

[54] **INJECTOR HEAD**

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[52] U.S. Cl. **239/400, 239/428, 60/39.74 A**

[51] Int. Cl. **B05b 7/10**

[58] Field of Search **60/258; 239/400, 403, 404, 239/428**

[56] **References Cited**

UNITED STATES PATENTS

3,242,670 3/1966 Buswell **60/258**
3,260,461 7/1966 Biber et al. **239/403**

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FOREIGN PATENTS OR APPLICATIONS

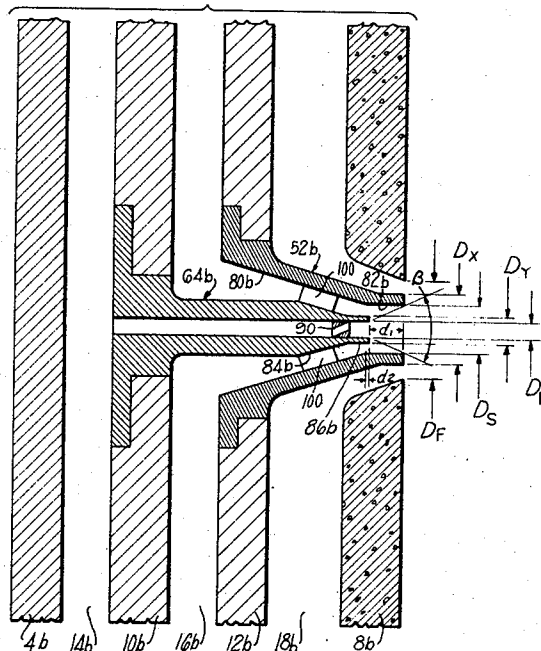
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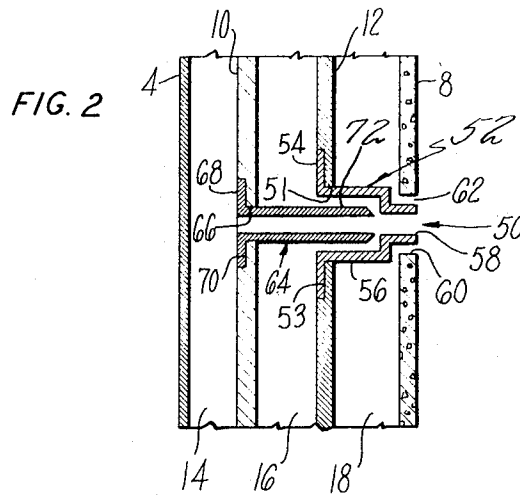
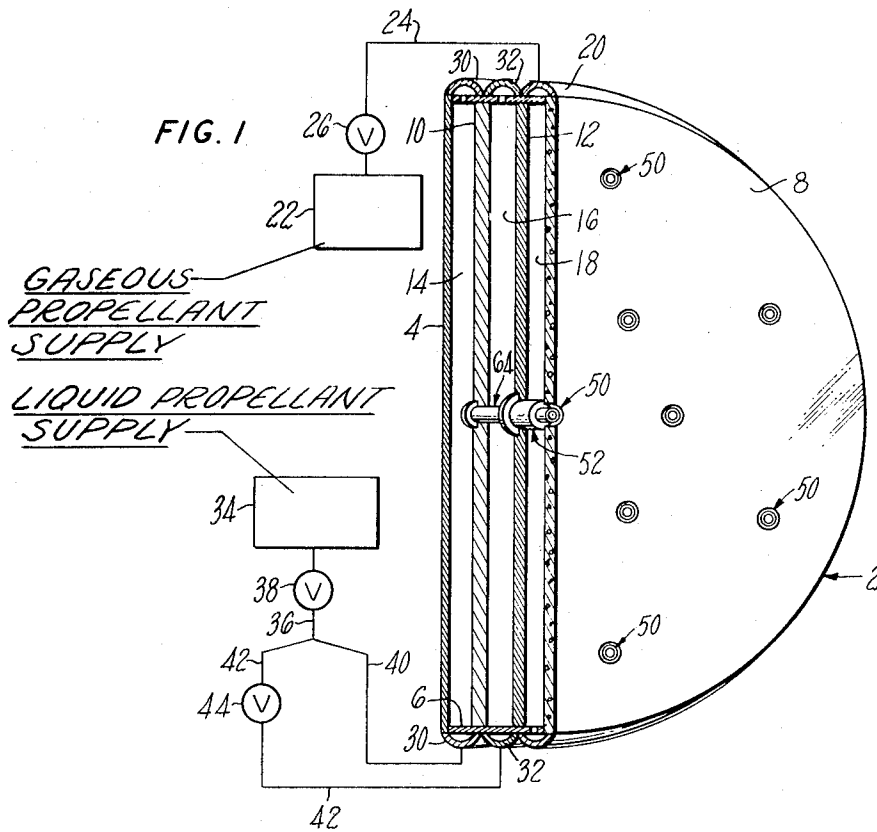
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[57] **ABSTRACT**

An injector head having a plurality of manifolds formed by plates spaced rearwardly of the face plate. The second plate spaced adjacent said face plate having a first projection which extends into an opening through the face plate. The third plate positioned rearwardly of said second plate having a second projection which extends into an opening which passes through the first projection of said second plate. An opening passes through said second projection. A back plate is located behind the third plate. Said projections having tapered sections to provide for a desired pressure drop.

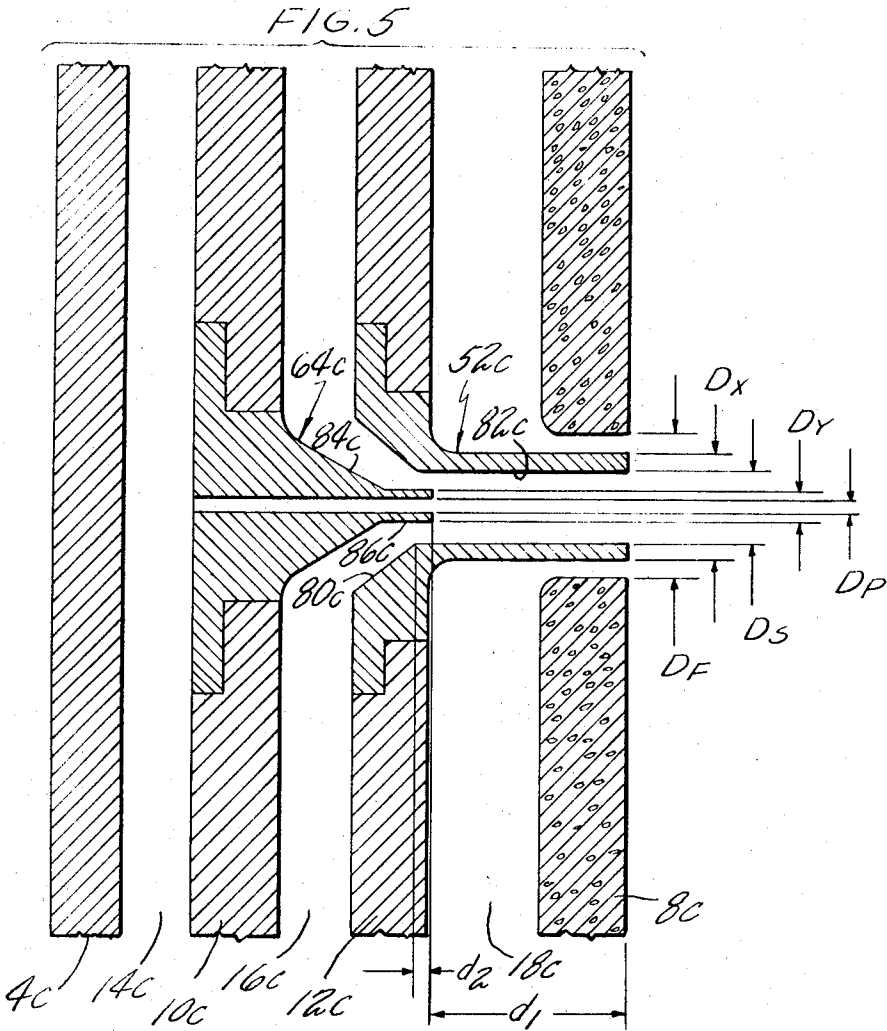
13 Claims, 5 Drawing Figures

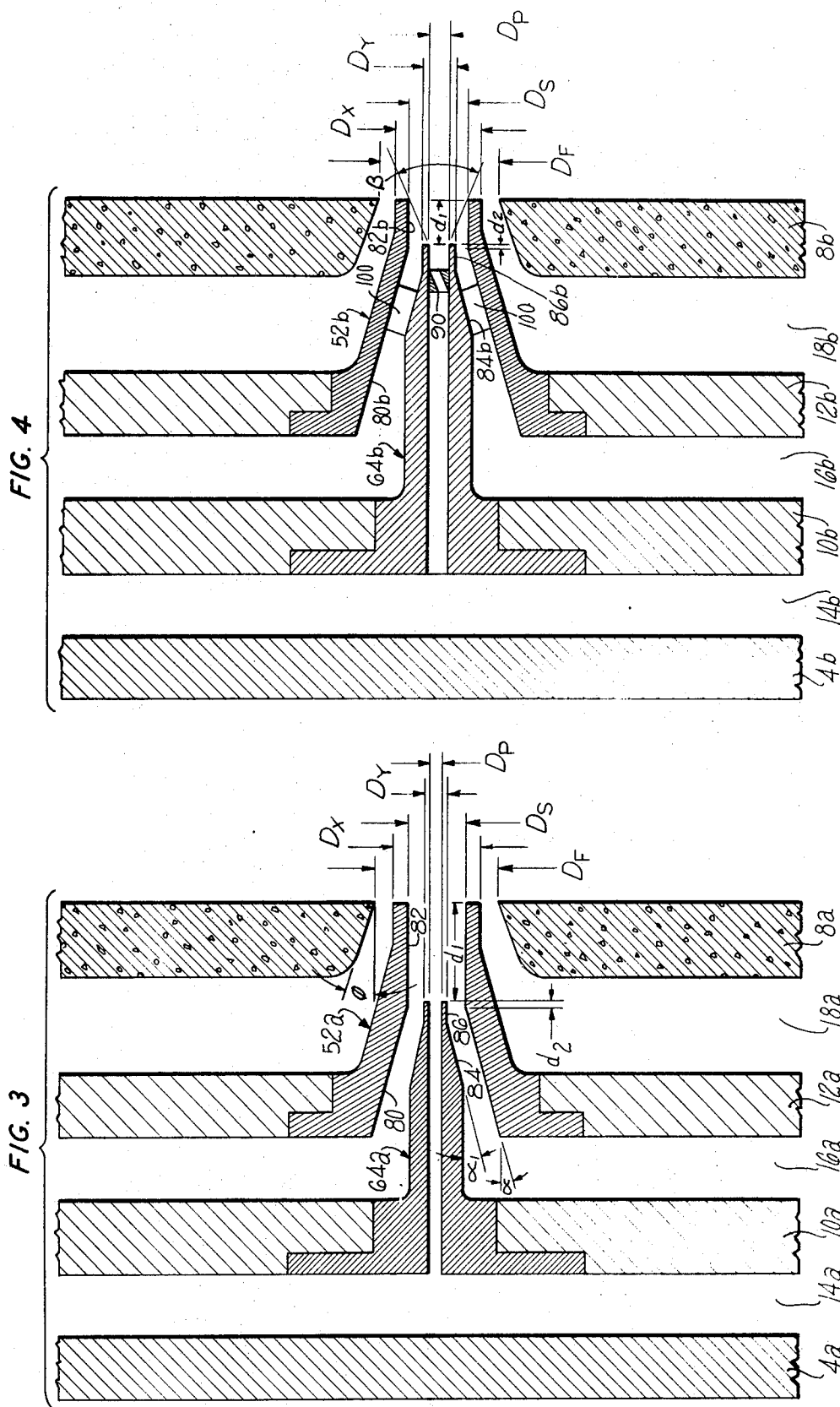




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INJECTOR HEAD

The invention herein described was made in the course of or under a contract with the Department of the Air Force.

This invention relates to an injector head for a bipropellant rocket engine wherein one propellant reaches the injector in a liquid state and the other propellant reaches the injector in a gaseous state.

An object of this invention is to provide an injector head with a dual orifice nozzle in the liquid side having a deep throttling capability with a range of propellants.

Another object of this invention is to provide an injector head including dual orifice nozzles which will provide high combustion performance in throttleable rocket engines.

A further object of this invention is to provide an injector head having a dual orifice nozzle wherein the primary nozzle is positioned for exhausting a flow therethrough into the secondary orifice such that its outlet is at the beginning of the secondary orifice or projecting thereinto and being retracted into the secondary orifice from the face of the injector head at least 0.5 times the diameter of the secondary orifice.

Another object of this invention is to provide a dual orifice nozzle wherein the conical inlet to the point where the primary nozzle is inserted in the secondary nozzle such that the major secondary pressure drop is approximately at the primary discharge plane.

A further object of this invention is to provide a dual nozzle for an injector head wherein the diameter of the primary orifice lies in the range of from 0.1 to 0.5 times the diameter of the secondary orifice.

Another object of this invention is to provide swirlers in the nozzles to affect higher mixing.

A further object of this invention is to provide an injector which could use gaseous hydrogen for its fuel and liquid fluorine for the oxidizer.

Another object of this invention is to provide an injector having a plurality of elements, each element having dual orifices for the injection of a liquid propellant from primary and secondary manifolds and a single orifice for the injection of a gaseous propellant.

A further object of this invention is to have a flow divider valve in the liquid propellant feedline to the manifold containing the secondary liquid propellant.

Another object of this invention is to provide a swirler in the end of the primary liquid propellant nozzle to enhance mixing between the flows from the primary and secondary orifices.

Other objects and advantages will be apparent from the specification and claims and from the accompanying drawings which illustrate embodiments of the invention.

FIG. 1 is a view showing a schematic perspective view of the injector head cut away through the middle for clarity.

FIG. 2 is a schematic showing of an enlarged sectional view taken through one of the complete nozzles.

FIG. 3 is an enlarged view showing another modification of a dual orifice nozzle.

FIG. 4 is an enlarged view showing still another modification of a dual orifice nozzle having swirlers.

FIG. 5 is an enlarged view showing still another modification of a dual orifice nozzle.

As seen in FIG. 1, an injector head 2 is formed having a back plate 4 and an annular end plate 6 which is faced with a plate of permeable material 8. Between the back plate 4 and face plate 8, two other plates 10 and 12 are fixedly positioned and sealed with end wall 6 to form three chambers 14, 16 and 18. The rearwardmost chamber 14 is connected to a primary manifold 30 for the liquid propellant and the center chamber 16 is connected to a secondary liquid propellant manifold 32. A gaseous propellant is admitted from a manifold 20 to the end chamber 18 having the permeable plate 8 as one side thereof.

The gaseous propellant is supplied to manifold 20 from a supply 22 through a conduit 24 having a control valve 26. A liquid propellant is directed to manifolds 30 and 32 from an oxygen supply 34 by a conduit system involving a feed conduit 36 being connected to two conduits 40 and 42 which are con-

nected to manifolds 30 and 32, respectively. Conduit 42 has a flow divider valve 44 connected therein. A control system as described here is also shown in U.S. application, Ser. No. 426,711, filed Jan. 15, 1965, to J. P. Mitchell et al.

The liquid propellant is brought to the face of the injector at a plurality of points by injector units 50. One part 52 of an injector unit 50 extends through a hole 51 in plate 12 and has a flange 54 which cooperates with a recess 53 formed in the back of plate 12 around the hole 51. This flange 54 is fixedly retained in the recess 53 by any means desired. The part 52 comprises a cylindrical member 56 which extends towards the plate 8 and is narrowed to a second cylindrical member 58 of smaller diameter. This member of smaller diameter, which is cylindrical in shape, projects into a hole 60 formed in the permeable plate 8. An annular opening 62 is formed around the portion 58, said annular opening being connected to the chamber 18.

Another part 64 of injector unit 50 extends through a hole 66 in plate 10 and has a flange 68 which cooperates with a recess 70 formed in the back of plate 10 around the hole 66. This flange 68 is fixedly retained in the recess 70 by any means desired. This part 64 comprises a cylindrical member 72 which extends into the cylindrical member 56 and is positioned substantially coaxial therewith. The free end of the member 72 is located adjacent the inner end of the second cylindrical member 58. This construction is arranged to provide an ejector effect between a flow passing through the cylindrical member 72 into the cylindrical member 58 so as to pull fuel from the annular passageway formed between the outer surface of cylindrical member 72 and inner surface of cylindrical member 56. Parts 52 and 64 can be made integral with plates 12 and 10, respectively, if desired. This would, of course, make sealing at that location unnecessary.

As seen in FIG. 3, this modification of the dual orifice nozzle sets forth a different configuration of the two main parts, 52a and 64a. While the manner in which these parts are connected to plates 10a and 12a remains the same, the configuration which projects outwardly differs. The part 52a is shown having an opening therethrough formed of a conical surface 80 and a cylindrical surface 82. The section 64a extends into the opening through section 52a and has a conical surface 84 formed on the exterior thereof adjacent the end which has a short cylindrical surface 86. The conical surface 84 and short cylindrical surface 86 cooperate with the conical surface 80 and cylindrical surface 82 to insure that the secondary flow between the parts 52a and 64a will not be restricted upstream of the primary discharge plane. The cylindrical surface 82 forms with cylindrical surface 86 the secondary orifice while the passageway through the part 64a forms the primary orifice. It can be seen, therefore, that the angle α of the conical surface 80 will never be less than the angle α_1 of the conical surface 84 ($\alpha > \alpha_1$).

The tip of the part 64a should extend to at least the inner end of the cylindrical surface 82, ($d_2 > 0$) and this tip of the part 64a should also be recessed from the open end of the cylindrical surface 82 a minimum of one-half times the diameter of the cylindrical surface 82 ($d_1 > 0.5D_2$). This length of the cylindrical surface 82 forming a mixing chamber between the tip of 64a and the tip of the part 52a. As mentioned before concerning the general nozzle arrangement in this specific construction, the primary orifice having a diameter D_p must be concentric with the cylindrical surface 82 having a diameter D_s , within 0.1 times D_s . Further, D_p should fall in a range of from 0.1 D_s to 0.5 D_s .

As seen in FIG. 4, this modification of the dual orifice nozzle sets forth a construction basically similar to that shown in FIG. 3; however, swirler vanes 100 have been located in the passageway between the conical surfaces 80b and 84b. Further, a swirler unit 90 has been placed in the passageway extending through the part 64b. The swirler 90 must have its vanes arranged so that the primary flow will not impinge the sides of the secondary orifice. As viewed in FIG. 4, it must be spread only as wide as the angle β shown drawn from the end

of the part 64b. This angle is determined neglecting secondary flow. In a modification having a swirler unit 90, the tip of the part 64b should extend to at least the inner end of the cylindrical surface 82b, and this tip of the part 64b can also extend to the outer end of the cylindrical surface 82b. It can be seen that at the location of this tip, the size is of maximum angle β for a particular nozzle. The use of the swirler unit 90 permits the mixing chamber to be reduced in length.

As seen in FIG. 5, this modification of the dual orifice nozzle sets forth a construction while basically similar to that shown in FIG. 3, has a long cylindrical surface 82. Two injectors were constructed and tested having a configuration such as shown in FIG. 5 for the dual orifice units. The injectors had the following dimensions:

Injector No. 1	Injector No. 2
$D_F = 0.170$ in.	$D_F = 0.170$ in.
$D_Y = 0.125$ in.	$D_Y = 0.125$ in.
$D_S = 0.072$ in.	$D_S = 0.072$ in.
$D_X = 0.036$ in.	$D_X = 0.031$ in.
$D_P = 0.010$ in.	$D_P = 0.0155$ in.
$d_1 = 0.200$ in.	$d_1 = 0.200$ in.
$d_2 = 0.015$ in.	$d_2 = 0.015$ in.

For both injectors, the permeable material used for the face plate 8 had a permeability of 40 scfm and the angle α was approximately 20° and the angle α_1 was approximately 15° . The oxidizer used was fluorine and the fuel was gaseous hydrogen.

An injector was constructed and tested having a configuration such as shown in FIG. 3. The injector had the following dimensions:

Injector No. 3
 $D_F = 0.127$ in.
 $D_Y = 0.106$ in.
 $D_S = 0.066$ in.
 $D_X = 0.035$ in.
 $D_P = 0.0151$ in.
 $d_1 = 0.150$ in.
 $d_2 = 0.015$ in.

For this injector, the permeable material used for the face plate 8 had a permeability of 40 scfm, and the angle α was approximately 15° and the angle α_1 was approximately 12.5° . The angle θ was approximately 20° . This was done to produce a sharp edged orifice and to permit nozzle assembly since the plate 8a forming the surface of the injector was formed as a cone and the parts 52a were formed integral with the plate 12a.

It is to be understood that the invention is not limited to the specific embodiments herein illustrated and described, but may be used in other ways without departure from its spirit as defined by the following claims.

We claim:

1. An injector head for a rocket engine having in combination: a face plate, said face plate having a hole therein, a second plate means spaced rearwardly from said face plate forming a first propellant manifold with said face plate, said second plate having a first projection which extends into said hole, said first projection having a first opening therethrough, a third plate means spaced rearwardly from said second plate means forming a secondary second propellant manifold with said second plate means, said third plate means having a second projection extending therefrom into said first opening passing through said first projection, said second projection

having a second opening passing therethrough, a fourth plate means spaced rearwardly from said third plate means forming a primary second propellant manifold with said third plate means, said first opening in said first projection having a conical section tapering inwardly from the secondary manifold to a cylindrical section at its free end, said second projection having a conical outer surface which tapers inwardly to a short cylindrical surface adjacent the end of the projection, and said mating conical section of said first projection and conical surface of said second projection forming a conical inlet such that the major pressure drop is downstream of the outlet end of the conical inlet.

2. An injector head as set forth in claim 1 wherein said first opening in said first projection has a cylindrical section at its free end, and said free end of said second projection extending into said first opening at least as far as the beginning of the cylindrical section.

3. An injector head as set forth in claim 1 wherein said first opening in said first projection has a cylindrical section at its free end, and said free end of said second projection being spaced from the free end of said cylindrical section a minimum distance of 0.5 times the diameter of the cylindrical section.

4. An injector head as set forth in claim 1 wherein said first opening in said first projection has a cylindrical section at its free end, and said free end of said second projection extending into said first opening to a point between the beginning of the cylindrical section and a point located a distance of 0.5 times the diameter of the cylindrical section from the free end of the cylindrical section.

5. An injector head as set forth in claim 2 wherein the diameter of the second opening lies in the range of from 0.1 to 0.5 times the diameter of the cylindrical section.

6. An injector head as set forth in claim 1 wherein said conical section of said first opening forms an angle in the range of from 30° to 40° and the conical surface of said second projection forms an angle in the range of from 25° to 30° .

7. An injector head as set forth in claim 1 wherein said first projection has an outer end which forms an annular opening with said face plate, said face plate being permeable.

8. An injector head as set forth in claim 7 wherein said face plate has a permeability of approximately 40 scfm.

9. An injector head as set forth in claim 1 wherein said first opening in said first projection has a cylindrical section at its free end, and said free end of said second projection extending into said opening approximately 0.015 in. and being spaced from the free end of the cylindrical section a distance of between 0.15 and 0.20 in.

10. An injector head as set forth in claim 1 wherein the first opening must be concentric with the second opening within 0.1 times the diameter of the first opening.

11. An injector head as set forth in claim 1 wherein the diameter of said first opening is approximately 0.072 in., the diameter of the second opening is approximately 0.010 in., and the diameter of said short cylindrical surface is approximately 0.036 in.

12. An injector head as set forth in claim 12 wherein a swirler unit is located within said second opening adjacent the end of the second projection.

13. An injector head as set forth in claim 1 wherein said swirler unit is arranged to direct a liquid therefrom having an increasing cone of flow which does not impinge upon the free end of the first opening.

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UNITED STATES PATENT OFFICE
CERTIFICATE OF CORRECTION

Patent No. 3,662,960 Dated May 16, 1972

Inventor(s) James P. Mitchell and Bruce T. Brown

It is certified that error appears in the above-identified patent and that said Letters Patent are hereby corrected as shown below:

Claim 12, Column 4, line 57, change "12" to --1--

Claim 13, Column 4, line 60, change "1" to --12--

Signed and sealed this 19th day of September 1972.

(SEAL)
Attest:

EDWARD M. FLETCHER, JR.
Attesting Officer

ROBERT GOTTSCHALK
Commissioner of Patents