIGH-IMPACT LONG-RANGE
JET FROM FAN-BLOWN AIR**

Mdl AF1-CAN-100 Air Cannon



(57) Blow-off nozzles are used for creating a high-energy air blast, for drying metal panels prior to painting. Depth or reach of penetration (in the atmosphere) is important. A bullet is provided in the centre of the nozzle. The bullet is aerodynamically faired, for minimum drag. The effect of the bullet is to create a low pressure area in the jet downstream of the nozzle. The low pressure area serves to hold the jet together, preventing spreading, to a degree that enables a significant increase in penetration distance. The bullet is mounted on faired arms, which are secured to the walls of the nozzle.

1 Abstract of the Disclosure

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3
4 Title: NOZZLE FOR PRODUCING A HIGH-IMPACT LONG-RANGE JET
5 FROM FAN-BLOWN AIR

6
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8 blast, for drying metal panels prior to painting. Depth or
9 reach of penetration (in the atmosphere) is important. A
10 bullet is provided in the centre of the nozzle. The bullet
11 is aerodynamically faired, for minimum drag. The effect of
12 the bullet is to create a low pressure area in the jet
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16 The bullet is mounted on faired arms, which are secured to
17 the walls of the nozzle.

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32 Docket: 878-11
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1 Title: NOZZLE FOR PRODUCING A HIGH-IMPACT LONG-RANGE JET
2 FROM FAN-BLOWN AIR

3
4 This invention relates to apparatus for producing an intense
5 jet of air from a nozzle. The jet of air is used
6 industrially for such purposes as blowing water, dust,
7 particulate material, etc, from surfaces, to clean and dry
8 the surfaces preparatory to painting, application of
9 adhesives, etc.

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11
12 BACKGROUND TO THE INVENTION

13
14 Conventionally, in automotive component painting
15 applications, for example, blow-off stations are provided
16 between the workpiece washing station and the paint spray
17 booth. The blow-off station includes several air-nozzles,
18 which are fed from a common fan, driven by an electric
19 motor. Typically, the fan supplies air at a flow rate of
20 2000 cfm or so, and at a pressure of around 1 psi (27" water
21 gauge). The air travels through flexible hoses or pipes to
22 the nozzles, the hoses being, typically, four inches in
23 diameter. The nozzles are mounted on a frame, and are
24 adjustable as to mounting position and angle.

25
26 The present invention is aimed at providing a manner of
27 designing the nozzle that enables the jet or stream of air
28 emanating from the nozzle to penetrate further, downstream
29 of the nozzle, for higher surface impact, without incurring
30 a penalty of increased energy requirements.

31
32
33 THE PRIOR ART

34
35 Patent publication US-5,636,795 (Sedgwick, Jun 1997) shows a

1 spray nozzle of the type with which the invention is
2 generally concerned.

3

4 It should be noted that the type of blowing-off to which the
5 invention refers is done by air at low pressures. That is
6 to say, the air-flow is generated by means of a air-fan,
7 rather than by means of an air-compressor.

8

9 It is of course possible to produce a vigorous jet of air by
10 blowing high pressure air (i.e air from a compressor, at
11 perhaps 80 psi) out of a nozzle. However, it would be
12 highly uneconomical to create the required huge flow rate
13 needed for air blow-off systems using air at 80 psi.

14

15 On the other hand, air at 80 psi is widely available as a
16 utility in factories generally, and there are a number of
17 technologies aimed at entraining atmospheric air into a high
18 pressure (80 psi) jet, to allow some of the energy of the
19 high pressure jet to be transferred to the surrounding air,
20 to give the jet the desired volumetric flow rate. However,
21 such systems are only economical because the high pressure
22 air compressor already exists.

23

24 Industrial purpose-designed air blow-off systems use a fan
25 that provides the air at low pressures, i.e at pressures in
26 the 0.5 to 2 psi region. In this case, entraining air from
27 the atmosphere into the jet is to be avoided. The invention
28 is concerned with applying as much as possible of the energy
29 derived from the fan into enabling the jet to penetrate more
30 deeply through the atmosphere, and such entrainment would,
31 in the present case, serve simply to dissipate the energy of
32 the jet, and detract from penetration.

33

34

35 THE INVENTION IN RELATION TO THE PRIOR ART

1 The invention provides a bullet, which is mounted in
2 position in the centre of the nozzle. The bullet serves, in
3 operation, to create a low pressure area downstream of the
4 nozzle.

5
6 It has been found that the low pressure area can be made to
7 extend so far downstream of the nozzle, under the conditions
8 as described herein, as to suck the jet in somewhat, and to
9 hold the jet together. The main reason a jet would fail to
10 penetrate a large distance is that the jet tends to spread
11 or widen, to strike the atmospheric air, and thereby to
12 dissipate its energy. The low pressure area created by the
13 bullet sucks the jet in, and keeps the jet together, for a
14 significantly increased distance. Thus, for example, where
15 a traditional low-pressure air nozzle might enable air to
16 penetrate a maximum of perhaps four feet, the nozzle with
17 the bullet can enable air to penetrate five or even six
18 feet.

19
20 Of course, it is always possible to create whatever strength
21 of jet is desired, simply by using a larger power source to
22 pump more air through a nozzle at higher pressure. But the
23 concern in this present case is with the efficiency at which
24 a given strength of jet can be provided. A high pressure
25 jet (as from a conventional factory air compressor) creates
26 such a high velocity in the emerging air as to create a low
27 pressure aura around the jet, which tends to suck in outside
28 air and entrain it in the jet. Thereby, the jet can impart
29 a portion of its energy to the surrounding air. With this
30 entrainment, instead of all the energy of the jet being in
31 the form of high-speed/ low-mass, the energy form of the jet
32 now becomes medium-speed/ medium-mass, which is more useful
33 for doing work. But still, a high-pressure system is
34 inefficient; it is inefficient to create high pressure, then
35 destroy it.

1 In the Sedgwick patent mentioned above, the emerging jet is
2 given a vigorous spin or rotational velocity. It might be
3 considered that a low pressure area exists on the inside of
4 the emerging jet, because of the cyclone effect arising from
5 the spin. However, it should be noted that a cyclone
6 creates a spinning vortex, with a low pressure area inside,
7 because of the presence of the low pressure; i.e in a
8 cyclone the low pressure core creates the spin, the spin
9 does not create the low pressure core. In Sedgwick, the
10 spin velocity has to be generated, and that takes energy.
11 Also, whatever spin velocity exists will be at its maximum
12 at the outside of the stream, where the stream hits the
13 stationary air. This interaction creates more friction, and
14 wastes more energy. In fact, whatever energy goes into
15 creating the rotation of the cyclone, must take away from
16 the energy available for the forwards penetration of the
17 jet.

18
19 It is an aim of the invention that the bullet should create
20 the downstream low pressure area aerodynamically, and
21 thereby cause only a minimum of disruption to the downstream
22 longitudinal penetration energy of the jet.

23
24 Nozzles are provided in many types of machine. Placing a
25 bullet in the centre of a nozzle would have a different
26 effect in different types of machine.

27
28 In the nozzle system as described herein, lowering the
29 pressure inside the jet has the effect of sucking the jet
30 together. By reaction, the low pressure area creates a
31 force on the bullet tending to draw the bullet downstream,
32 with the jet of air. In a jet engine, for example, the
33 purpose of the nozzle is to convert the energy of the
34 emerging stream of air into thrust for the aircraft, which,
35 it will be understood, is opposite to the purpose of

1 enabling the stream to penetrate as far as possible away
2 from the nozzle.

3
4 In a nozzle, air is accelerated up to jet speed by reducing
5 the cross-sectional area through which the air passes. It
6 might be considered that keeping the outside diameter of the
7 nozzle the same as the pipe, and making the bullet so large
8 that the bullet nearly fills the nozzle, would be a way of
9 creating the reduced area which is necessary for
10 accelerating the air. However, the overall or outside
11 dimensions of the jet should be kept small. If the bullet
12 is large, so that the jet becomes a thin annulus, the area
13 of the jet that is exposed to the outside air is
14 correspondingly large, and so, even though the jet might
15 emerge with good energy, the losses associated with the
16 interaction would be also large. Therefore, the bullet
17 should not be so large that the flow through the nozzle came
18 to have a configuration that could be considered annular to
19 a significant degree.

20
21
22 DETAILED DESCRIPTION OF PREFERRED EMBODIMENTS

23
24 By way of further explanation of the invention, exemplary
25 embodiments of the invention will now be described with
26 reference to the accompanying drawings, in which:

27 Fig 1 is a diagrammatic representation of a nozzle under
28 test, in which air passing through the nozzle contains
29 smoke, for visibility;

30 Fig 2 corresponds to Fig 1, and shows a nozzle that
31 incorporates the invention;

32 Fig 3 is a cross-section of the nozzle of Fig 2;

33 Fig 4 is a front elevation of a component of the nozzle of
34 Fig 2;

35 Fig 5 is a side elevation of the component of Fig 4;

1 Fig 6 is a pictorial view of the component of Fig 4;
2 Fig 7 is a pictorial view of the nozzle of Fig 2, in use.

3

4 The apparatuses shown in the accompanying drawings and
5 described below are examples which embody the invention. It
6 should be noted that the scope of the invention is defined
7 by the accompanying claims, and not necessarily by specific
8 features of exemplary embodiments.

9

10 Figs 1 and 2 illustrate the difference between a
11 conventional air-blow nozzle unit 20 and a nozzle unit 23
12 that incorporates an internal faired bullet, in accordance
13 with the invention. In both cases the muzzle of the nozzle
14 unit is about 2.5" in diameter and the nozzle unit is
15 supplied from a pipe of about 4" diameter. The difference
16 in the length of forceful penetration of the jets arises
17 because of the presence of the bullet in the nozzle of
18 Fig 2.

19

20 Figs 3 and 4 are cross-sections of the nozzle unit 23 of
21 Fig 2. The housing 24 is shaped to converge to a
22 cylindrical nose 25. The housing 24 is formed from aluminum
23 sheet, by spinning the sheet into a tubular form.

24

25 The bullet assembly 26 shown in Figs 4,5,6 fits
26 concentrically inside the nose 25, and includes two radial
27 arms 27,28. The arms terminate with bars 29,30. The bullet
28 unit, comprising the bullet 32, the arms 27,28, and the bars
29 29,30, are formed as a one-piece aluminum casting. The
30 bullet unit is mounted in place in the nose 25 by welding
31 the bars 29,30 to the internal cylindrical wall of the nose
32 25.

33

34 The bullet 32 is of an aerodynamically faired configuration,

1 the shape being so designed as to impart a minimum tendency
2 to cause drag and turbulence in the air flow passing through
3 the nozzle. The designer should take care to cause as
4 little energy as possible to be dissipated in the nozzle;
5 any energy that is dissipated as turbulence in the nozzle
6 takes away from the energy that would otherwise be available
7 for projecting the jet of air toward the work-piece. The
8 designer's aim is to create a low pressure area downstream
9 of the bullet, without creating turbulence.

10
11 The radial arms 27,28 are faired also, to minimise any
12 tendency of the arms to create turbulence. However, as
13 shown in Fig 5, the arm 27 is angled in the Fig 5 view.
14 Thus, air passing the arm 27 is given a velocity to the
15 left. The arm 28 is similarly angled, and deflects its
16 stream of air to the right. Thus, the air emerging from the
17 nose 25 has a degree of imparted helical twist or spin.
18 Again, the designer should take care, when imparting the
19 spin to the air flow. not to induce turbulence.

20
21 In the type of system as illustrated, air is blasted from
22 the muzzle 33 of the nozzle with a great deal of vigour.
23 Air-flows in the region of 450 cfm are typical. It is the
24 intention that the blast of air should be able to perform
25 useful work four, five, or even six, feet away from the 2.5
26 inch nozzle.

27
28 The presence of the bullet 32 means that the air jet flowing
29 from the nozzle contains a low pressure area 34, downstream
30 of the bullet. (Of course, no such low-pressure area is
31 present in a conventional nozzle, which has no bullet).
32 This low-pressure area gives rise to a suction force tending
33 to draw or hold the jet of air together. The low-pressure
34 area 34 tends to focus the jet, stopping the jet from
35 expanding or spreading; the more the jet can be prevented

1 from spreading, the further the jet can be made to
2 penetrate.

3
4 A jet of fast-moving air, as it emerges into, and interacts
5 with, the ambient air, starts to slow down. The outer
6 portions of the jet are retarded first. The molecules of
7 air in the outer portion start to spread out and become
8 dissipated. In other words the molecules of the outer
9 portion start to acquire an outwards or radial component to
10 their velocity. Gradually, as the jet travels further from
11 the nozzle, the whole air stream spreads and becomes
12 dissipated.

13
14 The low pressure area 34 provides a force acting on the jet,
15 which tends to stop the jet from spreading laterally. Thus,
16 because of the low-pressure area, the tendency of the outer
17 portions of the jet to acquire an outward velocity is
18 resisted. The air stream is held together by the low-
19 pressure area. Thus the stream remains in focus for a
20 significantly longer distance downstream from the nozzle,
21 and the depth of penetration at which the blast of the air
22 stream can do useful work is thereby increased.

23
24 The helical twist imparted to the stream by the angled arms
25 27,28, tends to make the stream a little more coherent, and
26 is also significant in increasing the depth of penetration
27 of the air stream.

28
29 The nozzle unit 23 is provided with a mounting fixture 36,
30 which comprises a short stub-tube 37 welded to the outside
31 of the housing 24. In a typical installation, several of
32 the nozzle units are provided (Fig 7), and directed around
33 the work-piece. Each nozzle unit is adjustable as to the
34 angle at which its jet is directed, and the unit is locked
35 in place by clamping the stub-tube 37 to a fixed frame.

1 As mentioned, a typical air flow through the nozzle would be
2 around 450 cfm. Such a flow would be supplied in the supply
3 pipe 39 at a pressure of about 1 psi. A one-horsepower
4 electric motor 38 would typically be required to power the
5 fan to supply air at that energy level.

6
7 The dimensions of the bullet are important. It might be
8 considered that the bullet should have a large cross-
9 sectional area in relation to the nozzle diameter, in order
10 that the low-pressure region 34 downstream of the bullet
11 might indeed be at low-pressure. It might be considered
12 that, the lower the pressure in the region 34, the more
13 marked the effect the low-pressure region has in preventing
14 the jet from spreading and holding the jet together.

15 However, there is a limit to the pressure reduction that can
16 be achieved in the region 34. If the diameter of the bullet
17 were too large, the air flow would be disrupted downstream
18 of the bullet, and turbulence would result, with consequent
19 loss of energy. For a nozzle having a nominal diameter of
20 2.5 inches, the bullet preferably should be no more than
21 about 1.0 inches in diameter.

22
23 On the other hand, the bullet should not be too small, or
24 the effect of the bullet in creating a low-pressure region
25 downstream of the nozzle will be negligible. Thus, the
26 bullet should have a diameter of at least 0.5 inches.

27
28 These dimensions can be expressed rather as percentages, i.e
29 the diameter of the bullet should be between 20 percent and
30 40 percent of the diameter of the nozzle. On a cross-
31 sectional area basis, the area of the bullet should be not
32 less than 4 percent, and not more than 16 percent, of the
33 area of the nozzle.

34
35 To minimize the aerodynamic drag caused by the bullet, the

1 downstream end of the bullet preferably should be conically
2 tapered to a point 40.

3
4 In some applications, for example in automotive spray
5 painting, it can be advantageous to apply a highlighting
6 liquid to the surface of the workpiece prior to painting.
7 The liquid highlights any surface defects, if present,
8 whereupon the workpiece can be removed from the production
9 line for remediation before paint is applied. In an
10 alternative construction (not shown), the bullet is provided
11 with a tube running down the centre of the bullet, and the
12 highlighting liquid can be applied to the surface of the
13 components by introducing the liquid through the tube,
14 whereby the liquid emerges at the point 40, and is carried
15 with the jet of air to the workpiece.

16
17 The location at which the bullet terminates is important.
18 If the bullet were to terminate short of the muzzle 33 of
19 the nozzle, the flow of air will start to conform to the
20 nozzle, rather than to the bullet, and the effect of the
21 bullet might be lost. On the other hand, if the bullet were
22 to protrude too far downstream of the muzzle, the stream
23 might tend to diverge upon emerging from the nozzle, because
24 of the presence of the protruding bullet, and the beneficial
25 effect of the low-pressure area would be lost.

26
27 The nozzle itself should be kept short. The reduced
28 diameter nose 25 of the nozzle is where the velocity of the
29 air is at its highest, and therefore also where the friction
30 is at its highest. Not only does the friction give rise to
31 direct loss of energy but the friction also causes
32 differential velocities within the jet, in that the
33 radially-outermost portions of the jet are retarded by the
34 friction, and so travel more slowly than the main area of
35 the jet. On the other hand, this tendency to differential

1 velocity, due to friction of the outer regions of the jet
2 against the walls of the nozzle, is offset by the fact that
3 the bullet creates some similar retardation of the centre
4 part of the jet. Both the nozzle and the bullet should be
5 kept short, to minimize aerodynamic friction losses.

6
7 The nozzle is most effective when the nose 25 of the nozzle
8 is right-cylindrical. If the nose were convergent, emergence
9 of the jet into the open air would be too abrupt and
10 turbulence might result. If the nose were divergent, part
11 of the energy of the jet would be lost creating back-
12 pressure against the nozzle. A right-cylindrical nozzle
13 enables a minimum energy loss of the jet in emerging from
14 the nozzle.

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Claims

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CLAIM 1. Apparatus for creating a jet of air, the jet being configured for a long distance of penetration, wherein: the apparatus includes a means for supplying air at a pressure not more than 2 psi, and at a flow rate not less than 300 cfm; the apparatus includes a convergent nozzle, and a means for passing the pressurised air therethrough; the apparatus includes a bullet, and a means for locating the bullet in the nozzle; the disposition and size of the bullet are such as to create a substantial low pressure area in the jet of air emerging from the nozzle, downstream of the nozzle.

CLAIM 2. Apparatus of claim 1, wherein the bullet is aerodynamically faired, for minimum drag and turbulence.

CLAIM 3. Apparatus of claim 2, wherein the bullet, on its downstream side, is cone shaped.

CLAIM 4. Apparatus of claim 3, wherein the bullet converges to a point on its downstream side.

CLAIM 5. Apparatus of claim 1, wherein the overall cross-sectional area of the bullet is no more than 16 percent of the area of the muzzle of the nozzle.

CLAIM 6. Apparatus of claim 1, wherein the overall cross-sectional area of the bullet is no less than 4 percent of the area of the muzzle of the nozzle.

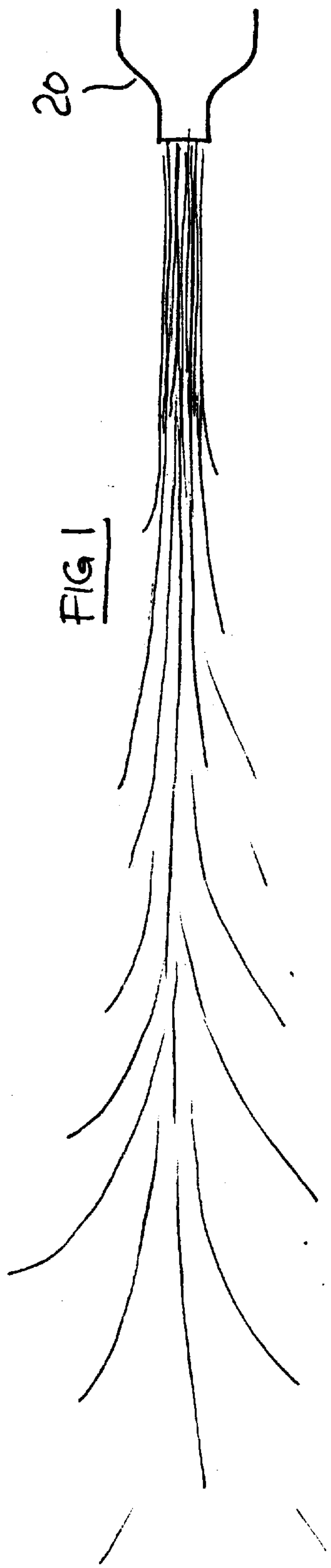
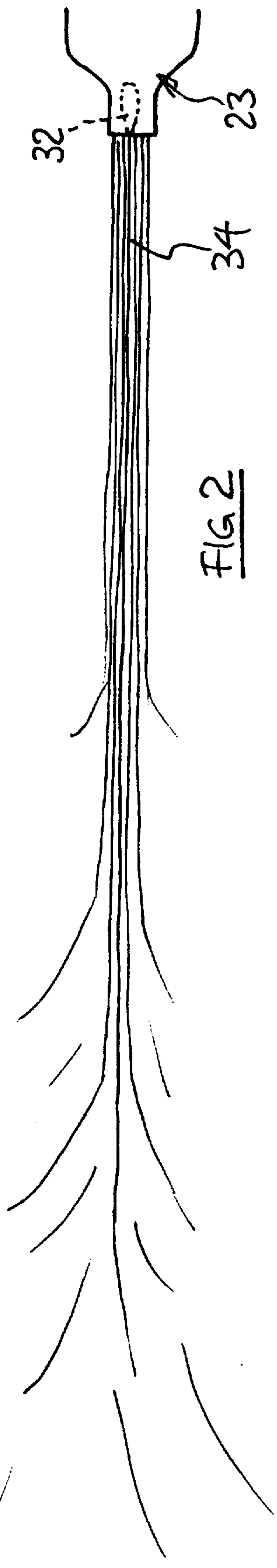
CLAIM 7. Apparatus of claim 1, wherein the means for locating the bullet in the nozzle is effective to position the bullet so that the downstream extremity of

1 the bullet is substantially in line with the muzzle of
2 the nozzle.

3

4 **CLAIM 8.** Apparatus of claim 1, wherein the means for
5 locating the bullet in the nozzle includes radial arms,
6 and a means for securing the arms to the walls of the
7 nozzle.

8



6 5 4 3 2 1 0

Mdl AF1-CAN-100 Air Cannon

FIG 3

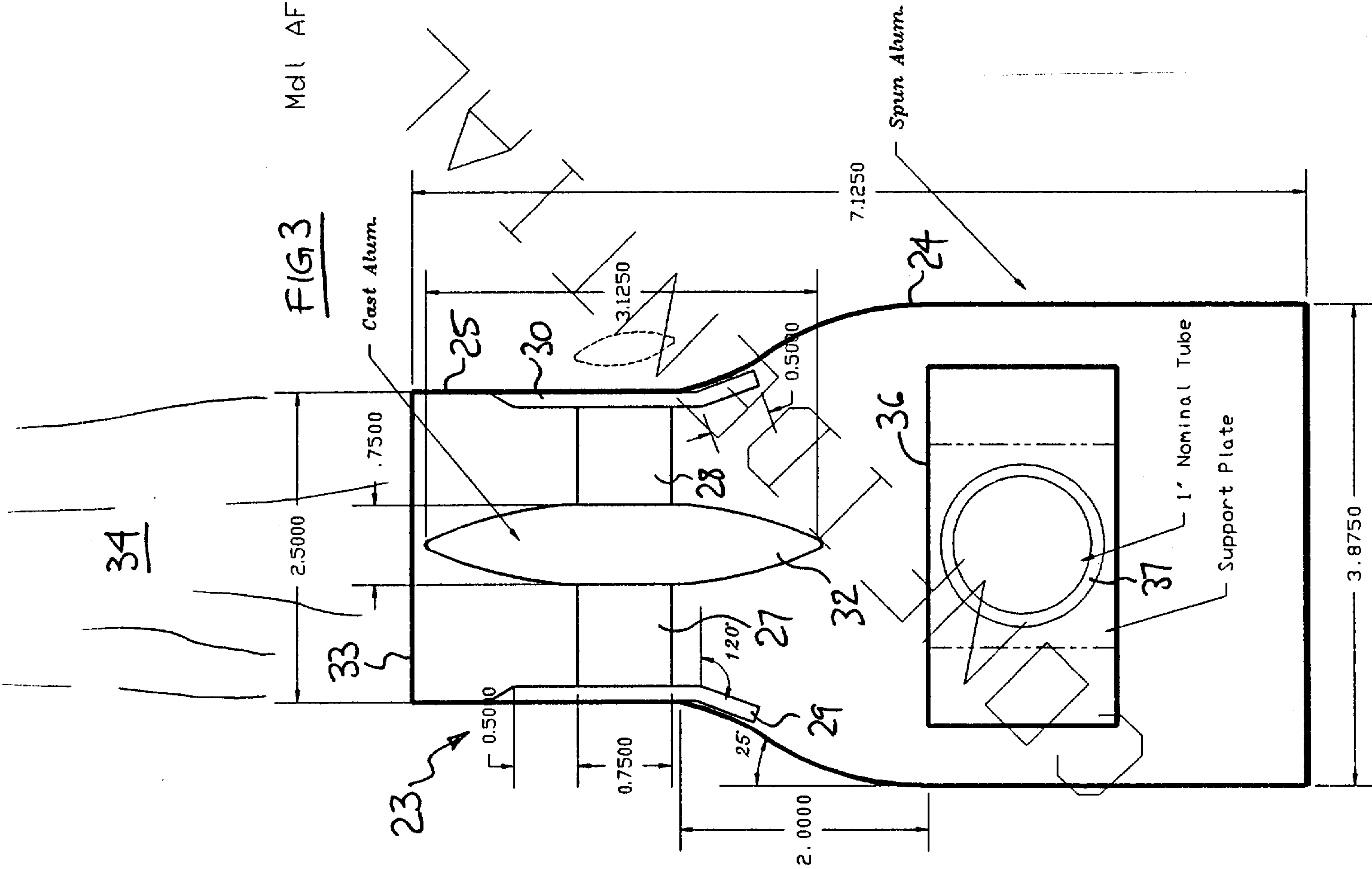


FIG 4

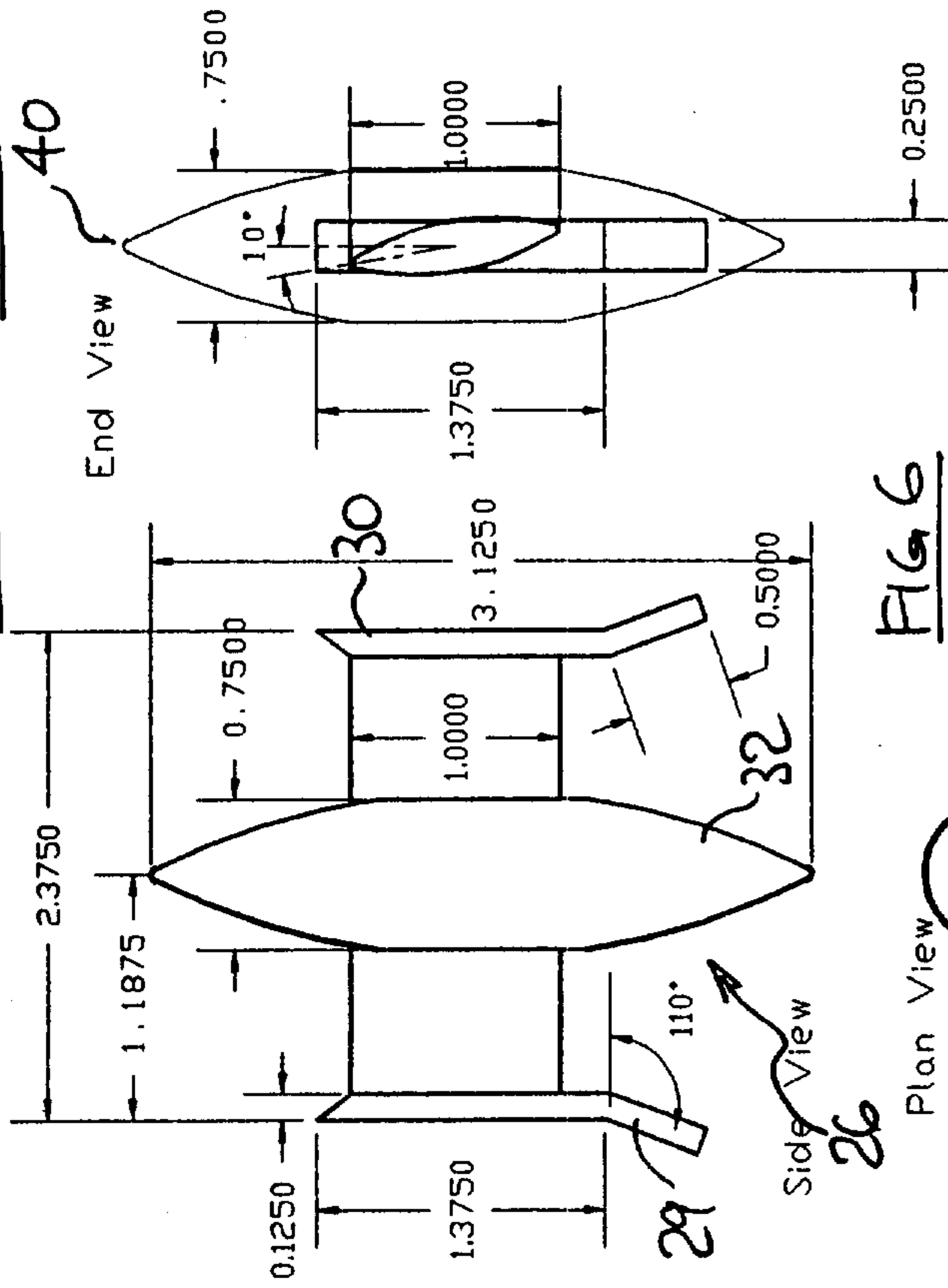


FIG 6

