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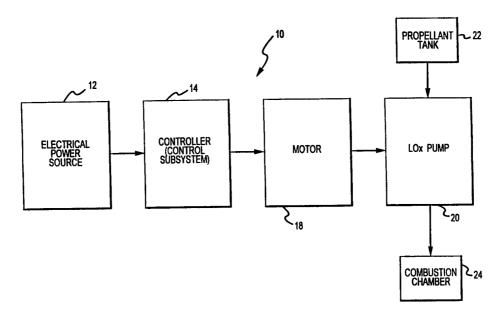
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(57) Abstract

A liquid propellant supply system (10) according to the invention is electrical in nature and avoids the need for a gas generator and a turbine assembly. In particular, the system (10) includes an electrical power source (12), a controller (14) and a motor (18) for driving the pump (20). The electrical power source (12) provides electrical power sufficient for high power, limited duration applications such as for launch vehicle applications. For launch vehicle applications, the power source (12) is preferably capable of providing very high power for at least 60 seconds. A number of different types of electrical power sources may be employed in this regard. For example, the power source (12) may include high energy density batteries, supercapacitors, or counter rotating flywheels. Multiple counter rotating motors may be employed in order to minimize precessional forces. The invention thereby eliminates problematic high pressure plumbing, reduces overall mass and allows for convenient mass flow rate throttling, as well as save abort options.

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C. DOC	UMENTS CONSIDERED TO BE RELEVANT							
Category*	Citation of document, with indication, where ap	opropriate, of the relevant passages	Relevant to claim No.					
X Y	US 3,077,073 A (KUHRT) 12 FEBR DOCUMENT.	UARY 1963, SEE ENTIRE	1, 3-5, 7-13, 15, 17-19, 21-25, 27-33, 36, 38					
			2, 6, 14, 16, 20, 26, 34, 35, 37					
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Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No
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