



INTERNATIONAL APPLICATION PUBLISHED UNDER THE PATENT COOPERATION TREATY (PCT)

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<p>(21) International Application Number: PCT/US93/01126 (22) International Filing Date: 9 February 1993 (09.02.93) (30) Priority data: 856,799 24 March 1992 (24.03.92) US (71) Applicant: BENDIX-ATLANTIC INFLATOR COMPANY [US/US]; c/o Allied-Signal Inc., 101 Columbia Road, Morristown, NJ 07962-2245 (US). (72) Inventors: BISHOP, Robert, J. ; 18329 Manorwood West, Mt. Clemens, MI 48044 (US). KREMER, Robert, M. ; 17490 Sewel, Fraser, MI 48026 (US). BROWN, Roy, G. ; 2105 Marion Anderson Road, Hot Springs, AR 71913 (US). RENFROE, Donald, W. ; P.O. Box 140, Haymarket, VA 22069 (US). FRANTOM, Richard, L. ; 67799 South Forest, Richmond, MI 48062 (US). OCKER, Klaus, F. ; 16273 Pine Ridge North, Fraser, MI 48026 (US). BAZEL, Teresa, L. ; 4828 Springbrook Drive, Annadale, VA 22003 (US).</p>		<p>(74) Agent: BLEEKER, Robert, A.; Allied-Signal Inc., Law Dept. (C.A. McNally), 101 Columbia Road, P.O. Box 2245, Morristown, NJ 07962-2245 (US). (81) Designated States: CA, JP, KR, European patent (AT, BE, CH, DE, DK, ES, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE). Published <i>With international search report.</i></p>
<p>(54) Title: MEANS FOR RENDERING A THRUST NEUTRAL INFLATOR SUITABLE FOR USE IN AIR BAG MODULES</p>		
<p>(57) Abstract</p> <p>An air bag module (50) comprising an inflator (40) for generating inflation gas; an air bag (12) disposed about the inflator to receive the inflation gas exiting the inflator; a structure (52) for supporting at least the inflator. The inflator comprises: a pressure vessel (42) and a plurality of exit ports (16). The exit ports are arranged to render the inflator in a thrust neutral condition when the inflator is not attached to the structure. A plug (60) or plugs is provided for closing a designated set of exit ports to render the inflator in a non-thrust neutral condition when supported by the structure. Each plug is received in each of the designated exit ports and is preferably received through a portion of the structure such that if the inflator is removed from the structure each plug must also be removed, thereby opening all of previously closed exit ports and rendering the inflator in its thrust neutral condition once again.</p>		

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**MEANS FOR RENDERING A THRUST NEUTRAL
INFLATOR SUITABLE FOR USE IN AIR BAG MODULES**

BACKGROUND AND SUMMARY OF THE INVENTION

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The present invention relates generally to supplemental inflatable restraint (air bag) systems and more particularly to a hybrid inflator.

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A hybrid inflator comprises a pressure vessel into which is stored a quantity of inert, pressurized gas. Means are provided to selectively open a passage leading from the pressure vessel to exit ports to permit the egress of the inflation gas. Such means typically includes a mechanism for braking a rupturable disk. Hybrid inflators may also include heat generators, such as a quantity of propellant, to heat the gas remaining in the pressure vessel thereby achieving a more efficient inflation of the air bag.

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It is a requirement that the orientation of the exit ports be arranged to maintain the hybrid inflator in a thrust neutral condition when gases exit the inflator and specifically when the inflator is not part of an air bag module. By thrust neutral it is meant that when gas exits the exit ports the resultant of the reaction forces is essentially zero. As can be appreciated, if the hybrid inflator were not thrust neutral and is accidentally energized or gas is discharged, such as from an over pressurization due to fire, etc., the inflator could be propelled about due to the non-zero resultant forces generated by the exiting gas flow.

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As such, the above thrust neutral requirement is beneficial when the hybrid inflator is being carried, shipped, etc. However, the above requirement, as will

be seen below, reduces the efficiency of the air bag inflation process. Accordingly, the invention comprises: an air bag module comprising: an inflator for generating inflation gas; an air bag disposed
5 about the inflator to receive the inflation gas exiting the inflator; a structure for supporting at least the inflator. The inflator comprising: a pressure vessel and a plurality of exit ports through which inflation gas exits the inflator. The exit
10 ports are arranged to render the inflator in a thrust neutral condition when the inflator is not attached to the structure. The module comprising first means for closing a designated set of exit ports to render the inflator in a non-thrust neutral condition when
15 supported by the structure. The first means includes a plug received in each of the designated exit ports. Each plug is preferably received through a portion of the structure such that if the inflator is removed from the structure each plug must also be removed,
20 thereby opening all of the previously closed exit ports and rendering the inflation in its thrust neutral condition once again.

Many other objects and purposes of the invention
25 will be clear from the following detailed description of the drawings.

BRIEF DESCRIPTION OF THE DRAWINGS

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In the drawings:

FIGURE 1 is illustrative of a prior art hybrid inflator.

35 FIGURES 2 through 5 illustrate exit port orientations of a hybrid inflator.

FIGURE 6 illustrate an inflator in an air bag module.

FIGURE 7 shows a cross-sectional view of a hybrid inflator and air bag module.

5 FIGURE 8 diagrammatically illustrates various elements of a hybrid inflator.

FIGURE 9 shows an alternative embodiment of the invention.

10 DETAILED DESCRIPTION OF THE DRAWINGS

FIGURE 1 illustrates a prior art hybrid inflator 10 for an air bag 12. The inflator 10 is secured within a manifold 14 or similar support structure. As can be seen, the inflator includes a plurality of exit ports 16, situated about one end thereof. What is not shown is the mechanism within the inflator which releases of the stored inflation gas. Regardless of how gas is released, it can be appreciated that gas exiting the equally spaced exit ports 16 will yield a thrust neutral condition. The concept of a thrust neutral inflator is beneficial, but when the inflator is installed in a module, the thrust neutral feature is a source of inefficiency. To achieve rapid inflation of an air bag, it is desirable that inflation gases exit the inflator and enter directly into the air bag. Since gas flow out from a hybrid inflator is multi-directional hybrid inflators typically include some mechanism to redirect certain parts of the gas flow into the air bag. This redirection is accomplished by the above mentioned manifold, or housing, etc. which may take a variety of configurations.

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Reference is made to FIGURE 2 which illustrates an end cross-sectional view of a typical hybrid inflator 10 having only two exit ports 16a and 16b. FIGURES 3 through 5 illustrate the utilization of the hybrid inflator having other discrete exist port configurations. The arrows 18 emanating from each of the ports illustrate the direction of the inflation gas stream as it exits the inflator. FIGURE 3 illustrates three equally spaced and sized exit ports 16a-16c. FIGURE 4 shows three asymmetrically located exit ports. The flow area of port 16c is greater than the flow area of either ports 16a or 16b to yield thrust neutrality. Typically the area of ports 16a and 16b are equal. The size of port 16c will also vary with the angular spacing between ports 16a and 16b. FIGURE 5 illustrates the use of four (4) equally spaced and sized exit ports 16a-16d.

Reference is now made to FIGURE 6 which illustrates an inflator secured to a housing 20 which is often referred to as a reaction can, housing, manifold or member in the art. The purpose of the housing 20 is to support the inflator as well as to intercept the inflation gas stream flow and redirect same into the airbag. By way of example, the inflator shown is therein uses two exit ports. As can be appreciated, one-half of the inflation gas will exit one of the ports such as 16a and directly impinge upon the housing 20 while the other one-half of the inflation gas exits the opposite port 16b and flows generally directly into the air bag. With regard to the gases exiting port 16a, it can be appreciated that such gases represent a heated, pressurized, high velocity stream. As the gases impinge upon the housing 20, a certain amount of heat energy is lost,

that is, absorbed by the housing which is typically at a lower temperature than the inflation gases. Since a relatively few exit ports are used, the high temperature inflation gases will impinge on a localized area of the housing causing perhaps, excessive heat, localized over pressurization, the need to reinforce the structure, and potential degradation of air bag material that may be located within the impingement area. As the reaction housing must be designed to redirect the stream of inflation gas it becomes more a complicated and costly component than may otherwise be needed resulting in increased weight, cost, and size of an air bag module.

Reference is now made to FIGURE 7 which shows a hybrid inflator 40 forming part of an air bag module 50. The cross-sectional view of the hybrid inflator is diagrammatically shown in FIGURE 8. Exemplary hybrid inflators are shown in United States Patents 5,022,674 and 5,076,607, which are incorporated herein by reference. The hybrid inflator 40 comprises a pressure vessel portion 42 into which is stored pressurized Argon gas. The pressure vessel portion includes a rupturable disk 44 that may be opened by any known means such as that illustrated in the above patent. The inflator 40 further includes a heat generating member 46, such as a gas generator also shown in the above patents. Upon rupturing of the disc 44, the gases enter into a non-pressure vessel portion 48 and then flow out of the plurality of exit ports 16. While two such exit ports 16a and 16b are illustrated, it should be appreciated that any number of port orientations can be used. In the case of two exit ports, the size of each exit port is chosen to be identical so as to maintain the thrust neutral feature

of the hybrid inflator when it is out of the module. The size of each exit port 16a and b is also chosen such that the area of one port is appropriate to fill the airbag at its prescribed fill rate and pressure. This feature is because the port facing away from the air bag will be closed. In the invention's most simple form, one port such as 16a, is internally threaded to accept a plug such as 60.

As is typical in the art, the hybrid inflator 40 may be secured to a housing 52 in the manner as described in the above patents. With the inflator 40 so positioned within the housing, the plug 60 is inserted through an opening 62 therein into the threaded exit port such as 16a. If needed a seal 64 such as an o-ring can be used to further seal the interface between the plug and the exit port. The housing, plug and inflator cooperate in a manner so that if the inflator is removed from the housing the plug must also be removed. As can be seen, with the plug 60 removed the inflator is once again rendered thrust neutral. If a port orientation such as that shown in FIGURE 5 is used two such plugs 60a and 60b are used to prevent reverse gas flow through the rearward facing ports 16c and 16d. FIGURE 9 illustrates as alternate embodiment of the invention in which the threads of the plug are covered with a thread sealant material 61 thereby replacing the o-ring. The inflator 40 of FIGURE 8 diagrammatically shows a gas generator housing 54. Such gas generator housing is shown in greater detail in United States Patent 5,076,607. The gas generator housing is located in the non-pressure vessel portion 48 of the inflator. As shown in FIGURE 9 plug 60 is inserted within the threaded port 16a and bottoms out on the

gas generator housing. The reaction force between the plug and the gas generator housing provides for a secure fit therebetween.

5 Accordingly, it can be seen that the present invention satisfies the requirement that the hybrid inflator be thrust neutral when it is not within an air bag module; and additionally, when installed, as part of an airbag module, inflation gases are directed
10 to flow only toward the airbag without requiring the physical redirection or deflection. As such, all of the inflation energy of the hybrid inflator is caused to flow outwardly through one or more exit port(s) to permit the direct flow of inflation gas into the air
15 bag.

 As can be seen from the above, the present invention permits the utilization of a pressure vessel/hybrid inflator as a thrust neutral gas
20 container and also when such inflator is installed in an air bag module provides for the improved inflation of the air bag. It should be appreciated that since inflation gases do not exit rearwardly from the inflator, heat energy is not lost due to the heating
25 up of closely spaced components such as the housing 52. The kinetic energy of the gas is maintained at a high level as the gas stream need not be redirected, the support structures can be reduced in weight, cost, etc. since they no longer have to be designed to
30 withstand the localized heating and pressurization resulting from the impinging inflation gases.

 Many changes and modifications in the above described embodiment of the invention can, of course,
35 be carried out without departing from the scope

thereof. Accordingly, the scope is intended to be limited only by the scope of the appended claims.

IN THE CLAIMS

1. An air bag module (50) comprising:
a inflator (40); an air bag (12) cooperatively disposed about the inflator to receive inflation gas exiting the inflator; a structure (52) for supporting at least the inflator; the inflator comprising:
a pressure vessel (42) and a plurality of exit ports (16) through which inflation gas exits the inflator, the exit ports arranged to render the inflator in a thrust neutral condition when the inflator is not attached to the structure, the module comprising first means (60) for closing a designated set of exit ports to render the inflator in a non-thrust neutral condition when supported by the structure.
2. The module as defined in Claim 1 wherein the exit ports are arranged about a portion of the inflator such that some of the inflation gas would normally exit the designated set of exit ports in a first direction generally opposite the direction of inflation of the air bag, and wherein the first means comprises means to prevent the inflation gas from flowing in such first direction.
3. The module as defined in Claim 2 wherein the first means comprises means to close all of the designated exit ports, including those exit ports which permit inflation gas to exit the inflator in the first direction.
4. The module as defined in Claim 3 wherein the first means includes a plug (60) received in each of the designated exit ports.

5. The module as defined in Claim 4 wherein each plug (60) is received through a portion of the structure such that if the inflator is removed from the structure each plug must also be removed, thereby opening all of previously closed exit ports and rendering the inflation in its thrust neutral condition once again.

6. The module as defined in Claims 4 wherein the inflator (40) includes a walled member (52) opposite the exit ports (16) and wherein an end of the plug 6 engages the walled member (52).

7. A hybrid inflator (40) for inflating an air bag, the inflator comprising a plurality of exit ports arranged thereon such that inflation gas may exit all such exit ports rendering the inflator thrust neutral and when the inflator is mounted proximate the air bag and in a support structure, certain exit ports are closed or sealed, thereby rendering the inflator in a non-thrust neutral condition.

8. The inflator (40) as defined in Claim 7, wherein the closure of the certain exit ports is achieved by inserting a plug therein during assembly of the inflator to the support structure.

9. The inflator, as defined in Claim 8, wherein when it is desired to separate the inflator from the cooperating support structure the plug must first be removed to permit removal of the inflator thereby rendering the inflator thrust neutral once again.

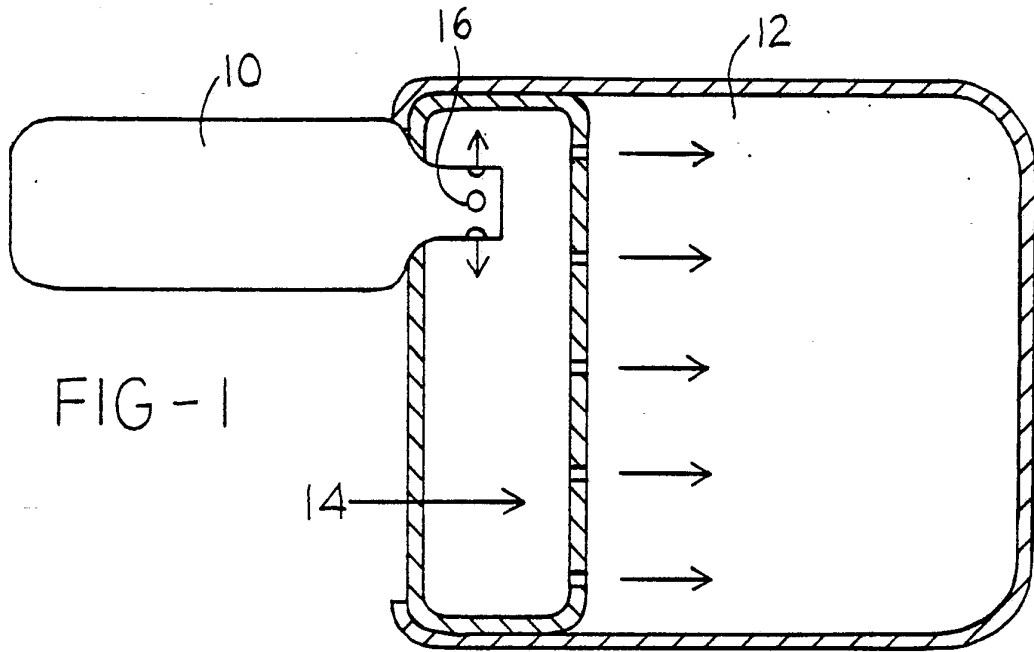


FIG - 1

PRIOR ART

FIG - 2

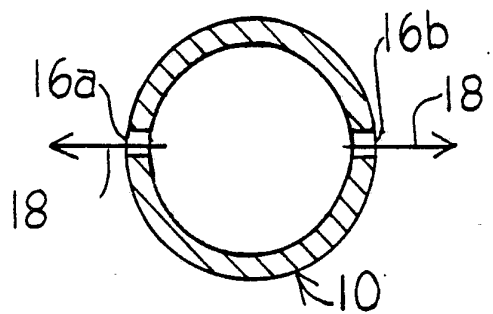


FIG - 4

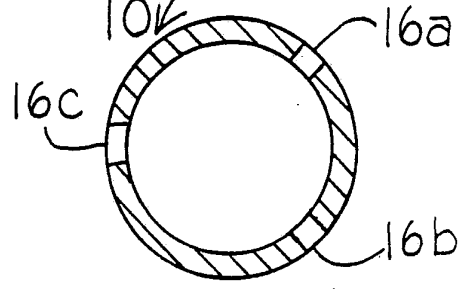


FIG - 3

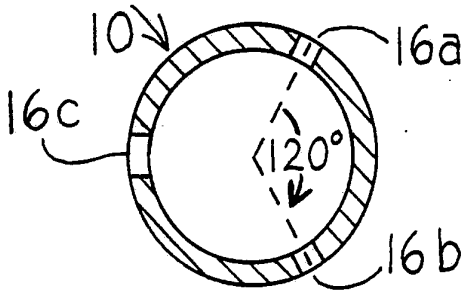
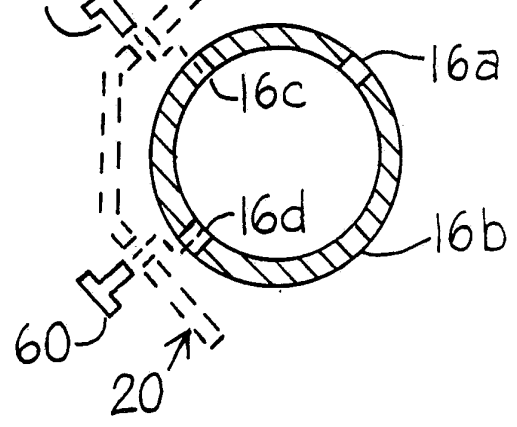
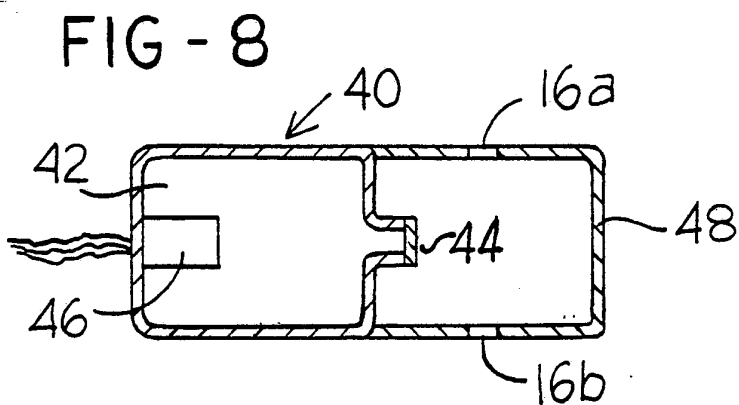
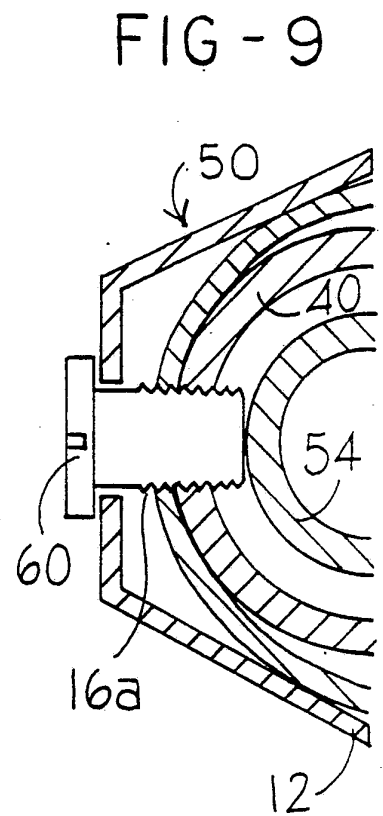
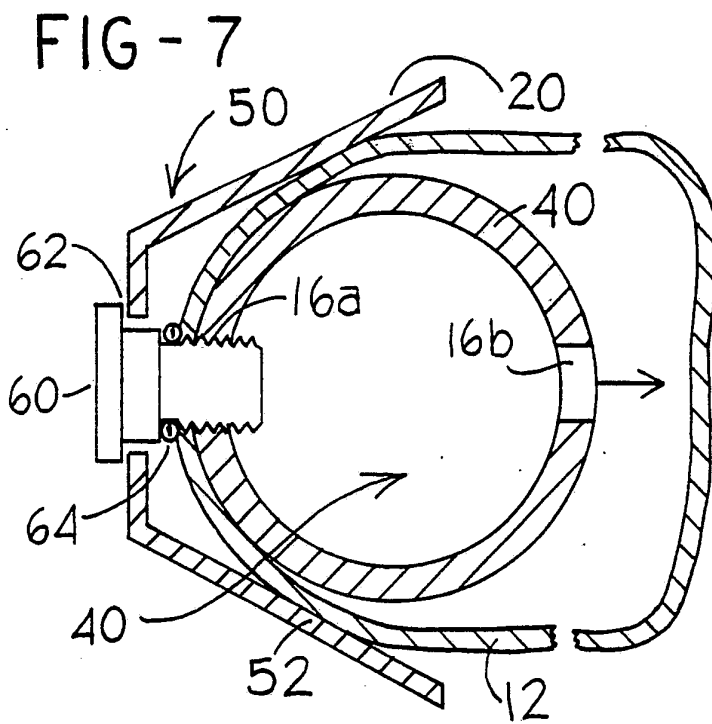
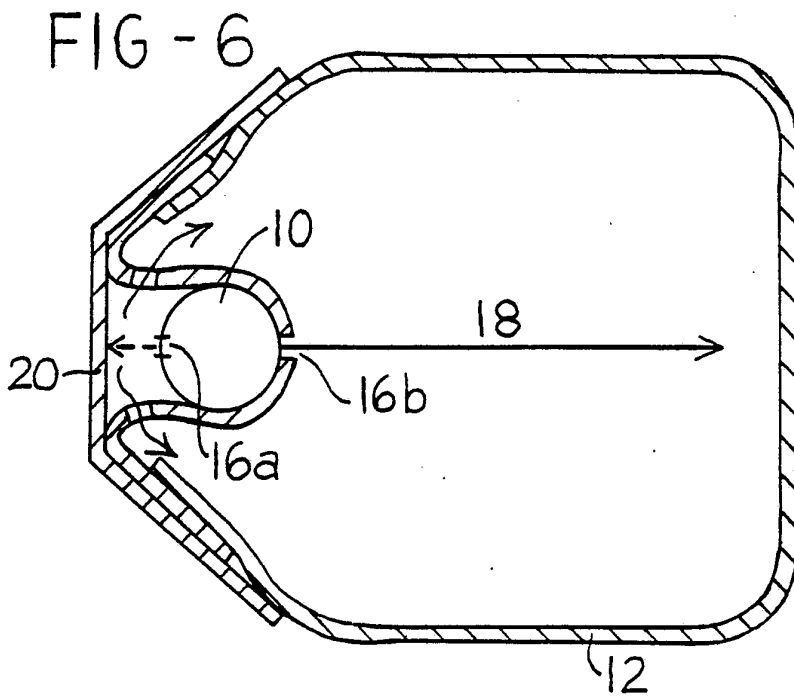


FIG - 5





INTERNATIONAL SEARCH REPORT

International Application No

PCT/US 93/01126

I. CLASSIFICATION OF SUBJECT MATTER (if several classification symbols apply, indicate all) ⁶		
According to International Patent Classification (IPC) or to both National Classification and IPC		
Int.Cl. 5 B6R21/26; F42B39/20		
II. FIELDS SEARCHED		
Minimum Documentation Searched ⁷		
Classification System	Classification Symbols	
Int.Cl. 5	B6R ; F42B ; F02K	
Documentation Searched other than Minimum Documentation to the Extent that such Documents are Included in the Fields Searched ⁸		
III. DOCUMENTS CONSIDERED TO BE RELEVANT⁹		
Category ¹⁰	Citation of Document, ¹¹ with indication, where appropriate, of the relevant passages ¹²	Relevant to Claim No. ¹³
P,A	FR,A,2 669 875 (TAKATA CORPORATION) 5 June 1992 see figures 1-6 see abstract see page 1, line 1 - line 11 see page 4, line 32 - page 11, line 34 ---	1-3,7
A	WO,A,9 013 458 (ALLIED-SIGNAL INC.) 15 November 1990 see figures 3-23 see abstract see page 7, line 1 - page 21, line 32 ---	1,7
A	DE,A,3 733 436 (TRW AUTOMOTIVE PRODUCTS) 21 April 1988 see figures 1-5,11,16-20 see column 6, line 57 - column 9, line 12 see column 17, line 8 - column 19, line 17 ---	1,7
-/--		
<div style="display: flex; justify-content: space-between;"> <div style="width: 45%;"> <p>¹⁰ Special categories of cited documents :</p> <p>"A" document defining the general state of the art which is not considered to be of particular relevance</p> <p>"E" earlier document but published on or after the international filing date</p> <p>"L" document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another citation or other special reason (as specified)</p> <p>"O" document referring to an oral disclosure, use, exhibition or other means</p> <p>"P" document published prior to the international filing date but later than the priority date claimed</p> </div> <div style="width: 45%;"> <p>"T" later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle or theory underlying the invention</p> <p>"X" document of particular relevance; the claimed invention cannot be considered novel or cannot be considered to involve an inventive step</p> <p>"Y" document of particular relevance; the claimed invention cannot be considered to involve an inventive step when the document is combined with one or more other such documents, such combination being obvious to a person skilled in the art.</p> <p>"&" document member of the same patent family</p> </div> </div>		
IV. CERTIFICATION		
Date of the Actual Completion of the International Search	Date of Mailing of this International Search Report	
07 JUNE 1993	16.06.93	
International Searching Authority	Signature of Authorized Officer	
EUROPEAN PATENT OFFICE		

III. DOCUMENTS CONSIDERED TO BE RELEVANT (CONTINUED FROM THE SECOND SHEET)		
Category °	Citation of Document, with indication, where appropriate, of the relevant passages	Relevant to Claim No.
A	US,A,3 441 290 (BERRYMAN, OLBERG, CARREY) 29 April 1969 see figures 1-4 see abstract see column 2, line 22 - column 4, line 24 ---	1,7
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**ANNEX TO THE INTERNATIONAL SEARCH REPORT
ON INTERNATIONAL PATENT APPLICATION NO.**

US 9301126
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This annex lists the patent family members relating to the patent documents cited in the above-mentioned international search report. The members are as contained in the European Patent Office EDP file on
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Patent document cited in search report	Publication date	Patent family member(s)	Publication date
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