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(54) Title: APPARATUS FOR CONTROLLING A TOILET SYSTEM

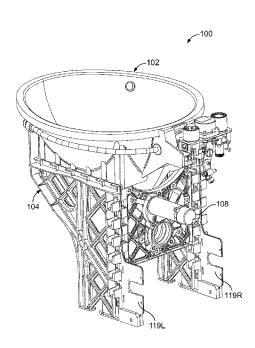


FIG. 1

(57) Abstract: A toilet system configured for operation on-board an aircraft includes, according to an implementation, a toilet bowl, a left side support bracket supporting the toilet bowl, a right side support bracket supporting the toilet bowl, a control box including a first set of pins, and a harness connector configured to electrically couple the toilet to the aircraft. The harness connector includes a set of pins, where the first set of pins of the control box are configured to be connected to the set of pins included in the harness connector, and the control box is configured to determine a platform of the aircraft based on the presence or absence of a pin in the set of pins included in the harness connector.



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APPARATUS FOR CONTROLLING A TOILET SYSTEM

Cross References to Related Applications

[0001] The present application claims the benefit of U.S. Provisional Application No. 62/140,260, entitled "VACUUM TOILET SYSTEM AND INSTALLATION METHOD THEREOF," and filed on March 30, 2015, which is incorporated herein by reference in its entirety.

Field of the Disclosure

[0002] The present disclosure is relates generally to toilets and, more particularly, an apparatus for controlling a toilet system.

Background

[0003] Designing a toilet for an aircraft poses challenges that do not generally occur in ground-based toilet designs. For example, in an aircraft, space and weight are at a premium, and using regular water-flush toilets is not practical. In addition, the toilet system must be compliant with aircraft specifications and must be configurable to interact with aircraft software platforms on different aircraft manufacturers.

Drawings

[0004] While the appended claims set forth the features of the present techniques with particularity, these techniques may be best understood from the following detailed description taken in conjunction with the accompanying drawings of which:

- [0005] FIG. 1 is a perspective view of a toilet configured according to an embodiment.
- [0006] FIG. 2 is a rear view of a controller mounted on a toilet configured according to an embodiment.
- [0007] FIG. 3 is a rear view of a controller mounted on a toilet system configured according to another embodiment.
- [0008] FIG. 4 is an exploded view of a controller pin configuration, according to an embodiment.
- [0009] FIG. 5 is a block diagram depicting logic circuitry that may be used in an embodiment.

Detailed Description

[0010] The following discussion is directed to various exemplary embodiments. However, one possessing ordinary skill in the art will understand that the examples disclosed herein have broad application, and that the discussion of any embodiment is meant only be exemplary of that embodiment, and not intended to suggest that the scope of the disclosure, including claims, is limited to that embodiment.

[0011] Certain terms are used throughout the following description to refer to particular features or components. As one skilled in the art will appreciate, different persons may refer to the same feature or component by different names. This document does not intend to distinguish between components or features that differ in name but not function. The drawing figures are not necessarily to scale. Certain features and components herein may be shown exaggerated in scale or in somewhat schematic form and some details of conventional elements may not be shown in interest of clarity and conciseness.

[0012] The disclosure is generally directed to a toilet system for use on an aircraft. According to an embodiment, the toilet system is configured for operation on-board an aircraft and includes a toilet bowl, a left side support bracket supporting the toilet bowl, a right side support bracket supporting the toilet bowl, a control box including a first set of pins, and a harness connector configured to electrically couple the toilet to the aircraft. The harness connector includes a set of pins, where the first set of pins of the control box are configured to be connected to the set of pins included in the harness connector, and the control box is configured to determine a platform of the aircraft (e.g., which vendor's software platform is being used or which "brand" of software platform is being used) based on the presence or absence of a pin in the set of pins included in the harness connector.

[0013] The term "logic circuitry" as used herein means a circuit (a type of electronic hardware) designed to perform complex functions defined in terms of mathematical logic. Examples of logic circuitry include a microprocessor, a controller, an application-specific integrated circuit, and a field-programmable gate array.

[0014] Turning to FIG. 1, a toilet configured according to an embodiment is shown. The toilet, generally labelled 100, is configured to be deployed on an aircraft, and may be housed within an external housing (not shown). The toilet 100 includes a bowl 102 attached to a frame 104. The toilet 100 also includes a motor 108 attached to the frame 104. The motor 108 drives a mechanical actuator to open or close a door to a waste outlet. The toilet 100 further

includes side support brackets 119L and 119R. According to an embodiment, the left side support bracket 119L is identical to the right side support bracket 119R. One skilled in the art will understand that the left side support bracket 119L and the right side support bracket 119R may not be identical in other embodiments. According to an embodiment, the side support brackets 119L and 119R each include a respective cut-out portion 120L and 120R. FIGS. 2 and 3 show rear views of a toilet system having a control box 118 mounted on the side support brackets 119L and 119R respectively.

[0015] Turning now to FIGS. 2 and 3, the toilet 100 further includes a control box 118 attached to the frame 104. The control box 118 contains logic circuitry 502 (discussed later with reference to FIG. 5) that controls the operation of the toilet 100. FIG. 2 depicts an embodiment of the toilet 100 in which the control box 118 is mounted on the left side support bracket 119L. FIG. 3 depicts another embodiment of the toilet 100 in which the control box 118 is mounted on the right side support bracket 119R. Although depicted on the left side support bracket 119L and the right side support bracket 119R according to the embodiments above, the control box 118 may be attached to the frame 104 at other locations.

10016] A cable 116 is attached to the control box 118 and to the motor 108. The cable 116 provides a transmission medium for electrical signals to travel from the control box 118 to the motor 108 (e.g., power and data) and for electrical signals to travel from the motor 108 to the control box 118 (e.g., data). For example, the logic circuitry in the control box 118 can transmit control signals to the motor 108 by way of the cable 116, and the motor 108 can transmit position signals to the logic circuitry by way of the cable 116. During operation, the door to the waste outlet is normally closed. When the logic circuitry in the control box 118 receives an activation signal (e.g., a signal generated by a "flush" button on the toilet 100), the logic circuitry sends a control signal to the motor 108 to open the door to the waste outlet. A pressure difference between the air in the waste pipe 106 and the air around the outside of the toilet 100 creates a suction at the waste outlet, which draws the waste from inside the bowl 102 out of the waste outlet. After a predetermined period of time, the motor 108 drives the mechanical actuator to close the door.

[0017] A cable 117 is also attached to the control box 118 and connects to the aircraft. The cable 117 provides a transmission medium for electrical signal to travel from the control box 118 to the aircraft (e.g., data) and for electrical signals to travel from the aircraft to the control box 118 (e.g., power and data). For instance, the logic circuitry in the control box 118 can transmit a flush count, total number of hours of operation of the toilet 100, and a

flush valve time to open/close to enable the aircraft to perform maintenance diagnostics. According to an embodiment, the control box 118 is mounted on one of the left side support bracket 119L and the right side support bracket 119R, and the cable 117 is configured to be disposed in a respective cut out portion 119L and 119R of the other of the left side support bracket 119L and the right side support bracket 119R.

[0018] Turning to FIG. 4, in an embodiment, the control box 118 includes a casing 302, a first set of pins $310_1...310_M$, and a second set of pins $312_1...312_M$. In an embodiment, the casing 302 is environmentally sealed and constructed from Electromagnetic Interference (EMI) resistant engineered polymer. In an embodiment, the casing 302 is formed of a nickel-impregnated polyethyleneimine (PEI) housing to provide electromagnetic compatibility mitigation.

[0019] During operation, the pins first set of pins $310_1...310_M$ are connected to the aircraft via a harness connector and the second set of pins $312_1...312_N$ are connected to the motor 108. In an embodiment, the control box 118 determines the aircraft specifications based on the presence or absence of a particular pin in the harness connector connecting the control box 118 to the aircraft.

[0020] For instance, in an embodiment, the control box 118 includes a first set of pins $310_1...310_M$, where M = 10. If, for example, the harness connector includes 10 pins that correspond to the first set of pins $310_1...310_{10}$ in the control box 118, the control box 118 may receive aircraft configuration information via the harness connector and determine that the aircraft includes a software platform of a first aircraft manufacturer and operate the toilet 100 accordingly. If, for example in another embodiment, the harness connector includes 9 pins that correspond to the first set of pins $310_1...310_9$ in the control box 118, the control box 118 may receive aircraft configuration information via the harness connector and determine that the aircraft includes a software platform of a second aircraft manufacturer and operate the toilet 100 accordingly. Of course, a person skilled in the art will understand that any number of pins may be provided in the first set of pins $310_1...310_M$ of the control box 118. Moreover, a person skilled in the art will understand that the harness connector may include any number of pins as determined by the configuration information of the respective aircraft platform and is not limited to the example discussed above.

[0021] Turning to FIG. 5, in an embodiment, the control box 118 includes logic circuitry 502 that generally controls the operation of the toilet 100. The logic circuitry 502 of the

control box 118 is configured to determine the platform of the aircraft on which the toilet 100 is installed based on the pin configuration of the first set of pins $310_1...310_M$. The control box 118 is configured to provide a transmission medium for electrical signal to travel from the toilet 100 to the aircraft (e.g., data) and for electrical signals to travel from the aircraft to the toilet 100 (e.g., power and data). For instance, the logic circuitry 502 in the control box 118 can transmit a flush count, total number of hours of operation of the toilet 100, and a flush valve time to open/close to enable the aircraft to perform maintenance diagnostics. One skilled in the art will understand that additional information may be shared between the toilet 100 and the aircraft via the control box 118. For instance, in an exemplary implementation, the control box 118 may communicate any failure issues (e.g., clogs, power loss, motor failure, etc.) to the aircraft.

[0022] By way of example, in an embodiment, the motor 108 includes logic circuitry that controls the operation of the motor 108. The logic circuitry 502 communicates with logic circuitry 504 of the motor 108. In an embodiment, when the logic circuitry 502 of the toilet 100 carries out a start-up operation (e.g., upon power-up of the toilet 100), the logic circuitry 502 transmits, to the logic circuitry 504 of the motor 108, a command to put the motor 108 in a zero position. In order to reach the zero position, the logic circuitry 504 causes the rotor 206 to rotate in the appropriate direction and for the appropriate number of revolutions in order to move the door 302 to a predetermined position, such as fully open or fully closed.

[0023] A feature of the control box 118 in accordance with the embodiments discussed above is that a common circuit board hardware and software (implementing logic circuitry 502) may be used in conjunction with platforms of various aircraft manufacturers. Additional features of the control box 118 in accordance with the embodiments discussed above will be readily apparent to one skilled in the art.

[0024] For the purposes of promoting an understanding of the principles of the disclosure, reference has been made to the embodiments illustrated in the drawings, and specific language has been used to describe these embodiments. However, no limitation of the scope of the disclosure is intended by this specific language, and the disclosure should be construed to encompass all embodiments that would normally occur to one of ordinary skill in the art.

[0025] The particular implementations shown and described herein are illustrative examples and are not intended to otherwise limit the scope of the disclosure in any way. For

the sake of brevity, conventional electronics, control systems, software development and other functional aspects of the systems (and components of the individual operating components of the systems) may not be described in detail.

[0026] The steps of all methods described herein are performable in any suitable order unless otherwise indicated herein or otherwise clearly contradicted by context. The use of any and all examples, or exemplary language (e.g., "such as") provided herein, is intended merely to better illuminate the disclosure and does not pose a limitation on scope unless otherwise claimed. Numerous modifications and adaptations will be readily apparent to those skilled in this art without departing from the spirit and scope of the disclosure.

[0027] It will also be recognized that the terms "comprises," "comprising," "includes," "including," "has," and "having," as used herein, are specifically intended to be read as openended terms of art. The use of the terms "a" and "an" and "the" and similar referents in the context of describing the invention (especially in the context of the following claims) are to be construed to cover both the singular and the plural, unless the context clearly indicates otherwise. In addition, it should be understood that although the terms "first," "second," etc. may be used herein to describe various elements, these elements should not be limited by these terms, which are only used to distinguish one element from another.

We claim:

1. A toilet system configured for operation on-board an aircraft comprising:

- a toilet bowl;
- a left side support bracket supporting the toilet bowl;
- a right side support bracket supporting the toilet bowl;
- a control box including a first set of pins; and
- a harness connector configured to electrically couple the toilet to the aircraft, the harness connector including a set of pins; wherein

the first set of pins of the control box are configured to be connected to the set of pins included in the harness connector; and

the control box is configured to determine a platform of the aircraft based on the presence or absence of a pin in the set of pins included in the harness connector.

2. The toilet system according to claim 1, wherein

the left side support bracket and the right side support bracket each contain a respective cut-out portion.

3. The toilet system according to claim 2, wherein

the control box is mounted on one of the left side support bracket and the right side support bracket, and

the harness connector is disposed within the cut out portion of the other of the left side support bracket and the right side support bracket.

4. The toilet system according to claim 1, further comprising

motor via a cable.

a motor configured to control a door to a waste outlet of the toilet bowl; wherein the control box includes a second set of pins configured to electrically control the

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5. The toilet system according to claim 2, wherein the left side support bracket is identical to the right side support bracket.

- 6. The toilet system according to claim 1, wherein the control box is configured to transmit at least one of a flush count, a total number of operational hours, and flush valve time to open/close, to the aircraft via the harness connector.
- 7. The toilet system according to claim 1, wherein the control box is disposed in a casing.
- 8. The toilet system according to claim 7, wherein the casing is formed of an electromagnetic interference resistant engineered polymer.
- 9. The toilet system according to claim 7, wherein the casing is formed of a a nickel-impregnated polyethyleneimine (PEI) housing.
- 10. The toilet system according to claim 1, wherein the control box is configured to provide at least one of diagnostic and fault information to the aircraft.
- 11. The toilet system according to claim 3, wherein the control box is mounted on one of the left side support bracket and the right side support bracket via a plurality of screws.

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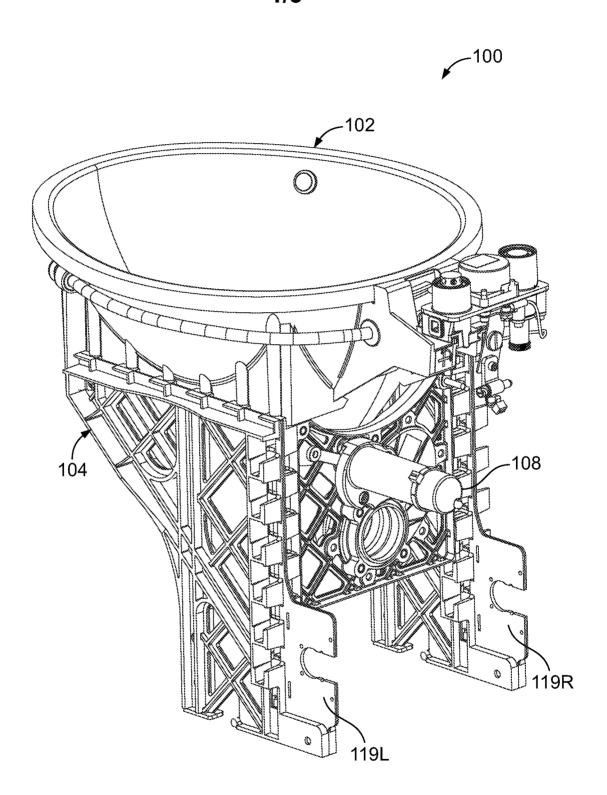
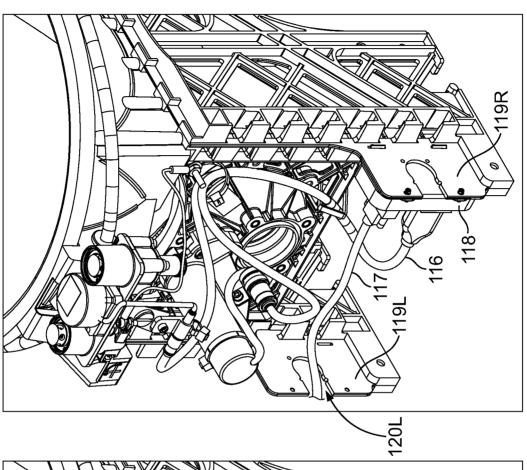


FIG. 1





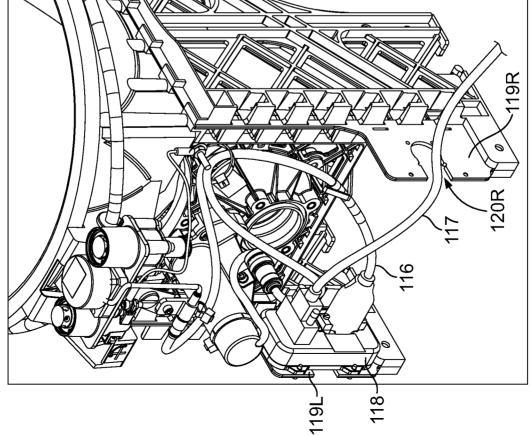


FIG. 2

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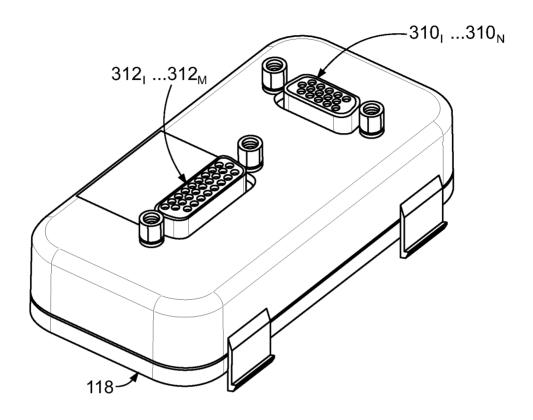


FIG. 4

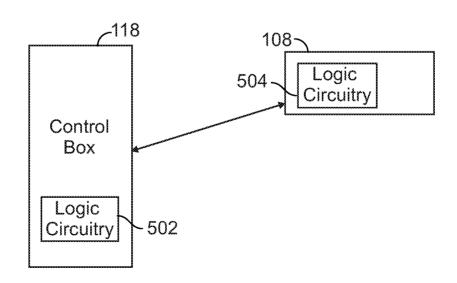


FIG. 5

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INTERNATIONAL SEARCH REPORT

International application No. PCT/US2016/024936

Α.	CLASSIFICATION	OF SUBJECT MATTE	R

IPC(8) - B64D 11/02; E03D 3/10; E03D 5/012; E03D 5/10; E03D 11/10 (2016.01)

CPC - B64D 11/02; B61D 35/007; E03D 5/012; E03D 5/024; E03F 1/0006 (2016.05)

According to International Patent Classification (IPC) or to both national classification and IPC

B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols)

IPC(8) - B64D 11/02; E03D 3/10; E03D 5/012; E03D 5/10; E03D 11/10 (2016.01) CPC - B61D 35/007; B64D 11/02; E03D 5/012; E03D 5/024; E03F 1/0006 (2016.05)

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched USPC - 4/252.1; 4/316; 4/431; 4/434; 4/435 (keyword delimited)

Electronic data base consulted during the international search (name of data base and, where practicable, search terms used)

PatBase, Google Patents, Google Scholar, Google

Search terms used: toilet, control, controller, connector, harness, harness connector, plural, multi, pin, aircraft, airplane, electrical, electronic, cable, processor, microprocessor, computer, control box, control module, bowl, bracket, vacuum toilet, connection, adaptable, different, support, specifications, determines

C. DOCUMENTS CONSIDERED TO BE RELEVANT

Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
4	US 2001/0034902 A1 (TYLER) 01 November 2001 (01.11.2001) entire document	1-11
4	US 2002/0069459 A1 (PONDELICK et al) 13 June 2002 (13.06.2002) entire document	1-11
	US 2010/0083435 A1 (HOANG et al) 08 April 2010 (08.04.2010) entire document	
	US 4,713,847 A (OLDFELT et al) 22 December 1987 (22.12.1987) entire document	
4	US 5,035,011 A (ROZENBLATT et al) 30 July 1991 (30.07.1991) entire document	1-11

	Further documents are listed in the continuation of Box C.	See patent family annex.		
* "A"	Special categories of cited documents: document defining the general state of the art which is not considered to be of particular relevance	"T" later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle or theory underlying the invention		
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"P"	means document published prior to the international filing date but later that the priority date claimed	being obvious to a person skilled in the art an "&" document member of the same patent family		
Date of the actual completion of the international search		Date of mailing of the international search report		
24 May 2016		1 6 J U N 2016		
Name and mailing address of the ISA/ Mail Stop PCT, Attn: ISA/US, Commissioner for Patents P.O. Box 1450, Alexandria, VA 22313-1450 Facsimile No. 571-273-8300		Authorized officer Blaine R. Copenheaver PCT Helpdesk: 571-272-4300 PCT OSP: 571-272-7774		

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