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A VIBRATION DAMPER ASSEMBLY FOR A VIBRATING STRUCTURE
GYROSCOPE RESONATOR

This invention relates to a vibration damper
5 assembly adapted for use with a Vibrating Structure
Gyroscope Resonator suitable particularly for use with a
substantially ring-like, substantially planar resonator.

It is known to build Vibrating Structure
Gyroscopes using planar ring members as the resonant
10 structure. The principles of operation and practical
implementation of ring structures in vibrating structure
gyroscopes are described in the present Application EP/UK-
B-0461761.

The ring is typically excited into a $\cos 2\theta$
15 resonance mode. This mode actually exists as a degenerate
pair of modes at a mutual angle of 45° . The vibration
patterns for these modes are shown schematically in Figure
1A and 1B. One of the modes is excited as the carrier
mode as shown in Figure 1A with the other used as the
20 response mode as shown in Figure 1B. The carrier mode
vibration is typically maintained at a constant amplitude
at the peak resonance frequency. When the gyro body is
rotated, Coriolis forces couple energy into the response
mode. The amplitude of motion of the response mode is
25 directly proportional to the applied rotation rate.

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A particular implementation of this technology uses a Nickel-Iron alloy (NiFe-45) planar metal ring as the resonator 1 as shown in Figure 2. The outer diameter of the ring 1 is 22mm with a rim width of 1mm and a depth of 1.2mm. The ring is attached to a central hub 2 by means of eight compliant legs 3 allowing substantially undamped motion of the ring 1. The ring is attached to a support pedestal (not shown) at the central hub 2 to leave the ring 1 and legs 3 freely suspended. The ring structure, together with location of the carrier mode drive elements 4 (to set the ring into oscillation) and pick-off elements (to detect the motion) around the ring are shown schematically in Figure 2. The carrier mode drives 4 excite the carrier mode resonance at two anti-nodal positions (i.e. points of maximum radial motion). The carrier mode pick-offs 5 detect this motion at the remaining anti-nodal points. The response mode pick-offs 6 are located at two of the response mode anti-nodal positions. The signal level on these pick-offs will be directly proportional to the applied rate.

Many potential sensor applications require the gyroscope to operate under quite demanding environmental conditions. One such requirement is that the rate output noise should remain within specified limits in response to an applied

vibration spectrum. The applied vibration spectrum for a typical automotive application is 10-800Hz at 2g rms. However, certain military applications specify significantly higher vibration levels over wider spectral ranges.

The principle effect of an applied vibration will be to cause motion of the resonator with respect to the gyro body. As the pick-off modules are rigidly attached to the gyro body this will inevitably cause a corresponding relative motion between the resonator and pick-offs. The sensor pick-offs 5, 6 are of capacitive design with the pick-off plate and the circumferential edge of the ring 1 forming the two capacitor plates. The ring 1 is maintained at high voltage with respect to the pick-off plates. Any change in the pick-off to ring gap will give rise to a signal. Applied vibration in the plane of the ring 1 will modulate the capacitor plate gap at the input vibration frequency. The pick-offs 5, 6 are arranged in pairs with the signals from pick-offs positioned diametrically opposite one another across the ring being summed at the input to the signal processing electronics. These signals will be in anti-phase and will thus tend to cancel one another provided that the pick-off gains are well matched.

Vibration along the axis normal to the plane of the ring 1 will cause the rim to move with respect to the pick-off plate 5, 6, as shown in cross-section in Figure 3. The net effect of this motion is also to modulate the pick-off to ring gap. The phase of the signals on the pick-off pair will be identical for this motion of the ring and will therefore sum together. Furthermore, if the pick-off plate and ring are not aligned precisely parallel this will also cause the gap size to change generating a vibration induced error signal.

Any relative motion of the resonator 1 with respect to the pick-offs 5, 6 will generate a signal on the pick-offs. The scheme for the derivation of the rate signal in the sensor means that only noise inputs around the ring resonance frequency, within the gyroscope bandwidth, pass through to the rate output. This is shown schematically in Figures 4A to 4F. The ring is vibrated at the $\cos 2\theta$ resonance frequency which provides the carrier signal at which demodulation takes place (Fig 4A). A rate input (Fig 4B) modulates the carrier giving sidebands on the pick-off signal output as shown in Figure 4C. The vibration induced noise will appear at the baseband frequency. In the absence of any filtering the gyroscope output will appear as in Figure 4D.

This is filtered by the effective system bandwidth (Figure 4E) to give the observed rate output shown in Figure 4F.

For the known gyroscope described here the carrier frequency is at 5kHz. This is well above any applied vibration input band and thus the gyroscope is inherently insensitive to vibration under normal circumstances. However, at extreme levels of vibration the amplitude of the ring motion may become sufficiently large to saturate the pick-off amplifier output at particular times in the vibration cycle. This intermittent saturation will cause instability in the gyro control loops with a consequent degradation in noise performance.

The in-plane vibration modes of the resonator 1 described are all well above the input vibration range. The planar nature of the resonator 1 results in the out-of-plane stiffness being significantly lower than the in-plane, however. The out-of-plane mode frequencies thus occur at lower frequencies with the lowest being the bending mode, as shown in Figure 3, which occurs at approximately 2kHz. This vibration mode may be excited directly by external vibration. The high Quality Factor of the ring 1 (approximately 2000) may result in a large amplitude of motion in this mode. In

the limit, extremely high levels of applied vibration may result in mechanical failure of the resonator 1.

The sensor rate output for this gyroscope is within the specified operating limits for typical civil applications vibration specification. Under some of the more demanding military specifications, saturation of the pick-off output due to the large amplitude of motion in the out-of-plane bending mode causes the noise level to rise to unacceptably high levels.

There is thus a requirement for an improved gyroscope capable of operating in these harsh vibration environments.

A known means of improving the vibration tolerance of a gyroscopic device is to fix it onto an anti-vibration mount. This restricts the g-level to which it is exposed by absorbing some of the relative motion in the external mount. A problem with this solution is that the mount needs to be large enough to accommodate the entire mass of the gyroscope. This increases the overall volume occupied by the sensor. For military applications, in particular, this is undesirable as the space envelope available for the sensor pack may be restricted. External anti-vibration mounts typically damp vibration inputs along every axis. For rapid angular accelerations the compliance of the mount will result in the

applied rate not being applied instantaneously to the gyroscope and will consequently restrict its bandwidth. Any motion of the external mount will also cause a misalignment of the sensor axis which is again
5 undesirable.

There is thus a need for a generally improved vibration damper assembly for a Vibrating Structure Gyroscope and for a vibrating structure gyroscope incorporating such a vibration damper assembly.

10 According to a first embodiment of the present invention there is provided a vibration damper assembly adapted for use with a substantially ring-like, substantially planar resonator of a vibrating structure gyroscope, including a female portion constructed so as to
15 be attachable to a body of a vibrating structure gyroscope to have limited resiliency with respect to the body in a direction substantially perpendicular to the resonator, and a male portion fixedly attachable at or adjacent one end thereof to a resonator of the vibrating structure
20 gyroscope, projectable into the female portion, and fixably attached at or adjacent its other end in and to the female portion, wherein the assembly is such as to permit, in operation, damping of vibrating movement of the resonator in a direction substantially perpendicular
25 thereto whilst retaining stiffness and freedom to vibrate for the resonator in the plane thereof.

Preferably the female portion includes an open ended axially bored tube-like metal collar locatable in an aperture through the body so that the longitudinal axis of
30 the collar bore extends substantially perpendicularly to the body, a pair of superimposed, spaced apart, apertured metal rings or plate which are arrangeable with the aperture in one ring or plate adjoining or surrounding one open end of the collar and with the aperture in the other
35 ring or plate adjoining and surrounding the other open end



of the collar, and with the outermost peripheral edge of
the two metal rings or plates being directly attachable to
the body which extends therebetween, and a resilient
apertured grommet engagable via its aperture around the
5 collar and locatable between the pair of spaced rings or
plates for engagement of the outer periphery of the
grommet with the innermost peripheral edge of the
resonator.

Conveniently, the pair of metal rings or plates
10 are made of beryllium-copper alloy and the resilient,
apertured grommet is made of silicone rubber.

Advantageously the female portion includes a pair
of axially bored bushes each having a shoulder extending



externally around one end thereof which bushes are locatable one at each end of the collar to project into the bore therein with the external shoulder on one of the said bushes engaging the adjoining ring or plate around the aperture therethrough and with the external shoulder on the other of said bushes engaging the adjoining other ring or plate around the aperture therethrough.

Preferably the pair of bushes are made from an electrically insulating material which more preferably is a plastics material.

Conveniently the male portion includes an open ended axially bored spindle having a first open ended region which is locatable in a bore through the resonator and which has a shoulder therearound for contact with the resonator around one open end of the bore therethrough, having a second open ended region which is engagable in a first of the bushes when the latter is located in the collar bore and having an intermediate flanged region having a diameter greater than the diameters of the first and second regions to bear at one surface against the resonator and at an opposite surface against the first bush shoulder.

Preferably the male portion includes an open ended axially bored metal bush having a shoulder extending

externally therearound at or adjacent one end, which metal bush is insertable in the second of said bushes, when the latter is located in the collar bore, so that the metal bush shoulder abuts the second bush shoulder, and a rod member having an enlarged head at one end, which rod member is projectable through the spindle bore, with the head engaging the spindle first region shoulder, through the collar bore and into fixable engagement at the other end of the rod member with the metal bush.

Conveniently the rod member other end has an externally threaded region for flexible engagement with a correspondingly threaded internal region of the metal bush bore.

According to a further aspect of the present invention there is provided a Vibrating Structure Gyroscope having a substantially ring-like, substantially planar resonator and a vibration damper assembly, according to any one of claims 1 to 10, attaching the resonator to an apertured substantially planar portion of a body of the gyroscope, wherein the resonator is made of metal, and the body has recessed portions surrounding the apertured therethrough for accommodating the metal rings or plates, which recessed portions taper towards the body aperture and terminate in a

reduced thickness rim which defines said body aperture, with the body rim engaging the resilient grommet which assists in the damping effect.

Preferably the components of the vibration damper assembly and the resonator and body are secured together by an adhesive which more preferably is a toughened epoxy adhesive.

For a better understanding of the present invention, and to show how the same may be carried into effect, reference will now be made, by way of example, to the accompanying drawings in which:

Figures 1A and 1B diagrammatically illustrate a carrier mode and response mode of vibration respectively for a conventional Vibrating Structure Gyroscope Ring Resonator excited into a $\cos 2\theta$ resonance mode,

Figure 2 is a diagrammatic planned view of a ring resonator, drive and pickoff means for a conventional Vibrating Structure Gyroscope,

Figure 3 diagrammatically illustrates in cross-section the vibration of the resonator of Figure 2 along an axis normal to the plane of the resonator,

Figure 4A graphically illustrates vibration of the resonator at the $\cos 2\theta$ resonance frequency with the carrier

signal being shown on a plot of applied drive versus frequency,

Figure 4B shows graphically as a plot of applied rate against frequency the rate input,

Figure 4C shows as a plot of sensor pick off output against frequency how the rate input amplitude modulates the carrier giving side bands on the pick off signal output, with the vibration induced noise at the baseband frequency,

Figure 4D shows as a graphic plot of unfiltered gyro output against frequency the output rate signal with the vibration induced noise in the absence of any filtering,

Figure 4E shows as a graphic plot of effective baseband system response against frequency the effective system bandwidth in the present of filtering,

Figure 4F shows as a graphic plot of baseband gyro output against frequency the observed rate output for the gyroscope of Figures 2 and 3,

Figure 5 is a vertical cross sectional view through part of a female portion of a vibration damper assembly according to the present invention in operative association with part of a body of a vibrating structure gyroscope.

Figure 6 is a vertical cross sectional view similar to that of Figure 5 showing a resonator located via a male

portion of the vibration damper assembly according to the present invention in the female portion thereof,

Figure 7A shows schematically how the relative motion of the gyro body pickoffs, vibration damper assembly and resonator according to the present invention may be modelled without internal damping,

Figure 7B shows how the same as in Figure 7A may respond with the presence of the internal vibration damper assembly according to the present invention,

Figure 8 is a graphical display of amplitude of motion against vibration frequency showing the motion of the resonator with respect to the pick offs and gyro body for a vibrating structure gyroscope without a vibration damper assembly according to the present invention and,

Figure 9 is a graphic display of amplitude of motion against vibration frequency showing the motion of the resonator with respect to the pickoffs and gyro body in the presence of a vibration damper assembly according to the present invention.

A vibration damper assembly illustrated generally in Figures 5 and 6 of the accompanying drawings at 7 is intended for use with a substantially ring-like, substantially planar resonator 1 of a vibrating structure gyroscope. The

vibration damper assembly 7 provides a means for damping the resonant response of the out of plane bending mode of the resonator 1 to prevent saturation of the pickoffs 5,6. This is achievable without increasing the overall vibrating structure gyroscope volume and with minimal or no introduction of misalignment or bandwidth problems. As shown in both Figures 5 and 6 the assembly 7 includes a female portion constructed so as to be attachable to a body 8 of a vibrating structure gyroscope to have limited resiliency with respect to the body 8 in a direction substantially perpendicular to the body 8. The assembly 7 also includes a male portion more particularly shown in Figure 6 fixedly attachable at or adjacent one end thereof to the resonator 1 of the gyroscope projectable into the female portion, and fixedly attachable at or adjacent its other end in and to the female portion. The assembly 7 is such as to permit in operation damping of the vibratory movement of the resonator 1 in a direction substantially perpendicularly thereto whilst retaining stiffness and freedom to vibrate for the resonator 1 in the plane thereof.

The female portion includes an open ended axially bored tube-like metal collar 9 locatable in an aperture through the body 8 so that the longitudinal axis of the bore of the

collar 9 extends substantially perpendicularly to the body 8. The female portion also includes a pair of superimposed, spaced apart apertured metal rings or plates 10 preferably made of beryllium-copper alloy. These rings or plates 10 are of relatively small size generally approximately 24mm in diameter with a 0.12 mm thickness and a central aperture of 3.75mm in diameter. The rings or plates 10 are arrangeable with the aperture in one ring or plate 10 adjoining and surrounding one open end of the collar 9 and with the aperture or plate 10 adjoining in the other ring or plate 10 adjoining and surrounding the other open end of the collar 9 as shown in Figures 5 and 6. The outermost peripheral edge 10a of the two rings or plates 10 are directly attachable to the body 8 which extends therebetween. The female portion further includes a resilient apertured grommet 11, preferably made of silicon rubber, for damping, engagable via its aperture around the collar 9 and locatable between the rings or plates 10 for engagement at the outer periphery of the grommet 11 with the innermost peripheral edge or rim 8a of the body 8.

Further forming part of the female portion of the assembly 7 is a pair of axially bored bushes 12 each having a shoulder 12a extending externally around one end thereof.

The bushes 12 are locatable one at each end of the collar 9 to project into the bore therein with the external shoulder 12a on one bush 12 engaging the adjoining ring or plate 10 around the aperture therethrough and with the external shoulder 12a on the other bush 12 engaging the adjoining other ring or plate 10 around the aperture therethrough. Preferably the bushes 12 are made from an electrically insulating material such as a plastics material. The collar 9 is rigid whereas the grommet 11 is resilient. The grommet 11 in a typical example of the invention is of 12mm diameter and 5mm thick. A recess typically 1mm in width and 2mm deep is located midway around the grommet 11 into which recess the rim or outer edge 8a of the body is fixed. The body 8 has recessed portions 8b surrounding the aperture therethrough for accommodating the metal rings or plates 10 as shown particularly in Figure 5. The recessed portions 8b of the body 8 taper towards the body aperture and terminate in the reduced thickness rim 8a which defines the body aperture. The tapered regions 8c of the body 8 define with the rings or plates 10 and the grommet 11 pockets 13 which may contain air or other medium to assist in the damping effect of the grommet 11.

The male portion of the assembly 7 includes an open ended axially bored spindle 14 having a first open ended region 15 which is locatable in a bore through the resonator 1 and which has a shoulder 15a therearound for contact with the resonator 1 around one open end of the bore therethrough. The spindle 14 has a second open ended region 14a which is engagable in a first of the bushes 12 as shown in Figure 6. Also forming part of the spindle 14 is an intermediate flanged region 16 having a diameter greater than the diameter of the first and second regions 15 and 14a to bear at one surface against the resonator 1 and at an opposite surface against the first bush shoulder 12a.

The bushes 12 which are electrically insulating serve electrically to isolate the resonator 1 and the male portion of the assembly, which are at a high voltage, from the gyroscope body 8.

Also forming part of the male portion is an open ended axially bore metal bush 17 having a shoulder 17a extending externally therearound at or adjacent one end. The bush 17 is insertable in the second of the bushes 12 which is in the collar bore so that the shoulder 17a abuts the second bush shoulder 12a and a rod member 18 having an enlarged head 18a at one end is projectable through the spindle bore with the

head 18a engaging the spindle first region shoulder 15a, through the collar bore and into fixable engagement of the other end of the rod member with the metal bush 17. Preferably as illustrated in Figure 6 the rod member other end has an externally threaded region 18b for releasably fixable engagement with a correspondingly threaded internal region of the metal bush bore.

The components of the vibration damper assembly of the present invention and the resonator and body of the Vibrating Structure Gyroscope preferably are fixed together using an adhesive which conveniently is a toughed epoxy adhesive such as Araldite AV119 (Trade Mark). The adhesive provides adequate bonding and peel strength with a glass transition temperature above 90°C so that its mechanical properties remain substantially constant over the operating temperature range of the gyroscope.

The assembly 7 is stiff both torsionally and in the plane of the resonator 1 and thus will not introduce any additional in-plane motion in response to external vibration. The rate response and in-plane vibration response will thus be substantially unaffected. The assembly is, however, compliant in the axis normal to the resonator plane. Under the influence of vibration in the axis normal to the plane of

the resonator 1 the resonator/damper assembly will move relative to the gyro body 8.

However, where the Silicone rubber grommet 11 is recessed around its mid-point between the plates 10 it is fixed to the gyro body 8., As the assembly moves upwards the Silicone rubber, below the section is compressed with the rubber above being put into the tension and vice versa as the assembly moves downwards. This straining of the Silicone rubber grommet serves to dissipate the vibration energy thus suppressing the mechanical quality factor of the damper resonance.

The damper assembly 7 is designed such that it has an out-of-plane resonance frequency of around 1kHz with a Quality Factor of approximately 60. The ring resonance frequency is around 2KHz with a Quality Factor of approximately 2000. The relative motion of the gyro body/pick-offs, damper assembly and resonator 1 may be modelled by considering the respective components to be a series of coupled, damped spring-masses undergoing forced vibration. The response of this system may be compared with that of a system without a damper assembly of the invention. The system without a damper assembly is shown in Figure 7A, M_1 is the gyro body mass with K_1 and C_1 being the spring constant

and damping coefficient, respectively, connecting the gyro to the external vibration source (e.g. vibration table). M_2 is the effective resonator mass with K_2 and C_2 the spring constant and damping coefficient for the out-of-plane mode. The vibration is applied to the gyro body 8 and expressions may be derived for the relative motion of the masses as a function of input vibration frequency. The corresponding case including a damper assembly according to the invention is shown in Figure 7B. For this system, M_1 , K_1 and C_1 are as in Figure 7A but M_2 now represents the combined damper and spindle and resonator support mass. K_2 and C_2 refer to the coupling between the gyro body and damper assembly. M_3 is the effective resonator mass with K_3 and C_3 representing the resonator to damper assembly coupling.

The motion of the resonator 1 with respect to the pick-offs 5, 6 and body 8, with no damper assembly, is plotted in Fig. 8. The plot shows the relative motion on the y-axis (arbitrary units) as a function of applied vibration frequency, for a constant input vibration level. The amplitude of motion increases sharply at the out-of-plane bending frequency of the ring 1. It is this motion which saturates the pick-off output and leads to a dramatic increase in noise level. Fig 9 shows the effect of

introducing a damper assembly. A second resonance feature appears associated with the damper assembly resonance. The coupling between the ring 1 and damper assembly oscillations results in both resonant frequencies being perturbed from their independent natural frequencies. The damper assembly resonance appears approximately 100Hz below its unperturbed natural frequency with the ring resonance appearing some 200Hz above its natural frequency. The peak amplitude of the ring resonance is drastically reduced, by over an order of magnitude, by the presence of the damper assembly. This reduction is sufficient to prevent saturation of the pick-off output at high vibration levels thus enabling the gyro to meet the high vibration input performance specification.

By mounting the resonator 1 on a damper assembly 7 which is internal of the gyro, vibration sensitivity may be drastically reduced. The assembly 7 replaces the usual solid metal body 8 of the gyro on which the resonator is normally mounted and thus does not increase the gyro size.

Any relative motion of the resonator 1 with respect to the pick offs 5 at baseband frequencies (noise) due to the inclusion of the relatively low frequency resilient mounting does not contaminate the required rate signals due to the processes described in respect of Figure 4.

THE CLAIMS DEFINING THE INVENTION ARE AS FOLLOWS:

1. A vibration damper assembly adapted for use with
5 a substantially ring-like, substantially planar resonator
of a vibrating structure gyroscope, including a female
portion constructed so as to be attachable to a body of a
vibrating structure gyroscope to have limited resiliency
10 with respect to the body in a direction substantially
perpendicular to the resonator, and a male portion fixedly
attachable at or adjacent one end thereof to a resonator
of the vibrating structure gyroscope, projectable into the
female portion, and fixably attached at or adjacent its
15 other end in and to the female portion, wherein the
assembly is such as to permit, in operation, damping of
vibrating movement of the resonator in a direction
substantially perpendicular thereto whilst retaining
stiffness and freedom to vibrate for the resonator in the
20 plane thereof.

2. An assembly according to claim 1 wherein the
female portion includes an open ended axially bored tube-
like metal collar locatable in an aperture through the
body so that the longitudinal axis of the collar bore
25 extends substantially perpendicularly to the body, a pair
of superimposed, spaced apart aperture metal rings or
plates which are arrangeable with the aperture in one ring
or plate adjoining and



surrounding one open end of the collar and with the aperture in the other ring or plate adjoining and surrounding the other open end of the collar, and with the outermost peripheral edge of the two metal rings or plates being directly attachable to the body which extends therebetween, and a resilient apertured grommet engagable via its aperture around the collar and locatable between the pair of spaced rings or plate for engagement of the outer periphery of the grommet with the innermost peripheral edge of the body.

3. An assembly according to claim 2, wherein the pair of metal rings or plates are made of a beryllium-copper alloy and wherein the resilient, apertured grommet is made of silicone rubber.

4. An assembly according to claim 2 or claim 3, wherein the female portion includes a pair of axially bored bushes each having a shoulder extending externally around one end thereof, which bushes are locatable one at each end of the collar to project into the bore therein with the external shoulder on one of said bushes engaging the adjoining ring or plate around the aperture therethrough and with the external shoulder on the other of said bushes engaging the adjoining other ring or plate around the aperture therethrough.

5. An assembly according to claim 4, wherein the pair of bushes are made from an electrically insulating material.

6. An assembly according to claim 5, wherein the pair of bushes are made from a plastics material.

7. An assembly according to any one of claims 1 to 6, wherein the male portion includes an open ended axially bored spindle having a first open ended region which is locatable in a bore through the resonator and which has a shoulder therearound for contact with the resonator around one open end of the bore therethrough, having a second open ended region which is engagable in the first of the bushes when the latter is located in the collar bore and having an intermediate flanged region having a diameter greater than the diameters of the first and second regions to bear at one surface against the resonator and at an opposite surface against the first bush shoulder.

8. An assembly according to claim 7, wherein the male portion includes an open ended axially bore metal bush having a shoulder extending externally therearound at or adjacent one end, which metal bush is insertable in the second of said bushes, when the latter is located in the collar bore, so that the metal bush shoulder abuts the second bush shoulder, and a rod member having an enlarged head at one end which rod

member is projectable through the spindle bore, with a head engaging the spindle first region should, through the collar bore and into fixable engagement at the other end
5 of the rod member with the metal bush.

9. An assembly according to claim 8, wherein the rod member other end has an externally threaded region for fixable engagement with a corresponding threaded internal
10 region of the metal bush bore.

10. A vibration damper assembly for a substantially ring-like, substantially planar resonator of a Vibrating Structure Gyroscope substantially as hereinbefore
15 described and as illustrated in Figures 5 and 6 as modified or not by Figures 7B and 9 of the accompanying drawings.

11. A Vibrating Structure Gyroscope having a
20 substantially ring-like, substantially planar resonator and a vibration damper assembly according to any one of claims 2-10, attaching the resonator to an apertured substantially planar portion of a body of the gyroscope, wherein the resonator is made of metal, and the body has
25 recessed portions surrounding the aperture therethrough for accommodating the metal rings or plates, which recessed portions taper towards the body aperture and terminate in a reduced thickness rim which defines said body aperture, with
30



the body rim engaging the resilient grommet which assists in the damping effect.

12. A Vibrating Structure Gyroscope according to claim 11, wherein the components of the vibration damper assembly and of the resonator and body are secured together with an adhesive.

13. A vibrating structure gyroscope according to claim 12, wherein the adhesive is a toughened epoxy adhesive.

14. A vibrating structure gyroscope according to any one of claims 11 to 13, substantially as hereinbefore described and as illustrated in any one of Figures 5 and 6 as modified or not by Figures 7B or 9 of the accompanying drawings.

Fig.1A.

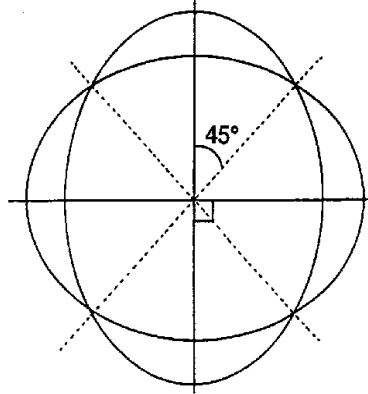


Fig.1B.

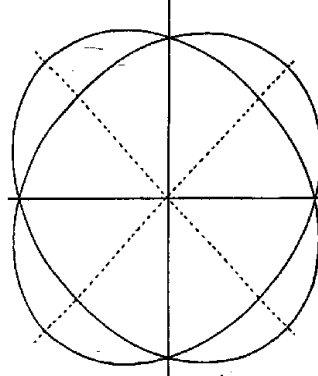


Fig.2.

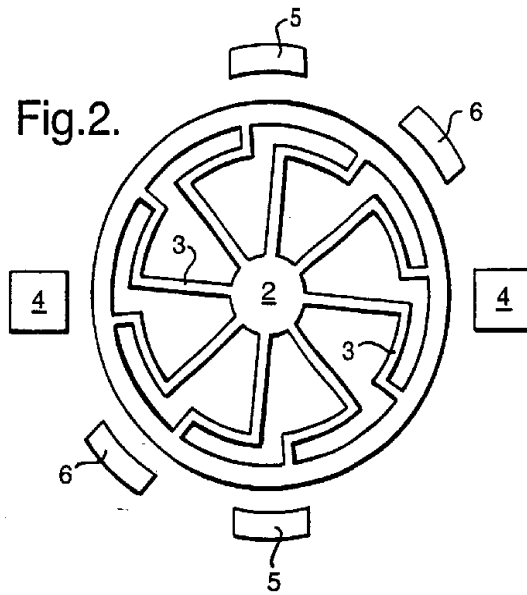
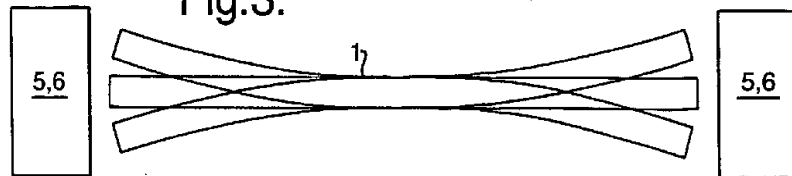


Fig.3.



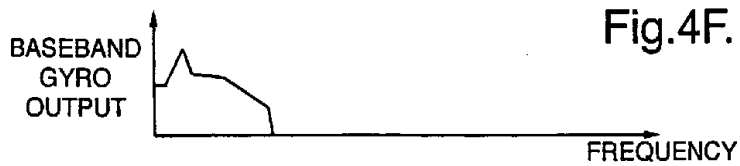
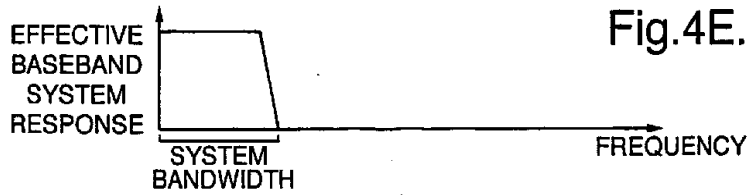
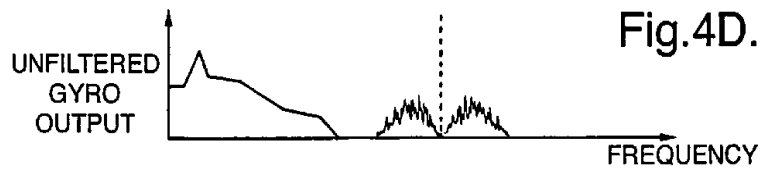
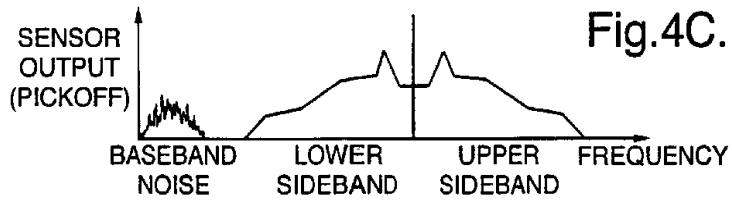
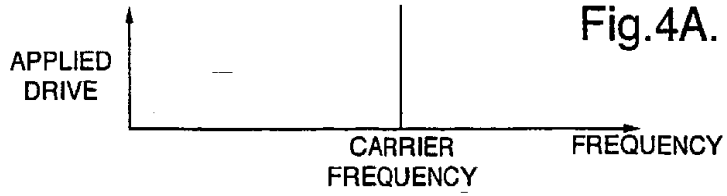


Fig.5.

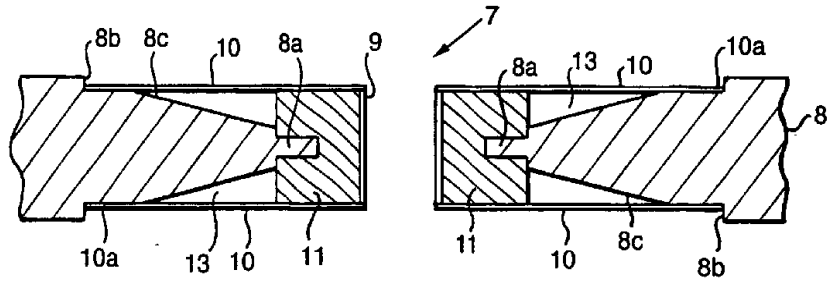


Fig.7A.

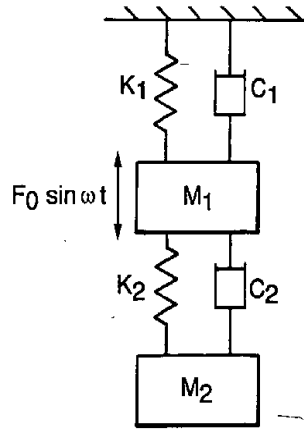


Fig.7B.

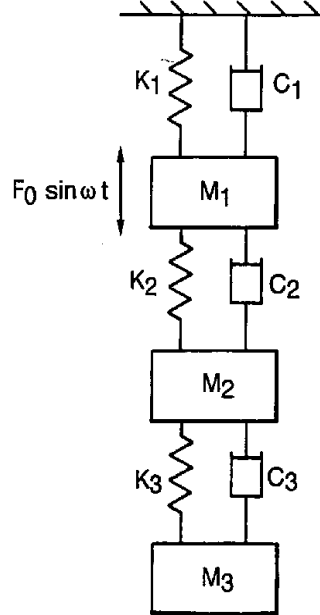


Fig.6.

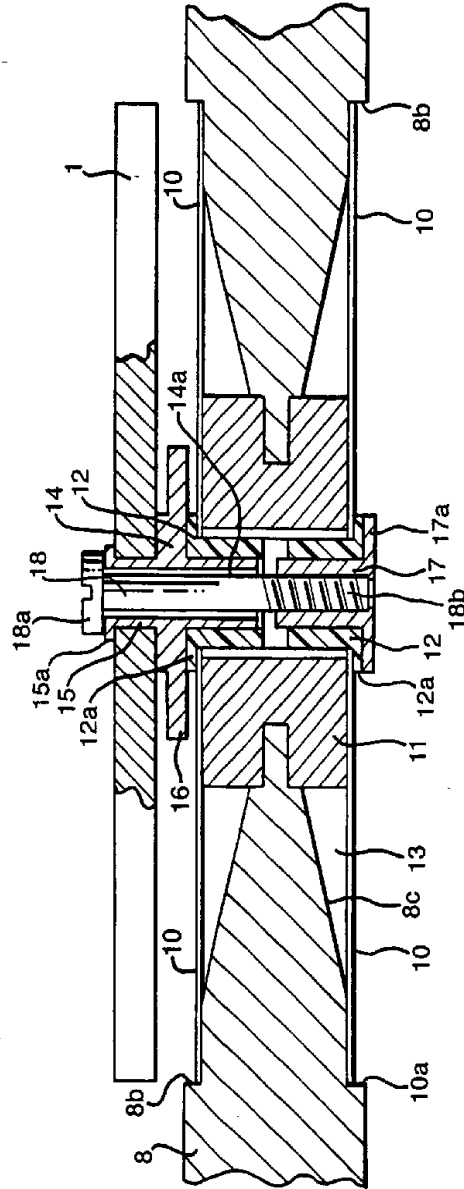


Fig.8.

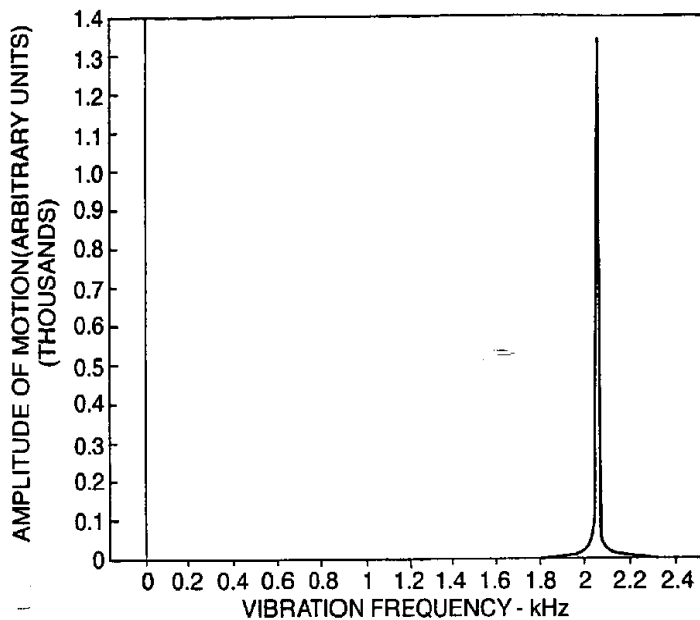


Fig.9.

