

(12) United States Patent

Palmer et al.

(54) SYSTEM AND METHOD OF WRAPPING FLOW IN A FLUID WORKING APPARATUS

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This patent is subject to a terminal dis-

claimer.

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CPC F04D 5/002 (2013.01); F04D 1/06 (2013.01); F04D 17/12 (2013.01); F04D

23/008 (2013.01); H02K 7/1823 (2013.01)

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(Continued)

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(56)References Cited

U.S. PATENT DOCUMENTS

945,742 A 1/1910 Boeckel et al. 3,007,306 A 11/1961 Martin et al. (Continued)

FOREIGN PATENT DOCUMENTS

3327838 A1 DE DE 198 04 845 A1 8/1999 (Continued)

OTHER PUBLICATIONS

International Preliminary Report on Patenability mailed Apr. 3, 2014, in application Serial No. PCT/US2012/056524 in the name of Harris Corporation.

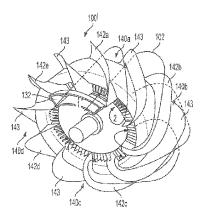
(Continued)

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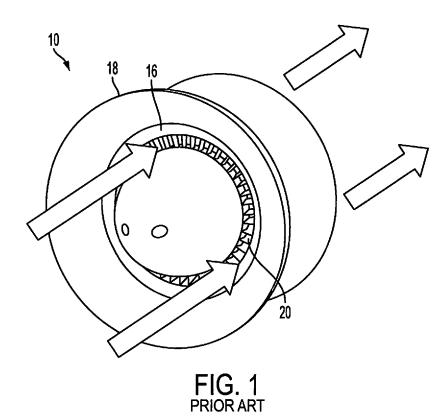
ABSTRACT

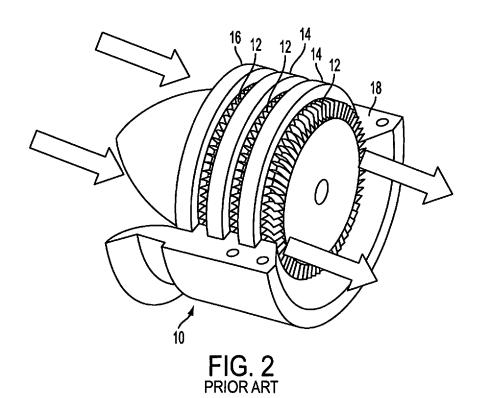
A fluid working apparatus (100) including a housing structure (130) with an inlet (132) and an outlet (133). A working assembly positioned in the housing (130) has an inlet side and an outlet side with the at least one rotor (114) having a plurality of blades (115) positioned between the inlet and outlet sides. At least one return assembly (140, 142) is configured to return fluid flow from the outlet to the inlet side of the working assembly whereby a working fluid passes through the housing inlet (132), from the inlet side of the working assembly to the outlet side thereof while workingly engaging a first subset of the rotor blades (115), through the at least one return assembly (140, 142), from the inlet side of the working assembly to the outlet side thereof while workingly engaging a second subset of the rotor blades (115), and out of the housing outlet (133). A method of working a fluid is also provided.

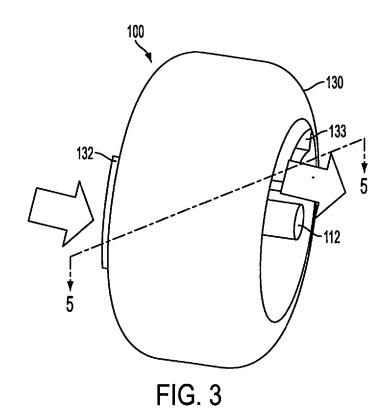
21 Claims, 34 Drawing Sheets



See application file for complete search history. See application file for for search file for for search file for for search file for for file for for file for fil	(51)	Int. Cl. F04D 1/06 F04D 17/12 F04D 23/00	(2006.01) (2006.01) (2006.01)	2005/01327 2009/02418 2010/00195	860 A1 10/200 590 A1* 1/201	5 Neary 9 Monacelli et al. 0 Guedes-Pinto H02K 1/02 310/53
(58) Field of Classification Search USPC 41571, 40, 553, 554, 571, 185; 60/39.43, 30.44, 269 See application file for complete search history. (56) References Cited U.S. PATENT DOCUMENTS 20100073580 A1 32010 A1 32010 Prodesbrge et al. 20100073580 A1 32010 Prodesbrg A1 20100073580 A1 20100 Prodesbrg et al. 20100073580 A1 102010 Prodesbrg A1 20100073580 A1 20100 Prodesbrg A1 20100073580 A1 20100 Prodesbrg A1 20100073580 A1 20100 Prodesbrg A1 20100073580 A1 2010007358		F04D 1/04	(2006.01)			
USPC . 415/1, 60, 55.3, 55.4, 57.1, 185; 60/39.43, 201000771362 A1 3/2010 Papile Search history. See application file for complete search history. (56) References Cited	(58)	Field of Classification	n Search			
See application file for complete search history. (56) References Cited U.S. PATENT DOCUMENTS 3.070.349 A 12/1962 Steward 4.070.349 A 12/1963 Steward 4.070.349 A 12/1		USPC 415/1, 60, 5	5.3, 55.4, 57.1, 185; 60/39.43,	2010/00713		
See application file for complete search history. 201109024374 At 1 2012015 1		· · ·				
Color		See application file for				
(56) References Cited U.S. PATENT DOCUMENTS 3.070.348 A 12/1902 Stewart 3.070.349 A 12/1902 Gardenier 3.070.349 A 12/1902 Gardenier 3.070.349 A 12/1902 Gardenier 3.070.349 A 12/1902 Gardenier 3.070.349 A 12/1903 Gardenier 4.070.939 A 6/1907 Mugle 4.070.939 A 6/1907 Mugle 4.070.939 A 6/1907 Wolff 4.070.939 A 12/2004 4.190.549 A 4/1939 Abbom 4.190.549 A 4/1939 Abbom 4.252.549 A 2/1981 Giles 4.070.190.190.190.190.190.190.190.190.190.19		see approamen into re	a complete senion metery.			
U.S. PATENT DOCUMENTS 2012/0139470 A1 6/2012 Indir et al. 2012/013959 A1 6/2012 Indir et al. 2003/013950 A1 2/2003 3075.73 A1 2/2003 4027.993 A 6/1977 Wolff W0 2004/033859 A1 4/2004 4027.993 A 6/1978 Indir et al. 4027.993 A1 6/2012 Indir et al. 402003/03859 A1 4/2004 402004/03859 A1 4/2004 40200	(56)	Referen	ices Cited			
U.S. PATENT DOCUMENTS 3,070,348 A 12/1962 Stewart 3,607,048 A 12/1962 Stewart 3,607,048 A 12/1962 Stewart 3,607,049 A 12/1972 Minto 3,704,570 A 12/1972 Minto 3,704,570 A 12/1972 Bov 3,704,570 A 12/1972 Bov 3,704,570 A 12/1972 Bov 4,009,587 A 3/1977 Robinson, Jr. et al. 4,009,587 A 3/1977 Robinson, Jr. et al. 4,009,587 A 3/1977 Wolff Wolff WO 2006/02844 A 1, 3/2006 4,102,508 A 19/198 Isahiki et al. 4,102,639 A 4/1980 Abom WO 2006/02844 A 1, 3/2006 4,102,508 A 2/1983 Ribis et al. 4,202,543 A 2/1983 Ribis et al. 4,203,543 A 3/1981 Ribis WO 2012/076902 Al. (6/2012 4,203,543 A 3/1981 Ribis et al. 4,203,634 A 1/1993 Bush et al. 4,304,675 A 1/1983 Newton et al. 4,372,124 A 2/1983 Newton et al. 4,372,124 A 1/1993 Scaring et al. 4,469,207 A 2/1985 Ameron 4,408,408 A 1/1984 Goldsberry 4,503,531 A 8/1999 Souring et al. 4,502,603 A 1/1993 Burg FOZR 948 4,502,603 A 1/1993 Burg FOZR 948 5,269,148 A 1/1994 Burg FOZR 948 5,269,148 A 1/1995 Bove Bekedam 5,269,148 A 1/1995 Scaring et al. 6,203,603 B 1 2/1993 Scaring et al. 5,269,148 A 1/1995 Scaring et al. 5,269,148 A 1/1995 Scaring et al. 6,203,603 B 1 2/1993 S	(50)	11010101				
3,070,348 A 12/1962 Stewart		U.S. PATENT	DOCUMENTS			
3,070,349 A 1,121972 Matto 3,704,570 A 1,121972 Gardenier 3,704,570 A 1,21972 Gardenier 3,704,570 A 1,21972 Gardenier 3,704,570 A 1,21973 Gardenier 3,834,133 A 9,1974 Bow P P 2000,282810 A 10,2000 3,935,710 A 2,1976 Dickinson P 2000,168201 A 4,2001 4,002,387 A 8,1978 Robinson, Jr. et al. WO 0,207,3007 A 9,2002 4,106,294 A 8,1978 Czaja WO 0,207,3007 A 9,2002 4,106,294 A 1,1908 A 1,1908 Gardenier 4,122,580 A 10,1978 Isabiki et al. WO 0,206,108389 A1 4,2004 4,126,890 A 1,1908 Gardenier 4,225,543 A 2,1981 Gites WO 2,006,10835 A1 1,1000 4,235,543 A 9,1981 Buckner 4,201,538 A 1,1998 Goldsberry 4,205,481 A 1,1998 Goldsberry 4,205,481 A 1,1998 Gardenier et al. 4,384,446 A 1,1994 Goldsberry 4,548,043 A 1,1998 Salins et al. 4,287,138 A 1,1998 Grandshore et al. 4,287,138 A 1,1998 Grandshore et al. 4,287,138 A 1,1998 Broafenbrenner et al. 5,137,681 A 8,1995 Salins et al. 4,287,287 A 1,12000 Johnston mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Patent Applin. No. PCTUS201303141506 to Harris Corporation. (1) pages). 11etrantional Search Report and Written Opinion mailed Oct. 10, 2013 for International Patent Applin. No. PCTUS20130314570 (1) 2013 for International Patent Applin. No. PCTUS20130314570 (1) 2013 for International Patent Applin. No. PCTUS201303047750 (1) 2013 for International Patent Applin. No. PCTUS20130304750 (1) 2013 for International Patent Applin. No. PCTUS20130304750 (1) 2013 for International Patent Applin. No. PCTUS20130314570 (1) 2013 for International Patent Applin. No. PCTUS20130314570 (1) 2013 for International Patent Applin. No. PCTUS20130314570 (1) 2013 for International Patent Applin. No. PCTUS2013031587 (1) 2013 for International Patent Applin. No. PCTUS2013031587 (1) 2013 f						
3,636,706 A 11972 Minto 3,704,870 A 121972 Gardenier 3,834,133 A 91974 Bow PP 2000;282810 A 10,2000 3,935,710 A 21976 Dickinson PP 2001;108201 A 42001 3,935,710 A 21976 Mugele KR 101045802 B1 7,2011 4,009,387 A 31977 Robinson, Jr. et al. WO 20,703,007 A2 9,2002 4,027,939 A 31977 Robinson, Jr. et al. WO 20,003,007 A2 9,2002 4,027,939 A 81978 Caja WO 20,040,33859 A1 4,2004 4,106,239 A 81978 Caja WO 20,040,33859 A1 4,2004 4,106,239 A 81978 Caja WO 20,061,05815 A1 10,2006 4,106,239 A 81978 Caja WO 20,061,05815 A1 10,2006 4,252,547 A 21981 Giles WO 20,121,070,007 A1 2,2006 4,252,547 A 21981 Giles WO 20,061,05815 A1 10,2006 4,253,531 A 31981 Rizi WO 20,121,070,007 A1 2,2006 4,253,531 A 31981 Rizi WO 20,121,070,007 A1 2,2006 4,253,531 A 31981 Rizi WO 20,121,070,007 A1 4,200,06 4,253,531 A 31981 Rizi WO 20,121,070,007 A1 10,2006 4,253,531 A 31981 Rizi WO 20,121,070,007 A1 4,2004 4,253,531 A 31981 Rizi WO 20,101,070,007 A1 4,2004 4,253,531 A 31981 Rizi WO 20,061,05815 A1 10,2006 4,253,531 A 31981 Rizi WO 20,101,070,007 A1 4,2004 4,253,531 A 31981 Rizi WO 20,061,05815 A1 10,2000 4,253,531 A 31981 Rizi WO 20,101,070,007 A1 4,2004 4,253,531 A 31981 Rizi WO 20,101,070,007 A1 2,2002 4,253,531 A 31981 Rizi WO 20,101,070,007 A1 2,200,007 4,007,007,007 A1 2,200,007 4,007,007 A1 2,200						
3,704,570 A 12/1972 Gardenier J. 22/2003 3,935,710 A 2/1976 Dickinson J. P 2000,1828210 A 10/2000 3,933,865 A 8/1976 Mugele R 10/104820 B1 7/2011 4,009,587 A 3/1977 Robinson, Jt. et al. WO 0,007,007 A2 9/2002 4,007,093 A 6/1977 Wolff WO 0,007,007 A2 9/2002 4,007,093 A 6/1977 Wolff WO 0,007,007 A2 9/2002 4,007,093 A 6/1977 Wolff WO 0,007,007 A2 9/2004 4,102,280 A 10/1978 Ishiki et al. WO 0,006/105815 A1 10/2006 4,196,599 A 4/1980 Abom WO 2012/076902 A1 6/2012 4,232,345 A 2/1981 Buckner 4,233,211 A 8/1981 Buckner 4,291,538 A 9/1981 Buckner 4,393,212 A 1/1998 Sederquist et al. 4,484,403 A 5/1994 Moss Sederquist et al. 4,484,403 A 1/1987 Anderson A/1987 Anderson A/290,5481 A 3/1999 Scarriage et al. 4,905,481 A 3/1999 Scarriage et al. 4,905,481 A 1/1998 Scarriage et al. 4,905,481 A 8/1995 Burgy F02K 9/48 5,444,981 A 7/1997 Burgy F02K 9/48 5,444,981 A 7/1997 Hubber Scarriage et al. 5,137,681 A 8/1995 Burgy F02K 9/48 5,444,981 A 7/1997 Burgy F02K 9/48 5,444,981 A 7/1997 Burgy F02K 9/48 5,444,981 A 7/1997 Hubber Scarriage et al. 5,135,237 A 11/1998 Buckner 5,277,393 A 3/1998 Mahmoutzudeh Scarriage et al. 5,144,941 A 7/1997 Hubber Scarriage et al. 5,144,948 B1 7/2002 Scar					FOREIGN PAT	ENT DOCUMENTS
3,834,133 A 9/1974 Bow						
3.933.710						
3.973.865 A 8.1976 Robinson, Jr. et al. WO 0.2073007 A2 92002 4.027,993 A 6.1977 Robinson, Jr. et al. WO 0.2073007 A2 92002 4.106,294 A 81978 Czaja WO 2006/033859 A1 42006 4.122,680 A 41980 Abom WO 2006/105815 A1 10/2006 4.125,541 A 21981 Giles WO 2012/151055 A2 11/2012 4.258,551 A 31981 Ritzi WO 2012/151055 A2 11/2012 4.258,551 A 31981 Buckner 4.291,538 A 91981 Buckner 4.291,538 A 91981 Buckner 4.291,538 A 91981 Buckner 4.291,538 A 91981 Husain et al. 4.372,124 A 21983 Nishioka 4 4.372,124 A 21983 Sederquist et al. 4.372,124 A 21983 Sederquist et al. 4.384,446 A 11/1984 Goldsberry 4.548,446 A 11/1984 Goldsberry 4.548,446 A 11/1984 Goldsberry 4.595,444,481 A 10/1985 Kalina 4.595,444,81 A 81995 Searinge et al. 4.905,481 A 31/990 Searinge et al. 4.905,481 A 31/990 Searinge et al. 4.905,481 A 31/990 Bonofenbrenner et al. 5.137,681 A 81/992 Dougherty 5.137,681 A 81/992 Bougherty 5.137,681 A 81/993 Boughert et al. 5.264,4911 A 7/997 Huber 5.277,7393 A 31/998 Mahmoudzadeh 5.564,4911 A 7/997 Huber 5.777,739 A 31/998 Mahmoudzadeh 6.141,384 B1 7/2002 Koch 6.141,385 A 11/200 Donovan et al. 6.141,385 A 11/200 Donovan et al. 6.141,385 A 11/200 Donovan et al. 6.141,385 B1 12/001 Kimble 6.413,488 B1 7/2002 Koch 6.414,385 B1 1/2004 MacAdam et al. 8.262,339 B2 92012 Chien 8.61,635 B1 5/2014 MacAdam et al. 8.766,635 B1 5/2014 MacAdam et						
4,093,887 A 61977 Wolff WO 2004/033859 A1 4/2004 4,027,938 A 61977 Wolff WO 2006/028444 A1 3/2004 4,102,2680 A 101978 Isshiki et al. WO 2006/02844 A1 3/2006 4,106,594 A 21980 Giles WO 2012/076902 A1 6/2012 4,252,543 A 21981 Ritzi WO 2012/076902 A1 6/2012 4,252,543 A 21981 Ritzi WO 2012/076902 A1 6/2012 4,252,543 A 21981 Buckner 4,257,138 A 91981 Buckner 4,291,538 A 91981 Buckner 4,291,538 A 91981 Buckner 4,291,538 A 91981 Buckner 4,291,538 A 91981 Buckner 4,372,739 A 21983 Setting tet al. 4,372,739 A 21983 Setting tet al. 4,372,739 A 11983 Gildsberry 4,348,446 A 11/1984 Goldsberry 4,368,030 A 5/1985 Goldsberry 4,368,031 A 4/1987 Anderson 4,568,031 A 4/1987 Anderson 4,568,031 A 8/1990 Goldsberry 5,137,681 A 11/1984 Goldsberry 5,137,681 A 8/1990 Goldsberry 5,137,681 A 8/1990 Goldsberry 5,137,681 A 11/1994 Goldsberry 5,137,681 A 11/1994 Goldsberry 5,137,681 A 11/1996 Goldsberry 5,137,681 A 11/1996 Goldsberry 5,137,681 A 11/1996 Goldsberry 5,137,683 A 8/1990 Goldsberry 5,137,681 A 11/1996 Goldsberry 6,137,600,061 B 10/2000 Goldsberry 6,137,600,061 B 10/2000 Akiyama et al. 6,131,438 A 9/1995 Goldsberry 6,137,600,061 B 10/2000 Akiyama et al. 6,131,438 A 11/1996 Bock Goldsberry 6,137,600,061 B 10/2000 Akiyama et al. 6,131,438 B 17/2002 Colds Goldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7/2002 Cold Goldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7/2002 Cold Goldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7/2002 Coldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7/2002 Coldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7/2002 Goldsberry 8,176,724 B2 5/2012 Smith 8,136,148 B1 7		3,973,865 A 8/1976	Mugele			
4,027,993 A 6,1977 Wolff WO 2004,033859 AI 4,2004						
4,122,680 A 10/1978 International Search Report and Written Opinion mailed Oct. 9, 2013 for International Search Report and Written Opinion mailed Oct. 9, 2013 for International Search Report and Written Opinion mailed Oct. 9, 2013 for International Search Report and Written Opinion mailed Oct. 9, 2013 for International Search Report and Written Opinion mailed Oct. 9, 2013 for International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Search Report and Written Opinion mailed Oct. 10, 2013 for International Patent Appla. No. PCT/US2013/041506 for Harris Corporation (13 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation (13 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation. 10 pages). International Patent Appla. No. PCT/US2013/041506 for Harris Corporation.		4,027,993 A 6/1977				
4.196.594 A 4/1980 Abom WO 2012/076902 Al 6.2012						
4,252,543 A 2/1981 Ritz WO 2012/15/055 A2 11/2012						
4,258,551 A 3/1981 Ehrlich et al. 4,287,138 A 9/1981 Buckner 4,291,538 A 9/1981 Husain et al. 4,372,759 A 1/1983 Newton et al. 4,372,759 A 1/1983 Newton et al. 4,372,759 A 1/1984 Goldsberry 4,548,030 A 5/1984 Moss 4,484,446 A 11/1984 Goldsberry 4,548,030 A 5/1984 Moss 4,484,446 A 11/1985 Kalina 4,569,207 A 2/1986 James 4,569,207 A 2/1986 James 4,569,207 A 2/1986 James 4,569,207 A 1/1997 Anderson 4,905,481 A 3/1990 Searinge et al. 4,935,221 A 6/1990 Bronfenbrenner et al. 5,137,681 A 8/1991 Dougherty 5,156,534 A * 10/1992 Burgy F02K 9/48 5,444,981 A 8/1995 Kakovitch 5,269,148 A 1/1995 Bekestam 5,269,148 A 1/1995 Bekestam 5,262,035 A 1/1999 Mei et al. 5,262,035 A 1/1999 Mei et al. 5,277,381 A 1/1999 Mei et al. 5,277,381 A 1/1998 Buck 6,032,467 A 3/200 Oshita et al. 6,144,295 A 11/2000 Donovan et al. 6,144,295 A 11/2000 Donovan et al. 6,145,295 A 11/2000 Donovan et al. 6,145,295 A 11/2000 Donovan et al. 6,143,384 B1 7/202 Koch 6,490,865 B1 8/2001 Sittinger et al. 7,010,20 B2 3/206 B1 8/2000 Sittinger et al. 7,000,666 B2 1/2002 Padalah 7,800,669 B2 1/2002 Podolalah 8,831,658 B2 5/2011 Weber 8,176,724 B2 5/2012 Simith 8,276,633 B1 5/2014 Dale 9,297,387 B2 * 3/2016 Palmer F04D 17/122 9,303,514 B2 * 4/2016 Palmer F04D 17/122 9,303,514 B2 * 4/2010 Gold Gold Pote Potential Palmer P04D 17/122 9,300,514 B2 * 4/2010 Gold Gold Pote P04D 17/122 9,300,514 B2 * 4/2010 Gold Gold Pote P04D 17/122 9,300,514 B2 * 4/2010 Gold Gold Pote P04D 17/122 9,300,514 B2 * 4/2010 Gold Gold Pote P04D 17/122 9,300,514 B2 * 4/2010 Gold Pote P04D 17/122 9,300,514 B2 * 4/2010 Gold Pote P04D 17/122 9,300,514 B2 * 4/2010						
4,283,211						
4.291,538 A 9/1981 Husain et al. 4.346,675 A 1/1983 Nishioka 4.372,124 A 2/1983 Newton et al. 4.372,129 A 2/1983 Gedrquist et al. 4.448,030 A 5/1984 Moss 4.448,043 A 10/1985 Kalina 4.569,207 A 2/1986 James 4.660,511 A 4/1987 Anderson 4.905,481 A 3/1990 Searinge et al. 4.905,481 A 3/1990 Johnston 4.935,221 A 6/1990 Bronfenbrenner et al. 5.137,681 A 8/1992 Dougherty 5.156,534 A * 10/1992 Burgy F02K 9/48 5.269,148 A 12/1993 Ludwig 5.444,981 A 8/1995 Eskedam 5.527,7881 A 11/1996 Bekedam 5.527,7881 A 11/1996 Bekedam 5.522,7339 A 3/1998 Mahmoutzadeh 5.522,7339 A 3/1998 Mahmoutzadeh 5.523,728 A 11/1998 Buck 6.032,467 A 3/2000 Oshita et al. 6.145,259 A 11/2000 Lonovan et al. 6.141,955 A 11/2000 Lonovan et al. 6.143,484 B1 7/2002 Convan et al. 6.143,484 B1 7/2002 Sairo 7.096,666 B2 2004 Kalina 6.490,865 B2 12/2002 Pauly 6.7935,180 B2 5/2011 Weber 7.1935,180 B2 5/2011 Weber 8.176,792,366 B1 8/2004 Kalina 8.263,339 B2 9/2012 Chien 8.263,339 B1 9/2012					2010,010333 111	5,2015
4.366,675 A 1/1983 Newton et al. 4.372,759 A 2/1983 Newton et al. 4.372,759 A 2/1983 Newton et al. 4.448,446 A 11/1984 Goldsberry 4.548,043 A 10/1985 Kalina 4.660,511 A 4/1987 Anderson 4.905,481 A 3/1990 Scaringe et al. 4.905,481 A 3/1990 Scaringe et al. 4.912,6.643 A 5/1990 Johnston 4.926,643 A 5/1990 Johnston 4.926,643 A 8/1992 Dougherty 5.156,532 A 2/1998 Buck 5.269,148 A 12/1993 Ludwig 5.444,981 A 8/1995 Kakowitch 5.444,981 A 7/1997 Huber 5.444,981 A 7/1997 Huber 5.522,155 A 4/1997 Mei et al. 5.622,155 A 4/1997 Michael Scarch Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Search Report and Written Opinion mailed Oct. 10, 2013 for International Patent Applic. No. PCT/US2013/051857 (11 pages). International Patent Applic. No. PCT/US2013/051857 (11 pa					OTHER D	LIDI ICATIONS
4,372,124 A 2/198 Sedequisit et al.		· · · · ·			OTHER P	UBLICATIONS
4,4372,759 A 21983 Sederquist et al. 4,448,030 A 51984 Moss 4,484,446 A 11/1984 Goldsberry 4,548,043 A 10/1985 Kalina 4,660,207 A 21986 James 4,660,207 A 21986 James 4,905,481 A 31990 Scaringe et al. 4,926,643 A 51990 Johnston 4,926,643 A 51990 Bougherty 5,137,681 A 81992 Burgy FOZK 9/48 5,137,681 A 10/1992 Burgy FOZK 9/48 5,269,148 A 12/1993 Ludwig 5,444,981 A 81995 Kakovitch 5,269,148 A 12/1993 Ludwig 5,444,981 A 81995 Kakovitch 5,5476,525 A 12/1995 Bekedam 5,5474,911 A 71997 Huber 5,544,911 A 71997 Huber 5,527,7393 A 31,998 Mahmoudzadeh 5,544,911 A 71997 Huber 6,032,467 A 3/2000 Oshita et al. 6,141,955 A 11/2000 Donovan et al. 6,141,955 A 11/2000 Donovan et al. 6,141,955 A 11/2000 Donovan et al. 6,143,484 B1 7/2002 Koch 6,493,865 B2 12/2002 Pauly 6,760,256 B1 8/2004 Kalina 7,906,665 B2 8/2006 Singer et al. 7,096,665 B2 8/2006 Singer et al. 7,096,665 B2 8/2006 Singer et al. 7,096,665 B2 8/2006 Singer et al. 8,726,635 B1 5/2011 Weber 7,935,180 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,262				International	Search Report an	d Written Opinion issued on Oct 9
4,448,434 A 11/1984 Goldsberry 4,548,043 A 10/1985 Kalina 2013 in International Search Report and Written Opinion mailed Oct. 9, 2013 in International Application No. PCT/US2013/047750 (11 pages). 4,940,548 A 31/1990 Searing et al. 4,926,643 A 51/1990 Johnston 4,935,221 A 61/1990 Bronfenbrenner et al. 5,137,681 A 87/1992 Dougherty 5,156,534 A 7 10/1992 Burgy FOZK 9/48 5,269,148 A 12/1993 Ludwig Alfson 5,444,981 A 87/1995 Rakovitch 5,476,525 A 12/1995 Bekedam 5,5474,831 A 11/1996 Hablanian 5,5474,911 A 7/1997 Huber 5,622,055 A 47/1997 Huber 5,624,913 A 3/1998 Mahmoudzadeh 5,832,728 A 11/1998 Buck 6,141,955 A 11/2000 Donyan et al. 6,143,555 A 11/2000 Sohita et al. 6,145,556 Bl 8 2004 Kalina 6,470,865 B2 12/2002 Pauly 6,6769,256 B1 8,2004 Kalina 7,7010,920 B2 3/2066 6,480,865 B2 12/2002 Pauly 6,769,256 B1 8,2004 Kalina 7,7010,920 B2 3/2066 6,884,021 B2 4/2005 Saito 7,7010,724 B2 5/2011 Weber 7,735,180 B2 5/2011 Weber 8,736,635 B1 5/2014 Wacadam et al. 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,262,339 B2 9/2012 Chien 8,262,339 B2 9/2012 Chien 8,262,339 B2 9/2012 Chien 8,736,635 B1 1 5/2014 MacAdam et al. 8,776,635 B1 1 5/2014 Dale 9,297,387 B2* 3/2016 Shimizar et al. 9,297,387 B2* 3/2016 Shimizar et al. 9,204,0412,050 A1 1/2000 Shimizar et al. 9,207,387 B2* 3/2016 Shimizar et al. 9,207,387 B2* 3/2016 Shimizar et al. 9,207,387 B2* 3/2016 Shimizar et al. 9,204,0412,060 A1 1/2000 Shimizar et al. 9,204,0412,060 A1 1/2000 Shimizar et al. 9,207,387 B2* 3/2016 Shimizar et al. 9,204,0412,060 A1 1/2000 Shimizar et al.					_	
Add A A 10 1085 Kalina Add A A 10 1085 Kalina Add A A A A A A A A						
4,569,207 A 2/1986 James Anderson James Anderson James Anderson James Anderson James J						
Af-600.511 A					ernational rippinet	adol 110. 1 C 17 C 520 1 570 177 50 (11
4,905,481 A 3/1990 Scaringe et al. 4,926,643 A 5/1990 Bronfenbrenner et al. 5,137,681 A 8/1992 Burgy FO2K 9/48 5,269,148 A 8/1995 Kakovitch 5,269,148 A 8/1995 Kakovitch 5,476,525 A 12/1993 Bekedam 5,577,881 A 11/1996 Hablanian 5,622,055 A 4/1997 Mei et al. 5,622,055 A 4/1997 Mei et al. 5,622,055 A 1/1998 Burg 5,727,393 A 3/1998 Mahmoudzadeh 5,832,728 A 11/1998 Buck 6,141,955 A 11/2000 Oshita et al. 6,141,955 A 11/2000 Oshita et al. 6,141,955 A 11/2000 Akiyama et al. 6,143,484 Bl 7/2002 Koch 6,434,841 Bl 7/2002 Koch 6,438,480 Bl 8/2004 Kalina 6,400,865 B2 10/2010 Wimble 6,413,484 Bl 7/2002 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,905,665 B2 8/2006 Saranchuk et al. 7,905,665 B2 8/2006 Saranchuk et al. 7,906,665 B2 10/2010 Ishikawa et al. 8,631,638 B2 10/2010 Sikinger et al. 8					Search Report as	nd Written Opinion mailed Oct. 10,
4,935,221 A 6/1990 Bronfenbrenner et al.						
5,137,681 A * 10/1992 Burgy						
5,156,534 A * 10/1992 Burgy						
5,269,148 A 12/1993 Ludwig S,444,981 A 8/1995 Kakovitch Sekedam S,476,525 A 12/1995 Bekedam Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/453,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/453,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/453,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/453,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Appl. No. 13/459,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/459,409, filed Apr. 9, 2013. Appl. No. 13/459,400,400,400,400,400,4					ernational Patent A	Appln. No. PCT/US2013/051857 (11
5.269,148 A 12/1993 Ludwig Skakovitch 5,444,981 A 8/1995 Kakovitch 5,476,525 A 12/1995 Bekedam Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 14/138,903, filed Dec. 23, 2013. Reference to U.S. Appl. No. 14/139,094, filed Dec. 23, 2013. Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/533,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/339,674, filed Sep. 22, 2011. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/556,387, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/556,387, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/556,387, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/556,387, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/556,387, filed Jun. 26, 2012. Appl. No. 13/556,387,		0,100,00111 10/12/2			IIC Anni No 1	2/009 602 flad May 2 2011
5,474,981 A 8/1995 Kakovitch 5,476,525 A 12/1995 Bekedam Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,355, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/533,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/533,497, filed Jun. 26, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. International Preliminary Report on Patentability mailed Dec. 4, 2014 for International Preliminary Report on Patentability mailed Jan. 8, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 7,096,6649 B2 10/2001 Saito Harris Corporation. International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Prel		5,269,148 A 12/1993				
5,577,881 A 11/1996 Hablanian 5,622,055 A 4/1997 Mei et al. 5,624,911 A 7/1997 Huber 5,727,393 A 3/1998 Mahmoudzadeh 5,832,728 A 11/1998 Buck 6,032,467 A 3/2000 Oshita et al. 6,141,955 A 11/2000 Akiyama et al. 6,141,955 A 11/2000 Akiyama et al. 6,141,955 A 11/2000 Expl. No. 13/239,674, filed May 22, 2012. 6,250,105 B1 6/2011 Kimble 6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,864,021 B2 4/2005 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,006,665 B2 8/2006 Singer et al. 7,006,665 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Smith 8,262,339 B2 9/2012 Chien 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,276,635 B1 5/2014 MacAdam et al. 8,276,635 B1 5/2014 MacAdam et al. 8,276,635 B1 5/2014 MacAdam et al. 8,276,635 B1 5/2014 Dale 9,297,387 B2 * 3/2006 Shimizu et al. 2002/0162330 A1 11/2002 Shimizu et al. 2002/0162330 A1 17/2004 Gottlieb Reference to U.S. Appl. No. 13/359,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2012. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, filed Apr. 9, 2013. Reference to U.S. Appl. No. 13/259,409, fi						
5,622,055 A 4/1997 Mei et al. 5,624,911 A 7/1997 Huber Reference to U.S. Appl. No. 13/859,409, filed Apr. 9, 2013. 5,644,911 A 7/1998 Mahmoudzadeh 5,832,728 A 11/1998 Buck Reference to U.S. Appl. No. 13/333,497, filed Jun. 26, 2012. 6,032,467 A 3/2000 Oshita et al. 6,141,955 A 11/2000 Akiyama et al. 6,141,955 A 11/2000 Donovan et al. 6,145,295 A 11/2000 Donovan et al. 6,250,105 B1 6/2001 Kimble 6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,769,256 B1 8/2004 Kalina 6,884,021 B2 4/2005 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,096,665 B2 10/2009 Abdallah 6,884,021 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Shimizu et al. 8,176,724 B2 5/2011 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 1/2014 MacAdam et al. 9,297,387 B2 * 3/2016 Palmer						
S.644,911 A 7/1997 Huber Reference to U.S. Appl. No. 13/533,497, filed Jun. 26, 2012.		5,577,881 A 11/1990 5,622,055 A 4/1997				
5,727,393 A 3/1998 Mahmoudzadeh Reference to U.S. Appl. No. 13/239,674, filed Sep. 22, 2011. Reference to U.S. Appl. No. 13/239,674, filed Sep. 22, 2011. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. Reference to U.S. Appl. No. 13/477,394, filed May 22, 2012. International Preliminary Report on Patentability mailed Dec. 4, 2014 for International Preliminary Report on Patentability mailed Dec. 4, 2014 for International Preliminary Report on Patentability mailed Jan. 8, 2015 for International Preliminary Report on Patentability mailed Jan. 8, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. Information about Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications. International Search Report and Written Opinion mailed Nov. 16, 2012; Application No. PCT/US2012/034199 in the name of Harris Corporation. International Patent Appln. No. PCT/US2012/034199 in the name of Harris Corporation. Information about Related Patents and Patent Applications. International Search Report and Written Opinion mailed Nov. 16, 2012; Application No. PCT/US2012/034199 in the name of Harr				Reference to	U.S. Appl. No. 1	3/533,497, filed Jun. 26, 2012.
6,032,467 A			Mahmoudzadeh	Reference to	U.S. Appl. No. 1	3/239,674, filed Sep. 22, 2011.
6,141,955 A 6,145,295 A 11/2000 Akiyama et al. Donovan et al. Donovan et al. 6,145,295 A 17/2002 Kimble 6,250,105 B1 6/2001 Kimble 6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,769,256 B1 8/2004 Kalina 8,2004 Kalina 8,2005 Saranchuk et al. 7,010,920 B2 3/2006 Stinger et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,262,339 B2 8,262,339						
6,145,295 A 11/2000 Donovan et al. 6,250,105 B1 6/2001 Kimble 6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,769,256 B1 8/2004 Kalina 6,884,021 B2 4/2005 Saito 7,010,920 B2 3/206 Saranchuk et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F04D 17/122 9,303,514 B2* 4/2						
6,250,105 B1 6/2001 Kimble 6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,769,256 B1 8/2004 Kalina 7,010,920 B2 4/2005 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2019 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F01D 1/122 9,303,514 B2* 4/2016 Palmer F01D 1/122 9,303,514 B2* 4/2016 Palmer F01D 1/12 2002/0162330 A1 1/2002 Shimizu et al. 2004/0123609 A1 7/2004 Formula in the Application in the A						
6,413,484 B1 7/2002 Koch 6,490,865 B2 12/2002 Pauly 6,769,256 B1 8/2004 Kalina 6,884,021 B2 4/2005 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F04D 17/122 9,303,514 B2* A/2016 Pal		6,250,105 B1 6/2001	Kimble			Appiii. No. FC1/032013/0141300 to
6,490,805 B2 6,769,256 B1 8/2004 Kalina 2015 for International Patent Appln. No. PCT/US2013/047750 to 6,884,021 B2 4/2005 Sairto Harris Corporation. 7,010,920 B2 3/2006 Saranchuk et al. 2015 for International Patent Appln. No. PCT/US2013/047750 to Harris Corporation. 1,004,061 B2 10/2009 Abdallah Silikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,061 B2 10/2010 Ishikawa et al. 2015 for International Patent Appln. No. PCT/US2013/051857 to Harris Corporation. 1,004,004,004,004,004,004,004,004,004,00		6,413,484 B1 7/2002	Koch			ort on Patentability mailed Ian 8
6,884,021 B2 4/2005 Saito 7,010,920 B2 3/2006 Saranchuk et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F01D 1/12 2002/0162330 A1 1/2002 Shimizu et al. 2004/0123609 A1 7/2004 Falmar Harris Corporation. Harris Corporation. International Preliminary Report on Patentability mailed Feb. 5, 2015 for International Patent Applications and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications. Information about Related Patents and Patent Applications. International Search Report and Written Opinion mailed Nov. 16, 2012; Application No. PCT/US2012/034199 in the name of Harris Corporation. U.S. Appl. No. 13/859,106, filed Apr. 9, 2013, System and Method of Wrapping Flow in a Fluid Working Apparatus.						
7,010,920 B2 3/2006 Saranchuk et al. 7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 8,176,724 B2 5/2011 Weber Section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications. 8,262,339 B2 9/2012 Smith Letter, which concerns Related Patents and Patent Applications. 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F01D 1/12 2002/0162330 A1 1/2002 Shimizu et al. 2004/0123609 A1 7/2004 Gottlieb						11
7,096,665 B2 8/2006 Stinger et al. 7,600,961 B2 10/2009 Abdallah 7,806,649 B2 10/2010 Ishikawa et al. 7,935,180 B2 5/2011 Weber 8,176,724 B2 5/2012 Smith 8,262,339 B2 9/2012 Chien 8,631,658 B2 1/2014 MacAdam et al. 8,726,635 B1 5/2014 Dale 9,297,387 B2* 3/2016 Palmer F04D 17/122 9,303,514 B2* 4/2016 Palmer F01D 1/12 2002/0162330 A1 1/2002 Shimizu et al. 2004/0123609 A1 7/2004 Stinger et al. 2015 for International Patent Applic. No. PCT/US2013/051857 to Harris Corporation. Information about Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications, see section 6 of the accompanying Information Disclosure Statement Letter, which concerns Related Patents and Patent Applications. Information about Related Patents and Patent Applica				International	Preliminary Rep	ort on Patentability mailed Feb. 5,
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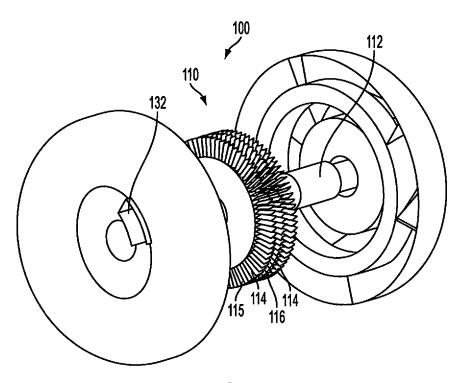
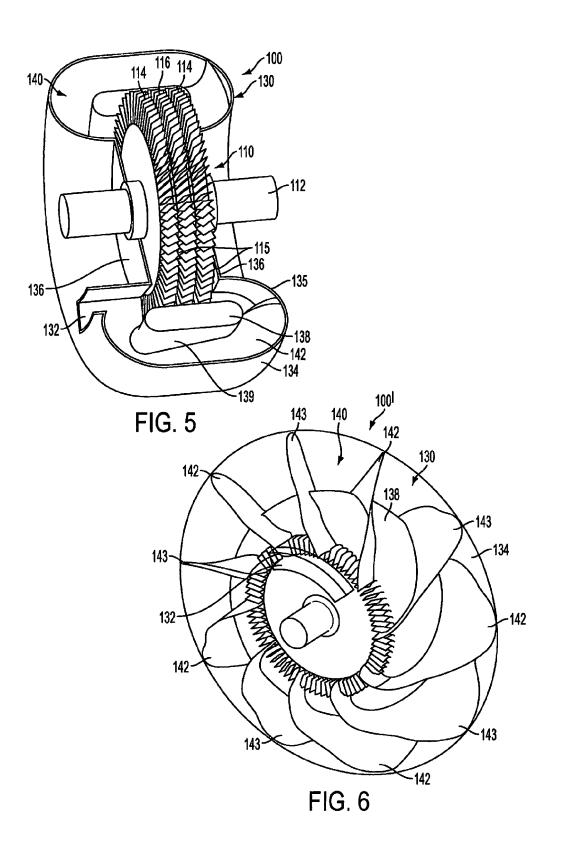
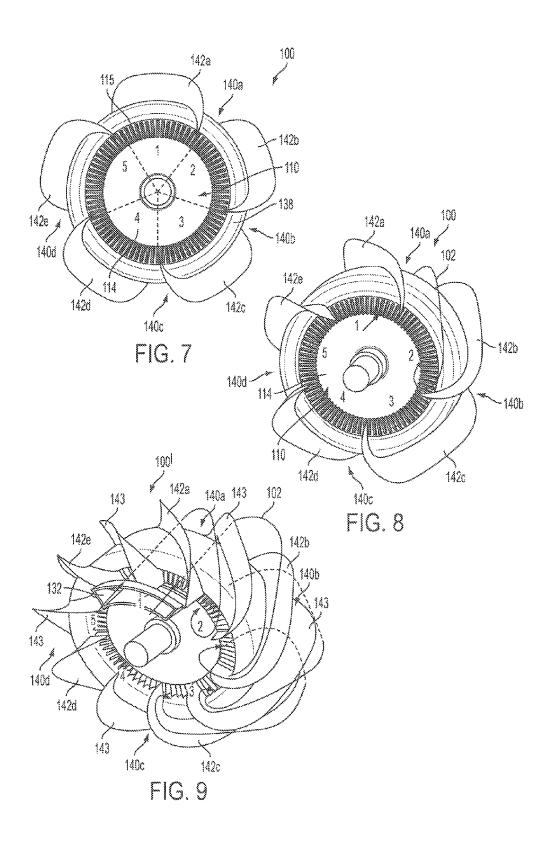
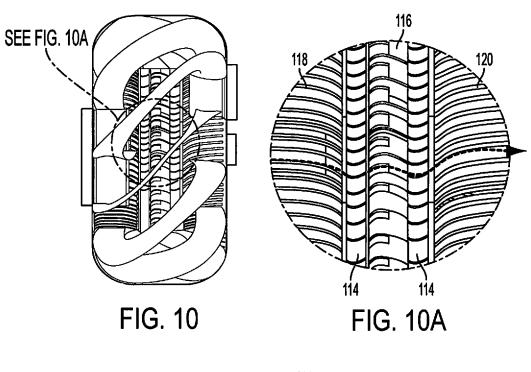


FIG. 4







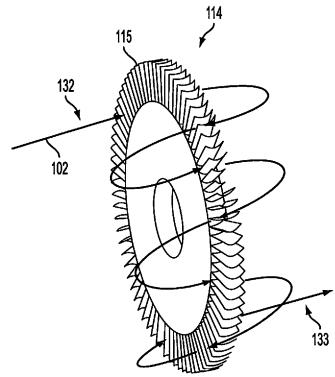
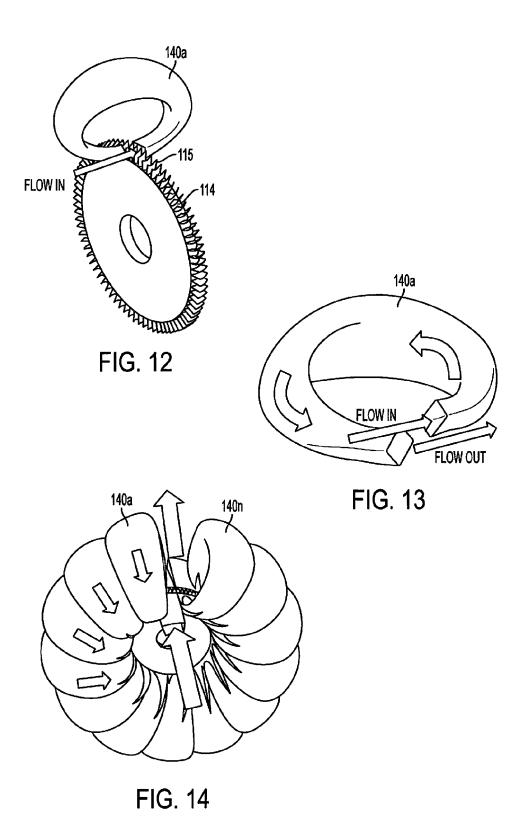


FIG. 11



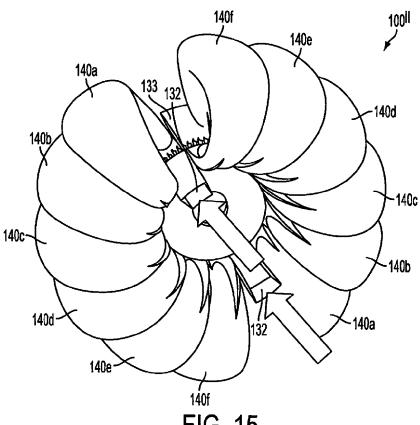


FIG. 15

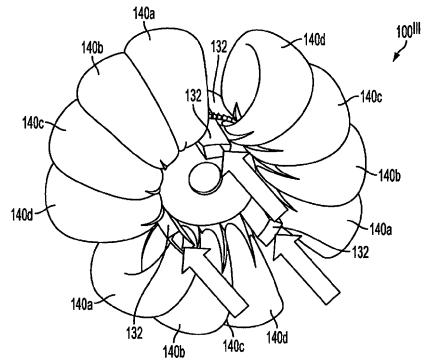


FIG. 16

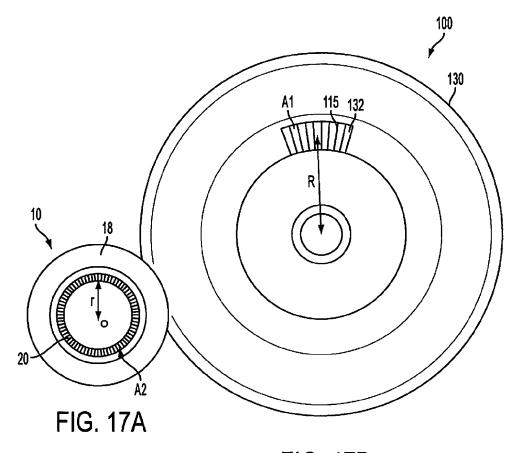
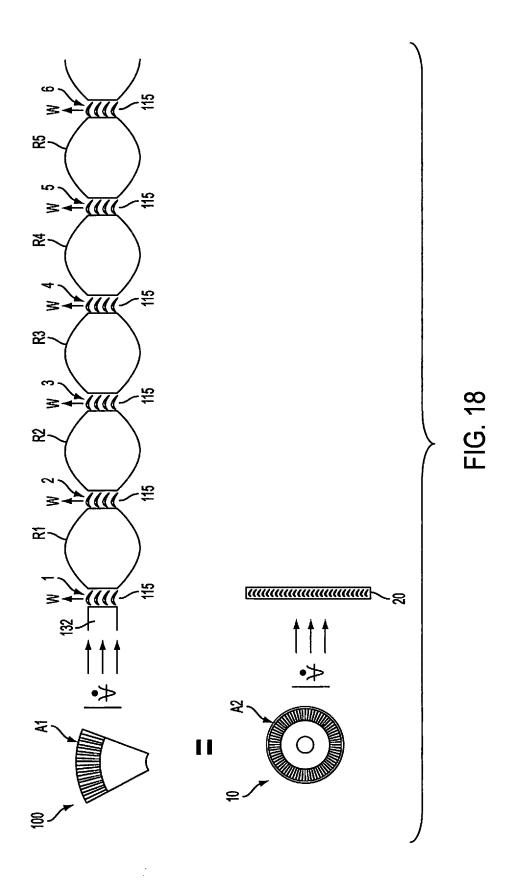
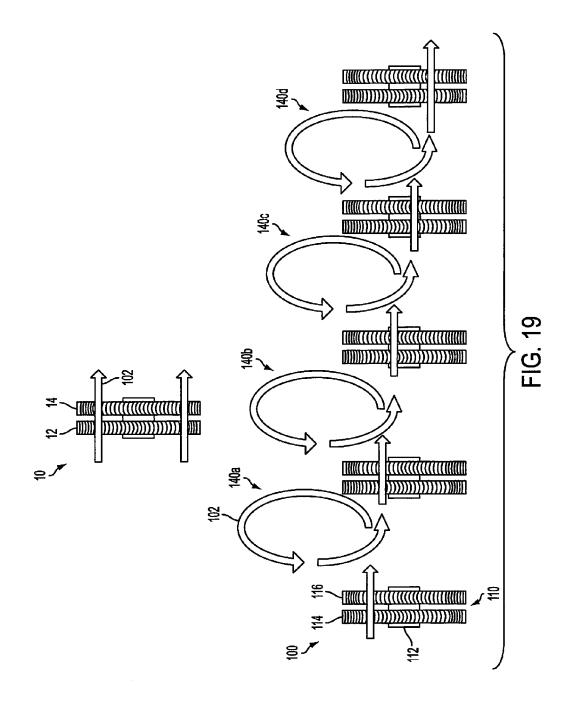
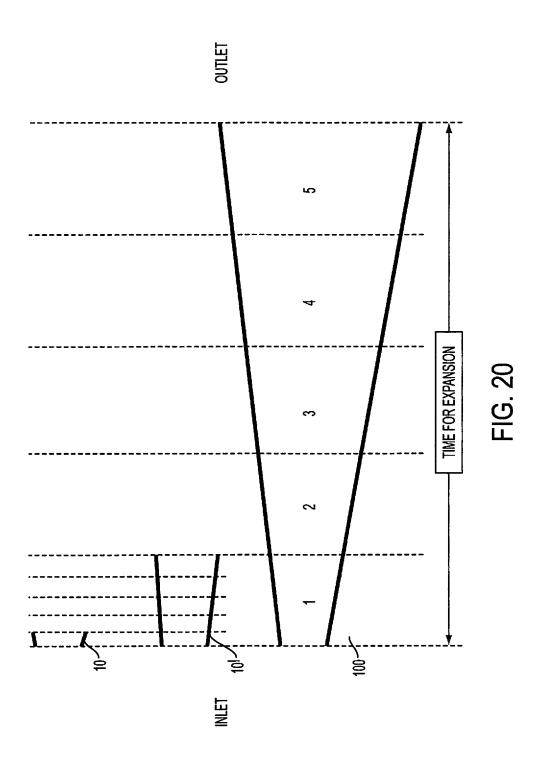


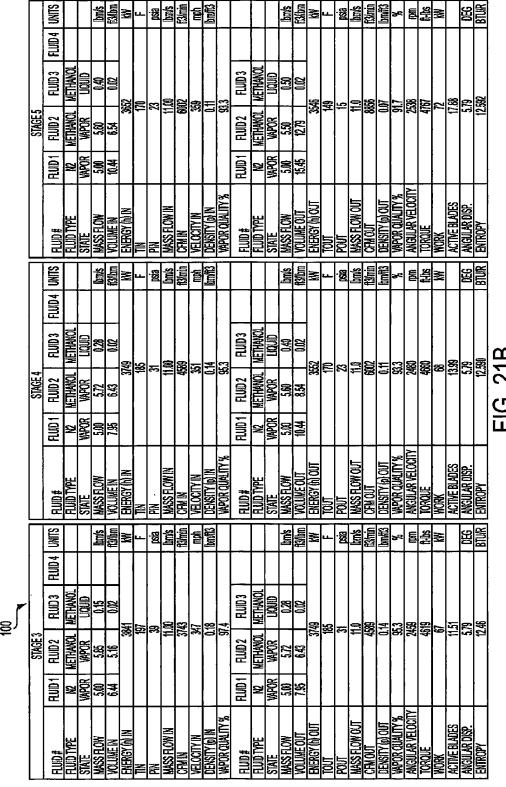
FIG. 17B

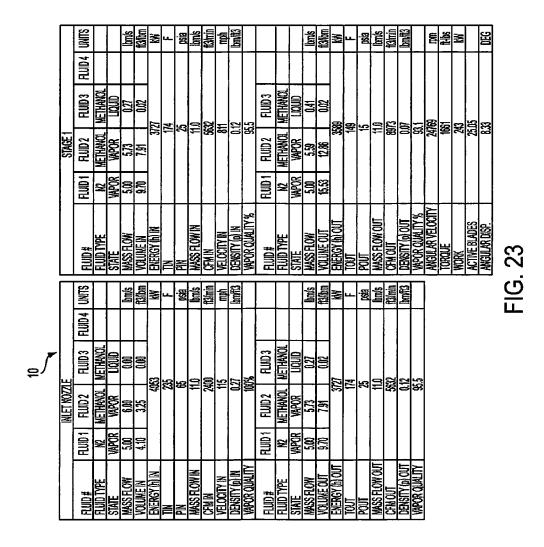


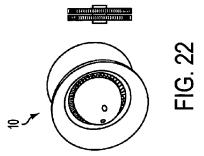




		SIND			Dms	ff3/lbm	Æ	<u></u>	DSia	puls	f3min	Ę						bms	f3/bm	E	u	-88- -88-	Dmls	f3min	EEEE	æ	8	fils	*		99	BTUR	
		FUID4																															
		Fum3	METHANOL	TIOND	0.01			1		Œ	ĝ	6	_		FLWD3	METHANOL	I amoin	0.15	0.02	-			. 0	3		ţ	99	1		*	9	<u>.</u>	
	STAGE 2	FLUID 2	METHANOL	WAPOR	5.99	4.31	3931	IZ	47	11.00	3179	æ	07	8	FLUID2	METHANO!	WAPOR	5.85	5.16	38	19	**	11.	31/2	0.	97.	拔	46	88	9.7	5.7	12.4	
		FUD 1	NZ	WPOR	5.00	5.43							L		PLUID 1	23	VAPOR	5.00	6.44										L				
		#01013	FLUID TYPE	SIATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	NLL ,	NId)	MASS FLOW IN	NIM	NETOCILA IN	DENSITY (p) IN	VAPOR QUALITY %	FUU)#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	1001	TUOA	INASS FLOWOUT	LNO PILO	DENSITY (p) OUT	VAPOR QUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	ENTROPY	
		SLIND			lbms	ft3/lbm	KW	4	psia	Siudi	13/min	щош	Drafts					Dms	(B)(bm	₩	Ŀ	psia	Sjudj	f3/min	bmft3	26	me.	ft-lbs	ΚN		930	BTUR	
		FUD04																															
		FLUD3	METHANOL		00:0		35	₽.	5	11.0		6	₽ŧ.	e:	FUID3	METHANOL		0.01		31	1	1	0	£2	7	8 0;	02	40	8	83	£	æ	21A
	STAGE	FUID2	METHANOL	VAPOR	900		39	77	25	#	1117		76	1	FWD2	METHANOL	WPOR	5.99	4.31	3931	R	7	#	31	07	66	24	46	9	8.6	5.	1238	FIG. 2
		FWD 1	Ø	WAPOR	600	47.4									1999	N2	WPOR	5.00	5.43														正
		FLUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	AL.	PIN	MASS FLOW IN	(CEM IN	VELOCITY IN	DENSITY (a) IN	VAPOR QUALITY %	FuD#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	TOUT	POUT	MASS FLOW OUT	CFM OUT	DENSITY (a) OUT	VAPOR QUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	ENTROPY	
		STIN			sug)	R3/lbm	3	u.	osia	lbmk	R3/min	цdш	[bmff3					bms	13/bm	**	-	DSia	Sund	(C) Minin	lbmf53	96						BTUR	
		FLUD4																															
8	E	FUUD3	METHANOL		000	000	4053	235	85	Ü.	2400	101	0.27	%0%	<u></u>	METHANOL		000	000	3995	30	9	11.0	1111	0.24	0:0						12.39	
	IN ET NOZZI	FLWD2	METHANOL	VAPOR	90.9	325	40	2	9	11.	77		=	=	FEED	METHANOL.	WAPOR	6.00		æ	7	3	1,	77	0	100						12	
		FLWD 1	₩	VAPOR	5.00	4.10									FLUID1	W.	WPOR	2.00	4.74														
		FUED#	FLUIO TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	1	PN.	MASS FLOW IN	CFMIN	VELOCITY IN	DENSITY (p) IN	VAPOR QUALITY	FLUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	TOUT	P0UT	MASS FLOW OUT	CFM OUT	DENSITY (b) OUT	VAPOR QUALITY						ENTROPY	







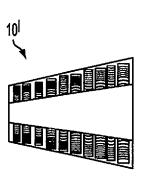
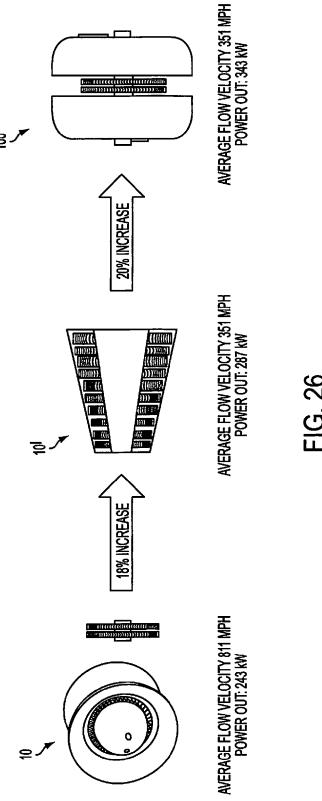


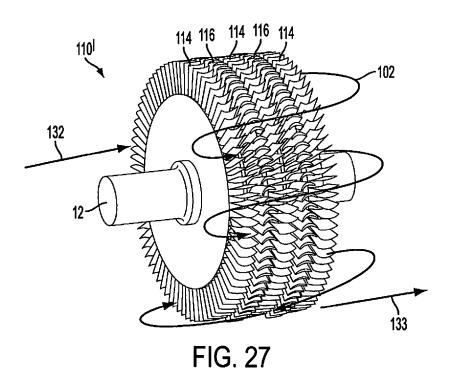
FIG. 24

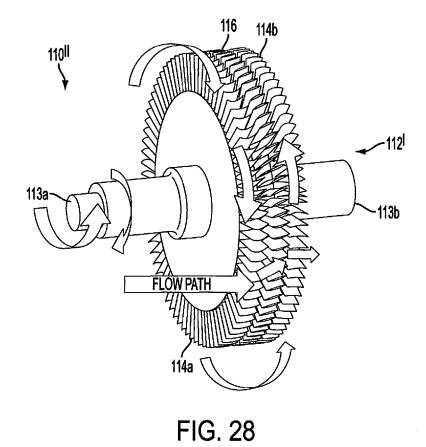
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					Bmis	f2/lbm	*	L	Să	Sund	f3/min	Ę	Pan ff3					Sm(s	f3/lbm	K	<u>.</u>	isi.	Dms	f3/min	lbmft3		Ē	Ξ	₹		93	
		FUID4																														
		FUID3	METHANOL.	CICOLD	0.01	0.02	Ħ	1		00	6j	3	<u>.</u>	∞	FLUD3	METHANOL	CIOCIT	0.15	0.02				0	e e	8	7	J.	æ		12	ĝ.	
	STAGE 2	FLUID 2	METHANOL	VAPOR	5.99	4.31	3331	Ø	4	11.00	31	35	70	868	FLUID 2	METHANOL	VAPOR	5.85		384	16	æ	110	3743	0,18	97.4	1999	S	7	32.	6.7	
		F.UID 4	27	NAPOR	200	5.43									FUID 1	N2	NAPOR	5.00	6.44													
		FLUID#	FLUID TYPE	STATE	MASS FLOW	VOLUMEIN	ENERGY (h) IN	TIN NI	NI	MASS FLOW IN	CEMIN	VELOCITY IN	DENSITY (b) IN	VAPOR QUALITY %	FLUID#	FLUID TYPE	STATE	MASSFLOW	VOLUME OUT	ENERGY (h) OUT	<u>10</u> 1	POUT	MASS FLOW OUT	CFM OUT	DENSITY (b) OUT	VAPOR QUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	
		SILIN			Sung	ff3/fbm	墨		DSia	Dms	(Calmin	de la	圖記					Spund)	ESIBM MISSION	Æ	Ŀ	DS:	lbm/s	f3/min	Dmoff3		ma	स-प्रि	₹		路	
		RUID4																														
		EUUD3	METHANO!	OMO1	0.00		22	0.	2	0:	11	33	ř.	100.0	ECUID3	METHANOL	OMOTI	100	1	31]]	7	0.	79	71	8:	9148	6	4	31.53	72	5A
	STAGE 1	FLUD 2	METHANOL	VAPOR	6.00		88	77	5	ļ	77	25	0	10	FUID 2	METHANOL.	WAPOR	5,99	4.31	æ	X	4	#	31	0	36	16	-8	†	31	9	FIG. 25A
		FWD1	2	WAPOR	5.00	1/1									FLWD 1	M	WAPOR	600	£#3													ĬĻ.
		#@P	FLUID TYPE	STATE	MASS FLOW	VOLUMEIN	ENERGY (h) IN	I TIN	PIN	MASS FLOW IN	CFMIN	VELOCITY IN	DENSITY (b) IN	VAPOR QUALITY %	FUU #	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	TOUT	POUT	MASS FLOW OUT	CFMOUT	DENSITY (b) OUT	WPORQUALITY %	AMGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	
	П	SIIS			Brus	nd Est	≩	ц.	3	(purk	13/min	and the	Dm/E3							≊	Ľ.	.8 <u>2</u>	SE SE	ft3/min	Bruff3							
		FUID4																														
- <u>p</u> /		FUID3	METHANOL		000	ı	23	5		. 0	Q	5	L	%	FLUID3	METHANOL.	LEGNO	000	000	3 5	0		0	7	7							
	IN ET NOZZI E	FLUD2	METHANO!	WAPOR	6.00		#	S	88		W	10	07	100%	FLUID 2	METHANOL.	WAPOR	609	3.76	XX	7	ਖ਼ਫ਼	11.	1111	0.7							
		FUO 1	ZZ.	WAPOR	5.00	4.10									FLWD 1	24	VAPOR	5.00	474					:								
		FUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	≅	Pin	MASS FLOW IN	CFRIN	VELOCITY IN	DENSITY (6) IN	VAPORQUALITY	FUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	TOUT	POJT	MASS FLOW OUT	CFMOUT	DENSITY (b) OUT							

		SIIN			Simql	ft3Npm	æ	щ.	· <u>8</u>	SEG Seg	f3hrin	巨	21 E		-			Smd	ft3/Jbm	W.	·	1930	Smal	f3hiri	lbm/ff3		튣	ths	×		贸	
		FUID4	П																													
		FUND3	METHANOL.		0,40	70'0		_		8	N			3	FUID3	METHANOL.	0001	050		97	ő	2	0.	S)7	.7	82	1770	0	3 2	74	
	STAGE 5	FUD2	METHANOL	VAPOR	5.60		38	J	2	Ħ,	3	**	O	93.3	FLUID 2	METHANOL	VAPOR	5.50		35	1	•	ĮĮ.	88	0	94	92	17	8	35	10	
		FLUID 1	¥	WAPOR	2.00	10.44									FWD1	2	VAPOR	2.00	15.45													
		FLUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	NL	PN	MASS FLOW IN	CFMIN	VELOCITY IN	DENSITY (5) N	VAPOR QUALITY %	FLUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	1001	POUT	MASS FLOW OUT	CFNOUT	DENSITY (b) OUT	VAPOR QUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	
		SJIND			bms	f3fbm	KA		Bela	Sung	£Smiri	щф	Shind					SEC SEC	130m	æ	L	psia	Studi	ft3/min	lbm/ff3			f SE	Æ		呂	
		FUUD4																														
		FLUID3	METHANOL		0.28		3749	38	-	8	88	0#	71.	95.3	FUID3	NETHANOL		0.40	_	3652	0/	ę.	11.0	02	11	3.3	61	1353	2	. 33	95	25B
	STAGE 4	FLUD 2	METHANO!	WAPOR	5.72		37	 	~	H	4	3	3	ð	FLUD2	METHANOL.	WAPOR	5.60		88	ļ	7	ţ	æ	0	25	76	13	9	33	9	FIG. 29
		FUID 1	22	I VAPOR	5.00	7.95									B	2	I VAPOR	5.00	10.44													正
		FLUID#	FLUID TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	NI NI	N.	MASS FLOW IN	CFILIN	VELOCITY IN	DENSITY (b) IN	VAPOR QUALITY %	#82	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	1001	POUT	MASS FLOW OUT	CFMOUT	DENSITY (b) OUT	VAPOR QUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	
		UNITS			Suud	f3lbm	₹	-	Sia	Sund	(S)min	la la	lbm/R3					Suc	f3jbm	M	Ч	Sá	buk	(Shrii)	bruff3		md.	f-lps	₹		路	
		FUID4													_																	
-p/		EUNT3	METHANOL	110000	0.15	0.02	41	187	cn	8	3743	34 5	8	1.	FLUID3	METHANOL	017011	0.28		3749	35	31	11.0	89	14	.3	88	1194	3	32.68	.78	
	STAGE 3	FLUID2	METHANOL	WAPOR	5.85	,	384)	3	Ŧ	37	25	ت	9.	FUID 2	METHANOL	WAPOR	5.72	6.43	37	185	63	1	46	0	<u>ج</u>	38	11	4,	32	7	
		FUID 1	2	VAPOR	5.00	6.44									E	ZV	I VAPOR	200	7.95													
		FLUID#	FLUID TYPE	STATE	MASS FLOW	VOLUME IN	ENERGY (h) IN	NI.	S.	MASS FLOW IN	CHAIN	VELOCITY IN	DENSITY (a) IN	VAPOR QUALITY %	FUD#	FLUID TYPE	STATE	MASS FLOW	VOLUME OUT	ENERGY (h) OUT	T00.T	POUT	MASS FLOW OUT	CFWOUT	DENSITY (p) OUT	VAPOR CUALITY %	ANGULAR VELOCITY	TORQUE	WORK	ACTIVE BLADES	ANGULAR DISP.	







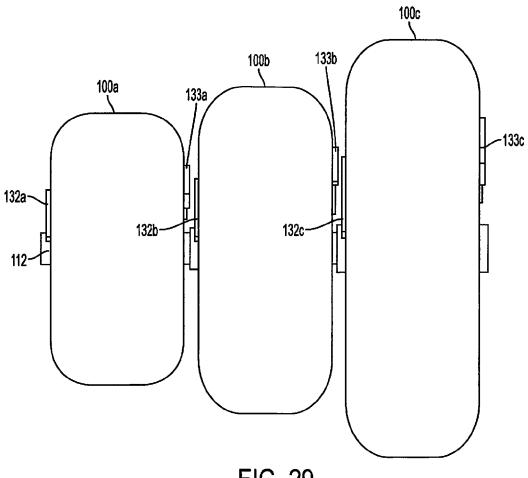
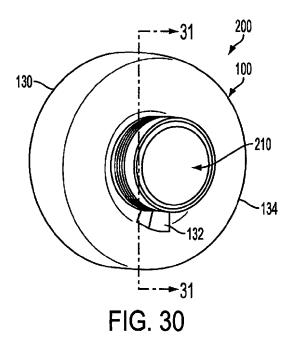
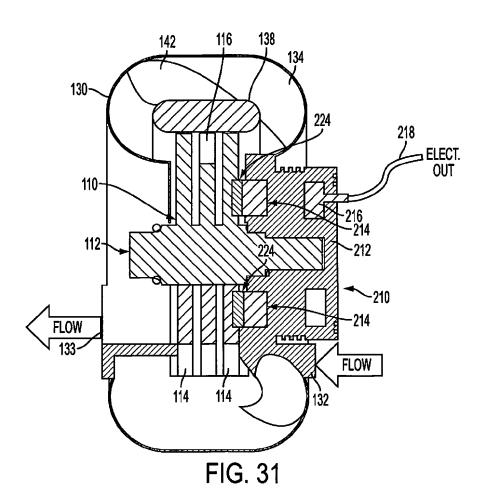
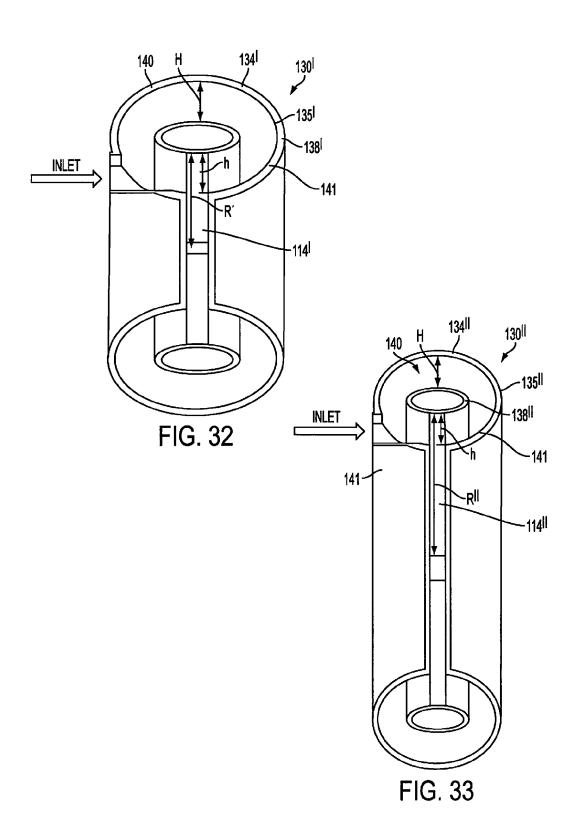


FIG. 29

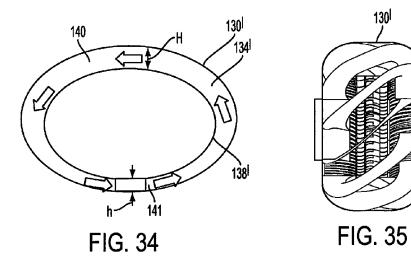


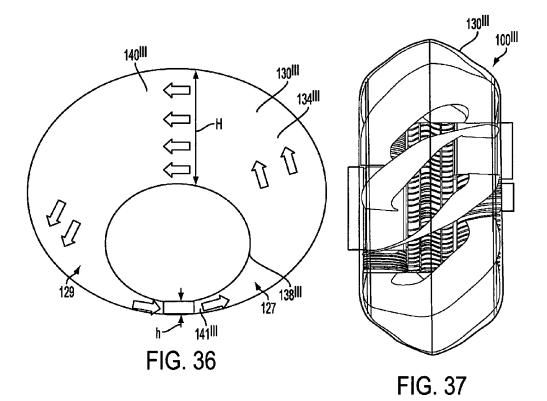


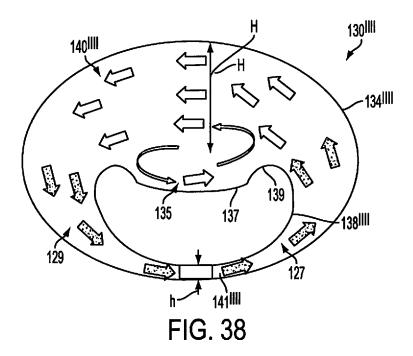


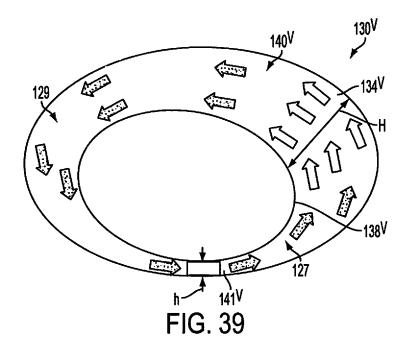
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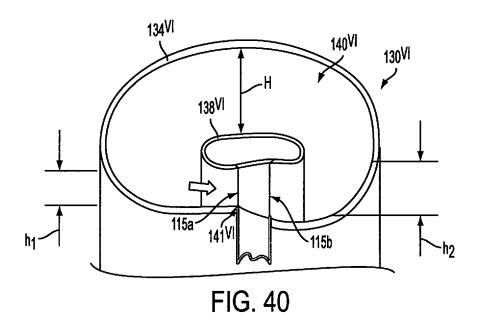
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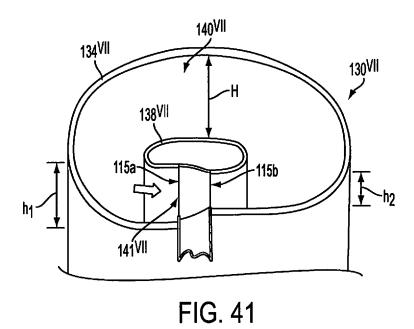












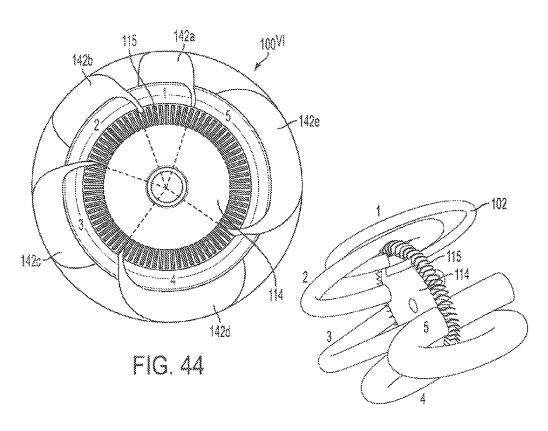
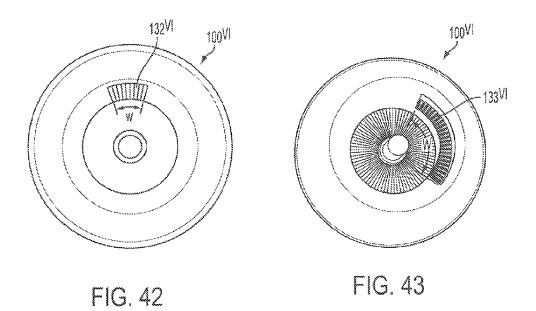
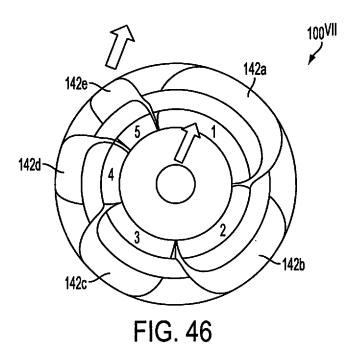


FIG. 45





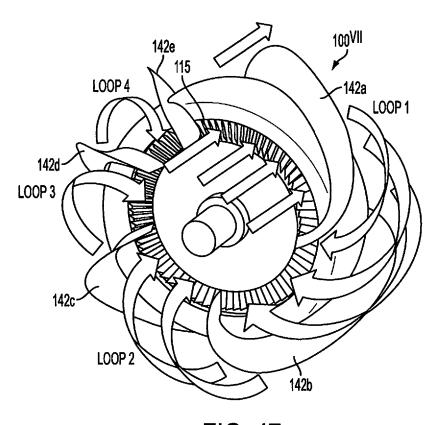
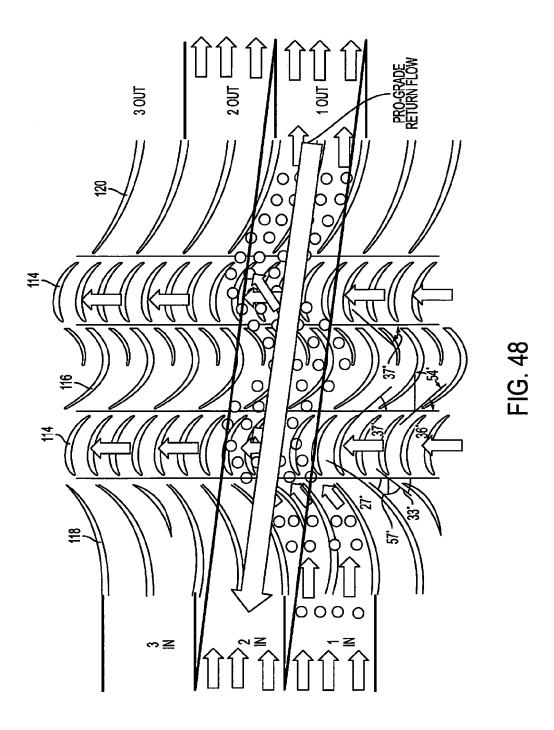
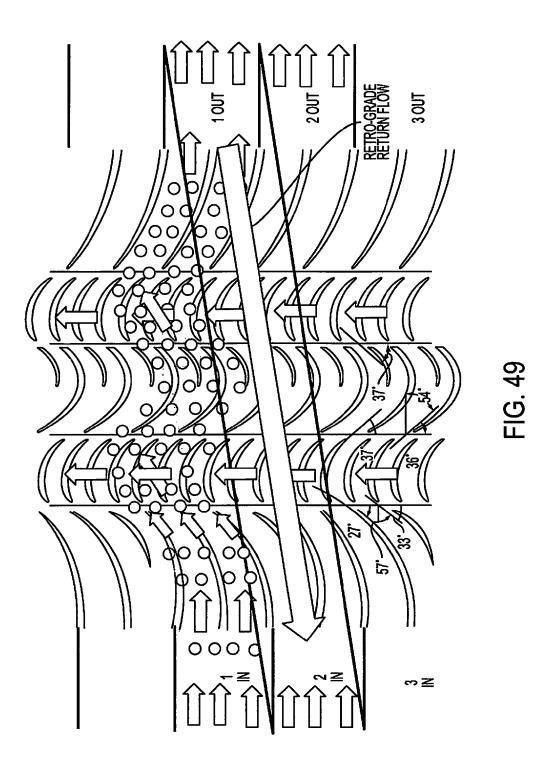


FIG. 47





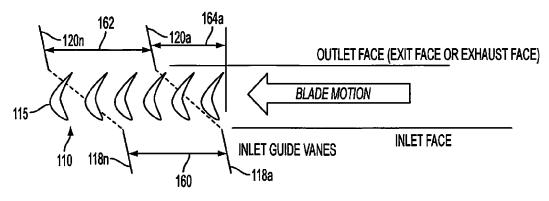


FIG. 50A

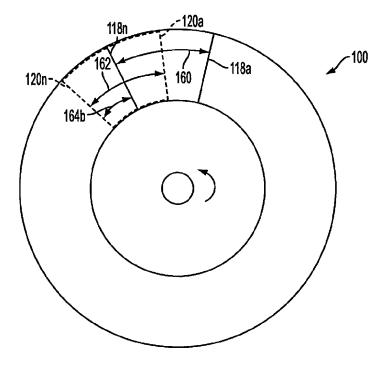


FIG. 50B

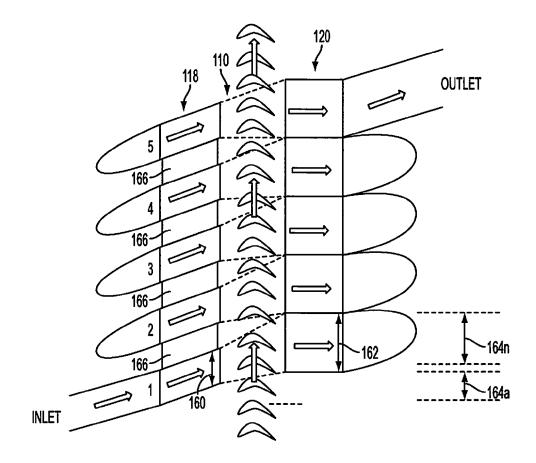


FIG. 51

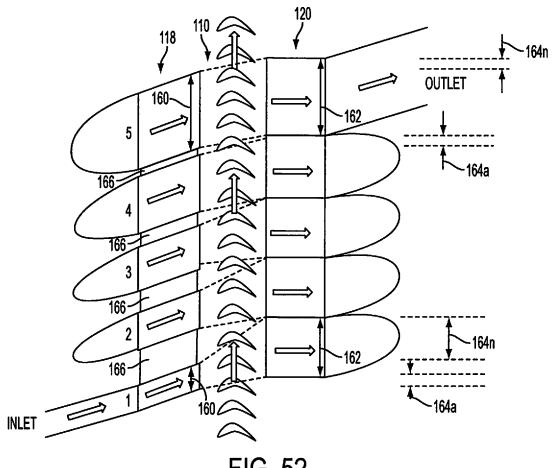


FIG. 52

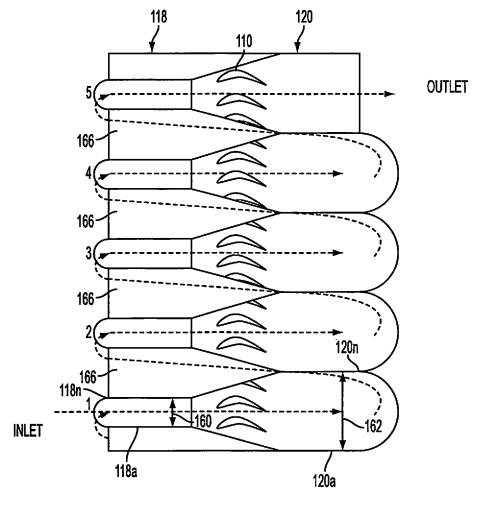
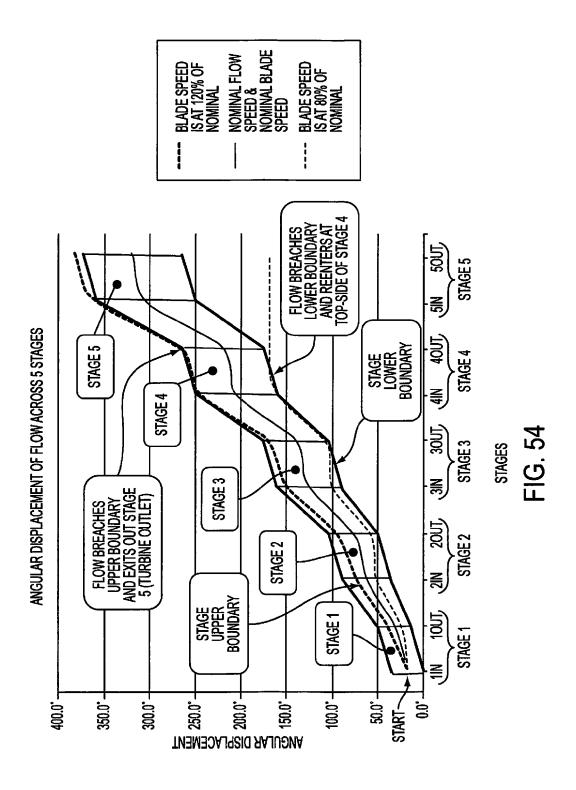


FIG. 53



SYSTEM AND METHOD OF WRAPPING FLOW IN A FLUID WORKING APPARATUS

BACKGROUND OF THE INVENTION

Statement of the Technical Field

The invention concerns fluid working, and more particularly systems and methods for wrapping fluid flow in a fluid working apparatus, for example an expander or compressor, which results in an increased capacity to perform work by 10 the fluid or on the fluid.

Description of the Related Art

A turbo-expander is a machine which continuously converts kinetic energy into mechanical energy by harnessing the pressure and heat of pressurized fluid to rotate a shaft. 15 FIGS. 1 and 2 show an exemplary axial turbo-expander 10. Each stage of the expander 10 includes a rotatable rotor 12 and a stationary stator 14. Inlet vanes 16 and outlet vanes (not shown) may be provided to help guide the path of the flowing fluid and the vanes may serve as the stator for one 20 or more of the stages. The rotors 12, stators 14 and vanes are supported in a housing 18. To ensure proper flow and rotation, each of the rotors 12 must be manufactured within tight tolerances relative to the housing 18. As illustrated by the arrows, the fluid passes through each stage a single time, 25 interacting with the rotor 12 and stator 14 for only the period of time it takes for the fluid to pass through the stage. As the fluid passes through a given stage, the fluid expands and exerts a force to rotate the rotor 12, which in turn rotates the shaft (not shown).

Turbo-expanders are utilized in various applications, for example, a compressor-drive, power generator, brake drive, or cooling system. In the first three examples, the power transmitted to the shaft is used to drive a compressor, drive an electrical generator or is dissipated through an oil brake or air brake, respectively. In a cooling or refrigeration system, the gas exiting the expander, which is colder and lower-pressure than it was when it went in, is directed to a heat exchanger. Expanders and compressors may comprise or take on many different physical configurations, all of which are easily found in literature. The axial flow example shown provides the most useful architecture for the purpose of contrasting the difference. These applications are for illustrative purposes only and are not intended to be limiting.

An axial compressor works just like the turbo expander 45 but in reverse. Power is supplied to the shaft which in turn rotates the rotors. The rotors accelerate the fluid and the stators diffuse the flow to obtain a pressure increase. That is, the diffusion in the stator converts the velocity increase gained in the rotor to a pressure increase. As with the 50 expander, the fluid passes through each stage a single time, interacting with the rotor and stator for only the period of time it takes for the fluid to pass through the stage.

SUMMARY OF THE INVENTION

Embodiments of the invention concern a fluid working apparatus. In at least one embodiment, the fluid working apparatus includes a housing structure with a housing inlet and a housing outlet. A working assembly is positioned in 60 the housing with a rotor thereof rotatably supported in the housing structure. The working assembly has an inlet side and an opposite outlet side with the at least one rotor having a plurality of blades positioned between the inlet and outlet sides. At least one return assembly is configured to return 65 fluid flow from the outlet side of the working assembly to the inlet side of the working assembly a working fluid

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passes through the housing inlet, then from the inlet side of the working assembly to the outlet side thereof while workingly engaging a first subset of the rotor blades, then through the at least one return assembly, then from the inlet side of the working assembly to the outlet side thereof while workingly engaging a second subset of the rotor blades, and thereafter out of the housing outlet.

Embodiments of the invention concern a method of re-circulating a working fluid. The method includes passing the fluid from an inlet side of a working assembly to an outlet side thereof, the working assembly including at least one rotor having a plurality of blades positioned between the inlet and outlet sides, whereby the fluid while workingly engages a first subset of the rotor blades; passing the fluid through a return assembly whereby the fluid flows from the outlet side of the working assembly to the inlet side of the working assembly; and passing the fluid from the inlet side of the working assembly to the outlet side thereof while workingly engaging a second subset of the rotor blades.

The present invention provides multi-pass recirculation of the working fluid that is unique relative to the current art.

BRIEF DESCRIPTION OF THE DRAWINGS

Embodiments will be described with reference to the following drawing figures, in which like numerals represent like items throughout the figures, and in which:

FIG. 1 is a front isometric view of an exemplary prior art axial turbo-expander.

FIG. 2 is a rear isometric view of the axial turbo-expander of FIG. 1 in partial cutaway.

FIG. 3 is an isometric view of a fluid working apparatus in accordance with an exemplary embodiment of the present invention.

FIG. **4** is an exploded view of the fluid working apparatus of FIG. **3**.

FIG. 5 is a cross-sectional view along the line 5-5 in FIG.

FIG. 6 is an isometric view of an exemplary fluid working apparatus with the outer housing shown transparently.

FIG. 7 is a front elevation view of the fluid working apparatus of FIG. 3 with the outer housing omitted.

FIG. 8 is an isometric view of the fluid working apparatus of FIG. 3 with the outer housing omitted.

FIG. 9 is an isometric view of the fluid working apparatus of FIG. 6 with the outer housing omitted.

FIG. 10 is a top plan view of the fluid working apparatus of FIG. 6 with the outer housing shown transparently.

FIG. 10 A is an expanded view of a portion of the fluid working apparatus of FIG. 10.

FIG. 11 is an isometric view of an exemplary rotor illustrating an exemplary flow path of a fluid working apparatus in accordance with the invention.

FIGS. 12-14 are drawings that are useful for understanding flow paths of the fluid in accordance with one or more embodiments of the invention.

FIGS. **15-16** are drawings that are useful for understanding alternative flow paths of the fluid in accordance with one or more embodiments of the invention.

FIGS. 17A and 17B illustrate the inlet area of an exemplary fluid working apparatus of the present invention relative to the inlet area of a prior art axial turbo-expander.

FIG. **18** is a drawing that is useful for understanding the flow volume and work output of an exemplary fluid working apparatus of the present invention relative to that of a prior art axial turbo-expander.

FIG. 19 is a drawing that is useful for understanding the flow path of an exemplary fluid working apparatus of the present invention relative to that of a prior art axial turbo-expander.

FIG. **20** is a drawing that is useful for understanding the time for expansion of an exemplary fluid working apparatus of the present invention relative to that of a prior art axial turbo-expander.

FIGS. 21A and 21B present a table showing measurements from an exemplary single rotor, five zone fluid working apparatus of the present invention.

FIG. 22 shows an exemplary prior art, single rotor axial turbo-expander and FIG. 23 is a table showing exemplary measurements therefore.

FIG. 24 shows an exemplary prior art, five rotor axial turbo-expander.

FIGS. 25A and 25B present a table showing measurements from a prior art, five rotor axial turbo-expander as illustrated in FIG. 24.

FIG. 26 is a drawing representing the data from the various tables from FIGS. 21A, 21B, 23, 25A and 25B.

FIG. 27 is an isometric view of a working assembly of an alternative exemplary embodiment of the present invention.

FIG. $\bf 28$ is an isometric view of a working assembly of an 25 alternative exemplary embodiment of the present invention.

FIG. **29** is a drawing illustrating multiple fluid working apparatuses connected in series.

FIG. 30 is an isometric view of a generator device incorporating an exemplary fluid working apparatus of the present invention.

FIG. 31 is a cross-sectional view along the line 31-31 in FIG. 30.

FIG. 32 is a cross-sectional view of an exemplary housing $_{35}$ in accordance with an embodiment of the invention.

FIG. 33 is a cross-sectional view of another exemplary housing in accordance with an embodiment of the invention.

FIG. 34 is a drawing illustrating fluid flow through the housing of the exemplary fluid working apparatus of FIG. $_{40}$

FIG. **36** is a drawing illustrating fluid flow through the housing of the exemplary fluid working apparatus of FIG. **37**

FIGS. **38-41** are drawings illustrating fluid flow through 45 other exemplary housings.

FIG. **42** is a front elevation view of a fluid working apparatus in accordance with an exemplary embodiment of the present invention.

FIG. **43** is a rear elevation view of the fluid working ⁵⁰ apparatus of FIG. **42**.

FIG. 44 is a front elevation similar to FIG. 42 with a portion of the outer housing omitted.

FIG. **45** is a drawing illustrating fluid flow through the fluid working apparatus of FIG. **42**.

FIG. **46** is a front elevation similar to FIG. **44** illustrating another exemplary embodiment of the present invention representing a compressor configuration.

FIG. 47 is an isometric view of the fluid working apparatus of FIG. 46.

FIGS. **48** and **49** are drawings that are useful for understanding different flow options through the exemplary fluid working apparatus of the present invention.

FIGS. 50A and 50B are top and elevation views, respectively, illustrating an exemplary stage offset configuration in accordance with an embodiment of the invention.

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FIGS. **51-53** are drawings that are useful for understanding different stage transitions through the exemplary fluid working apparatus of at least one embodiment of the present invention.

FIG. **54** is a graph illustrating exemplary angular displacement of flow across a five zone fluid working apparatus of at least on embodiment of the present invention.

DETAILED DESCRIPTION

The invention is described with reference to the attached figures. The figures are not drawn to scale and they are provided merely to illustrate the instant invention. Several aspects of the invention are described below with reference to example applications for illustration. It should be understood that numerous specific details, relationships, and methods are set forth to provide a full understanding of the invention. One having ordinary skill in the relevant art, however, will readily recognize that the invention can be 20 practiced without one or more of the specific details or with other methods. In other instances, well-known structures or operation are not shown in detail to avoid obscuring the invention. The invention is not limited by the illustrated ordering of acts or events, as some acts may occur in different orders and/or concurrently with other acts or events. Furthermore, not all illustrated acts or events are required to implement a methodology in accordance with the invention.

Referring to FIGS. 3-14, fluid working apparatuses 100, 100' in accordance with exemplary embodiments of the present invention will be described. As used herein, the term fluid working apparatus refers to an apparatus with a rotatable rotor which works on a working fluid or is worked on by a working fluid. Examples include expanders and compressors, but are not limited to such.

Each fluid working apparatus 100, 100' includes a working assembly 110 supported within a housing 130. The working assembly 110 includes a shaft 112 which supports at least one rotor 114 with a plurality of blades 115. In the present embodiments, the working assembly 110 includes a pair of rotors 114 with a stator 116 positioned therebetween. As shown in FIG. 10A, inlet vanes 118 and outlet vanes 120 may be provided to guide fluid flow across the rotors 114 and stator 116. The invention does not require 2 rotors, it could be accomplished with 1, 3, 4 or more depending on the specifics of the design application.

The housing 130, shown in FIG. 5, includes an outer housing member 134 and an inner housing member 138. The outer housing member 134 includes a hollow tubular portion 135 with side walls 136 extending within the center of the tubular portion 135. The side walls 136 support the shaft 112 with the rotors 114, stator 116 and the inlet and outlet vanes 118, 120 positioned therebetween. The stator 116 may be rotationally fixed through connection to the inner housing member 138 or otherwise. The rotors 114 and stator 116 extend radially into the tubular portion 135 of the outer housing member 134. A housing inlet 132 extends through one side of the outer housing member 134 aligned with the blades 115 of the rotors 114. As illustrated in FIGS. 4 and 6, the housing inlet 132 extends only a portion of the housing 130 circumference. A housing outlet 133 extends through the opposite side of the outer housing member 134 in alignment with the rotor blades 115. Similarly, the housing outlet 133 extends only a portion of the housing 130 circumference. As will be described hereinafter, the housing inlet 132 and outlet 133 may have the same or different circumferential widths. The surface of the working assembly

110 adjacent the inlet 132 may be referred to the inlet surface and the surface of the working assembly 110 adjacent the outlet 133 may be referred to as the outlet surface.

The inner housing member 138 is positioned within the tubular portion 135 of the outer housing 134 and has a 5 tubular outer surface 139 spaced from the inside surface of the outer housing 134 such that a return chamber 140 is defined between the inner housing 138 and the outer housing 134. The inner housing member 138 is illustrated as a solid structure, but may instead be completely or partially hollow. 10 The inner housing member 138 is maintained in position relative to the outer housing member 134 by a plurality of boundary vanes 142, alone or in conjunction with guide vanes 143, extending between the inner surface of the tubular portion 135 and the outer surface 139 of the inner 15 housing member 138.

The boundary vanes 142 extend helically and divide the return chamber 140 into distinct return zones 140a, 140b, 140c, 140d as illustrated in FIGS. 7-9. For example, the boundary vanes 142a and 142b define the return zone 140a. 20 vanes 142b and 142c define the return zone 140b, vanes 142c and 142d define the return zone 140c, and vanes 142dand 142e define the return zone 140d. The vanes 142, and thereby the return zones 140a-140d, extend from the outlet surface of the working assembly 110 to the inlet surface of 25 the working assembly 110. Each return zone 140a-140d represents a return assembly. As illustrated in FIGS. 6 and 9, one or more guide vanes 143 may be positioned between the boundary vanes 142. The guide vanes 143 do not define a given return zone 140a-140d, but instead assist in guiding 30 fluid as it travels through the given zone, for example by reducing turbulence.

The working fluid enters through the housing inlet 132 and passes through a first working zone 1 of the rotor blades 115. The working fluid acts on the rotors 114 as it passes 35 through and then exits the rear of the working assembly 110 as shown in FIGS. 10 and 10A. Upon exiting, the working fluid is directed through the return zone 140a and back to the inlet surface of the working assembly 110 whereafter it passes through a second zone 2 of the rotor blades 115. The 40 working fluid again acts on the rotors 114 as it passes through and then exits the rear of the working assembly 110. Upon exiting, the working fluid is directed through the return zone 140b and back to the inlet surface of the working assembly 110 whereafter it passes through a third zone 3 of 45 the rotor blades 115. The working fluid again acts on the rotors 114 as it passes through and then exits the rear of the working assembly 110. Upon exiting, the working fluid is directed through the return zone 140c and back to the inlet surface of the working assembly 110 whereafter it passes 50 through a fourth zone 4 of the rotor blades 115. The working fluid once again acts on the rotors 114 as it passes through and then exits the rear of the working assembly 110. Upon exiting, the working fluid is directed through the return zone **140***d* and back to the inlet surface of the working assembly 55 110 whereafter it passes through a fifth zone 5 of the rotor blades 115. The working fluid once again acts on the rotors 114 as it passes through and then exits the rear of the working assembly 110. The working fluid, after acting on the rotors 114 five times, then exits through the housing outlet 60 133. Each pass of the working fluid across a rotor 114 may be referenced as a stage. In the presently described embodiment, the working fluid passes across two rotors 114 five times, thereby achieving ten stages of potential work on the

FIG. 11 illustrates an exemplary path 102 of the working fluid as it enters through the housing inlet 132 and passes the

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rotor blades 115 multiple times before exiting through the housing outlet 133. FIGS. 12-14 provide further simplified illustrations of the exemplary fluid flow. The return zones 140a or return assemblies are illustrated as independent tubular members, which is conceivable, however, the zones are preferably defined by the housing 130 and vanes 142 as described above. The flow in for a given blade zone passes the rotor 114 and then is returned through the return zone 140a. The entrance to the return zone 140a is circumferentially offset from the exit from the return zone 140a such that the flow out of the working fluid is delivered to the next zone of blades 115.

As illustrated in FIG. 14, the number of return zones 140a-140n is not limited to five as in the previously illustrated embodiments, but may be any number of return zones one or more such that the fluid passes across the working assembly 110 at least twice. Furthermore, as illustrated in FIGS. 15 and 16, the fluid working apparatus 100", 100"" may have more than one housing inlet 132 and corresponding housing outlet 133. In the embodiment illustrated in FIG. 15, the fluid working apparatus 100" includes two housing inlets 132, with each housing inlet providing a flow path through six return zones 140a-140f such that the fluid working traveling each path passes the working assembly 110 seven times. In the embodiment illustrated in FIG. 16, the fluid working apparatus 100" includes three housing inlets 132, with each housing inlet providing a flow path through four return zones 140a-140d such that the fluid working traveling each path passes the working assembly 110 five times. The number of inlets as well as the number of return zones for each flow path may be varied as desired.

Having described the general configuration of exemplary embodiments of the fluid working apparatus 100, a comparison relative to an axial flow device will be provided with reference to FIGS. 17A-26. Referring to FIGS. 17A and 17B, an exemplary fluid working apparatus 100 is show with a prior art axial turbo-expander 10 with an equivalent flow rate. The fluid working apparatus 100 has an inlet 132 which extends over a partial circumference while the turbo-expander 10 has an inlet 20 extending about the complete circumference. As such, if the inlet areas A1 and A2 are to be equal, the apparatus 100 has a larger radius R and larger blade height than the radius r and blade height of the turbo-expander 10. The larger rotor diameter can be tuned to provide increased torque output in the fluid working apparatus 100.

FIG. 18 illustrates the effective work performed by the fluid working apparatus 100 compared to that of the turboexpander 10 with equivalent flow rates V1 and V2. As illustrated here, the apparatus 100 has five return zones R1-R5 such that work is performed on the rotor blades 115 six times as shown 1-6. In this example, the apparatus 100 has twenty four larger blades relative to the turbo-expander 10 having a single stage with twenty-four smaller blades, each addressing the same initial working fluid flow conditions. Due to the larger radius R, the rotor blades 115 are three times the size as the rotor blades of the turbo-expander 10. In relative and simplistic terms, the performance of the device may be viewed as W=F*B*S*N*P wherein F is the work flow, B is the number of blades, S is the relative size of the blade, N is the number of passes and P is the available drive pressure ratio. For a work flow of 100, the apparatus 100 work W=100*4*1*6*1/2=1200 while the turbo-expander 10 work W=100*24*1/3*1*1=800. In this example, the fluid working apparatus 100 achieves fifty percent more work than the turbo-expander 100 with an equivalent work flow. This is a simplified comparison for illustration pur-

poses only. It is recognized that a broad range of complex equations are used to calculate turbine performance. It is known generally that for low pressure, slower speed flows it is desirable to have a larger blade area to interface with the working fluid. The exemplary apparatus 100 provides this 5 feature.

FIGS. 19 and 20 illustrate the respective flow paths 102, and associated time of expansion, as the working fluid flows through the axial turbo-expander 10 and an exemplary fluid working apparatus 100 of the present invention. As shown in 10 FIG. 19, the flow 102 through the turbo-expander 10 is a single axial flow across the single stage. As illustrated in FIG. 20, this flow provides a minimal amount of time for expansion of fluid passing through the turbo-expander 10. Even in a five stage axial turbo-expander 10', the time for 15 expansion is relatively minimal. In the illustrated fluid working apparatus 100, the flow travels through four return zones 140a-140d such that five working stages 1-5 are utilized, as illustrated in FIG. 19. With the multiple passes of the working fluid along with the time of travel through the 20 return zones 140a-140d, the time for expansion through the apparatus 100 is substantially greater than even the five stage turbo-expander 10'. The fluid working apparatus 100 of the present invention provides a simpler construction, which is easier and less costly to produce, which achieves 25 higher performance.

FIGS. 21A, 21B, 23, 25A and 25B provide tables illustrating exemplary data of a fluid working apparatus 100 and axial turbo-expanders 10, 10' based on computer models. The five stage configurations are based on common inlet 30 nozzle, with pressure drop of 20 psi from a starting Temp of 235 F and pressure of 65 psi. All configurations use a mixed working fluid comprising 6 lbm/s flow of methanol and 5 lbm/s nitrogen. For the five stage configurations, the average flow velocity is equilibrated at 351 MPH. All comparisons 35 start with an initial enthalpy energy of 4,053 kW.

FIGS. 21A and 21B illustrate the projected data of a single rotor fluid working apparatus 100 with five working zones defined about its circumference to achieve five stages. With a blade diameter of 14.2" ID and 19" OD and an inlet flow 40 velocity of 349 mph, the apparatus 100 achieves a power output=343 kW.

In comparison, FIGS. 23 and 25A and 25B illustrate projected data for a single stage axial turbo-expander 10 similar to that illustrated in FIG. 22 and a five stage axial 45 flow expander 10' similar to that illustrated in FIG. 24, respectively. These comparisons are relative performance profiles without regard to design specifics, e.g. drag losses in the flow channels or the like. The turbo-expander 10 has a blade diameter of 5.0" ID×6.6" OD, and turbo expander 10' 50 has a blade diameter of 6.6" ID×8" OD (growing progressively to 10"OD in the last stage). As shown in FIG. 23, with an inlet flow velocity=811 mph, the turbo-expander 10 achieves a power output=243 kW. As shown in FIGS. 25A and 25B, with an average flow velocity=351 mph, the five 55 stage turbo-expander 10' achieves a power output=287 kW.

FIG. 26 summarizes the exemplary data of FIGS. 21A, 21B, 23, 25A and 25B. As shown, the single rotor fluid working apparatus 100 provides a 23% increase over the five stage turbo-expander 10' and an 18% increase over the 60 single stage turbo-expander 10. It is noted that the constructs of the present invention provide the opportunity to convert power from lower pressure flows. This is achieved do to the larger blade area and slower rotational speeds for an equivalent volume of flow.

Referring to FIGS. 27 and 28, additional exemplary working assemblies 110' and 110" are illustrated. In the

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embodiment of FIG. 27, the working assembly 110' has a shaft 112 with three rotors 114 and two stators 116. The working assemblies of the present invention may have any desired number of rotors and stators. In the illustrated embodiment, the flow path 102 is such that the flow 102 is returned three times and thereby passes across all three rotors 114 four times between the inlet 132 and outlet 133. As such, the working assembly 110' provides twelve working stages.

The working assembly 110" of FIG. 28 illustrates a multi-part shaft 112'. The inner shaft 113A is associated with the rotor 114A and the outer shaft 113B is associated with the rotor 114B. In the illustrated embodiment, the rotors 114A and 114B have opposite configurations such that the fluid flow causes the inner shaft 113A to rotate counter-clockwise while the outer shaft 113B rotates clockwise. Alternatively, the shafts 113A and 113B may rotate in the same direction and may be selectively coupled or decoupled to one another. Any desired shaft configuration is within the scope of the present invention.

FIG. 29 illustrates multiple fluid working apparatuses 100A-110C connected to one another in series. Fluid passes into the first apparatus 100A via inlet 132A, loops through multiple stages and exits through outlet 133A. Fluid from outlet 133A then travels into the second apparatus 100B via inlet 132B, loops through multiple stages and exits through outlet 133B. Fluid from outlet 133B then travels into the third apparatus 100C via inlet 132C, loops through multiple stages and exits through outlet 133C. Fluid couplings, not shown, are provided between each outlet and the next inlet The apparatuses 100A-100C may have different configurations to facilitate different conditions, for example, high pressure fluid entering inlet 132A while progressively lower pressure fluid enters inlets 132B and 132C. The apparatuses 100A-100C may share a common shaft 112 as illustrated or may have separate shafts. In all other aspects, the fluid working apparatuses 100A-100C operate in accordance with the various embodiments described herein.

Referring to FIGS. 30 and 31, a generator or motor device 200 incorporating a fluid working device 100 in accordance with the invention is illustrated. The fluid working device 100 is substantially as described above and includes a working assembly 110 positioned within a housing 130. The generator device 200 further includes a generating or motor unit 210 supported by the housing 130. The unit 210 includes a housing 212 which is positioned within the center of the tubular outer housing member 134. Preferably the housing 112 encapsulates one end of the shaft 112 and is fluidly sealed relative to the housing 130. The opposite end of the shaft 112 may be sealed relative to the housing 130 such that the generator device 200 is a sealed unit, similar to a refrigeration compressor.

An embodiment of such a device configuration may include one or more fixed magnets 214 supported within the housing 212 adjacent to one of the rotors 114. The magnets 214 are aligned with corresponding magnets 224 mounted on the rotor 114 such that the magnets 224 rotate therewith. The configuration would allow the outer housing 134 and the housing 212 to provide a complete enclosure isolated from the generator or motor unit. In a generator configuration, conversion unit 216 within the housing 212 converts the mechanical energy generated by the rotating rotors 114 to electrical energy in a known manner. The electrical energy is then transferred by an electrical outlet 218, for example, an electrical wire, to a desired circuit. Inversely, if used as a motor driven compressor or the like, electrical energy is

received in the conversion unit and it is then converted and the interaction between the magnets 214 and 224 cause the rotor 114 to rotate.

Various modifications may be made to the components of the fluid working apparatus 100 to achieve a desire output 5 based on variable conditions. The performance of the overall apparatus 100 is dictated by many artifacts of the fluids being used to drive the device including but not limited to: the inlet fluid pressure, exit fluid pressure, the density, the velocity of the flow, the overall configuration of the housing that defines the loops, and the physical properties that make up the working fluid. These properties can include temperature, and available heat that affect the density and therefore volume of the flow. In general terms, the ability for the apparatus to transmit the energy within the working fluid to 15 the rotors relies on a plurality of relationships between the housing, inlet guide vanes, the blades, the stators if used and the exit guide vanes. In addition, the working fluid expansion chambers, created by the housing, provides a better opportunity for the thermal energy in the working fluid to be 20 converted to kinetic energy in the flow. Specifically, the longer distance from outlet to inlet of a stage enables a longer acceleration period. Slower acceleration rates to achieve the equivalent fluid velocity at the next inlet requires less energy to produce, and this can be equated to requiring 25 less drive pressure between the stages.

For the same inlet area and working fluid flows it is possible to reconfigure the physical architecture of the housing to provide unique (different) shaft output properties. Referring to FIGS. 32 and 33, two exemplary housings 130', 30 130" are illustrated. Both housings 130', 130" include an outer housing member 134', 134" with an inner housing member 138', 138" positioned within the tubular portion 135', 135" thereof. In each case, a working flow chamber 141 having a height h is defined between the radially inward 35 portions of the housing members 134', 134" and 138', 138" and a return chamber 140 having a height H is defined between the radially outward portions of the housing members 134', 134" and 138', 138". In these illustrated embodiments, the heights h and H are substantially the same. The 40 difference between the housings 130' and 130" is that housing 130' has a smaller radius R' than the radius R" of the housing 130", each with correspondingly sized rotors 114', 114". Assuming a constant (or the same) working fluid flow rate for both, with the same gross inlet area 132' and 132", 45 an apparatus with the housing 130' with the smaller radius R' would operate at a higher rotational speed with less torque. Conversely, an apparatus with the housing 130" with the larger radius R" would operate at a slower turbine rotational velocity, providing a higher torque to the shaft.

Referring to FIGS. **34-39**, other exemplary housing configurations are illustrated. The embodiment of FIGS. **34** and **35** are similar to that of FIG. **32** and show the housing **130**' having a working flow chamber **141** having a height h that is substantially the same as the maximum height H of the 55 return chamber **140**. This configuration of FIG. **34** provides a constant or near constant cross sectional flow area.

In the housing 130" of FIGS. 36 and 37, the outer housing member 134" and the inner housing member 138" are configured such that the working flow chamber 141" has a 60 height h that is substantially smaller than the maximum height H of the return chamber 140". As a result, the return chamber 140" defines a diffuser portion 127 and a nozzle portion 129. The diffuser portion 127 will slow the flow and allow the fluid a longer period to exchange thermal energy 65 to motive energy (expansion) which is desirable for creating drive motive force later in the nozzle of the next pass

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through the blades 115. This is accomplished by enabling the flow a brief period of expansion (and therefore cooling) which results in an increased volumetric flow rate at a lower pressure. The nozzle portion 129 affords the opportunity to speed the flow up. Speed in the flow is desirable for transferring the fluid motive force into motion of the blades by means of transferring inertia from the flow to rotors.

FIG. 38 illustrates a housing 130"" similar to the housing 130" in that the outer housing member 134"" and the inner housing member 138"" are configured such that the working flow chamber 141"" has a height h that is substantially smaller than the maximum height H of the return chamber 140"". In the present embodiment, the outer surface 139 of the inner housing member 138"" includes a recessed portion 137 such that a chamber 135 is defined adjacent the diffuser portion 127. The chamber 135 may be configured to facilitate greater mixing of the working fluid as it travels through the return chamber 140"". Other profiles of the chambers or configurations of the housing members may be utilized to create turbulence or swirling that may be beneficial in certain applications. It is important to note that both the outer housing profile and the inner housing profile can be changed to create the desired working flow chamber. Further they do not need to be the same profile from zone to zone. It is therefore possible to have the chamber of the first zone look similar to FIG. 34 and the chamber of the last zone could look like FIG. 36.

FIG. 39 illustrates a housing 130° similar to the housing 130° in that the outer housing member 134° and the inner housing member 138° are configured such that the working flow chamber 141° has a height h that is substantially smaller than the maximum height H of the return chamber 140° . In the present embodiment, the maximum height H, and thereby the diffuser portion 127, is radially offset such that the flow experiences a more rapid expansion followed by a longer nozzle 129. The housing configurations are not limited to those illustrated and it is understood that various other housing configurations may be utilized to control flow through the housing.

FIG. 40 illustrates a housing 130^{vi} similar to the housing 130" in that the outer housing member 134^{vi} and the inner housing member 138vi are configured such that the maximum height H of the return chamber 140^{vi} is greater than the height h1, h2 of the working flow chamber 141vi. In the present embodiment, the height h1 of the leading portion of the working flow chamber 141^{vi} is smaller than the height h2 of the trailing portion of the working flow chamber 141vi. Such configuration of the housing 130^{vi} facilitates a structure wherein the blades have a varying configuration with a smaller leading edge 115a and a larger trailing edge 115b. The mass flow rate through a turbine may be assumed constant and as the velocity of the flow changes, as it passes over the blade, the flow cross sectional area is allowed to change as well. In this exemplary embodiment, with the change in blade width and flow cross sectional area going from smaller to larger, the flow velocity will slow down.

FIG. 41 illustrates a housing 130^{vii} is similar to the housing 130^{vi} in that the outer housing member 134^{vii} and the inner housing member 138^{vii} are configured such that the maximum height H of the return chamber 140^{vii} is greater than the height h1, h2 of the working flow chamber 141^{vii} . In the present embodiment, the height h1 of the leading portion of the working flow chamber 141^{vii} is larger than the height h2 of the trailing portion of the working flow chamber 141^{vii} . Such configuration of the housing 130^{vii} facilitates a structure wherein the blades have a varying configuration with a larger leading edge 115a and a smaller trailing edge

115b. In this exemplary embodiment, with the change in blade width and flow cross sectional area going from larger to smaller, the flow velocity will speed up.

The housing configurations are not limited to those illustrated and it is understood that various other housing configurations may be utilized to control flow through the housing.

Referring to FIGS. 42-45, an alternative method of controlling flow through the fluid working apparatus 100^{vi}. In the present embodiment, the circumferential spacing of the 10 boundary vanes 142a-142e is varied such that the circumferential width of the working zones 1-5 correspondingly varies. Referring to FIG. 42, the vanes 142a and 142b are spaced such that the working zone 1 has a circumferential width encompassing nine rotor blades 115. The vanes 142b and 142c are spaced such that the working zone 2 has a circumferential width encompassing eleven rotor blades 115. The vanes 142c and 142d are spaced such that the working zone 3 has a circumferential width encompassing sixteen rotor blades 115. The vanes 142d and 142e are 20 spaced such that the working zone 4 has a circumferential width encompassing twenty rotor blades 115. The vanes 142e and 142a are spaced such that the working zone 5 has a circumferential width encompassing twenty-five rotor blades 115. FIG. 45 illustrates how the volume of the flow 25 102 increases as it passes through the stages of the present embodiment. As shown in FIGS. 42 and 43, the housing inlet 132vi has a width w corresponding to the width of the first working zone 1 and the housing outlet 133vi has a width W corresponding to the width of the last working zone 5, in this 30 case growing in width from zone to zone (or chamber to

While the widths in the current embodiment progressively increase, the invention is not limited to such and the position of the vanes 142 may be varied in any desired manner. For 35 example, the width of the zones may increase every other zone, with the width of the intermediate zone remaining constant. FIGS. 46 and 47 show a fluid working apparatus 100^{vii} wherein the circumferential width decreases from the first working zone 1 to the last working zone 5. Such a 40 configuration may be utilized when the fluid working apparatus 100^{vii} is utilized as a compressor. Other combinations of increasing or decreasing widths may be utilized to achieve desired flow patterns. Furthermore, it is noted that in certain applications, the working fluid may be a non-expansive or compressive working fluid and the flow chambers will have a constant or near constant cross section.

Referring to FIGS. 48 and 49, another manner of controlling the flow through the apparatus 100 is to configure the helical nature of the boundary vanes 142 such that the 50 return flow is either pro-grade or retro-grade. FIG. 48 shows a pro-grade return flow wherein the inlet of the next working zone is circumferentially offset from the outlet of the previous working zone in the same direction as the rotors 114 rotate. FIG. 49 shows a retro-grade return flow wherein the 55 inlet of the next working zone is circumferentially offset from the outlet of the previous working zone in a direction opposite the direction the rotors 114 rotate. In some applications, pro-grade architecture is preferred as it may offer a shorter flow path, however, in some applications the retro- 60 grade configuration may advantageously provide a longer flow path between looping stages where the fluids interact with the rotor blades.

The flow may be further controlled or optimized by altering the configuration of the inlet and outlet vanes 118, 65 120. FIGS. 50A and 50B illustrate an exemplary embodiment wherein the outlet vanes 120*a*-120*n* for a given work-

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ing zone are circumferentially offset from the inlet vanes 118a-118n of that working zone. More specifically, the inlet guide vanes 118 for a given zone extend a width 160 between the first inlet vane 118a of the zone and the last inlet vane 118n of the zone (the intermediate vanes are not shown). Similarly, the outlet guide vanes 120 for a that zone extend a width 162 between the first outlet vane 120a of the zone and the last outlet vane 120n of the zone (the intermediate vanes are not shown). The widths 160 and 162 may be equal as illustrated in FIGS. 50A and 50B or may be different as illustrated in FIGS. 51 and 52. Furthermore, the widths 160 or 162 between zones 1-5 may be different as illustrated in FIG. 52. As shown in FIGS. 51 and 52, the different spacing may be addressed by closing or sealing portions 166 defined between the inlet vane zones 1-5. The closing or sealing portions 166 may be defined by portions of the housing 130 or may be separate components.

The first outlet vane 120a is circumferentially offset a distance 164a from the first inlet vane 118a and the last outlet vane 120n is circumferentially offset a distance 164n from the last inlet vane 118n. In the embodiment of FIGS. 50A and 50B, the distances 164a and 164n are equal, however, FIG. 51 illustrates that the distances 164a and 164n may be different. Furthermore, as shown in FIG. 52, the difference between distances 164a and 164n may vary between working zones 1-5.

FIG. 53 illustrates an embodiment wherein the inlet and outlet vanes 118 and 120 are not circumferentially offset, but instead are generally coaxial. For each working zone 1-5, the distance 160 between the first inlet vane 118a and the last inlet vane 118n is less than the distance 162 between the first outlet vane 118a and the last outlet vane 118n. In this way, the outlet vanes 120a-120n circumferentially overlap the inlet vanes 118a-118n in both circumferential directions and a diffuser configuration is defined from the inlet vanes 118 to the outlet vanes 120. Again, a closing or sealing portion 166 may be provided between the inlet vanes 118 of adjacent working zones 1-5. As well, the above overlap approach may be accomplished while incorporating some offset.

It is noted that flow through adjacent working zones 1-5 will be at different flow rates. The difference in fluid speed between adjacent zones will typically self seal along the pressure lines, similar to an air shield or air knife. That is, the high velocity flow of fluid prevents or minimizes fluid in one zone from transitioning to another. Under ideal operating conditions, the fluid flow will not spill over from one zone to another zone. However, the apparatus 100 typically remains operational even if the flow spills over between

FIG. 54 illustrates the angular displacement of flow across five stages or zones of the apparatus 100. The figure shows the cumulative effects of the flow of the working fluid as the blade speed becomes disproportionate to the design speed (flow channel prescription). The solid center line shows the nominal flow of the working fluid as it would be contained dominantly within the flow channels when the blade speed is best matched to the housing configuration for a particular application of working fluid flow conditions. The upper dashed line represents the shifting position of the flow as the blade speed becomes faster than the design speed. This might occur when load is removed from the shaft, and the rotors would therefore likely speed up, until the working fluid flows were cut back. As noted, if the flow goes above a boundary level, the flow may spill over into the forward zone. The spilled over flow may then simply provide work within the next zone until the fluid flow is corrected and/or re-balanced.

Likewise, the lower dashed line represents the condition where the rotor speed is slower than the proposed housing configuration nominal. This condition would likely occur when load (or additional load) is applied to the shaft, and the load increase causes slowing of the working assembly, until 5 such a point when the operating parameters are adjusted to bring operation back to nominal. If the flow goes below a boundary level, the flow may spill over and reenter the same stage. Again, the design is tolerant of this condition as the spillover will be useful as it has the potential to perform work in the next successive pass until the fluid flow is corrected.

Unless otherwise defined, all terms (including technical and scientific terms) used herein have the same meaning as commonly understood by one of ordinary skill in the art to 15 which this invention belongs. It will be further understood that terms, such as those defined in commonly used dictionaries, should be interpreted as having a meaning that is consistent with their meaning in the context of the relevant art and not be interpreted in an idealized or overly formal 20 sense unless expressly so defined herein.

All of the apparatus, methods and algorithms disclosed and claimed herein can be made and executed without undue experimentation in light of the present disclosure. While the invention has been described in terms of preferred embodiments, it will be apparent to those of skill in the art that variations may be applied to the apparatus, methods and sequence of steps of the method without departing from the concept, spirit and scope of the invention. More specifically, it will be apparent that certain components may be added to, 30 combined with, or substituted for the components described herein while the same or similar results would be achieved. All such similar substitutes and modifications apparent to those skilled in the art are deemed to be within the spirit, scope and content of the invention as defined.

What is claimed:

- 1. An apparatus, comprising:
- a housing structure with a first housing inlet and a first housing outlet;
- a working assembly having an inlet side and an outlet side 40 with at least one first rotor having a plurality of rotor blades positioned between the inlet and outlet sides, the working assembly positioned in the housing structure such that the at least one first rotor is rotatably supported therein;

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- at least one single return chamber returning fluid flow from the outlet side of the working assembly to the inlet side of the working assembly; and
- a whole quantity of working fluid which flows through elements of the apparatus in the sequential order and 50 manner specified below:
 - (a) the first housing inlet in a direction towards a first circumferential portion of a circumferential inlet area defined between the housing structure and the inlet side of the working assembly;
 - (b) the first circumferential portion of the circumferential inlet area in a direction towards the outlet side of the working assembly while workingly engaging a first subset of the plurality of rotor blades;
 - (c) a first circumferential portion of a circumferential 60 outlet area defined between the housing structure and the outlet side of the working assembly;
 - (d) the at least one single return chamber in a direction towards a second circumferential portion of the circumferential inlet area;
 - (e) the second circumferential portion of the circumferential inlet area in a direction towards the outlet

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side of the working assembly while exclusively workingly engaging a second subset of the plurality of rotor blades; and

- (f) the first housing outlet.
- 2. The apparatus according to claim 1, wherein the working assembly further includes at least one stator positioned adjacent to the at least one first rotor.
- 3. The apparatus according to claim 1, wherein the working assembly includes at least one second rotor.
- **4**. The apparatus according to claim **3**, wherein a stator is positioned in between each adjacent pair of rotors.
- **5**. The apparatus according to claim **3**, wherein each of the first and second rotors is supported on the same shaft.
- 6. The apparatus according to claim 3, wherein the at least one first rotor supported on a first shaft and the second rotor is supported on a second shaft, with the first and second shafts rotating in the same direction or opposite direction.
- 7. The apparatus according to claim 6, wherein the first and second shafts are configured to be selectively coupled or uncoupled from one another.
- 8. The apparatus according to claim 1, further comprising at least N return assemblies, wherein N is an integer equal to 2 or more, such that the working fluid passes from the inlet side to the outlet side at least N+1 times and thereby workingly engages at least N+1 subsets of rotor blades before passing out of the first housing outlet.
- **9**. The apparatus according to claim **8**, wherein the number of rotor blades in each subset of rotor blades is equal.
- 10. The apparatus according to claim 8, wherein the number of rotor blades in at least one of the subsets of rotor blades is different from the number of rotor blades in another of the subsets of rotor blades.
 - 11. The apparatus according to claim 1, wherein the working assembly is an expander.
 - 12. The apparatus according to claim 1, wherein the working assembly is a compressor.
 - 13. The apparatus according to claim 1, wherein the housing structure includes an inner housing member and an outer housing member, and the at least one single return chamber is defined by boundary vanes extending radially between the inner and outer housing members.
 - 14. The apparatus according to claim 13, wherein each boundary vane is circumferentially offset from the inlet side to the outlet side in a direction which is the same as the direction of rotation of the at least one first rotor to create a pro-grade return flow.
 - 15. The apparatus according to claim 13, wherein each boundary vane is circumferentially offset from the inlet side to the outlet side in a direction opposite from the direction of rotation of the at least one first rotor to create a retro-grade return flow.
 - 16. The apparatus according to claim 1, further comprising:
 - X second housing inlets and X second housing outlets, wherein X is an integer equal to 2 or more, with a distinct flow path defined between each said second housing inlets and a corresponding one of said second housing outlets; and
 - at least one return assembly defined along each flow path such that the working fluid traveling along each flow path passes from the inlet side to the outlet side at least two times and thereby workingly engages at least two subsets of rotor blades before passing out of the respective first or second housing outlet.

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- 17. The apparatus according to claim 1, further comprising a generating or motor unit supported within the housing structure and operating in association with rotation of the at least one first rotor.
- 18. The apparatus according to claim 17, wherein the 5 generating or motor unit is electrically associated with one or more fixed magnets supported within the housing adjacent to the working assembly and one or more rotating magnets mounted on the at least one first rotor in alignment with the fixed magnets.
- 19. The apparatus according to claim 17, wherein the generating or motor unit converts mechanical energy to electrical energy.
- **20**. The apparatus according to claim **17**, wherein the generating or motor unit is sealed within the housing.
- 21. A method of causing a fluid to flow through an apparatus, comprising:
 - causing a whole quantity of the fluid to flow through elements of the apparatus in the sequential order and manner specified below:
 - (a) a housing inlet of a housing structure in a direction towards a first circumferential portion of a circum-

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ferential inlet area defined between the housing structure and an inlet side of a working assembly, the working assembly including at least one rotor having a plurality of rotor blades;

- (b) the first circumferential portion of the circumferential inlet area in a direction towards an outlet side of the working assembly while workingly engaging a first subset of the plurality of rotor blades;
- (c) a first circumferential portion of a circumferential outlet area defined between the housing structure and the outlet side of the working assembly;
- (d) a single return chamber in a direction towards a second circumferential portion of the circumferential inlet area;
- (e) the second circumferential portion of the circumferential inlet area in a direction towards the outlet side of the working assembly while workingly engaging a second subset of the plurality of rotor blades; and
- (f) a housing outlet of the housing structure.

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